# DETERMINATION OF HEAT TRANSFER COEFFICIENTS IN STIRRED TANK SYSTEMS

FOR REFERENCE

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by

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# DETERMINATION OF HEAT TRANSFER COEFFICIENTS IN STIRRED TANK SYSTEMS

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## KEY WORDS

HEAT TRANSFER CORRELATIONS HEAT TRANSFER DATA HEAT TRANSFER COEFFICIENT (- In agitated vessels - In coiled vessels)

iii

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#### ABSTRACT

v

Heat transfer coefficients for helically coiled and continuously stirred tanks were determined using flat paddle turbine impellers. Heat transfer experiments have been carried out by varying the inlet temperature, flowrate of hot stream and agitator speed in the two flat bottom cylindrical stirred tanks in series. Experiments were also repeated for a single tank.

The investigation was conducted under both steady-state and unsteady-state conditions with water being used, as the liquid to be mixed. The individual film heat transfer coefficients were calculated from the experimental overall heat transfer coefficients by using the Modified-Wilson Graphical Method. Two computer programs were also developed to check the validity of the analytical model used, and to compute film heat transfer coefficients and temperatures.

A Reynolds number range of 19439 to 76727 was used for the agitated side, that there has been very little investigation in this region. In the coil side, the Reynolds number range was varied from 1100 to 14000. The convective heat transfer coefficients for unsteadystate conditions were found to have the same value as for steady-state conditions. The values of the individual film heat transfer coefficients found were in agreement with the previously published values for similar cases and the analytical model for heat transfer in two consecutive stirred tanks gave same results in agreement with the experimental results.

Agitator speed was changed between 200 and 500 rpm. Higher values of heat transfer coefficients were observed with increasing agitator speeds.

**ΰ**ΖΕΤ

Serpantinli ve sürekli karıştırıcılı sistemlerde açık düz bıçaklı turbin karıştırıcılar kullanılarak ısı transfer katsayıları hesaplanmıştır. İkili tank sisteminde yapılan deneylerde karıştırıcı hızı, sıcak su giriş sıcaklığı ve akış hızı değiştirilmiş ve bu deneyler tek tank için tekrar edilmiştir.

Deneyler sırasında gerek tank içinde gerekse serpantin içinde su kullanılmış, kararlı ve kararsız durumlar incelenmiştir. Doğrudan deneysel değerler kullanılarak hesaplanan bileşik ısı transfer katsayısından, Wilson'un Grafik Yöntemi kullanılarak iç ve dış ısı transfer katsayıları hesaplanmıştır. Biri tek tank, diğeri iki tanklı sistem için olmak üzere iki bilgisayar programı hazırlanmış, ampirik denklemler ve geliştirilen analitik modeller kullanılarak ısı transfer katsayıları ve sıcaklıklar hesaplanmıştır. Bu çalışmada karıştırıcılı taraf için Reynolds sayısı aralığı 19439-76727 olup, şimdiye kadar az denenmiş bir aralıktır. Serpantin içinde ise Reynolds sayısı aralığı 1100-14000'dir. Kararsız durum için hesaplanan konvektif isi transfer katsayilarinin kararli durum için hesaplananlarla aynı değerde olduğu bulunmuştur. Bu çalışmada bulunan film 151 transfer katsayısı değerleri daha önce aynı durum için yayınlanan değerlere uymaktadır. Ayrıca, deneysel sonuçlar ile modelden hesaplanan sonuçların da uyumlu olduğu görülmüştür.

Karıştırıcı hızı 200'den 500 rpm'e kadar değiştirilmiştir. Artan karıştırıcı hızı ile yükselen ısı transfer katsayıları gözlemlenmiştir.

# TABLE OF CONTENTS

	Page
KEYWORDS	iii
ACKNOWLEDGEMENT	iv
ABSTRACT	V
σzet	vii
LIST OF TABLES	xi
LIST OF FIGURES	xiii
LIST OF SYMBOLS	xv
I. INTRODUCTION	1
I.A. Heating Systems for Stirred Tanks	3
I.A.1 Direct Systems I.A.2 Indirect Systems	3 4
I.B. Equipment and Operation in Liquid Mixing	5
I.B.1 Turbines and Propellers I.B.2 Paddle Agitators I.B.3 Anchor Agitators I.B.4 Helical Screw Agitators I.B.5 Double Helical Ribbon Agitators	6 9 12 12
II. HEAT TRANSFER IN STIRRED TANK SYSTEMS	15
<ul><li>2.A. Forced Convection</li><li>2.B. Inside Film Coefficient</li><li>2.C. Outside Film Coefficient</li><li>2.D. Methods to Obtain Individual Coefficients.</li></ul>	16 17 20 27

	19 - A.		Page
III.	MATHE	MATICAL MODELLING	30
	3.A.	Mathematical Modelling for a Single Continuous Stirred Tank	30
	3.B.	Mathematical Modelling for two Stirred Tanks in Series	32
	3.C.	Description of the Computer Programs	35
IV.	EXPER	IMENTAL WORK	41
	4.A.	Experimental Equipment	41
		<ul> <li>4.A.1 Tanks</li> <li>4.A.2 Coils</li> <li>4.A.3 Stirrers</li> <li>4.A.4 Thermocouples</li> <li>4.A.5 Digital Thermometer</li> <li>4.A.6 Hot Water Reservoir</li> <li>4.A.7 Cold Water Reservoir</li> <li>4.A.8 Heaters</li> <li>4.A.9 Rotary Pump</li> </ul>	41 42 42 45 45 45 46 46 46
	4.B. 4.C.	• •	46 50
		<pre>4.C.1 Steady-State Experiments 4.C.2 Unsteady-State Experiments</pre>	51 53
۷.	RESUL	TS	55
	5.A.	Evaluation of Experimental Results	55
· · ·		5.A.1 Steady-State Results	55
		5.A.l.a Determination of Overall Heat Transfer Coefficient 5.A.l.b Graphical Estimation of Internal and External Film	56
		Heat Transfer Coefficient	56
		5.A.2 Unsteady-State Results	78
	5.B.	Computer Analysis of Data	83

6.A. Discussion of Results and Further Work Suggestion856.B. Discussion of Errors93REFERENCES99APPENDICES101APPENDIX I. EXPERIMENTAL DATA AND COMPUTED TEMPERATURES102APPENDIX II. MATHEMATICAL MODELLING117APPENDIX III. LIST OF COMPUTER PROGRAMS126APPENDIX IV. PHYSICAL PROPERTIES OF WATER134			Page
Suggestion856.B. Discussion of Errors93REFERENCES99APPENDICES101APPENDIX I. EXPERIMENTAL DATA AND COMPUTED TEMPERATURES102APPENDIX II. MATHEMATICAL MODELLING117APPENDIX III. LIST OF COMPUTER PROGRAMS126APPENDIX IV. PHYSICAL PROPERTIES OF WATER134	VI. DISCU	SSION	85
REFERENCES99APPENDICES101APPENDIX I.EXPERIMENTAL DATA AND COMPUTED TEMPERATURES102APPENDIX II.MATHEMATICAL MODELLING117APPENDIX III.LIST OF COMPUTER PROGRAMS126APPENDIX IV.PHYSICAL PROPERTIES OF WATER134	6.A.		85
APPENDICES101APPENDIX I. EXPERIMENTAL DATA AND COMPUTED TEMPERATURES102APPENDIX II. MATHEMATICAL MODELLING117APPENDIX III. LIST OF COMPUTER PROGRAMS126APPENDIX IV. PHYSICAL PROPERTIES OF WATER134	6.B.	Discussion of Errors	93
APPENDIX I.EXPERIMENTAL DATA AND COMPUTED TEMPERATURES102APPENDIX II.MATHEMATICAL MODELLING117APPENDIX III.LIST OF COMPUTER PROGRAMS126APPENDIX IV.PHYSICAL PROPERTIES OF WATER134	REFERENCES		99
APPENDIX II. MATHEMATICAL MODELLING117APPENDIX III. LIST OF COMPUTER PROGRAMS126APPENDIX IV. PHYSICAL PROPERTIES OF WATER134	APPENDICES		101
APPENDIX III. LIST OF COMPUTER PROGRAMS 126 APPENDIX IV. PHYSICAL PROPERTIES OF WATER 134	APPENDIX I.	EXPERIMENTAL DATA AND COMPUTED TEMPERATURES	102
APPENDIX IV. PHYSICAL PROPERTIES OF WATER 134	APPENDIX II.	MATHEMATICAL MODELLING	117
134	APPENDIX III.	LIST OF COMPUTER PROGRAMS	126
APPENDIX V. DERIVATION OF ERROR EQUATIONS 136	APPENDIX IV.	PHYSICAL PROPERTIES OF WATER	134
	APPENDIX V.	DERIVATION OF ERROR EQUATIONS	136

X

# LIST OF TABLES

			Page
TABLE	2.1	Heat transfer to immersed helical coils in agitated vessels	23
TABLE	2.2	Summary of correlations	24
TABLE	2.3	Comparison of exponential of correlations	26
TABLE	4.1	Range of parameters for single tank and two tanks in series	52
TABLE	4.2	Range of parameters for single tank and two tanks in series	53
TABLE	4.3	Range of parameter for steady-state runs	54
TABLE	5.1	Sample temperature-time data - observed	67
TABLE	5.2	Sample temperature-time data - computed	68
TABLE	5.3	Comparison of experimental results and computed results	69
TABLE	5.4	Comparison of experimental results and computed results	70
TABLE	5.5	Comparison of experimental results and results from the computer	74
TABLE	5.6	Comparison of experimental results and computed results	75
TABLE	5.7	Time constant - Single tank	81
TABLE	5.8	Time constant - Two tanks in series	82
TABLE	6.1	Comparison of geometry factors with experimental results	86
TABLE	6.2	Determination of geometry constant	88

xi

			Page
TABLE	6.3	Sample comparison between observed temperatures and calculated temperatures	91
TABLE	6.4	Numerical error values	97
TABLE	A.I.1	Agitator speed is variable (Single tank)	103
TABLE	A.I.2	Hot water flowrate is variable (Single tank)	104
TABLE	A.I.3	Agitator speed is variable (Two tanks in series)	105
TABLE	A.I.4	Hot water flowrate is variable (Two tanks in series)	106
TABLE	A.I.5	Temperature-Time data - observed	107
TABLE	A.I.6	Temperature-Time data - observed	108
TABLE	A.I.7	Temperature-Time data - observed	109
TABLE	A.I.8	Temperature-Time data - observed	110
TABLE	A.I.9	Temperature-Time data - observed	111

# LIST OF FIGURES

			Page
FIGURE	1.1	Viscosity ranges for agitators	7
FIGURE	1.2	Six blade flat blade turbine with removable blades	7
FIGURE	1.3	Disk mounted curved blade turbine	7
FIGURE	1.4	Hub mounted curved blade turbine	7
FIGURE	1.5	Hub mounted pitched blade turbine	11
FIGURE	1.6	Marine propeller	11
FIGURE	1.7	Radial and axial flow patterns	11
FIGURE	1.8	Anchor agitators	11
FIGURE	1.9	Gate type anchor agitator	13
FIGURE	1.10	Helical screw agitator	13
FIGURE	1.11	Helical ribbon agitator	13
FIGURE	2.1	Agitated vessel with immersed coil for heat transfer	22
FIGURE	3.1	Single tank	30
FIGURE	3.2	Two tanks in series	33
FIGURE	3.3	Flowchart for the single tank systems	37
FIGURE	3.4	Flowchart for the two tanks in series	39
FIGURE	4.1	Experimental set-up - Single tank	43
FIGURE	4.2	Experimental set-up - Two tanks in series	44
FIGURE	4.3	Dimensions of the tank system	47

58 FIGURE 5.1 Temperature profile for single tank FIGURE 5.2 Temperature profile for two tanks in series 58 FIGURE 5.3 Modified Wilson plot for single tank Run 1-4 62 FIGURE 5.4 Modified Wilson plot for single tank Run 5-10 62 FIGURE 5.5 63 Modified Wilson plot for single tank Run 11-15 FIGURE 5.6 Modified Wilson plot for single tank Run 16-20 63 FIGURE 5.7 Modified Wilson plot for two tanks in series Run 21-24 64 FIGURE 5.8 Modified Wilson plot for two tanks in 64 series Run 25-29 FIGURE 5.9 Modified Wilson plot for two tanks in series Run 30-34 65 FIGURE 5.10 Modified Wilson plot for two tanks in series Run 34-39 65 FIGURE 5.11 Relationship between agitator speed and external heat transfer coefficient (Single tank) 66 FIGURE 5.12 Relationship between agitator speed and external heat transfer coefficient (Two tanks in series) 66 FIGURE 5.13 Temperature change with time for unsteadystate heat transfer 79 FIGURE 6.1 Final correlation of data with previously reported data 90 FIGURE A.I.1 Temperature change with time for unsteady-state heat transfer (Single tank) 112 FIGURE A.I.2 Temperature change with time for unsteady-state heat transfer (Single tank) 113 FIGURE A.I.3 Temperature cpahge with time for unsteady-state heat transfer (Single tank) 114 FIGURE A.I.4 Temperature change with time for unsteady-state heat transfer (two tanks in series) 115 FIGURE A.I.5 Temperature change with time for unsteady-state

heat transfer (two tanks in series)

Page

116

xiv

## LIST OF SYMBOLS USED

A <sub>o</sub> ,A <sub>i</sub> ,A <sub>m</sub>	•	Outside, inside and average heat transfer areas respectively $(m^2)$
C <sub>h</sub>	:	Specific heat of hot water (J/kg <sup>0</sup> C)
C <sub>W</sub>	:	Specific heat of cold water (J/kg <sup>O</sup> C)
D <sub>T</sub>	:	Tank diameter (m)
DA	:	Agitator diameter (m)
D <sub>H</sub>	:	Diameter of coil tubing (m)
D <sub>c</sub>	:	Diameter of coil
ff <sub>i</sub> , ff <sub>ci</sub>	:	Fouling factor inside vessel and on coil side respectively (m <sup>2</sup> . <sup>o</sup> C/W)
HI	•	Film heat transfer coefficient inside the coil $(W/m^2.^{O}C)$
НО	:	Film heat transfer coefficient in agitated side $(W/m^2.^{OC})$
H <sub>&amp;</sub>	:	Liquid height in the vessel (m)
Н <sub>с</sub>	:	Coil height from the vessel bottom (m)
<b>k</b>	•	Thermal conductivity of water (W/m <sup>O</sup> C)
k <sub>w</sub>	:	Thermal conductivity of copper coil wall (W/m $^{O}$ C)
L <sub>c</sub>	:	Length of coil(m)
M	:	Cold water flowrate (m <sup>3</sup> /sec)
N	:	Agitator speed (rpm)
N <sub>Nu</sub>	:	Nusselt number, hD/k
N <sub>Re</sub>	:	D <sub>A</sub> Nρ/μ

xv

N <sub>pr</sub>	:	Prandtl number , C <sub>p</sub> µ/k
N <sub>Gz</sub>	:	Greatz number, N <sub>Re</sub> N <sub>pr</sub> D <sub>H</sub> /L
N <sub>Gr</sub>	:	Grashoff number, $(L^3g^2\rho\beta\Delta T)/\mu^2$
N <sub>st</sub>	:	Stanton number, N <sub>Nu</sub> /N <sub>Re</sub> N <sub>pr</sub>
Q	:	Heat transfer rate, Cal/hr
τ <sub>1</sub> , ΤΗΊ	•	Inlet temperature of hot stream to tank II ( $^{\circ}$ C)
T <sub>2</sub> , TH2	•	Inlet temperature of hot stream to tank I ( $^{\circ}$ C)
т <sub>з</sub> , тнз		Exit temperature of hot stream from tank I ( $^{\circ}$ C)
t <sub>o</sub> , TCO	:	Inlet temperature of cold stream to tank I ( $^{\circ}$ C)
t <sub>1</sub> , TC1	:	Cold water temperature in tank I ( <sup>O</sup> C)
t <sub>2</sub> , TC2	:	Cold water temperature in tank II ( <sup>O</sup> C)
SINGLE TANK		
T <sub>1</sub> , TH1	•	Inlet temperature of hot stream ( <sup>O</sup> C)
T <sub>2</sub> , TH2	•	Exit temperature of hot stream ( <sup>O</sup> C)
t <sub>o</sub> , TCO	:	Inlet temperature of cold stream ( <sup>O</sup> C)
t <sub>2</sub> , TC1	:	Cold water temperature in the tank ( <sup>O</sup> C)
U	:	Overall heat transfer coefficient (W/m <sup>2 O</sup> C)
V <sub>H</sub>	:	Velocity of hot stream (m/sec)
W	•	Hot stream flowrate (m <sup>3</sup> /sec)
W <sub>ib</sub>	:	Impeller blade width (m)
ρ <sub>w</sub>	•	Density of cold water (kg/m <sup>3</sup> )
<sup>₽</sup> h	:	Density of hot stream (kg/m <sup>3</sup> )
μ.	:	Viscosity at bulk temperature (kg/m sec)
μ <sub>W</sub>	•	Viscosity at wall temperature (kg/m sec)
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# I. INTRODUCTION

1

One of the fundamental steps in numerious chemical processes is mixing by agitation. In mixing applications, the transfer of heat to or from a fluid in agitated vessels is a common operation. The rate of heat transfer is a function of the physical properties of the agitated liquid and of the heating or cooling medium, the vessel geometry, the material and the thickness of the wall and the degree of agitation. When processing is controlled by heat transfer, such variables as log-mean temperature difference and heat transfer surface area will usually predominate over the agitator variables. Mixing can only affect the outside film resistance.

In engineering work it is often necessary to estimate the overall heat transfer coefficient for a particular operation. While it is possible to calculate individual film coefficients for condensing films and for heating and cooling fluids in the jacket or coil as well as to calculate the resistance of heat transfer wall, limited data are available for the kettle side of the heat transfer surface for the agitated side of the vessel.

Common heat transfer surfaces are jackets and coils of pipe.

The common jacketed vessel provides heat surface around the periphery of the vessel, but offers the minimum area for a given volume of liquid. Immersed coils are used to provide heat transfer surface in process vessels and augment available jacketed surface. An immersed coil in an agitated vessel with or without baffle, takes one of two forms (a) helical coil type or (b) the vertical baffle type. Coils are preferred to jacketed surfaces becouse of lower cost and accomodate higher pressures in a coil or circulate fluids at higher velocities and thus attain higher heat transfer coefficients.

Numerous types of liquid mixing equipment are available for use in chemical processing. The theoretical development of liquid mixing has been slow, and in the absence of the specific design information, equipment type has quite needlessly multiplied during the years. All liquid mixing systems, however, have three factors in common: liquid being mixed, a vessel to confine the liquid and a mechanical device to generate turbulence within the system.

The aim of the present work is to determine both the internal and external film coefficients of heat transfer for a coil immersed in a stirred liquid flowing through two tanks in series, or through a single tank.

This chapter contains a brief review of heating systems for stirred tanks followed by equipment and operation in liquid mixing. In Chapter 2 the theory for heat transfer in stirred tank vessels and a literature survey are given. Method of data analysis is also discussed in the same Chapter. Mathematical modelling for a single tank and two tanks in series are presented and the computer programs

are described in Chapter 3. The equipment used, and the details on the experimental study are reported in Chapter 4. Chapter 5 contains the results of the experiments and the computer analysis of data. Chapter 6 is a discussion of the results and the possible errors.

### 1.A HEATING SYSTEMS FOR STIRRED TANKS

Heating systems for stirred tanks can be classified as either direct or indirect. The former include direct firing and electrical heating. In indirect systems, carrier fluids are used to transport heat to the process. Carrier fluids are also used to extract heat from a process. Also indirect systems can be divided into two categories, as liquid systems and vapor systems.

In the following sections of this part, direct and indirect systems and the advantages and disadvantages of these will be discussed [1].

#### 1.A.1 Direct Systems

Open flame direct firing is nowadays seldom used for the heating of stirred tanks. The heating tends to be uneven and the temperature is difficult to control. In addition, the processing of combustible material is hazardous. The units, although relatively inexpensive, are unwieldy since the burners, fuel supply and controls must be at the process location.

There are various electrical methods of heating a stirred tank. The two most common are resistance heating and induction

heating. In the former, heat is produced by the direct application of a voltage to a resistor. The resistor material is usually an alloy of nickel and chromium. Electrical resistance heating has the disadvantages of low efficiency, limited life of the resistors, and the inconvenience and expense associated with the removal of insulation to replace burried but resistors.

Induction heating systems work on the principle that disturbances in the molecular structure of material caused by a varying magnetic field, produce heat. Induction heaters can be added to the vessel which is already installed. This can be placed outside the insulation on the vessel. The initial cost of induction heating system is quite low. In addition, they are extremely reliable and require almost no maintenance. However, all electrical heating systems tend to have high operating costs. This may be offset by the following advantages: safety, convenience, cleanliness, precision and ease of control and no possibility of contaminating the process liquid.

### 1.A.2 Indirect Systems

In these systems, carrier fluids are used to transport heat to the process. The most widely used heat transport fluids are steam and water. The principle disadvantage is the high pressure involved in high temperature systems. Water in its natural state is never pure. It frequently contains dissolved materials which decrease in solubility with increasing temperature. This may result in the formation of scale over heat transfer areas and could lead to a decrease

in heat transfer rate. Water may also contain corrosive and foamproducing substances. A further disadvantage of water is that in cold weather, the water may freeze when the system is not in use. Since water expands in volume on freezing, serious demage to equipment can result.

Other fluids commanly used to transport heat are mercury, molten salt mixtures, mineral oils and a number of organic compounds having a variety of trade names. These organic compounds are high boiling materials so that the heat transfer systems can be operated at low pressures. Both liquid and vapor phase systems are used.

In liquid systems, the heat transferred is from sensible heat. The faster the liquid is pumped through the system, the less is its change in temperature over the heat transfer surface and the more uniform is the heating or cooling.

In vapor systems, the heat transferred is from latent heat. Since all the heat transfer takes place at the saturation temperature, the entire transfer surface is at a uniform temperature.

A variety of heating systems used differ from the basic systems. In general, heat transfer coefficients are better for condensing vapor systems than for liquid systems.

In this investigation an indirect liquid heat transfer system was used. Water was carrier fluid to transport heat to the system.

#### 1.B EQUIPMENT AND OPERATION IN LIQUID MIXING

Numerious types of liquid mixing equipment are available for use in chemical processing. The theoretical development of liquid

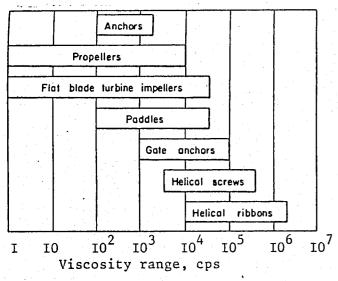
mixing has been slow, and in absence of specific design information, equipment types quite needlessly multiplied during the years. All liquid mixing systems, however, have three factors in common: liquid(s) being mixed, a vessel to confine the liquid(s) and a mechanical device to generate turbulance within the system.

The classification of mixing equipment is usually made on the basis of liquid viscosity since viscosity is a prime contributor to the forces tending to dampen flow through a mixing system. Figure 1.1 shows the recommended viscosity ranges for a number of common agitator types [1].

#### 1.B.1 Turbines and Propellers

For mixing low-to-medium viscosity liquids the flat blade turbine or marine type propeller is recommended. These are general purpose agitators and can be used under a variety of processing conditions. One of the most common turbines is the six blade flat blade, disk mounted type shown in Figure 1.2. A large variety of turbine agitators are available which are modifications of the flat blade design. The disk mounted curved blade turbine (Figure 1.3) and the hub mounted curved blade turbine (Figure 1.4) are useful where the general characteristic of the flat blade type are desired but at a lower shear.

Another modification of the flat blade design is the pitched blade hub mounted turbine (Figure 1.5) with straight blades set at less than 90°C from the horizontal. This design provides reduced



FigureI-I. Viscosity ranges for agitators.

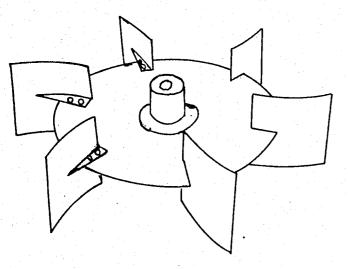


Figure I-3. Disk mounted curved blade turbine.

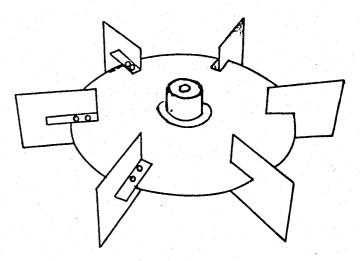


Figure I-2. Six blade flat blade turbine with removable blades.

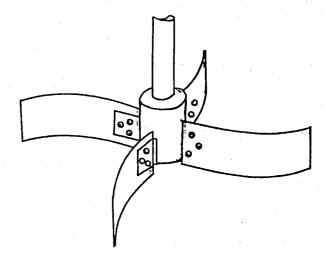


Figure I-4. Hub mounted curved blade turbine.

power requirements and is useful when mixing liquids having a highgravity solids content. Turbine impellers are made either as single piece castings or in more than one piece. The letter are used for entry into vessels having narrow openings.

A common marine propeller has 3 blades, with a blide pitch equal to the propeller diameter. The marine propeller is shown in Figure 1.6.

The 6 blade flat blade turbine and the marine propeller may be mounted in a vessel in various positions. For general processing service, it is recommended that these agitators be mounted 1 impeller diameter from the tank bottom and have diameters 1/3 of the tank diameter.

Agitators are used to produce flow and, subsequently, turbulence in a liquid mass. Each type of agitator causes high velocity liquid to flow through a vessel in a specific path, referred to as a flow pattern. Flat and curved blade turbines, mixing low or moderate viscosity liquids, produce radial flow patterns when used in a baffled vessel. Radial flow, shown in Figure 1.7A is primarily perpendicular to the vessel wall.

The marine type propeller and pitched blade turbine produce axial flow patterns when centered in a baffled vessel containing low or moderate viscosity liquids. Axial flow, shown in Figure 1.7B is primarily flow parallel to the tank wall.

If either the marine propeller or any of the impeller agitators are used in an unbaffled vessel containing low viscosity liquid(s), vortexing develops. The liquid swirls in the direction of the agitator

rotation, causing a drop in liquid level around the agitator shaft. Vortexing increases with impeller speed until eventually the vortex passes through the agitator. The mixing efficiency of vortexing systems is usually lower than for geometrically similar nonvortexing systems.

#### 1.B.2 Paddle Agitators

Paddles are low speed, large blade area agitators, which function by pushing or carrying liquid in a circular path around the vessel. No high speed liquid streams are produced as with turbines, and very little top-to-bottom turnover takes place unless baffles are placed in the vessel and multiple paddles are used. Unbaffled paddles operating in low viscosity liquids produce severe vortexing, even at moderate Reynolds numbers. This leads to inefficient mixing. Since paddles do not mix by high velocity streams, they are well suited for high viscosity service.

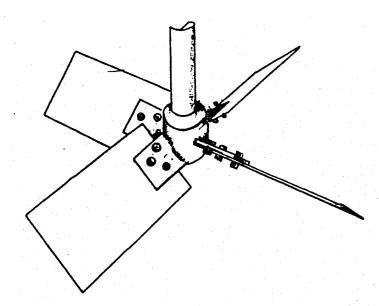
Modified versions of the basic 2 blade paddle are available. The choice depends on the viscosity of the liquid and the degree of shear required.

### 1.B.3 Anchor Agitators

Anchor agitators have been sucessfully used in the batch mixing of liquids having viscosities approaching 100,000 cp Uh1 [2] compared the relative effectiveness of an anchor to that of large paddles and turbines operating in viscous fluids. At 40 rpm, the

anchor adequately mixed 40,000 cp liquid, as compared to a maximum of 15,000 cp for the turbine and paddle agitators. The anchor agitator is generally slow moving, large surface area device; in close proximity to the vessel wall. For heat transfer applications, wall scrappers are utilized, which prevent the built-up of a stagnant film between the anchor and the vessel wall. For liquids of low viscosity (100-1000 cp) the plain horseshoe type anchor Figure 1.8A provides adequate agitation. As viscosity increases, however, cross-members Figure 1.8B or auxiliary paddles Figure 1.8C are requires to overcome viscous drag forces and maintain motion in the core of the liquid mass. For very viscous liquids double motion anchor-paddle combinations are useful. The same effect is obtained when the basic horseshoe anchor is equipped with additional vertical members. This type of agitator is commonly known as a gate type anchor and is shown in Figure 1.9.

It is generally recognized that slow moving agitators with large surface areas are needed to mix and perform other operations in the processing of viscous materials. The inadequacy of high-speed agitators for this task has been effectively demonstrated. The anchor and its modifications occupies the position of the appropriate agitator for materials having consistencies between mobile fluids and extremely high consistency materials which are generally plastic in character.



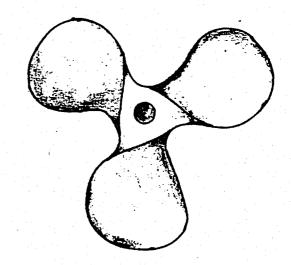
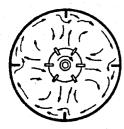
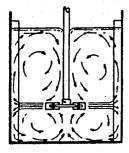
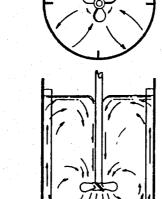


Figure I-5. Hub mounted pitched blade turbine.





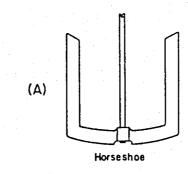


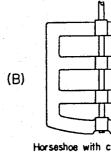
Radial flow pattern

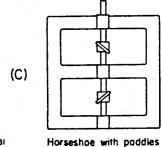
Axial flow pattern

Figure I-7. Radial and axial flow patterns.

Figure I-6.Marine propeller.







Horseshoe with cross-member

Figure I-8. Anchor agitators.

#### 1.B.4 Helical Screw Agitators

The helical screw agitator Figure 1.10 is an effective device when used in high viscosity liquid(s), since it does not depend on high velocity liquid streams to accomplish mixing.

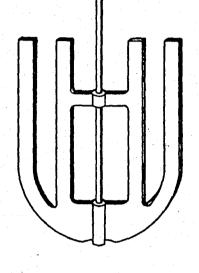
Helical screws may be obtained in numerious sizes and geometries. All screws, however, are classified on the basis of the following parameters: diameter, pitch, flight number, flight depth and screw length.

The selection of the proper screw for any specific process application is a function of numerious system variables. The screw diameter to tank diameter ratio, the screw pitch to screw diameter ratio, the number of flights and the flight depth each contribute to the overall capabilities of the mixing configuration. The screw length is usually predetermined by processing conditions, since the screw should extend from the tank bottom to the liquid surface.

The screw normally functions by carrying liquid from the vessel bottom to the liquid surface. The liquid is then discharged and returns to the tank bottom to fill the void created when fresh liquid is carried to the surface. Alternatively, screws may be operated in the reverse direction to pull liquid to the bottom of the vessel less power is required in this case.

## 1.B.5 Double Helical Ribbon Agitators

For extremely high viscosity applications (1,000,000 cp) specialized agitators must be utilized to obtain any degree of



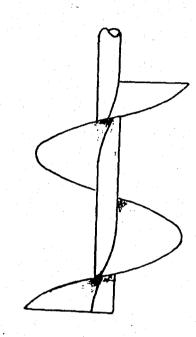


Figure I-9.Gate type anchor agitator.

Figure I-IO.Helical screw agitator.

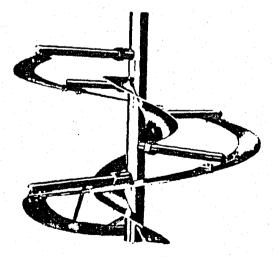


Figure I-II.Helical ribbon agitator.

top-to-bottom turnover. The helical ribbon agitator Figure 1.11 is well suited for ultra high viscosity mixing. Mixing occurs through the forced downward action of the central helix and subsequent bottomto-top motion brought about by the outer helical ribbon. The helical ribbon gives complete mixing with no stagnant areas. Since the outer ribbon is in close proximity to the vessel wall, the scrapping action increases the heat transfer rate in jacketed tanks.

# II. HEAT TRANSFER IN STIRRED TANK SYSTEMS

The heating or cooling of an agitated liquid mass is a common industrial practice. Heat transfer may take place by radiation, conduction or convection, or by a combination of all the three processes. Radiation occurs when energy, in the form of high frequency electromagnetic waves, is emitted from a heat source. Conduction is the transfer of energy between vibrating molecules which remain in a fixed position relative to each other. Convection, both natural and forced, occurs between colliding molecules at different degrees of excitation as they change position and move through a liquid [3].

In a vessel containing an agitated liquid, heat transfer is brought about primarily through conduction and forced convection [4]. The resistance or film, theory conveniently describes this process by

> Driving Force = Rate (2.1) Resistance

#### where

Driving Force: Temperature difference, ∆T in °C Resistance : 1/conductance: 1/U. where U is in watt/m<sup>2</sup>°C Rate : Heat flow per unit area, Q/A in watt/m<sup>2</sup>

Therefore equation (2.1) can be written as;

$$\frac{\Delta T}{1/U} = Q/A$$

or as,  $Q = U A \Delta T$ .

The overall heat transfer coefficient, U, is determined from a series of five resistances to transfer of heat with internal coils, the coefficient U must be referred to the inner and outer coil surface. The overall heat transfer coefficient,  $U_0$ , based on the outside surface of the coil can be written as

$$\frac{1}{U_{0}} = \frac{1}{h_{0}} + ff_{i} + \frac{x}{k_{w}} \left(\frac{D_{To}}{D_{TAV}}\right) + \frac{1}{h_{i}} \left(\frac{D_{To}}{D_{Ti}}\right) + ff_{ci} \qquad (2.2)$$

If fouling factors (ff<sub>i</sub>, ff<sub>ci</sub>) are neglected, contributions of the operating variables can be separated into three heat transfer resistances. The outside convective heat transfer coefficient,  $h_0$ , lumps together convective and conductive effects in the fluid. This coefficient is usually the limiting resistance and is affected by agitator operation.

In the following sections of this chapter forced convection, inside, outside film coefficient, and methods to obtain individual coefficients will be discussed.

## 2.A A FORCED CONVECTION

Heat transfer by convection is due to fluid motion. Cold fluid adjacent to a hot surface receives heat which it imparts to the bulk of the cold fluid by mixing with it. Free or natural convection occurs when the fluid motion is not implemented by mechanical agitation. But when the fluid motion is mechanically agitated, the heat is transferred by forced convection. The mechanical agitation may be supplied by stirring, although in most process applications it is induced by circulating the hot and cold fluids at rapid rates on the opposite sides of pipes or tubes. Free and forced convection heat transfer occur at very different rates, the latter being the more rapid and therefore more common. Factors which promote high rates for forced convection do not necessarily have the same effect on free convection.

Forced convection heat transfer is the most frequently employed mode of heat transfer in the process industries. Hot and cold fluids, separated by a solid boundary, are pumped through the heat transfer equipment, the rate of heat transfer being a function of the physical properties of the fluids, of the operating conditions and the geometry of the system. Theoretical analyses of forced-convection heat transfer have been limited to relatively simple geometries and to laminar flow. Analysis of turbulent flow heat transfer have been based upon some mechanistic model and have not generally yielded relationships which are suitable for design purposes. Usually for complicated geometries only empirical relationships are available, and frequently these are based upon limited data and special operating conditions [5,6].

#### 2.B INSIDE FILM COEFFICIENT-INTERNAL COIL

Internal coils should be designed for turbulent flow. For special cases involving viscous heat-transfer fluids, laminar or

transition flow may be unavoidable.

Numerous relationships have been proposed for predicting heat transfer for turbulent flow in tubes. A number of observers report data for heating or cooling various fluids with 0.7 <  $N_{pr}$  < 700, for 10.000 <  $N_{Re}$  < 120.000, for  $L/D_{H} \ge 60$ . These data [7] for moderate  $\Delta T$  have been correlated by three types of equations. Dittus and Boelter [8] evaluates all physical properties at the bulk temperature

$$\frac{n_{i} D_{H}}{k_{b}} = 0.023 \left(\frac{D_{H} V_{H}}{\mu_{b}}\right)^{\circ} \cdot \left(\frac{C_{p} \mu}{k}\right)^{\circ}_{b} \cdot \left(\frac{C_{p} \mu}{k}\right)^{\circ}_$$

Colburn [9] evaluates all properties, except  $C_p$  in the Stanton modules, at the film temperature,  $t_f = (t_w + t_b)/2$ :

$$\frac{h_{i}}{C_{p_{b}}V_{H}} \left(\frac{C_{p^{\mu}}}{k}\right)_{f}^{2/3} = \frac{0.023}{(D_{H}V_{H}/\mu_{f})^{0} \cdot ^{2}}$$
(2.4)

Sieder and Tate [10] evaluates all physical properties at the bulk temperature, except  $\mu_W$  in a viscosity-ratio term:

$$\frac{h_{i}D_{H}}{k_{b}} = 0.023 \left(\frac{C_{p}\mu}{k}\right)_{b}^{0} \cdot {}^{33} \left(\frac{D_{H}V_{H}}{\mu_{b}}\right)^{0} \cdot {}^{8} \left(\frac{b}{\mu_{W}}\right)^{0} \cdot {}^{14}$$
(2.5)

As pointed out by Colburn [9] the form of the last two equations has several advantages over that of Eq. (2.3).

At a given Reynolds number, heat transfer coefficient of coils, particularly with turbulent flow, are higher than those of long, straight pipes, due to the greater friction. Jaschke [11] found that the results for long straight tubes should be multiplied by a turbulent-flow coil correction factor  $(1 + 3.5(D_H/D_C))$ . The Sieder-Tate equation for straight pipes, multiplied by a turbulent flow coil correction factor, can be used to calculate the inside convective heat transfer coefficient of coil

$$\frac{h_i D_H}{k_b} = 0.023 \left(\frac{C_p}{k}\right)_b^{0.33} \left(\frac{D_H V_H}{\mu_b}\right)^{0.8} \left(\frac{\mu_b}{\mu_W}\right)^{0.14} \left(1 + 3.5 \frac{D_H}{D_c}\right) \quad (2.6)$$

The transition region lies in the range  $2100 < N_{Re} < 10.000$ . No simple equation exists for accomplishing a smooth mathematical transition from laminar flow to turbulent flow. Of the relationships proposed, Hausen's equation [12] fits both the laminar and fully turbulent extremes quite well.

$$\frac{h_{i}D_{H}}{k_{b}} = 0.116((\frac{D_{H}V_{H}}{\mu_{b}})^{2/3} - 125)(\frac{C_{p}\mu}{k})^{1/3}[1 + (\frac{D_{H}}{L})^{2/3}](\frac{\mu_{b}}{\mu_{w}})^{0.14}$$
(2.7)

between 2100 and 10.000. It is customary to represent the probable magnitude of coefficients in this region by hand-drawn curves [13].

Normally laminar flow occurs when  $N_{Re} < 2100$ . Laminar flow heat transfer has been subjected to extensive theoretical study. The energy equation has been solved for a variety of boundary conditions and geometrical configurations. However, true laminar flow heat transfer very rarely occurs. Natural convection effects are almost always present, so that the assumption that molecular conduction alone occurs is not valid. Therefore, empirically derived equations are most reliable. For circular tubes several relationships are applicable depending upon value of Graetz number,  $N_{Gz} = (N_{Re}N_{pr} D_{H}/L)$ , where,  $N_{Re} = (D_{H}V_{H})/\mu$ ,  $N_{pr} = (C_{p}\mu/k)$ . For  $N_{Gz} < 100$ , Hausen's [12] equation is recommended:

$$\frac{h_i D_H}{k} = 3.66 + \frac{0.085 N_{GZ}}{1 + 0.047 N_{GZ}^{2/3}} (\frac{\mu_b}{\mu_w})^{0.14}$$
(2.8)

For  $N_{GZ} > 100$ , the Sieder-Tate [10] relationship is satisfactory for small diameters and  $\Delta T$ 's:

$$\frac{h_i D_H}{k} = 1.86 N_{GZ}^{1/3} \left(\frac{\mu_b}{\mu_W}\right)^{0 \cdot 14}$$
(2.9)

A more general expression covering all diameters and  $\Delta T$ 's is obtained by including an additional factor 0.87(1 + 0.015N $_{Gr}^{1/3}$ ) on the right side of Eq. (2.9), where, N<sub>Gr</sub> =Grashoff number. An equation published by Oliver [14] is also recommended.

In the present study, to calculate film heat transfer coefficient inside the coils, taking into consideration the experimental conditions, the following equations were used; for turbulent flow, Eq. (2.6), for transition region, Eq. (2.7) and for laminar flow, Eq. (2.9).

### 2.C OUTSIDE FILM COEFFICIENT-INTERNAL COIL

During the past 35 years, there has been considerable investigation of heat transfer coefficients for the agitated side of the coil tank vessels. Many of these heat transfer data are confined to very specific conditions and it is difficult to apply them under conditions other than those from which they were obtained. Most of the authors present the result of their investigation by an equation of the following form

$$N_{Nu} = C_0 \cdot N_{Re}^A \cdot N_{pr}^B V_{is}^E$$
(2.10)

The constant  $C_0$  values differ for various types of mixers and were expressed by a product of new constant C value and dimensionless modul representing the effects of the geometrical mixer parameters so;

$$N_{Nu} = C.N_{Re}^{A}.N_{pr}^{B}.V_{is}^{E}.X_{1}.X_{2}....$$
 (2.11)

where,  $X_1, X_2, \ldots$  are dimensionless moduli representing the effect of mixer geometry on heat transfer in a mixing vessel.

$$N_{Re} = \frac{\rho \cdot N \cdot D_A^2}{\mu}$$
 is modified Reynolds number (for mixing).

$$N_{pr} = \frac{C_p \cdot \mu}{k}$$
 is the Prandtl number,

 $N_{Nu} = \frac{h_0 \cdot D_H}{k}$  is the Nusselt number,

$$V_{is} = \frac{\mu_b}{\mu_w}$$
 is the ratio of the process-liquid viscosities  
at bulk temperatures and at wall temperatures.

The results of more extensive experimental investigations are outlined in this section. Except for the scattered early data of Pierce and Terry [15] and of Rhodes [16], the work of Chilton et al. [17] was the first study to use conventional groups for correlating heat transfer to jackets or coils. Since then, considerable progress in this area has been made despite the many variables required to describe the geometry of a typical agitated system. Known heat transfer data for helical coils are summarized in Table 2.1. The manner of presentation of this indicates both, that there is no overall correlation of the available data, and that there are fragments of information giving the effect of many variables. The correlating equation of each separate work is tabulated in Table 2.2.

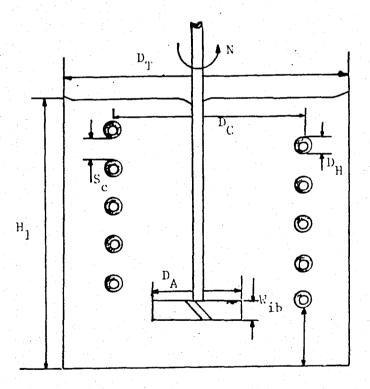


Figure 2-1. Agitated vessel with immersed coil for heat transfer.

TABLE 2.1 -	Heat	Transfer	to	Immersed	Helical	Coils	in	Agitated	Vessels

Impell	er			Helical C	oil				V	essel	
Investigator	D <sub>A</sub> (in)	w <sub>ib</sub> /D <sub>A</sub>	H <sub>c</sub> /D <sub>A</sub>	Material	D <sub>H</sub> (in)	S <sub>c</sub> (in)	D <sub>c</sub> (in)	H <sub>c</sub> (in)	D <sub>T</sub> (in)	H <sub>l</sub> /D <sub>T</sub>	Bottom Type
1) CHILTON et.al	7.2	0.17	0.33	Copper	1/2	1/4	9.60	5.75	12	0.83	Dished
2) PRATT	5.5-14	various	various	Stainless steel and lead	3/4-3.5	0.48-2	7.75	18-28	18-23.5	-	Flat
3) CUMMING and WEST	12	0.17		stainless steel	1.0	1/2	24	23	30	1.0	Dished
4) KRAUSSOLD	20	1.4		-		25/32	•	-	40	1.1	Dished
5) OLDSHUE and GRETTON	20-28	0.2		Copper Stainless steel	7/8 1.75	7/8 1.75	3.15 34.2	31.5 31.5	48 48	ן ז	Flat Flat
6) SKELLAND and DABROWSKI	36.9	-		Copper	3/4	1.25	13	15	-	H :20"	Flat
7) JHA and RAO	4 <u>B</u>	affle width D <sub>A</sub>	- = 0.25	Copper	1/2	-	10	<b></b>	14	1.4	Flat

Investigator	Correlation	Range of N <sub>Re</sub>
1) CHILTON Et al.	$\frac{h_0 D_T}{k} = 0.87 (N_{Re})^{0.62} (N_{pr})^{0.33} (\mu_b / \mu_w)^{0.14}$	300 - 4x10 <sup>5</sup>
2) PRATT	$\frac{h_o D_T}{k} = 34(N_{Re})^{\circ \cdot 5}(N_{pr})^{\circ \cdot 3}(S_c/H_c)^{\circ \cdot 8}(W_{ib}/D_c)^{\circ \cdot 25}((D_A D_T)/D_H)^{\circ \cdot 1}$	2x10 <sup>4</sup> - 5x10 <sup>5</sup>
3) CUMMING and WEST	$\frac{h_0 D_T}{k} = 1.01 (N_{Re})^{0.62} (N_{pr})^{1/3} (\mu_b/\mu_w)^{0.14}$	2x10 <sup>3</sup> - 7x10 <sup>5</sup>
4) KROUSSOLD	$\frac{h_o D_T}{k} = 1.01 (N_{Re})^{0.62} (N_{pr})^{1/3} (\mu_b / \mu_w)^{0.14}$	0.3 - 1x10⁵
5) OLDSHUE and GRETTON	$\frac{h_0^{D_H}}{k} = 0.17(N_{Re})^{0.67}(N_{pr})^{0.37}(D_A/D_T)^{0.1}(D_H/D_T)^{0.5}$	400 - 1.5x10 <sup>6</sup>
6) SKELLAND and DABROWSKI	$\frac{h_0^{D}H}{k} = 0.345(N_{Re})^{0.62}(D_T/C_p)^{0.27}$	2.5x10⁵ - 10 <sup>6</sup>
7) JHA and RAO	$\frac{h_0 D_T}{k} = 0.18(N_{Re})^{0.67}(N_{pr})^{0.33}(D_H/D_T)^{-0.48}(D_c/D_T)^{-0.27}(H_c/D_T)^{0.14}$	1 - 2.5x10 <sup>5</sup>

TABLE 2.2 - Summary of Correlations

From Table 2.1, the designer can obtain dimensions for geometric similarity. Although Pratt's work offers the choice of more dimensional variables, the different values for some of the exponents in Pratt's equation [18] introduce some question concerning the accuracy obtained by using this relationship directly. Very little of the other work has been directed towards finding the effect of dimensional variations. Cummings-West [19] ran tests under steady-state conditions using six different liquids having widely varying properties. These tests indicated that film coefficients would be the same for a given Reynold number whether one or two turbines were on the shaft. The tests conducted with the pitched-blade turbine instead of retreating-blade turbine showed on approximate 10% reduction in film coefficient. No significant change in the film coefficient resulted from reversing the pitched-blade turbine. It is apparent that only two of the available correlations, Oldshoe's and Skelland's [21,22] indicate the effect of coil tube diameter,  $D_{H}'$ . Also, since  $D_{T} >> D_{H}$  the values of  $h_{O}'$ obtained from the same Nusselt number for different correlations, is different. Since the Nusselt number describes heat transfer occurring mainly at the coil, it is more logical to define Nusselt number based on the coil tube diameter,  $D_{H}'$ , rather than the tank diameter  $D_{T}$  [24]. Of the dimensional parameters, only the effect of  $D_A^{}$ ,  $D_T^{}$ , and to an extent 'D $_{\rm H}$ ' in addition to fluid properties are indicated in Table 2.3. None of these can be considered to be fairly established [2]. It may be remarked that the parameters  $'D_{H}'$  and  $'S_{c}'$  have much greater influence on  $h_0'$ . These affect considerably the flow distribution through and around the coil or the degree of by passing.

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25

Variable	CHILTON et al.	PRATT	CUNNING and WEST	OLDSHUE and GRETTON	SKELLAND and DABROWSKY	JHA and RAO
D <sub>A</sub>	1.24	1.20	1.24	1.43	1.24	1.34
N	0.62	0.5	0.62	0.67	0.62	0.67
D <sub>T</sub>	-1.0	-0.9	-1.0	-0.6	0.27	-0.39
D <sub>H</sub>		-0.3		-0.5	-1.0	-0.48
S <sub>c</sub>		0.8	-	-	-	-
µ*/µ <sub>w</sub>	0.14	0.14		0.14	0.14	0.14

TABLE 2.3 - Comparison of Exponentials of Correlations

\*Most have assumed this in accordance with Sieder and Tate.

The influence of a number of other factors can be expressed qualitatively. At lower speeds the effect of agitation can be overshadowed by free convection [2], and at increasingly higher speeds the heat transfer rates will level off. The impeller blade length and width, obviously affect the area of shear, turbulence, hence degree of mixing. This consequently is revealed as nonuniformities is temperature. The parameters  $H_{ib}$ ,  $H_{\&}$ ,  $D_c$  again affect the degree of turbulence and mixing, thus influencing ' $h_0$ '. The influence of other factors, such as type of impeller, number of impellers, and other factors on ' $h_0$ ' can not be established and the need for more data is obvious.

#### 2.D METHODS TO OBTAIN INDIVIDUAL FILM COEFFICIENTS

Values of the individual film coefficients are obtained experimentally by one of three methods:

- 1. Use of modified Wilson plot [25,19,2]
- 2. Measurement of the temperature of the heat transfer barrier by imbedded thermocouples, [17,26,21]
- 3. Measurement of mechanical energy losses and use of momentum and heat transport analogies [27,28].

The use of the Modified Wilson plot [29] bears further discussion. In this part the Modified Wilson plot which was used in this investigation will be discussed in detail.

a) <u>Case I</u> (determination of inside film coefficient, h<sub>i</sub>)

For the case where outside resistance of coil is controlling, and when conditions of flow and temperature inside the coil do not change appreciably, it can be shown that Eq.(2.2) reduces to

$$\frac{1}{U_0} = \beta + \frac{1}{h_0}$$
(2.12)

For a given vessel and coil geometry, it can be recognized that  $h_0$  is the some way dependent on degree of mixing, hence speed of the agitator. Thus for above conditions, it is satisfactory to say that [30]

$$h_0 = K N^0$$

Experimental observations seem to indicate that  $\alpha$  is 0.62 [19] so Eq. (2.12) becomes

$$\frac{1}{U_0} = \beta + \frac{1}{K N^0 \cdot 6^2}$$
(2.13)

Thus a plot of  $1/U_0$  vs  $1/N^{0.62}$  will be linear with a slope of 1/K and an intercept of  $\beta$ , after determining  $\beta$  and using its definition

$$\beta = \frac{1}{h_i} \frac{A_o}{A_i} + \frac{x}{k} \frac{A_o}{A_m} \quad (constant) \quad (2.14)$$

such that k and  $h_i$  do not vary appreciably, one can evaluate " $h_i$ " as all the remaining terms are known.

## <u>Case II</u> (determination of outside film coefficient, $h_0$ )

For the case where resistance of the coil is controlling, and when degree of mixing, and temperature outside the coil do not change appreciably, it can be shown that [30]

$$\frac{1}{U_{i}} = \gamma + \frac{1}{h_{i}}$$
(2.15)

For a given coil geometry, it can be imagined that "h<sub>i</sub>" like the friction factor inside the coil must in the some way be related to the flow pattern, hence the velocity of flow of the fluid through the coil. Thus for this conditions it can be assumed that

$$h_i = K'V^{\alpha'}$$

b)

Experimental observations seem to indicate that  $\alpha'$  is 0.8 and for fully developed turbulent flow then Eq. (2.15) becomes

$$\frac{1}{U_{i}} = \gamma + \frac{1}{K'V^{0} \cdot B}$$
(2.16)

Thus a plot of  $1/U_{\rm i}$  vs  $1/V^{0\, \cdot\, 8}$  will be linear with a slope of 1/K' and intercept of  $\gamma$ 

$$\gamma = \frac{1}{h_0} \frac{A_i}{A_0} + \frac{x}{k} \frac{A_i}{A_m}, \qquad (2.17)$$

 $h_o$ , and k do not vary appreciably. Thus  $h_o$  can be evaluated.

In the present study the Wilson graphical method was used to calculate individuals coefficients from the experimental overall co-efficients.

## III. MATHEMATICAL MODELLING

In this chapter mathematical models were developed for a continuous stirred single tank and for two stirred tanks in series respectively. Only the original differential equations, the boundary conditions and final solutions have been presented in this chapter. The details on solutions are explained in Appendix II.

# 3.A MATHEMATICAL MODELLING FOR A SINGLE CONTINUOUS STIRRED TANK

For the single continuous stirred tank, where the flow of cold water is in the tank side and that of hot water is in the coil side, counter current flow operation may be performed.

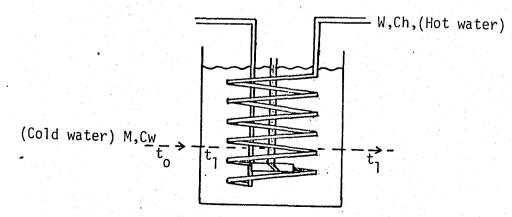


Figure 3.1 - Single tank

Then, the heat balance for the tank shown in Figure 3.1 is as follows

{Input} - {Output} = {Accumulation}

 $WC_{h}\rho_{h}T_{1} - WC_{h}\rho_{h}T_{2} + (MC_{w}\rho_{w}t_{0} - MC_{w}\rho_{w}t_{1}) = VC_{w}\rho_{w}(dt_{1}/d\theta) \quad (3.1)$ 

where;

M : Cold water flowrate

W : Hot water flowrate

 $\mathbf{C}_{\mathbf{h}} \colon \mathbf{Hot} \text{ water heat capacity}$ 

U : Overall coefficient of heat transfer

A : Heat transfer area of coil

V : Volume of the tank

 $C_w$ : Cold water heat capacity

T<sub>1</sub>: Hot water inlet temperature

T<sub>2</sub>: Hot water exit temperature

t<sub>o</sub>: Cold water inlet temperature

 $t_1$ : Cold water exit temperature

 $\boldsymbol{\rho}_h \text{: Hot water density}$ 

 $\theta$  : Time

Assuming proper agitation, (i.e. uniform temperature in the tank,  $t_1$ ) the heat transfer rate equation for the tank is

$$\frac{U_1 A(T_2 - T_1)}{\ln \frac{T_2 - t_1}{T_1 - t_1}} = WC_h \rho_h(T_1 - T_2)$$

this equation simplifies to

$$\frac{T_2 - t_1}{T_1 - t_1} = \exp[-\frac{UA}{WC_h\rho_h}] = \alpha_1 \text{ (Alpha}_1) \tag{3.2}$$

Using the following initial condition

at 
$$\theta = 0$$
  $t_1 = t_{1s}$  (After reaching a steady-state)

the final solution is

$$t_{1} = [t_{1s} - \frac{WC_{h}\rho_{h}(1-\alpha)T_{1} + MC_{w}\rho_{w}t_{o}}{MC_{w}\rho_{w} + WC_{h}\rho_{h}(1-\alpha)}]exp^{-\left[\frac{MC_{w}\rho_{w}+WC_{h}\rho_{h}(1-\alpha)}{VC_{w}\rho_{w}}\right]}$$

+ 
$$\frac{WC_{h}\rho_{h}(1-\alpha)T_{1} + MC_{w}\rho_{w}t_{o}}{MC_{w}\rho_{w} + WC_{h}\rho_{h}(1-\alpha)}$$
(3.3)

where,  $T_1$  and  $t_0$  are known temperatures, and  $t_1$  and  $T_2$  can be calculated.

## 3.B MATHEMATICAL MODELLING FOR TWO STIRRED TANKS IN SERIES

Consider the system, where cold water at temperature  $t_0$  is fed to tank I where it is well-stirred in contact with heating coils. The continuous discharge from this tank at temperature  $t_1$  flows into stirred tank II and leaves tank II at temperature  $t_2$ . Heating water at temperature  $T_1$  flows into the coil of the tank II and then leaves tank II at temperature  $T_2$  enter the coil of the first tank. Finally the hot water is at temperature,  $T_3$  as it leaves the coil of tank I. Symbols used in the modelling are similar to those of the single tank, the only difference being that subscripts 1 and 2 indicate the properties which belong to tank I and tank II respectively.

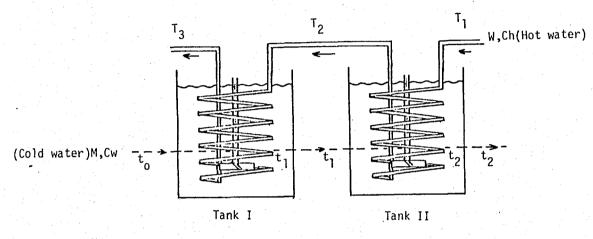


Figure 3.2 - Two tanks in series

Then the heat balances for the two tanks in series shown in Figure 3.2 are as follows;

TANK I: 
$$WC_{h}\rho_{h}(T_{2}-T_{3}) - MC_{w}\rho_{w}(t_{1}-t_{0}) = VC_{w}\rho_{w}\frac{dt_{1}}{d\theta}$$
 (3.4)

TANK II: 
$$WC_{h}\rho_{h}(T_{1}-T_{2}) - MC_{w}\rho_{w}(t_{2}-t_{1}) = VC_{w}\rho_{w}\frac{dt_{2}}{d\theta}$$
 (3.5)

Assuming proper agitation (i.e. uniform temperature in tank I,  $t_1$  and in tank II,  $t_2$ ) the heat transfer rate equations for the tanks I and II, are

TANK I 
$$\frac{T_3 - t_1}{T_2 - t_1} = \exp[-\frac{U_1 A}{W C_h \rho_h}] = \alpha$$
 (Alpha) (3.6)

TANK II 
$$\frac{T_2 - t_2}{T_1 - t_2} = \exp[-\frac{U_2 A}{W C_h \rho_h}] = \beta$$
 (Beta) (3.7)

Using systematic elimination technique [31] the amount of algebra involved may be reduced to a minimum. These four equations are solved simultaneously.

Initial conditions are;

at

$$\theta = 0$$
  $t_1 = t_{1s}$  (After reaching a steady-state)  
 $t_2 = t_{2s}$ 

Final solutions are

$$t_{1} = \frac{V}{M} [m_{1}Ae^{m_{1}\theta} + m_{2}Be^{m_{2}\theta}] + [\frac{WC_{h}P_{h}}{WC_{w}P_{w}}(1 - \beta) + 1] *$$

$$(Ae^{m_{1}\theta} + Be^{m_{2}\theta} + CP) + [\frac{WC_{h}P_{h}}{MC_{w}P_{w}}(\beta - 1)]T_{1} \qquad (3.8)$$

and

$$t_2 = Ae^{m_1\theta} + Be^{m_2\theta} + CP$$
 (3.9)

where;

$$CP = \frac{MC_{w}\rho_{w}t_{o} + (WC_{h}\rho_{h} - WC_{h}\rho_{h}\alpha\beta + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}}{MC_{w}\rho_{w}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\alpha}{MC_{w}\rho_{w}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\alpha\beta}{MC_{w}\rho_{w}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\alpha\beta}{MC_{w}\rho_{w}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}\rho_{w}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\beta}{MC_{w}} + \frac{W^{2}C_{h}^$$

$$BP = \frac{2WC_{h}\rho_{h}}{VC_{w}\rho_{w}} + \frac{2M}{V} - \frac{WC_{h}\rho_{h}}{VC_{w}\rho_{w}} - \frac{WC_{h}\rho_{h}}{VC_{w}\rho_{w}}$$

$$DP = \frac{M^{2}}{V^{2}} + \frac{MWC_{h}\rho_{h}}{V^{2}C_{w}\rho_{w}} - \frac{W^{2}C_{h}^{2}\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}}{V^{2}C_{w}^{2}\rho_{w}^{2}} + \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\alpha}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{h}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}} - \frac{W^{2}C_{h}^{2}\rho_{w}^{2}\alpha\beta}{V^{2}C_{w}^{2}\rho_{w}^{2}} - \frac$$

 $A = t_{2s} - B - CP$ 

#### 3.C DESCRIPTION OF THE COMPUTER PROGRAMS

Computer programs for both the single tank system and the two tanks in series were prepared using the analytical model developed in part (3.A) and (3.B) to facilitate the calculation of steady-state and unsteady-state temperatures and of heat transfer coefficients. Each program contains a main program and four subprograms which calculate physical properties of water (Appendix IV).

The program flowcharts are given in Figure 3.3 for the single tank and in Figure 3.4 for two tanks in series. Symbols used are given in Appendix III.

Inlet temperatures and flowrates of hot and cold water are given as data input SI units have been used in both programs, as well as in the data and the results. In both programs, data are read and written first. Time is taken large enough to obtain steady-state temperatures and then physical properties of water (i.e.  $C_p$ ,  $\mu$ ,  $\rho$ , k) are evaluated at inlet hot and cold water temperatures. New temperatures are calculated until the desired convergence is obtained. These temperatures are written as steady-state temperatures. Secondly, using the steadystate values as initial values, transient temperatures are calculated at thirty second time intervals.

These programs calculate the film heat transfer coefficient inside the coils, using the following equations for turbulent flow Eq. (2.6), for transition region, Eq. (2.7), for laminar flow, Eq. (2.9). They also make use of the Cumming-West's correlation [Table 2.2, Eq. 3], but geometry factor "1.40" to calculate outside film heat transfer coefficient.

In the "Single Tank" program, Equations 3.2 and 3.3 are used to calculate temperatures.

In the "Two Tanks in Series" program, Equations 3.6, 3.7, 3.8 and 3.9 are used to calculate unknown temperatures.

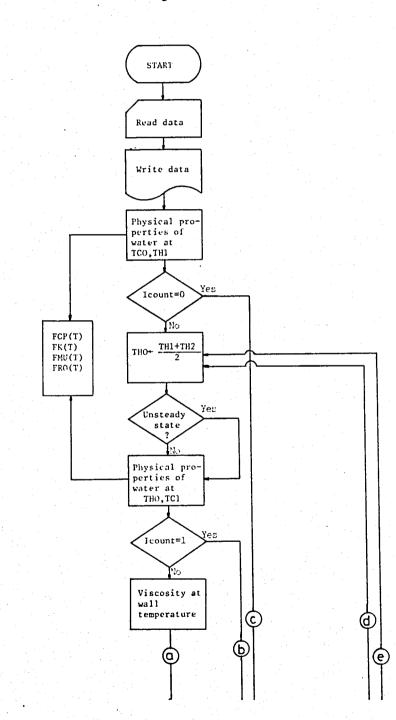
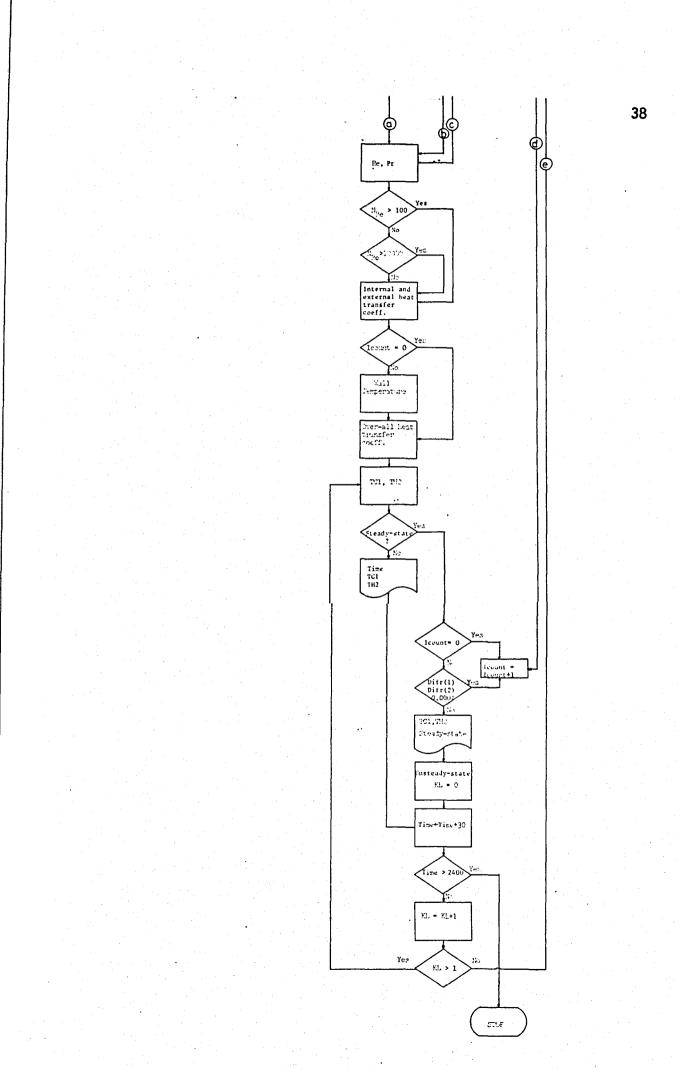
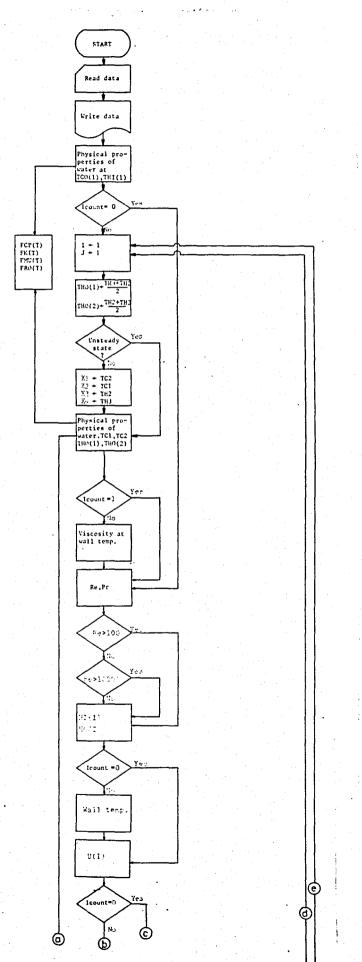


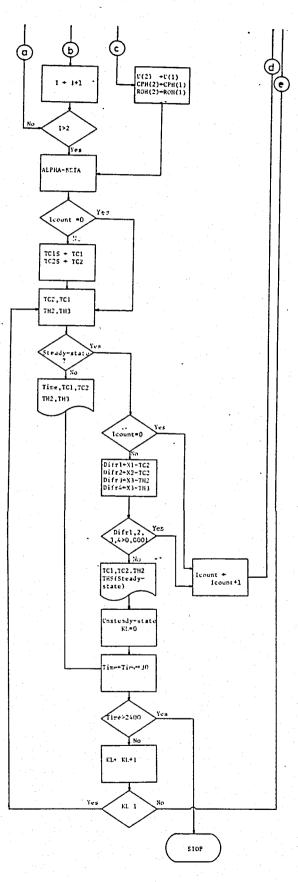
Figure 3-3.Flow chart of single tank.

37





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## IV. EXPERIMENTAL WORK

In this chapter experimental equipment and experimental procedure are described.

## 4.A EXPERIMENTAL EQUIPMENT

The experimental set-up used in this study consisted of two dimensionally similar tanks, coils, stirrers, a number of thermocouples, a digital thermometer, pumps, a hot and cold water reservoir and heaters as shown in Figure 4.1 for single tank and Figure 4.2 for two tanks in series. The connections were provided by specially insulated pipes.

### 4.A.1 Tanks

Each galvanized tank having  $20.0\pm0.1$  cm (7.9 in) inside diameter, was cylindrical, flat bottomed and capable of holding  $4.08\times10^3$  cm<sup>3</sup> (1.08 ga). Also in front of the each tank, there was a glass liquid level indicator pipe.

## 4.A.2 Coils

The coils were made of clean copper tubing and were fixed by means of corks to the lids of the tanks so that they stood firmly inside the vessels. The coiling was circular and the gap between individual turns of coils was sufficiently great to allow for free circulation of liquid over the complete surface of the tubing. Each gap was 0.635±0.1 cm (0.250 in). The copper tubes used were 0.635±0.01 cm (0.250 in) outer diameter and 0.470±0.01 cm (0.185 in) inside diameter.

#### 4.A.3 Stirrers

Agitation was supplied by flat blade turbine impeller,  $7.99\pm0.01$  cm (3.15 in) in diameter. Each impeller contained six blades, each being 1.99±0.01 cm (0.82 in) in width. The impellers were placed at a height of 6.0±0.2 cm (2.4 in) from the bottom of the vessel.

### 4.A.4 Thermocouples

Temperature measurements were measured using iron-constantan thermocouples, 0.10 cm in diameter. In order to minimize response time, thermocouple junctions were inserted into water without protecting tubes. In order to prevent thermocouples from the corrosive effect of water, first, all thermocouple junctions were cleaned with a kind of detergent, known as Lissapol NX, to remove oil and other dirts, then coated with nail lacquer.

Iron-constantan is the most widely used of all thermocouples.

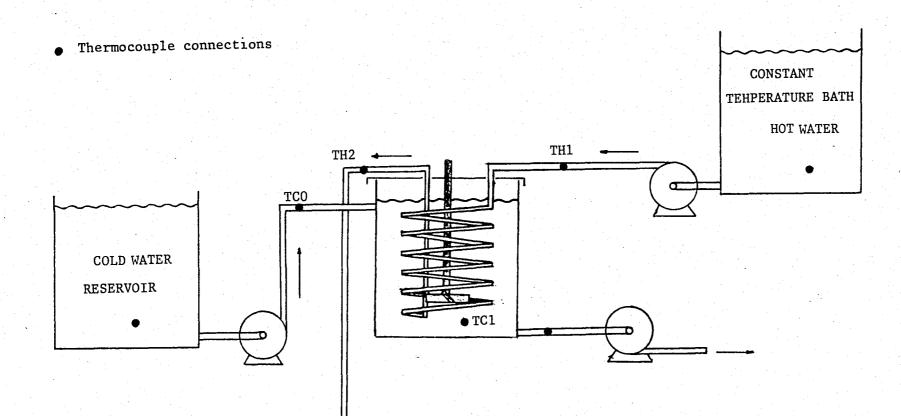


Figure 4-I.Experimental set up single tank.

43

5.5

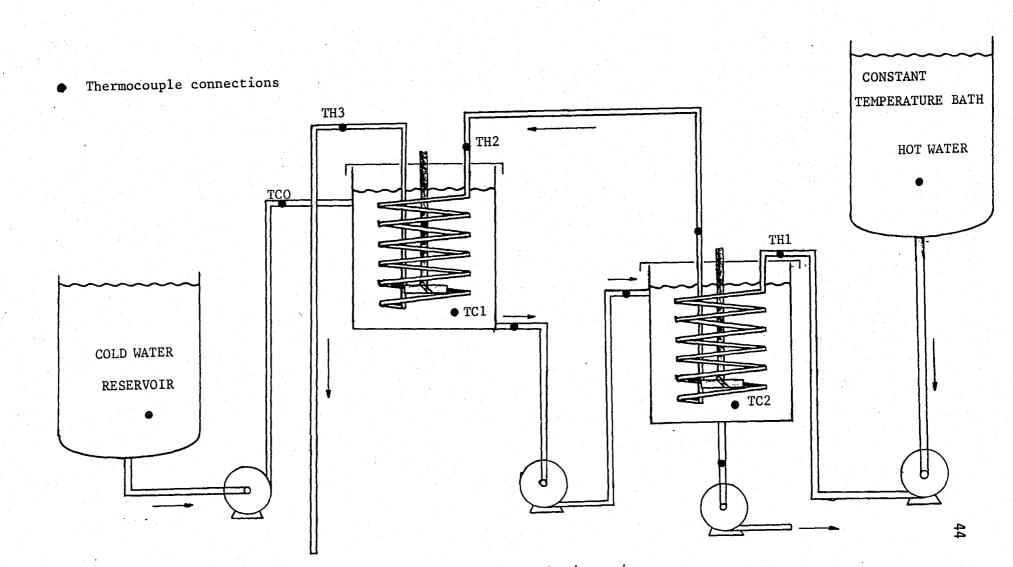


Figure 4-2. Experimental set up two tanks in series.

It gives a higher emf at a given temperature than any of the other common couples, and is the least expensive. Their usual temperature range of operation is -200 to  $750^{\circ}$ C (-300 to  $1400^{\circ}$ F).

4.A.5 Digital Thermometers

Temperature measurements were directly read out by means of a digital thermometer which had twelve separate thermocouple connections. Some technical specifications of this instruments are given below.

Line voltage	: 220 V AC ±15%, 48, 52 Hz
Operating temperature	: -10 to +50°C
Power consumption	: 4 VA
Measurement sensitivity	: ±0.1% (full scale)
Break protection	: If the sensor probes are open the
	display is blanked.

4.A.6 Hot Water Reservoir

The hot water reservoir was a rectangular galvanized tank, approximately 50 cm long, 25 cm wide and 50 cm high. It was insulated carefully with glass wool and two layers of nylon were wrapped around it.

### 4.A.7 Cold Water Reservoir

A 50 lt plastic container was used as cold water feed tank. In order to prevent variation of the head of pump connected to this container, liquid level in the container was kept constant throughout the experiment.

#### 4.A.8 Heaters

Two heating mantles were placed inside the hot water reservoir. One, had a resistance thermometer which could be set at a desired temperature (CENCO, 1010 W, 110 V), the other an electrical resistance (2500 W, 220 V). The temperature was also controlled by an inserted thermocouple.

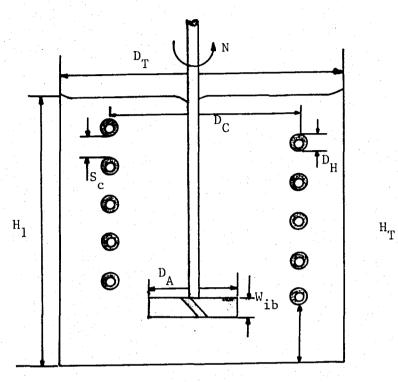
### 4.A.9 Rotary Pumps

Two types of rotary pump were used, a 50 W,200 U/min pump and a 90 W, 2000 U/min pump to feed hot and cold water streams.

Dimensions of the tank system used in the experimental work are shown in Figure 4.3.

## 4.B EXPERIMENTAL ARRANGEMENT

Two types of arrangements were used for the experiments. Firstly, single tank arrangements as shown in Figure 4.1 and secondly two tanks in series, Figure 4.2, are described below.



4.1

Figure 4-3. Dimensions of the tank system.

The single tank was insulated by glass wool first and wrapped by nylon sheets in order to prevent insulater from getting wet. Flat blade turbine impellers were mounted on the shaft coaxial with the tanks. The lower edge of each agitator was placed at a height of 6.0±0.2 cm (2.4 in) from the bottom of the vessel. Agitator speed was set at the desired speed by means of the speed button on a Heidolph stirrer whose stirring speed could be changed over a range of 45 to 2000 rpm. This stirring speed was checked using a revolution counter and also stop-watch for low speeds.

Inside the tanks clean copper coils where also mounted with their axis coincident with the axis of the vessels. Mean diameter of the coils were  $16.00\pm0.2$  cm (63 in) and height of each coil from the vessel bottom was  $2.0\pm0.1$  (0.8 in).

Tap water was fed to cold water reservoir. The level of the water in the reservoir was kept constant to prevent variation of the head to the feed pump. By manipulating the value on the exit side of the pump, water level in the tank and the cold water flowrate were adjusted. Also a continuous discharge was obtained by another pump at the exit of the tank.

Hot water was pumped from a constant temperature bath into the coil also by means of a rotary pump.

Directions of flow of the cold and hot water were opposite, i.e. counter-current. For all the runs reported a liquid depth of  $17.0\pm0.1 \text{ cm} (6.7 \text{ in})$  was used and the area of heating surface exposed the mixing liquid for each coil was 598.5 cm<sup>2</sup> (92.8 in<sup>2</sup>). Cold and hot water flowrates were measured by collecting water in a graduated

48

cylinder for periods of one or two minutes.

For two tanks in series, another dimensionally similar tank was connected to the previous single tank, the arrangement of the coil, stirrer and as well as water level in the tanks were kept same. As shown in Figure 4.2, outlet of the tank I was 50 cm higher from the inlet of the tank II to maintain the flow of cold water. Unfortunately, water level in the tanks changed with time. In order to prevent this situation, another pump was placed between two tanks. Again all connections were insulated carefully.

Figure 4.1 shows the thermocouple arrangements for the single tank. Seven thermocouples were used in this system. Three were installed into tank, hot water reservoir and cold water reservoir. And others were used to measure hot and cold water inlet and outlet temperatures. For two tanks in series twelve thermocouples were placed as shown in Figure 4.2. Four were installed into tank I and II, hot water and cold water reservoir. Others were used to measure hot and cold water inlet and outlet temperatures for tank I and tank II. All thermocouples were placed in small holes on the rubber connection pipes and sealed by liquid joint. Before putting thermocouples in their places, they were connected to the digital thermometer and controlled against one another in water at room temperature and at 60, 80°C and also checked with another digital thermometer.

No baffles were present in the vessel. However some baffling was obtained by the presence of the coil thermocouples and the stirrer.

All time measurements were taken with a stop-watch and at least three readings were taken for each measurement.

In this thesis study with the vessels having the configurations showing in Figure 4.3, the range of the parameters were as follows:

- 1. Reynolds number range,  $1100 < N_{Re} < 14000$  (for coil side)
- 2. Ratio of the agitator diameter to tank diameter,

 $D_{\Delta}/D_{T}$  : 0.400

3. Ratio of agitator height to tank diameter,

 $H_{\Delta}/D_{T}$  : 0.300

4. Ratio of impeller blade width to impeller diameter,

 $W_{ib}/D_{\Delta}$ : 0.200

5. Ratio of coil diameter to tank diameter,

 $D_{c}/D_{T}$  : 0.800

6. Ratio of coil length to tank diameter,

 $L_{c}/D_{T}$  : 0.700

7. Ratio of coil tube diameter to coil diameter,

 $D_{\rm H}/D_{\rm c}$  : 0.040

8. Ratio of the space between coils to coil tube diameter,

 $S_{c}/D_{H}$  : 1.00

9. Ratio of coil height from vessel bottom to tank diameter,

 $H_{C}/D_{T}$  : 0.10

#### 4.C EXPERIMENTAL PROCEDURE

The experiments were performed in two sections as steady-state and unsteady-state cases for both two tanks in series and a single tank. In this investigation a Reynolds number range of 1100 to 14000 was used for flow in the coil.

## 4.C.1 Steady-State Experiments

In the first part this set of experiments, data were taken without changing cold water conditions and agitator speed. Only hot water flowrates were varied at different hot water temperatures and the procedure was as given below,

- 1. The mixer was set at desired speed and turned on.
- Cold water was pumped into tank(s) at a certain flowrate. The water level in the tank(s) was kept constant and the same for both tanks.
- Hot water at a given temperature was fed to coil(s) by a pump at a given flowrate.
- Temperatures were recorded when steady-state conditions were reached.
- After reaching steady-state and adjusting the hot water flowrate to new values, new steady-state temperatures were reached and recorded.
- All these were repeated at different hot water temperatures.

For this part, the range of parameters are given in Table 4.1 and thirtyone runs were performed.

In the second part of the experiments, only agitator speed was changed for given constant hot water and cold water conditions. The complete procedure was,

- 1. The agitator was set at desired speed and turned on.
- Tap water from the reservoir was pumped into tank(s) at a given flowrate.
- Hot water at a given temperature was fed to coil(s) by a rotary pump at a given flowrate.
- Sufficient time was allowed for a steady-state conditions to be achieved before recording the temperatures.
- 5. Changing only agitator speed (four times) this procedure was repeated.

The range of parameters for this part are given in Table 4.2 and eight runswere performed.

TABLE 4.1 - Range of Parameters for Single Tank and Two Tanks in Series (Hot stream flowrate is variable)

Parameters	Single Tank	Two Tanks in Series
Agitator speed (rpm)	200	200
Hot water inlet temperature (°C)	83.5-60.3	80.8-60.4
Cold water inlet temperature (°C)	17.5- 8.9	17.4-12.5
Hot water flowrate (ml/sec)	27.5- 5.6	16.6- 5.7
Cold water flowrate (ml/sec)	22.5-18.5	24.0-28.0

Parameters	Single Tank	Two Tanks in Series
Agitator speed (rpm)	200-500	200-500
Hot water inlet temperature (°C)	75.6-75.0	74.5-70.3
Cold water inlet temperature (°C)	12.6-12.5	15.3-12.2
Hot water flowrate (ml/sec)	26.5-26.3	15.1-15.0
Cold water flowrate (ml/sec)	23.5-22.3	30.2-30.0

TABLE 4.2 - Range of Parameters for Single Tank and Two Tanks in Series (Agitator speed is variable)

#### 4.C.2 Unsteady-State Experiment

In this set of experiments, data were taken at given cold and hot water flowrates and agitator speed. Only the hot water inlet temperature was changed. Temperatures were recorded at every thirty seconds until steady-state was reached. The procedure given below was followed.

- 1. Agitator was set at the desired speed and turned on.
- Top water at a given temperature and flowrate was pumped into tank(s).
- Hot water at a given temperature was fed to coil(s) by pump at a given flowrate.
- 4. When steady-state was reached, temperatures were recorded. The hot water inlet temperature changed suddenly and the temperatures were recorded at every thirty seconds until new steady-state conditions were reached.

whom had held by the

For this part the range of parameters are given in Table 4.3 and six runs were performed.

TABLE 4.3 -	Range	of Parameters	for	Unsteady-State Runs
	•			-

SINGL	E TANK						
Set	Hot Water Inlet Temperature (°C)	Cold Water Inlet Temperature (°C)	Cold Water Flowrate (ml/sec)	Hot Water Flowrate (ml/sec)			
]	81.3-67.0	15.2	20.0	18.2			
2	69.0-59.8	13.7	20.0	20.0			
3	83.8-63.5	12.9	20.1	20.0			
TWO	TWO TANK IN SERIES						
1	81.0-69.3	12.5	22.0	15.0			
2	71.0-61.8	13.2	24.0	15.5			
3	82.7-60.9	13.7	28.5	13.5			
		 	I	·			

\*Agitator speed was kept at 200 rpm for all sets.

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## V. RESULTS

Within the context of the work, overall heat transfer coefficient, external and internal film heat transfer coefficients were determined. First a graphical method known as the Modified Wilson plot [29] was used to evaluate the heat transfer coefficients and secondly on iterative computer program for each system, both single tank and two tanks in series was developed calculating heat transfer coefficients from the empirical equations and also temperatures. Complete sets of experimental data are presented in Appendix I.

## 5.A EVALUATION OF EXPERIMENTAL RESULTS

Both steady-state and unsteady-state results are analyzed in the following sections separately.

## 5.A.1 Steady-State Results

The results of the steady-state experiments were studied in two sections.

- a) Determination of the overall heat transfer coefficient.
- b) Graphical estimation of internal and external film heat transfer coefficients.

#### 5.A.l.a Determination of the Overall Heat Transfer Coefficient

Overall heat transfer coefficients were calculated from the experimental data directly by using the fundamental equations given below:

$$U = \frac{Q}{A_o \Delta T_{lm}}$$
(5.1)

where,  $\Delta T_{gm}$  is the logarithmic mean temperature difference, driving force between tank liquid and heating water. Temperature profiles are shown in Figures 5.1 and 5.2 for both single tank and two tanks in series.

A<sub>o</sub>, heat transfer area of coil is the total wetted area, based on outside surface of the coil. The outside heat transfer area was calculated knowing the total wetted length and the outside diameter of the coil tube.

Q is the total heat transferred and was calculated on the basis of both the hot stream and the cold stream to check heat loss. Calculations were made using total heat based on hot stream.

## 5.A.1.b <u>Graphical Estimation of External and Internal Film</u> <u>Heat Transfer Coefficient</u>

Since only overall coefficients were measured directly, it was necessary to calculate internal and external coefficients from the data obtained. The results were interpreted by a graphical method similar to that proposed by Wilson [29]. This method is described in Section 2.D in detail. Figures for single tank are given through 5.3 to 5.6. for two tanks in series through 5.7 to 5.10.

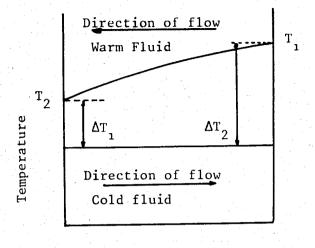
As shown in Section 2.D for case I, (i.e. conditions of flow and temperature inside the coil not changing appreciably, only agitator speed changing). External and Internal heat transfer coefficients were determined by the use of Eq. 2.13, i.e.

$$\frac{1}{U} = \beta + \frac{1}{K N^{0.62}}$$

and Eq. (2.14, i.e.

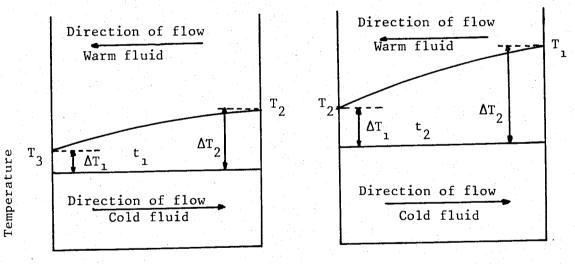
$$3 = \frac{1}{h_1} \frac{A_0}{A_1} + \frac{x}{k} \frac{A_0}{A_m} = \text{Constant}$$

Because of very thin wall of the coil tubing,  $A_i$ ,  $A_o$ , and  $A_m$ , the outside, inside and average areas respectively, were taken as being equal to each other. By plotting 1/U versus  $1/N^{0.62}$  a substantially straight line is obtained. To determine the value of the slope of the straight line, the method used is the least squares fit. Figure 5.3 shows the data for single tank, while Figure 5.7 represents the data for the two tanks in series. In this way from slope "1/K" the external film coefficient H0 = K N<sup>0.62</sup> was determined as a function of agitator speed, and, from intercept "B" the internal film coefficient HI =  $1/(\beta - (x/k))$ , was determined. Here x is thickness of the coil tubing and k, thermal conductivity of copper was taken as 385 W/m<sup>2</sup>oC [6]. In Figure 5.3 from the intercept the internal film coefficient was calculated as 6711 W/m<sup>2</sup>o<sup>c</sup>C, and from the slope the external film coefficient was calculated for runs 1 to 4 in Table 5.3. Using Figure 5.7 the internal film coefficients



71

Figure 5-1. Temperature profile for single tank.





TANK II

Figure 5-2. Temperature profile for two tanks in series.

for tank I  $3225 \text{ W/m}^{2} \text{°C}$ , for tank II  $4269 \text{ W/m}^{2} \text{°C}$  were found. These values are almost half of those of a single tank. The reason is that hot water flowrate in two tanks in series system is half of that of single tank. Results representing heat transfer coefficients are tabulated in Table 5.5 from run 21 to 24.

As described in Section 2.D for Case II (i.e. conditions of flow, temperature outside the coil and agitator speed not changing appreciably, and only water flowrate inside the coil (changing) external and internal heat transfer coefficients were determined by using Eq. 2.15. i.e.

$$\frac{1}{U} = \gamma + \frac{1}{K'V^{0} \cdot {}^8}$$

and Eq. 2.16, i.e.

 $\gamma = \frac{1}{HO} \frac{A_i}{A_o} + \frac{x}{k} \frac{A_i}{A_m}$  (constant)

by thin wall assumption (i.e.  $A_i = A_0 = A_m$ ). A plot of 1/U versus  $1/V^{0.8}$  yields a straight line with an intercept of at the ordinate  $\gamma$  and a slope of 1/K'. Again the least square fit was used to determine the value of the slope of the straight line. In this way from the slope, the internal film coefficient HI =  $K'V^{0.8}$  was calculated as function of hot stream velocity. From the intercept, the external film coefficient, HO =  $1/(\gamma - (x/k))$  was determined as a constant. For each set at least five runs were performed. For the single tank system, the hot water flowrate inside the coil changed from 27.5 ml/sec to 5.7 ml/sec, while the hot water inlet temperature

was varied as 81.4, 73.0 and 61.8°C respectively for each set. For two tanks in series, the hot water flowrate was varied from 16.6 ml/sec to 5.7 ml/sec and hot water inlet temperature changed as 80.5, 71.5 and 60.9°C for each set.

In single tank experiments represented in Figure 5.4, (runs 5 to 10) from the intercept, the external film coefficient was found as 4255 W/m<sup>2</sup>°C (749 Btu/hr.sq.ft) at N<sub>Re</sub>= 29.000 (average in agitated side). From the slope the internal film coefficient was calculated to be 2928 W/m<sup>2</sup>°C (516 Btu/hr.sq ft.°F - at a water velocity of one foot per second), at N<sub>Re</sub> = 4112 (inside the coil). As, for Figure 5.5 (runs 11 to 15) at N<sub>Re</sub> = 26.000 (agitated side) the external film heat transfer coefficient is 4219 W/m<sup>2</sup>°C (743 Btu/hr.sq ft.°F), and the internal film heat transfer coefficient, at N<sub>Re</sub> = 3761 (inside the coil) is 2916 W/m<sup>2</sup>°C (513 Btu/hr.sq ft.°F) at a water velocity of one foot per second. For Figure 5.6 (runs 16 to 20) the external film coefficient, at N<sub>Re</sub> = 24.000 (agitated side) is 4016 W/m<sup>2</sup>°C (707 Btu/hr.sq ft.°F) and the internal film coefficient. at N<sub>Re</sub> = 3165 (inside the coil), is 2610 W/m<sup>2</sup>°C (460 Btu/hr.sq ft.°F - at a water velocity of one foot per second). All results for the single tank are given in Table 5.4.

For two tanks in series (Figure 5.8 - runs 25 to 29) for tank I, at  $N_{Re} = 27.000$  (agitated side) external film heat transfer coefficient is 4266 W/m<sup>2</sup>°C (751 Btu/hr.sq ft.°F) and internal film coefficient at  $N_{Re} = 2736$  (inside the coil) is, 2720 W/m<sup>2</sup>°C (479 Btu/hr.sq ft.°F) at a water velocity one foot per second. For tank II, at  $N_{Re} = 38.100$ (agitated side) external film heat transfer coefficient is 4830 W/m<sup>2</sup>°C (850 Btu/hr.sq ft.°F) and internal film coefficient at  $N_{Re} = 4353$  (inside the coil), is 2760 W/m<sup>2</sup>°C (486 Btu/hr.sq ft.°F) at water velocity one foot per second. All results for two tanks in series are given in Table 5.6 and also are represented on Figures 5.9 and 5.10. The speed of the agitator remained constant at 200 rpm for both single tank and and two tanks in series.

Also to observe the effect of agitator speed on external film heat transfer coefficient. Logarithmic plot of agitator speed vs external film coefficient was plotted in Figure 5.11 for two tanks in series. Figure 5.12 for single tank. First this data were fitted a straight line, slopes were found 0.60 for single tank, for two tanks in series, tank I and tank II 0.59, 0.62 respectively. But when these data were fitted a second order polinomial, (i.e.  $\ln HO = A + BlnN +$  $C(lnN)^2$ ) coefficient of the second order variable was found unexpectedly 0.33 for single tank, for two tanks in series -0.26, -0.31 for tank I and tank II respectively. So, as a result it is impossible to observe asymptotic values for agitator speed from these data.

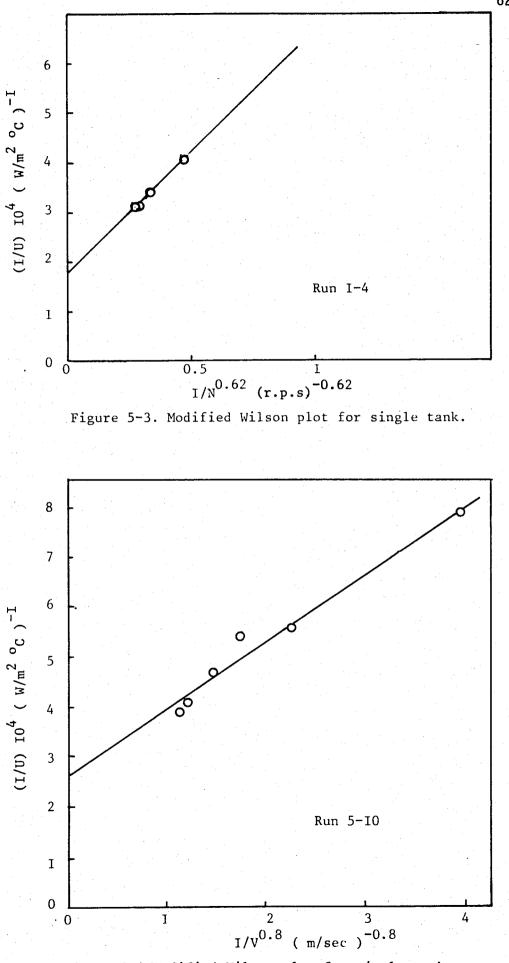
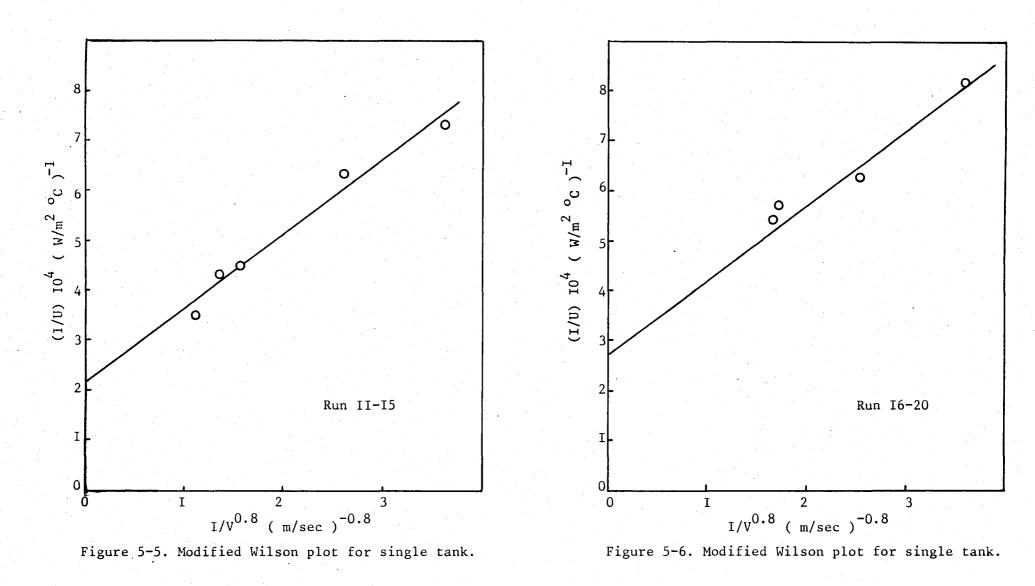
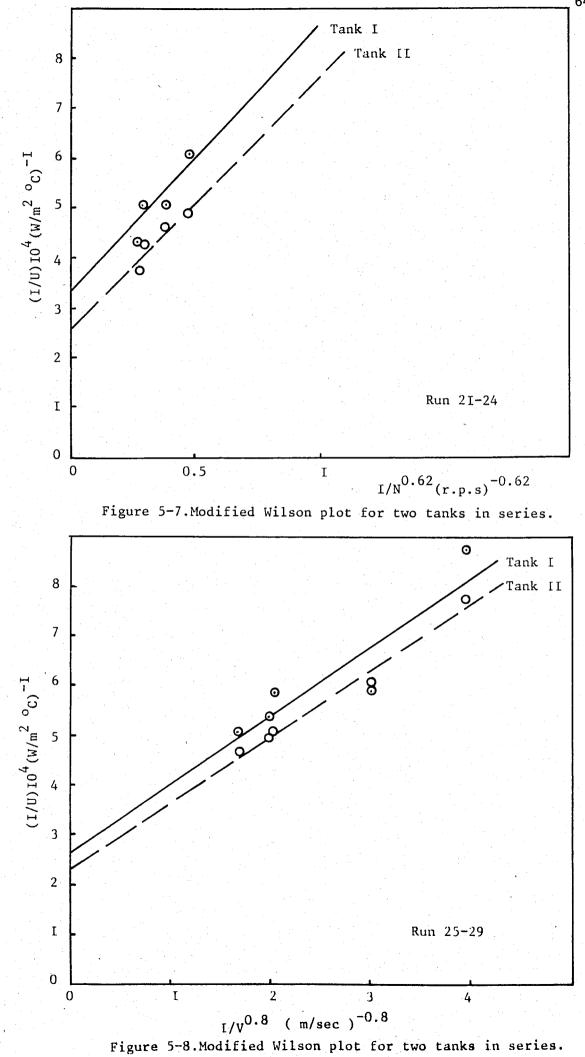
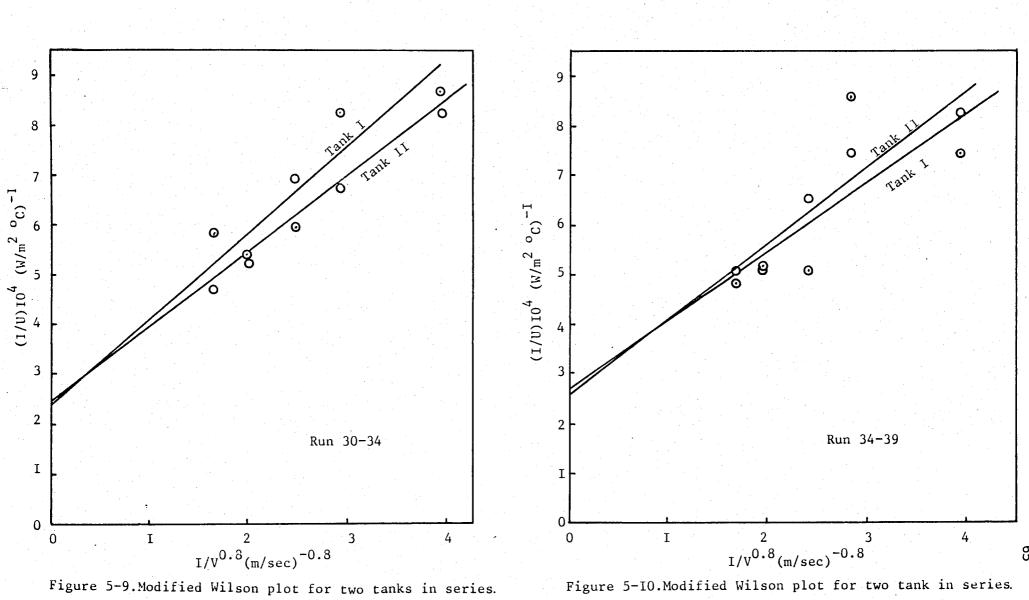


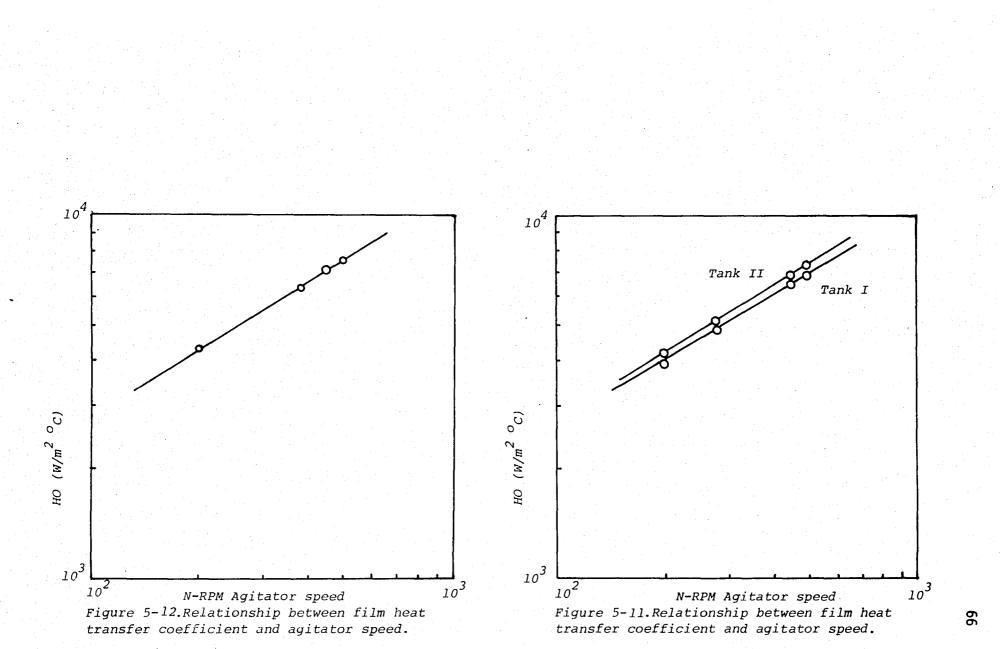
Figure 5-4. Modified Wilson plot for single tank.







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TABLE 5.1 - Sample Temperature - Time Data - Observed

N : W :	200 rpm 15.0 ml 22.0 ml 12.5°C	i /sec	in series	) · · · · · · · · · · · · · · · · · · ·
	TCIS	TC2S	TH2S	TH3S
	26.0	45.9	49.6	29.8
T(0(2))	12.5°C			

TCO(2): 12.5°C TH1(2): 69.3°C

Time (sec)	TC1 °C	TC2 °C	TH2 °C	TH3 °C	ALPHA*	BETA**
30 60 90 120 150 180 210 240 270 300 330 360 390 420 450 450 480 510 540 570 600 630	25.8 25.6 25.4 25.3 25.2 25.0 24.9 24.8 24.7 24.6 24.5 24.4 24.5 24.4 24.3 24.2 24.0 23.9 23.9 23.8 23.7 23.7 23.7	45.1 44.5 43.9 43.6 43.1 42.7 42.5 42.2 42.0 41.8 41.6 41.5 41.3 41.1 41.1 41.1 41.0 40.9 40.9 40.9 40.7 40.7	47.8 47.1 46.8 46.6 46.1 45.8 45.7 45.5 45.3 45.3 45.3 45.1 45.0 45.0 45.0 45.0 45.0 44.8 44.7 44.6 44.5 44.4 44.3 44.3	29.3 29.0 28.8 28.6 28.5 28.3 28.1 28.0 27.9 27.7 27.6 27.5 27.4 27.3 27.3 27.2 27.2 27.2 27.1 27.1 27.1 27.1 27.1	0.159 0.158 0.159 0.155 0.155 0.158 0.159 0.154 0.155 0.155 0.155 0.155 0.150 0.151 0.151 0.150 0.151 0.159 0.159 0.159 0.159 0.159 0.165	0.112 0.105 0.114 0.116 0.114 0.116 0.119 0.121 0.120 0.127 0.126 0.125 0.132 0.137 0.127 0.127 0.130 0.126 0.126 0.125 0.125
660	23.7	40.7	44.3	27.1	0.165	0.125

\* <u>TH3 - TC1</u> TH2 - TC1

\*\* <u>TH2 - TC2</u> TH1(2) - TC2

 $\mathbf{H}$ 

TABLE 5	.2 - Sam	ple Temp	erature	- Time Da	ata - C
Set No: N : W : M : TCO(1): TH1(1):	1 200 rp 15.0 m 22.0 m 12.5°C 81.0°C	l/sec l/sec			
	TC1S	TC2S	TH2S	TH3	S
	26.23	46.64	50.8	34 30.	55
TCO(2): TH2(2):	12.5°C 69.3°C				
Time <u>(sec)</u>	TC1 °C	TC2 °C	TH2 °C	TH3 °C	
30 60 90 120 150 180 210 240 270 300 330 360 390 420 450 480 510 540 570 600 630 660 630 660 690 720 750 750 780 810 840 870 900 930 960 990	$\begin{array}{c} 26.09\\ 25.92\\ 25.74\\ 25.56\\ 25.38\\ 25.22\\ 25.07\\ 24.94\\ 24.81\\ 24.71\\ 24.61\\ 24.53\\ 24.45\\ 24.39\\ 24.33\\ 24.28\\ 24.24\\ 24.20\\ 24.17\\ 24.14\\ 24.20\\ 24.17\\ 24.14\\ 24.20\\ 24.17\\ 24.14\\ 24.20\\ 24.02\\ 24.01\\ 24.02\\ 23.99\\ 23.99\\ 23.98\end{array}$	$\begin{array}{r} 45.63\\ 44.83\\ 44.19\\ 43.66\\ 43.23\\ 42.87\\ 42.58\\ 42.32\\ 42.11\\ 41.93\\ 41.78\\ 41.65\\ 41.53\\ 41.65\\ 41.53\\ 41.28\\ 41.22\\ 41.17\\ 41.22\\ 41.08\\ 41.04\\ 41.01\\ 40.99\\ 40.96\\ 40.91\\ 40.90\\ 40.91\\ 40.90\\ 40.91\\ 40.90\\ 40.88\\ 40.87\\ 40.86\\ 40.86\\ 40.86\\ \end{array}$	$\begin{array}{r} 48.64\\ 47.94\\ 47.38\\ 46.92\\ 46.55\\ 46.24\\ 45.98\\ 45.76\\ 45.57\\ 45.41\\ 45.28\\ 45.17\\ 45.07\\ 45.07\\ 45.07\\ 45.07\\ 44.98\\ 44.91\\ 44.85\\ 44.91\\ 44.85\\ 44.79\\ 44.61\\ 44.67\\ 44.61\\ 44.67\\ 44.61\\ 44.59\\ 44.57\\ 44.55\\ 44.51\\ 44.52\\ 44.51\\ 44.52\\ 44.51\\ 44.48\\ 44.48\\ 44.48\\ 44.48\end{array}$	30.04 29.78 29.53 29.30 29.09 28.90 28.73 28.58 28.45 28.34 28.23 28.14 28.07 27.89 27.89 27.89 27.80 27.71 27.69 27.67 27.67 27.67 27.61 27.60 27.59 27.58 27.58	

Calculated

# TABLE 5.2 - Continued..

Time <u>(sec)</u>	TC1 _°C	TC2 °C	TH2 °C	TH3 °C
1020 1050 1080 1110 1140 1170 1200 1230 1260 1290 1320 1350 1380 1410	23.98 23.97 23.97 23.97 23.97 23.97 23.96 23.96 23.96 23.96 23.96 23.96 23.96 23.96	40.85 40.85 40.84 40.84 40.83 40.83 40.83 40.83 40.83 40.83 40.83 40.82 40.82 40.82	44.47 44.46 44.46 44.46 44.45 44.45 44.45 44.45 44.45 44.45 44.45 44.45 44.45	27.57 27.56 27.56 27.56 27.56 27.56 27.55 27.55 27.55 27.55 27.55 27.55 27.55 27.55
				· · · · · · · · · · · · · · · · · · ·

Run Agitated Speed (rpm)	Agitated	Experimental Re	sults	Computed Results			
	HI HO (W/m <sup>2</sup> °C) (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)	HI HO (W/m <sup>2</sup> °C) (W/m <sup>2</sup> °C) (W	U /m <sup>2</sup> °C)			
1	200	6711 4334	2465	6788 4692	2588		
2	380	6711 6365	3011	6738 6983	3148		
3	450	6711 7023	3151	6676 7741	3278		
4	500	6711 7544	3252	6674 8225	3362		
	ļ	<b>_</b>		<u>↓ ····································</u>			

<u>171.1.25, 1512 5.5176</u> (\*\*\*\*\*\*\*

TABLE 5.3 - Comparison of Experimental Results and Computed Results (SINGLE TANK) - Agitator speed was changed -

\* Hot water flowrate (ml/sec) : 26.5

\* Hot water inlet temperature (°C) : 75.5

\* Cold water inlet temperature (°C): 12.6

\* Cold water flowrate (ml/sec) : 22.7

	Hot Stream	Ex	perimental Re	sults	Computed Results		
Run		HI (W/m <sup>2</sup> °C)	HO (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)	HI (W/m <sup>2</sup> °C)	HO (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)
5	27.5	6663	4255	2432	7395	4892	2735
6	25.0	6219	4255	2370	6700	4801	2607
7	19.8	5077	4255	2183	5397	4662	2349
8	16.0	4289	4255	2020	4530	4616	2158
9	11.2	3302	4255	1773	3284	4452	1801
10	5.6	1894	4255	1267	1763	4164	1200

TABLE 5.4 - Comparison of Experimental Results and Computed Results (SINGLE TANK) - Hot stream flow rate was changed -

\*Agitator speed (rpm) : 200.0

\*Hot water inlet temperature (°C) : 81.4

\*Cold water inlet temperature (°C): 10.5

\*Cold water flowrate (ml/sec) : 19.5

TABLE 5.4 - Continued..

Dum	Run Flowrate (m1/sec)	Exp	Experimental Results			Computed Results			
Run		HI (W/m <sup>2</sup> °C)	HO (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)	HI (W/m <sup>2</sup> °C)	HO (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)		
11	27.5	6663	4219	2420	6988	4738	2630		
12	21.5	5487	4219	2246	5693	4726	2420		
13	18.5	4815	4219	2124	4932	4613	2244		
14	9.6	2870	4219	1635	2744	4303	1605		
15	6.3	2062	4219	1337	1909	4225	1274		

*Agitator Speed (rpm) :	200.0
*Hot water inlet temperature (°C) :	73.0
*Cold water inlet temperature (°C):	15.9
*Cold water flowrate (ml/sec) :	20.3

TABLE 5.4 - Continued..

Run Flowrate (ml/sec)	Ext	perimental Re	sults	Computed Results			
	HI (W/m <sup>2</sup> °C)	HO (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)	HI (W/m <sup>2</sup> °C)	HO (W/m <sup>2</sup> °C)	U (W/m <sup>2</sup> °C)	
16	27.1	5880	4016	2247	6379	4490	2464
17	19.0	4003	4016	1905	4658	4372	2130
18	16.1	3063	4016	1662	4063	4355	1993
19	10.0	2637	4016	1529	2613	4129	1536
20	6.4	1851	4016	1226	1774	4006	1192

\*Agitated Speed (rpm) : 200.0

- \*Hot water inlet temperature (°C) : 61.8
- \*Cold water inlet temperature (°C): 10.5
- \*Cold water flowrate (ml/sec) : 19.7

			ЕХРІ	ERIMENTAL	RESULTS		
		7	ANK I			TANK II	
Run	Agitator Speed (rpm)	HI <sub>l</sub> (W/m <sup>2</sup> °C)	HO <sub>l</sub> (W/m <sup>2</sup> °C)	U <sub>l</sub> (W/m <sup>2</sup> °C)	HI <sub>2</sub> (W/m <sup>2</sup> °C)	HO <sub>2</sub> (W/m <sup>2</sup> °C)	U <sub>2</sub> (W/m <sup>2</sup> °C)
21	200	3225	3962	1699	4269	4194	2005
22	280	3225	4901	1851	4269	5188	2208
23	450	3225	6421	2033	4269	6799	2455
24	500	3225	6897	2079	4269	7302	2517
			C	OMPUTED F	ESULTS		
21	200	3246	3960	1705	4202	4247	2002
22	280	3250	4890	1858	4225	5289	2213
23	450	3226	6540	2045	4127	7098	2443
24	500	3243	7059	2100	4125	7646	2505

TABLE 5.5 - Comparison of Experimental Results and Results from the Computer (TWO TANKS IN SERIES) - Agitated speed was changed -

Hot stream flowrate (ml/sec): 15.0\*Cold stream flowrate (ml/sec): 30.0\*Cold water inlet temperature (ml/sec): 13.4\*Hot water inlet temperature (ml/sec): 72.3

			EXPER	IMENTAL R	ESULTS		
<u>.</u>			TANK I			TANK II	
Run	Hot Stream Flowrate (ml/sec)	<sup>HI</sup> 1 (W/m <sup>2</sup> °C)	HO <sub>l</sub> (W/m <sup>2</sup> °C)	U <sub>l</sub> (W/m <sup>2</sup> °C)	HI <sub>2</sub> (W/m <sup>2</sup> °C)	HO <sub>2</sub> (W/m <sup>2</sup> °C)	U <sub>2</sub> (W/m <sup>2</sup> °C)
25	16.6	4257	4266,	2019	4536	4830	2206
26	13.5	3605	4266	1859	3826	4830	2024
27	13.1	3464	4266	1821	3713	4830	1991
28	8.0	2332	4266	1450	2500	4830	1580
29	5.7	1794	4266	1222	1922	4830	1328
			СОМ	PUTED RE	SULTS	<u></u>	
25	16.6	3940	4265	1945	4813	4853	2274
26	13.5	3207	4167	1731	3999	4742	2054
27	13.1	3123	4160	1705	3908	4737	2028
28	8.0	1933	3983	1259	2482	4455	1530
29	5.7	1415	3910	1011	1835	4303	1245

TABLE 5.6 - Comparison of Experimental Results and Computed Results (TWO TANKS IN SERIES) - Hot stream flowrate was changed -

\*Cold water inlet temperature (°C): 17.5

\*Cold water flowrate (ml/sec) : 24.0

TABLE 5.6 - Continued...

		EXI	PERIMEN	TAL RES	ULTS	<u> </u>	
			TANK I			TANK II	
Run •	Hot Water Flowrate (ml/sec)	HI <sub>l</sub> (W/m <sup>2</sup> °C)	HO <sub>l</sub> (W/m <sup>2</sup> °C)	U <sub>l</sub> (W/m <sup>2</sup> °C)	HI <sub>2</sub> (W/m <sup>2</sup> °C)	HO <sub>2</sub> (W/m <sup>2</sup> °C)	U <sub>2</sub> (W/m <sup>2</sup> °C)
30	16.2	3444	4386	1837	3992	4587	2022
31	13.4	2858	4386	1656	3313	4587	1832
32	10.2	2305	4386	1454	2672	4587	1617
33	8.2	1951	4386	1305	2261	4587	1457
34	5.7	1451	4386	1060	1682	4587	1192
	<b>.</b>	C 0	MPUTED	RESULT	S		
30	16.2	3605	4099	1827	4381	4589	2118
31	13.4	3009	4030	1649	3732	4518	1941
32	10.2	2314	3945	1405	2895	4374	1667
33	8.2	1890	3894	1231	2383	4281	1473
34	5.7	1364	3839	981	1729	4151	1183

\*Agitator speed (rpm) : 200.0

\*Hot water inlet temperature (°C): 71.5

\*Cold water inlet temperature(°C): 15.9

\*Cold water flowrate (ml/sec) : 28.0

TABLE 5.6 - Continued..

	••••••••••••••••••••••••••••••••••••••	Е Х	PERIMEN	TAL RES	SULTS		
			TANK I			TANK II	
Run	Hot water Flowrate (ml/sec)	HI <sub>l</sub> (W/m <sup>2</sup> °C)	HO <sub>l</sub> (W/m <sup>2</sup> °C)	U <sub>l</sub> (W/m <sup>2</sup> °C)	HI <sub>2</sub> (W/m <sup>2</sup> °C)	HO <sub>2</sub> (W/m <sup>2</sup> °C)	U <sub>2</sub> (W/m <sup>2</sup> °C)
35	16.5	4226	4132	1981	3865	4329	2068
36	13.5	3644	4132	1843	3313	4329	1843
37	10.4	2951	4132	1647	2683	4329	1561
38	8.6	2506	4132	1497	2278	4329	1369
39	5.7	1802	4132	1215	1648	4329	1041
		C	ОМРИТЕР	RESULT	ΓS		
35	16.5	3469	3968	1766	4137	4394	2020
36	13.5	2884	3920	1593	3487	4333	1840
37	10.4	2256	3849	1372	2755	4222	1598
38	8.6	1892	3805	1223	2322	4145	1433
39	5.7	1312	3748	949	1616	4014	1118

\*Hot water inlet temperature (°C) : 60.9

\*Cold water inlet temperature (°C): 12.8

\*Average values.

#### 5.A.2 Unsteady State Results

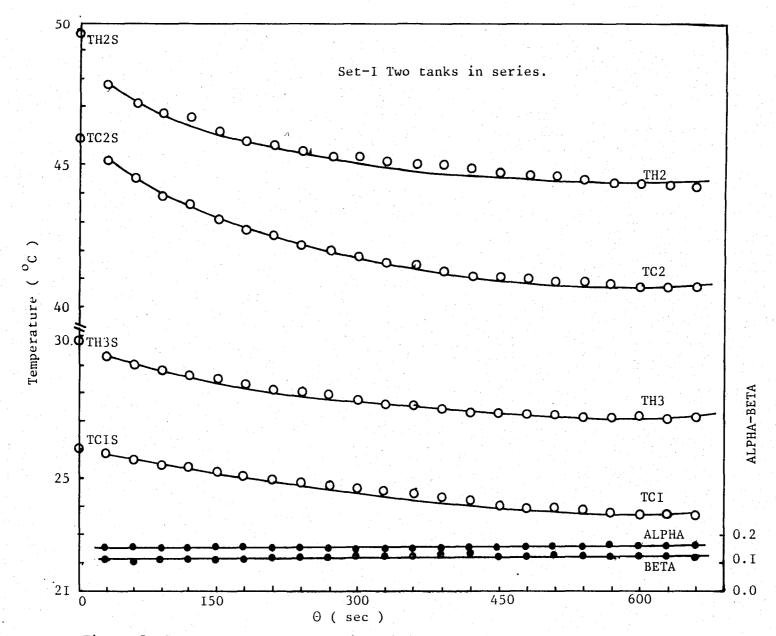
From the transient data obtained, the change in temperature with time was plotted in Figure 5.13 and in Appendix I, in Figures A.I.1 through A.I.5. And for single tank time versus Alpha =  $\frac{TH2-TC1}{TH1-TC1}$  and for two tanks in series time versus Alpha =  $\frac{TH3-TC2}{TH2-TC2}$  as well as time versus Beta =  $\frac{TH2-TC2}{TH1-TC2}$  were plotted to check whether, the overall coefficient remains constant. At the same time Alpha<sub>1</sub> is given by Alpha<sub>1</sub> = exp[-U<sub>A</sub>/WC<sub>h</sub> $\rho_h$ ] and Alpha and Beta are equal to Alpha =  $exp[-U_1A/WC_h\rho_h]$ , Beta =  $exp[-U_2A/WC_h\rho_h]$ . For example single tank set 2 average Alpha<sub>1</sub> was found to be 0.198 and is shown in Figure A.I.2 and in Table A.I.6 (Appendix I). Two tanks in series for set 1 Alpha = 0.157 and Beta = 0.122 values were calculated and also are tabulated in Table 5.1 and shown on Figure 5.13.

Time constants were determined for both single tank and two tanks in series. The value of tank temperature (TC1, TC2) reaches 63.2 percent of its ultimate value when the time elapsed is equal to one time constant [32]. For example considering two tanks in series and experiment set 1

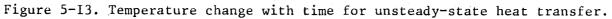
for tank I

 $TC_0 = 26.0^{\circ}C$  (value of tank temperature at the beginning)  $TC_{\infty} = 23.7^{\circ}C$  (ultimate value of the tank temperature)  $TCl_{\tau_1} = 24.5^{\circ}C$  (value of tank temperature after one time constant elapsed)

from Figure 5.13 time constant was found to be  $\tau_1 = 330$  sec.



2.1



1.15 T 11.7 T X

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For tank II

 $TC2_{0} = 45.9^{\circ}C$  $TC2_{\infty} = 40.7^{\circ}C$  $TC2_{\tau_{2}} = 42.6^{\circ}C$ 

again from Figure 5.13 time constant was found as  $\tau_2$  = 195 sec.

From computed temperatures for the same set;

Tank I

and

$$TCl_{0} = 26.23^{\circ}C$$
  
 $TCl_{\infty} = 23.96^{\circ}C$   
 $TCl_{\tau_{1}} = 24.75^{\circ}C$   
 $\tau_{1} = 288 \text{ sec}$ 

Tank II

and

and

 $TC2_{o} = 46.64^{\circ}C$   $TC2_{\infty} = 40.82^{\circ}C$   $TC2_{\tau_{2}} = 42.96^{\circ}C$  $\tau_{2} = 175 \text{ sec}$ 

Tabulated results are presented in Tables 5.7 and 5.8. Other tables and graphs for unsteady-state sets are given in Appendix I.

TABLE	5.7	-	Time	Constant	-	Single	Tank
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Set No.	Observed				Computed			
	TC1 °C <sup>0</sup>	TC1 °C <sup>∞</sup>	τc1 °C <sup>τ</sup>	τ (sec)	TC1 °C <sup>0</sup>	TC1 °C <sup>∞</sup>	ΤC1 °C <sup>τ</sup>	τ (sec)
1	43.4	36.9	39.3	99	43.37	37.09	39.40	116
2	38.7	33.8	35.6	110	38.31	34.16	35.70	113
3	44.7	35.6	38.9	108	44.86	35.62	39.02	112

- $TCl_0 = Value of tank temperature at beginning$
- $\text{TCl}_{\infty}$  = Ultimate value of tank temperature
- $\text{TCl}_{\tau}$  = Value of tank temperature after one time constant elapsed
  - $\tau$  = Time constant

	· · · · · · · · · · · · · · · · · · ·			BSE	RVED	<u> </u>		
	TA	NK I			Т	ANK II		
Set No.	TC1 °C <sup>0</sup>	TC1 °C <sup>∞</sup>	τc1 °C <sup>τ</sup>	τ <sub>ι</sub> sec	TC2 °C <sup>o</sup>	TC2 °C <sup>∞</sup>	τc2 °C <sup>τ</sup>	τ <sub>2</sub> Sec
1	26.0	23.7	24.5	330	45.9	40.7	42.6	195
2	24.2	22.3	23.0	240	40.6	36.4	37.9	225
3	22.8	19.8	20.9	230	39.8	31.6	34.6	160
		·. · ·		СОМР	UTED		· · · · ·	
1	26.23	23.96	24.75	288	46.64	40.82	42.96	175
2	24.08	22.40	23.02	245	40.70	96.36	37.92	150
3	22.55	19.81	20.81	233	40.37	31.83	34.97	145

TABLE 5.8 - Time Constants - Two Tank in Series

 $TCl_0$ ,  $TC2_0$  = values of tank I and tank II temperatures at the beginning  $TCl_{\infty}$ ,  $TC2_{\infty}$  = ultimate values of tank I and tank II temperatures  $TCl_{\tau_1}$ ,  $TC2_{\tau_2}$  = value of tank I and tank II temperatures after one time constant elapsed.

 $\tau_1, \tau_2$  = time constants for tank I and tank II, respectively.

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#### 5.B COMPUTER ANALYSIS OF DATA

Two iterative computer programs for both single tank and two tanks in series were developed. Calculations were made in three stages. In contrast to the approach used for experimental calculations, first external and internal film heat transfer coefficients (Chapter 2.B, Eq. 2.6, 2.7, 2.9) were calculated using the empirical equations. Since a dimensionally similar system to Cumming-West [19]; was used to calculate external film heat transfer coefficient, Cumming-West's empirical approach was chosen

(i.e. H0:1.01(K/D<sub>T</sub>)(ND<sub>A</sub><sup>2</sup> $\rho/\mu$ )<sup>0.62</sup>(C<sub>p</sub> $\mu/K$ )<sup>1/3</sup>( $\mu/\mu_W$ )<sup>0.14</sup>).

Secondly overall heat transfer coefficient as calculated by combining the three series resistances (inside coefficient, tube wall, and outside coefficient). Eq. 2.2 was used neglecting fouling factors and with thin wall assumption (i.e.  $1/U = (1/HI)+(1/HO)+(x/K_w)$ ), and the third stage deals with calculation of temperatures for both steadystate and unsteady-state cases.

Comparison of computed temperatures and experimentally observed temperatures for steady case are given in Appendix I and Tables A.I.1 through A.I.4.

Also computed external, internal and overall heat transfer coefficients with those of graphically found from the experimental data are given in Tables 5.3 through 5.6.

As an example of unsteady state case computed temperaturetime results are given in Table 5.2.

For example steady-state case for single tank in Table 5.4 (run: 9) at  $N_{Re}$ : 29110 (agitated side) external film coefficient was

calculated to be W/m<sup>2</sup>°C (784 Btu/hr.sq ft.°F) and internal film coefficient at  $N_{Re}$  = 4800 (inside the coil) is 3284 W/m<sup>2</sup>°C (578 Btu/hr.sq ft.°F).

Thermal resistance of tube, in Eq. 2.2 is  $x/K_w$ . The thermal conductivities of the tube wall was taken as 386 W/m °C [6] for copper.

The physical properties of water (C  $_p,\,\mu,\,k,\,\rho)$  as a function of temperature as given in Appendix IV.

## VI. DISCUSSION

### 6.A A DISCUSSION OF RESULTS AND FURTHER SUGGESTIONS

The system used in this investigation is a system that is dimensionally similar to that used by Cumming-West [19].

Cumming-West's correlation for heating and cooling fluids in a vessel with a single coil as heat transfer surface is given in Table 2.2, i.e. HO =  $1.01(D_T)^{-1}k(D_A^{2}N\rho/\mu)^{0.62}(C_p\mu/k)^{0.33}(\mu/\mu_w)^{0.14}$ .

The individual film heat transfer coefficients were calculated from the experimental overall coefficients by using the Modified-Wilson graphical method and two computer programs were developed for computing film heat transfer coefficients and temperatures. These computed values were compared with the values obtained experimentally. It was reconfirmed that using the geometry factor "1.40", as pointed out by Ackley [24] instead of "1.01" gave better results as shown by the sample calculation in Table 6.1. The calculated values of "HO" from the correlation agree within experimental error with the value of "HO" from Wilson's graphical method. Hence the method of interpretation used appears to be sound for the data obtained. Film heat transfer coefficients for the various conditions of testing and for the steady-state case are given for both the single tank and two tanks in series in Tables 5.3 through 5.6.

				·							
Geome	Geometry factor: 1.01										
Run	TC1 °C	TC2 °C	HI W/m <sup>2°</sup> C	HO W/m2°C	U W/m2°C						
11	43.04	52.70	7031	3410	2167						
12	43.48	51.05	5725	3405	2023						
13	39.77	47.15	4965	3328	1894						
14	30.24	35.36	2766	3115	1411						
15	28.69	31.80	1922	3060	1145						
Geome	try facto	or: 1.40									
11	44.42	51.64	6088	4738	2492						
12	44.57	50.15	5693	4726	2420						
13	40.70	46.15	4932	4613	2244						
14	30.68	34.46	2744	4303	1605						
15	28.86	31.18	1909	4225	1274						
Exper	imental	results									
11	44.8	51.2	6663	4219	2420						
12	44.1	50.3	5487	4219	2246						
13	40.7	46.3	4815	4219	2124						
14	30.9	34.9	2870	4219	1635						
15	29.0	30.9	2062	4219	1337						

TABLE 6.1 - Comparison of Geometry Factors With Experimental Results (SINGLE TANK)

The model for heat transfer in two consecutive stirred tanks give result which are in agreement with experimental results.

Unsteady state experimental results are presented in Table 5.1 and Figure 5.13 and in Appendix I, Tables A.I.5 through A.I.9 and on Figures A.I.1 through A.I.5. As shown in the figures for single tank Alpha<sub>1</sub> and for two tanks in series Alpha and Beta remained almost constant, i.e. the convective heat transfer coefficient for unsteady-state conditions has the same value as for steady-state conditions.

The values of the individual film heat transfer coefficients found in this study were in agreement with the previously published values [17] and [19] for similar cases as shown in Figure 6.1.

In this investigation, a Reynolds Number range of 19,439 to 76,727 has been used for the agitated side, which is a region that has not been investigated much until today. In the coil side the range of Reynolds numbers was 1100 to 14000. For low velocities, i.e. expected laminar flow and transition region, theoretically obtained temperatures were highly different from the experimentally obtained. This problem was solved using the equation applicable to turbulent region for low velocities too. A sample is shown in Table 6.3. It is concluded that because of the geometric effects, turbulent flow is observed even for low Reynolds numbers.

The effect of agitator speed on the heat transfer coefficient is shown on Figures 5.11 and 5.12. It could be reasoned that higher agitation speeds up to a certain value should produce higher heat transfer coefficients and this was found to be true in both cases for single tank and two tanks in series.

	(JINGLE TANK)		
Run	* $N_{Re} = \frac{D_A^2 N \rho}{\mu}$	$Y = \frac{(H_0 D_T / k) (\mu_W / \mu)^{0 \cdot 14}}{(C_p \mu / k)^{0 \cdot 33}}$	С
	''Re µ	(C <sub>p</sub> µ/k)°• <sup>33</sup>	
1	33490 64300	886 1341	1.39 1.40
2 3 4	69200	1361	1.36
4	73500	1459	1.40
5 6 7	37100 35221	941 929	1.39 1.40
	32121	860	1.38
8 9	31815 28709	849 788	1.37 1.37
10	24562	732	1.39
11 12	34100	920	1.42
12	34685 32571	912 882	1.40 1.40
14 15	26600	762 760	1.38 1.40
100 A	25834		н. 1919 - П. С. С. С. С. С. С. С. С. С. С. С. С. С.
16 17	30023 26812	767 745	1.30 1.34
18	26003	741	1.37
19 20	24151 22409	720 698	1.38 1.40

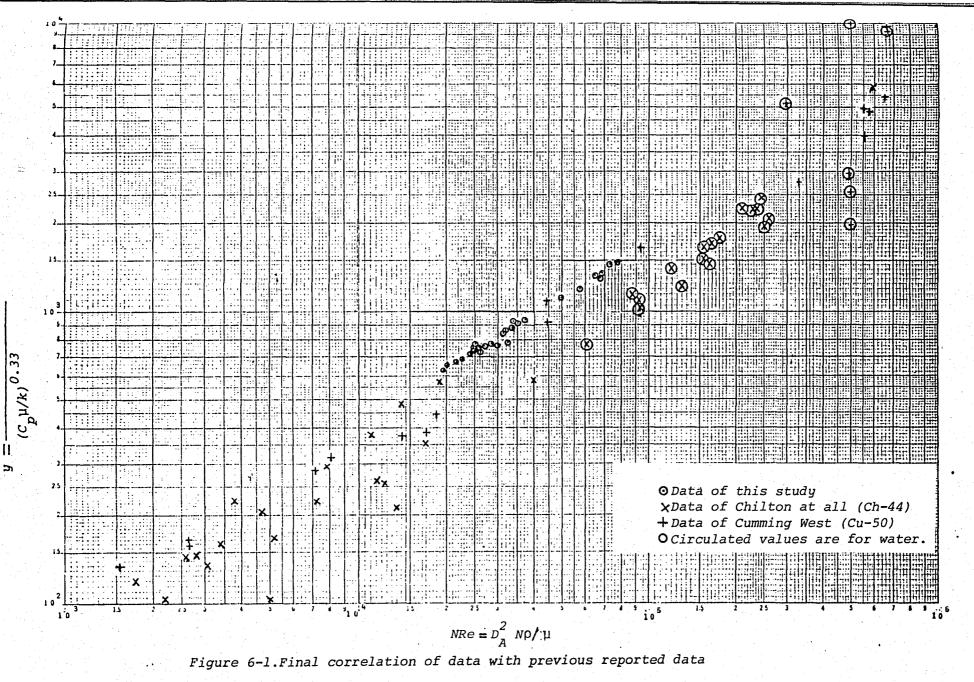
TABLE	6.2	 Determination	of	Geometry	Constant	(Fig.	6.1)
		(SINGLE TANK)					

Run		TANK I			TANK I	I
	N*Re	Ŷ	Cl	N <sub>Ře</sub>	Y	C <sub>2</sub>
21	21842	686	1.40	2985	823	1.38
22	31009	855	1.40	42826	1022	1.37
23	49862	1123	1.37	67841	1327	1.34
24	57248	1218	1.37	76727	1478	1.38
25	26630	776	1.40	38039	963	1.39
26	25364	766	1.42	35467	944	1.41
27	25072	756	1.41	35339	939	1.42
28	23900	740	1.42	33112	900	1.41
29	22895	721	1.42	31780	875	1.41
30	26033	769	1.40	32309	875	1.40
31	25074	762	1.42	30824	862	1.42
32	24300	746	1.42	30114	843	1.40
33	23355	722	1.41	29638	831	1.40
34	22711	711	1.41	28620	816	1.41
35	22053	685	1.39	28861	801	1.37
36	21472	677	1.40	27646	793	1.39
37	20609	660	1.40	25767	769	1.40
38	20086	648	1.39	24544	751	1.42
39	19439	635	1.39	23868	729	1.40

TABLE 6.2 - Continued... (TWO TANKS IN SERIES)

\*Modified Reynolds number

C Geometry constant



Experi	mentally obser	ved temperat	ures			
Run	N <sup>*</sup> Re <sub>tank</sub> I	N <sub>Re</sub> tank II	TC1	TC2	TH2	тнз
25	5506	7959	30.9	49.5	53.7	33.9
26	4159	6294	27.6	45.6	49.6	29.9
27	4018	6145	27.3	45.4	48.1	29.5
28	2068	3419	22.1	36.6	38.8	22.9
29	1352	2327	20.1	31.1	32.9	20.8
Calcu	lated results	using lamina	r and trar	nsition i	regions	equations
25	6913	8545	26.77	37.45	64.95	51.38
26	5351	6849	25.56	35.72	62.50	47.81
27	5190	6682	25.57	35.74	62.62	47.64
28	2708	3815	22.84	31.27	54.29	38.08
29	1732	2610	21.24	28.61	48.59	32.23
Calcu	lated results	using turbul	ent regior	n equatio	ons	
25	5402	8002	30.38	49.09	53.36	34.61
26	4080	6296	27.55	45.24	49.11	30.93
27	3998	6205	27.36	45.04	48.86	30.65
28	2067	3421	22.34	35.96	38.69	24.04
29	1360	2330	20.23	31.23	33.27	21.25

TABLE 6.3 - Sample Comparison Between Observed Temperatures and Calculated Temperatures (TWO TANKS IN SERIES)

<sup>\*</sup>Coil side.

During the experiments no measurements were made of coil wall temperatures; hence no direct data are available for the temperature difference between the coil wall and mixed water. But in computer programs wall temperatures were estimated to calculate viscosity at wall temperatures.

Equation of Cumming-West [19] and Chilton et.al [17] (Table 2.2) incorporate all the apparatus dimensions in the geometric constant 1.01 and 0.87 respectively which therefore restrict their application to apparatus geometrically similar to their experimental equipment. It is of interest to note that the exponentials of the Reynolds Number and Prandtl Number are not only constant in any given system, but seem to change very little with widely different mixing systems (Table 2.2, Chapter 2). If this conclusion can be substantiated, a greatly simplified approach to mixing heat transfer problems will be indicated.

The systems used in the present work were small. In scaling up heat transfer systems, all the different quantities that can be used to describe flow pattern and mixing performance are numerically different from a small to a large system. It is, therefore, important to carefully consider the effect of each these variables on scale-up and to make sure that, these adjustments are made properly. Further work on large scale apparatus is needed to check the film heat transfer equation.

The following work seems to desirable in this area of heat transfer.

- to obtain generalized correlation for external film heat transfer coefficient, extensive available data with numerical and computer techniques.
- to study the effect of finned coil geometry on external film heat transfer coefficient.
- to determine the effect of agitator speed and type on external film heat transfer coefficient for finnedcoiled tank.
- to obtain additional experimental data using different liquids in agitated side of tank.

### 6.B DISCUSSION OF ERRORS

Any experimentally determined quantity is subject to error and hence any calculated result which is based on experimental evidance is also limited in accuracy. The determination of the derived error from the observed error is a calculation of some importance since it indicates the point where experimental technique should be improved.

There are many sources of error. For this study sources of errors can be classified as follows:

- 1. Errors of measurement
- 2. Precision errors

The first type of errors are due to physical limitations of reading scales. In this study this type of error arised from measurements of diameters (tank, coil, coil tube agitator), flowrates and agitator speed. Precision error are the "built in" errors of the apparatus, this type of errors arised from uncalibrated scale on digital thermometer and graduated cylinder. For example the digital thermometer was capable of measuring temperature to one decimal point.

All of above errors can only be estimated and rarely measured. Estimated values of errors are given below (Estimation of length were dependent on measurement made at least five different points).

> Absolute error in diameter of agitator (  $\left( \delta D_{A} \right)$ 0.01 cm : Absolute error in diameter of coil tubing  $(\delta D_{\mu})$ : 0.01 cm Absolute error in diameter of coil  $(\delta D_{r})$ : 0.20 cm Absolute error in diameter of tank  $(\delta D_T)$ 0.10 cm Absolute error in agitator speed ( $\delta N$ ) 3 rpm : Absolute error in hot water flowrate ( $\delta W$ ) 0.05 ml/sec : 0.4°C Absolute errors in temperatures ( $\delta T$ ) Absolute errors in coil outside wall temperature  $(\delta T_w)$ : 0.5°C

In the following part of this chapter, a general functional relationship for the film heat transfer coefficient in agitated side of vessel and in the coil side is developed to calculate the error margin.

Empirical equation used to calculate the film heat transfer coefficient in agitated side of the vessel is,

94

H0 = 1.40(D<sub>T</sub>)<sup>-1</sup>k(
$$\frac{D_A^2 N \rho}{\mu}$$
)<sup>0.62</sup>( $\frac{C_p^{\mu}}{k}$ )<sup>0.33</sup>( $\frac{\mu}{\mu_W}$ )<sup>0.14</sup> (6.1)

where;

$$\rho = f(T), \quad \mu = f(T), \quad C_p = f(T), \quad \mu_w = \mu(T_w)$$

and external film coefficient is a function of below variables.

HO = 
$$f(D_T, D_A, N, T, T_w)$$

In the simplified form of equation (6.1)

$$H0 = 1.40 D_{C}^{-1} k^{0} \cdot {}^{77} D_{A}^{1} \cdot {}^{24} N^{0} \cdot {}^{62} \rho^{0} \cdot {}^{62} C_{D}^{0} \cdot {}^{33} \mu_{W}^{-0} \cdot {}^{14} \mu^{-0} \cdot {}^{15}$$
(6.2)

so;

Taking the partial derivatives of Eq. (6.2) final equation to calculate error margin can be written as (Appendix V)

$$\frac{\delta H0}{H0} = \left| -\frac{\delta D_{T}}{D_{T}} \right| + \left| 1.24 - \frac{\delta D_{A}}{D_{A}} \right| + \left| 0.62 - \frac{\delta N}{N} \right| + \left| -0.14(\mu_{W})' - \frac{\delta T_{W}}{\mu_{W}} \right| + \left| 0.77(k)' - \frac{\delta T}{k} \right| + \left| 0.62(\rho)' - \frac{\delta T}{\rho} \right| + \left| 0.33(C_{p})' - \frac{\delta T}{C_{p}} \right| + \left| -0.15(\mu)' - \frac{\delta T}{\mu} \right|$$

where;

(k)', (C\_p)', ( $\rho$ )', ( $\mu$ )', ( $\mu_W$ )' are derivatives.

Empirical equations used to calculate film heat transfer coefficient inside the coil is;

$$HI = 0.023k \ D_{H}^{-1}(C_{p}\mu/k)^{0} \cdot {}^{33}(D_{H}V_{H}/\mu)^{0} \cdot {}^{8}(\mu/\mu_{w})^{0} \cdot {}^{14}(1 + 3.5(D_{H}/D_{c}))$$

$$(V_{H} = W/3.1414(D_{H}^{2}/4))$$

$$HI = 0.028k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_{p}^{0} \cdot {}^{33}\mu_{w}^{-0} \cdot {}^{14}D_{H}^{-1} \cdot {}^{8}W^{0} \cdot {}^{8} + 0.028(3.5)k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}$$

$$\cdot \ C_{p}^{0} \cdot {}^{33}\mu_{w}^{-0} \cdot {}^{14}D_{c}^{-1}W^{0} \cdot {}^{8} \qquad (6.3)$$

$$HI = f_1(D_H, W, T, T_w) + f_2(W, D_c, T_w, T, D_H)$$

taking partial derivative of Eq. (6.3) the following equation is finally developed to calculate error (Appendix V).

$$\begin{split} \delta HI &= HI[|0.8(\delta W/W)| + |-0.14(\mu_W)'(\delta T_W/\mu_W)| + |0.33(C_p)'(\delta T/C_p)| \\ &+ |0.77(k)'(\delta T/k)| + |0.33(\mu)'(\delta T/\mu)| + |-0.8(\delta D_H/D_H|] \\ &+ |0.028k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_p^{0} \cdot {}^{33}\mu_W^{-0} \cdot {}^{14}W^{0} \cdot {}^{8}D_H^{-1} \cdot {}^{8}(-\delta D_H/D_H) \\ &+ 0.098k^{0} \cdot {}^{77}C_p^{0} \cdot {}^{33}\mu^{-0} \cdot {}^{33}\mu_W^{-0} \cdot {}^{14}W^{0} \cdot {}^{8}D_H^{-0} \cdot {}^{8}D_C^{-1}(-\delta D_C/D_C)| \end{split}$$

Numerical error values depending on the estimated error values are given on Table 6.4.

SINGLE TANK					
Run δHI(±)	δΗΙ%	δHO(±)	δ <b>Η0</b> %	δ <b>U(±)</b>	δU%
1 23.7	0.35	61.9	1.32	22.2	0.8
2 23.2	0.34	61.4	0.88	17.5	0.5
3 22.6	0.34	62.3	0.81	16.6	0.5
4 22.6	0.34	63.0	0.77	16.2	0.4
5       30.0         6       22.6         7       9.0         8       0.8         9       13.9         10       30.6	0.41	63.4	1.29	23.9	0.8
	0.34	62.8	1.31	21.9	0.8
	0.17	61.6	1.32	17.3	0.7
	0.02	61.2	1.32	13.5	0.6
	0.42	59.6	1.34	13.9	0.7
	1.73	56.5	1.35	18.8	1.5
1125.71212.3134.71418.21527.8	0.37	62.3	1.32	22.8	0.8
	0.21	62.1	1.31	18.5	0.7
	0.10	61.2	1.32	15.4	0.6
	0.66	58.1	1.35	14.3	0.8
	1.45	57.2	1.36	17.5	1.3
1620.3173.7182.81916.52025.7	0.32	60.0	1.34	21.1	0.8
	0.08	58.8	1.34	14.7	0.6
	0.07	58.6	1.34	12.9	0.6
	0.63	56.1	1.36	13.5	0.8
	1.45	54.7	1.36	16.4	1.3
TWO TANKS IN SEF	RIES	a dia 2010 amin'ny fisiana Ny INSEE dia mampina mampina mampina mampina mampina mampina mampina mampina mampina mampina mampina mampina ma Ny INSEE dia mampina mampina mampina mampina mampina mampina mampina mampina mampina mampina mampina mampina ma			
TANK I         21       22.5         22       22.6         23       22.4         24       22.5	0.69	54.1	1.36	16.2	0.95
	0.70	53.9	1.10	15.1	0.82
	0.69	55.7	0.85	14.4	0.70
	0.69	57.1	0.81	14.5	0.69
2528.72622.42721.72811.7297.5	0.73	57.6	1.35	19.0	0.97
	0.70	56.5	1.36	16.3	0.94
	0.69	56.5	1.36	15.9	0.93
	0.60	54.4	1.37	10.4	0.83
	0.53	53.5	1.37	7.4	0.73
30       25.6         31       20.6         32       14.9         33       11.4         34       7.1	0.71	55.8	1.36	17.6	0.96
	0.68	54.9	1.36	15.4	0.93
	0.65	53.9	1.37	12.3	0.88
	0.60	53.3	1.37	10.2	0.83
	0.52	52.6	1.37	7.1	0.73

TABLE 6.4 - Numerical Error Values

Run	δHI(±)	δHI%	δH <b>O(</b> ±)	δH0%	δU(±)	δU%
35 36 37 38 39	24.3 19.6 14.5 11.5 6.8	0.70 0.68 0.64 0.61 0.52	54.2 53.6 52.7 52.2 51.4	1.36 1.37 1.37 1.37 1.37 1.37	17.0 14.8 12.0 10.2 6.8	0.96 0.93 0.88 0.83 0.72
TANK II						
21	31.3	0.77	59.9	1.33	19.9	0.98
22	31.5	0.77	59.4	1.07	18.7	0.85
23	30.5	0.76	60.7	0.83	17.8	0.73
24	30.3	0.75	61.7	0.78	17.8	0.72
25	39.7	0.82	63.2	1.30	22.7	1.00
26	31.7	0.79	62.3	1.31	20.0	0.97
27	30.8	0.79	62.3	1.31	19.7	0.97
28	17.0	0.68	59.6	1.34	13.5	0.88
29	11.0	0.60	58.1	1.35	9.9	0.80
30	34.1	0.78	60.9	1.33	20.9	0.99
31	28.2	0.76	60.3	1.34	18.7	0.96
32	20.4	0.70	58.5	1.35	15.3	0.92
33	15.8	0.66	57.8	1.35	12.9	0.87
34	10.0	0.58	56.4	1.36	9.2	0.78
35	30.9	0.75	59.0	1.34	19.8	0.98
36	25.3	0.73	58.4	1.35	17.5	0.95
37	18.8	0.68	57.2	1.35	14.5	0.91
38	15.1	0.65	56.3	1.36	12.4	0.87
39	9.0	0.56	54.8	1.36	8.5	0.76

TABLE 6.4 - Continued...

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100

# APPENDICES

### APPENDIX I

### EXPERIMENTAL DATA AND COMPUTED TEMPERATURES

#### SYMBOLS USED

SINGLE TANK

TCO: Cold water inlet temperature (°C)
TH1: Hot water inlet temperature (°C)
TC1: Cold water exit temperature (°C)
TH2: Hot water exit temperature (°C)

TWO TANKS IN SERIES

TCO: Cold water inlet temperature to tank I (°C)
TH1: Hot water inlet temperature to tank II (°C)
TC1: Cold water exit temperature from tank I (°C)
TC2: Cold water exit temperature from tank II (°C)
TH2: Hot water exit temperature from tank II (°C)
TH3: Hot water exit temperature from tank I (°C)

				Observed Temperatures	Computed Temperatures
Run (	N W rpm) (ml/sec)	M (ml/sec)	TH1 TCO (°C) (°C)	TC1 TH2 (°C) (°C)	TC1 TH2 (°C) (°C)
1.	200 26.5	22.3	75.6 12.6	41.6 50.5	42.35 50.36
2	380 26.5	22.5	75.7 12.5	42.9 49.3	43.48 49.19
3	450 26.3	22.5	75.5 12.6	43.5 49.0	43.60 48.79
4	500 26.5	23.5	75.0 12.6	42.2 47.9	42.88 47.95

TABLE A.I.1 - Agitator Speed is Variable - Single Tank - Steady State

						Observed	Temperatures	Computed Te	emperatures
Run	W	TH1	M	N	TCO	TC1	TH2	TC1	TH2
	(ml/sec)	(°C)	(ml/sec)	(rpm)	(°C)	(°C)	(°C)	(°C)	(°C)
5 6 7 8 9 10	27.5 25.0 19.8 16.0 11.2 5.6	83.5 81.2 80.0 82.2 81.5 80.1	19.2 18.5 19.0 19.5 20.5 20.0	200 200 200 200 200 200 200	11.3 8.9 8.5 10.3 12.1 12.1	48.5 45.0 41.5 40.0 34.6 26.3	57.0 53.5 47.9 45.7 38.9 28.2	48.92 45.72 41.34 39.88 34.89 26.33	57.01 53.72 48.20 45.81 39.38 28.73
11	27.5	73.5	21.0	200	16.0	44.8	51.2	44.42	51.64
12	21.5	73.3	18.0	200	17.1	44.1	50.3	44.57	50.15
13	18.5	72.5	20.0	200	16.5	40.7	46.3	40.70	46.15
14	9.6	73.5	20.0	200	12.1	30.9	34.9	30.68	34.46
15	6.3	72.1	22.5	200	17.5	29.0	30.9	28.86	31.18
16	27.1	62.8	19.5	200	10.0	36.9	43.4	36.56	43.57
17	19.0	62.3	20.2	200	10.9	33.2	38.4	32.93	38.72
18	16.1	62.9	19.0	200	11.3	33.0	37.1	32.58	37.62
19	10.0	60.3	20.0	200	10.1	25.8	28.8	25.50	29.28
20	6.4	60.6	20.0	200	10.3	22.8	24.0	21.79	24.42

### TABLE A.I.2 - Hot Water Flowrate is Variable - Single Tank - Steady-State

TABLE A.I.3 - Agitator Speed is Variable - Two Tanks in Series - Steady-State

••••••••••••••••••••••••••••••••••••••		Observed Temperatures	Computed Temperatures	
Run N (rpm)	W M TH1 TCO (ml/sec) (ml/sec) (°C) (°C)	TC1 TC2 TH2 TH3 (°C) (°C) (°C) (°C)	TC1 TC2 TH2 TH3 (°C) (°C) (°C) (°C)	
21 200	15.0 30.2 73.4 12.2	20.8 36.6 41.7 25.1	20.52 36.25 41.48 24.60	
22 280	15.0 30.0 74.5 12.8	21.3 37.4 42.0 24.4	21.27 37.50 41.80 24.72	
23 450	15.1 30.0 71.0 12.3	21.0 36.5 40.1 23.9	21.34 36.79 40.07 24.00	
24 500	15.0 30.0 70.3 15.3	22.8 37.1 39.6 25.1	22.92 37.66 40.61 25.28	

						Obse	rved Te	emperatu	ires	Comput	ted Tempe	eratures	
Run	W (ml/sec)	TH1 (°C) (	M ml/sec)	N (rpm)	TCO (°C)	TC1 (°C)	TC2 (°C)	TH2 (°C)	TH3 (°C)	TC1 (°C)	TC2 (°C)	TH2 (°C)	TH3 (°C)
25 26 27 28 29	16.6 13.5 13.1 8.0 5.7	80.6 80.8 81.5 79.9 80.0	24.0 24.0 24.0 24.0 24.0	200 200 200 200 200 200	17.5 17.4 17.5 17.5 17.4	30.9 27.6 27.3 22.1 20.1	49.5 45.6 45.4 36.6 31.1	53.7 49.6 48.1 38.8 32.9	33.9 29.9 29.5 22.9 20.8	30.38 27.55 27.36 22.34 20.23	49.09 45.23 45.04 35.96 31.22	53.36 49.11 48.86 38.68 33.27	34.61 30.94 30.65 24.04 21.25
30 31 32 33 34	16.2 13.4 10.2 8.2 5.7	71.0 72.8 71.2 71.3 71.1	28.0 28.0 28.0 28.0 28.0	200 200 200 200 200 200	16.2 15.8 15.9 15.8 15.9	26.6 23.5 19.8 18.9 17.9	40.8 38.8 32.5 30.1 26.7	44.9 41.7 36.8 33.1 28.7	30.6 26.1 22.8 20.6 18.6	25.25 23.24 20.88 19.40 17.89	40.28 37.83 33.30 30.41 26.46	44.87 42.09 36.84 33.43 28.67	29.11 26.44 23.08 21.03 18.80
35 36 37 38 39	16.5 13.5 10.4 8.6 5.7	60.7 61.7 61.2 60.9 60.4	27.0 27.0 27.0 27.1 27.0	200 200 200 200 200	12.5 12.9 12.9 12.8 12.9	21.4 19.5 17.9 16.1 15.3	34.8 31.8 28.9 25.6 22.3	39.4 35.8 32.8 29.8 24.1	24.4 21.3 19.2 18.1 15.6	20.99 19.67 17.65 16.38 14.76	34.31 32.27 28.73 26.30 22.28	38.97 36.36 32.25 29.41 24.52	24.81 22.73 19.84 18.07 15.66

TABLE A.I.4 - Hot Water Flowrate is Variable - Two Tanks in Series - Steady-State

TABLE A.I.5 - Temperature - Time Data - Observed

Time sec	TC1 °C	TH2 °C	*ALPHA <sub>1</sub>
30	41.8	45.9	0.162
60	40.7	44.9	0.160
90	40.0	44.3	0.159
120	37.5	43.8	0.156
150	38.9	43.5	0.163
180	38.0	43.1	0.164
210	37.8	42.9	0.169
240	37.7	42.7	0.168
270	37.5	42.5	0.164
300	37.4	42.2	0.159
330	37.3	42.1	0.159
360	37.2	42.0	0.158
390	37.1	41.9	0.158
420	37.0	41.9	0.161
450	37.0	41.8	0.160
480	36.9	41.8	0.160
510	36.9	41.7	0.159
540	36.9	41.7	0.159
570	36.9	41.7	0.159
600	36.9	41.7	0.159

\*ALPHA<sub>1</sub> = (TH2 - TC1)/(TH1 - TC1)

TABLE A.I.6 - Temperature - Time Data - Observed

Time sec	TC1 °C	TH2 °C	ALPHA
30	37.2	42.0	0.212
60	36.2	41.9	0.216
90	36.0	40.7	0.198
120	35.4	40.6	0.213
150	35.1	40.1	0.202
180	35.0	40.0	0.202
210	34.8	39.8	0.200
240	34.7	39.7	0.199
270	34.6	39.6	0.198
300	34.4	39.3	0.193
330	34.3	39.3	0.196
360	34.2	39.1	0.191
390	34.1	39.1	0.195
420	34.0	39.0	0.194
450	34.0	38.9	0.190
480	33.9	38.8	0.189
510	33.8	38.8	0.192
540	.33.8	38.8	0.192
570 ·	33.8	38.8	0.192
600			

TABLE A.I.7 - Temperature - Time Data - Observed

Set No. : 3 (Single tank)

N - Agitator speed : 200 rpm W - Hot water flowrate : 20.0 ml/sec M - Cold water flowrate: 20.0 ml/sec TCO(1) : 12.9°C TH1(1) : 83.8°C TC1S TH2S

44.7 °C 50.5 °C

TCO(2) : 12.9°C TH1(2) : 63.5°C

Time	TC1	TH2	ALPHA <sub>1</sub>
sec	°C	°C	
30	42.8	46.1	0.160
60	41.1	44.6	0.160
90	39.6	43.9	0.180
120	38.4	42.8	0.175
150	37.9	42.5	0.180
180	37.5	41.8	0.165
210	37.2	41.6	0.167
240	36.9	41.1	0.158
270	36.4	40.9	0.166
300	36.3	40.8	0.166
330	36.1	40.7	0.168
360	35.9	40.7	0.174
390	35.8	40.6	0.173
420	35.8	40.6	0.173
450	35.7	40.6	0.173
480 510 540 570 600	35.7 35.7 35.6 35.6 35.6	40.5 40.5 40.4 40.4 40.4	0.172 0.172 0.172 0.172 0.172 0.172

TABLE A.I.8 - Temperature - Time Data - Observed

Set No. : 2 (Two tanks in series)

N - Agitator speed : 200 rpm W - Hot water flowrate : 15.5 ml/sec M - Cold water flowrate: 24.0 ml/sec

TCO(1) : 13.2°C TH1(1) : 71.0°C

TC1S	TC2S	TH2S	TH3S
24.2°C	40.6°C	45.3°C	27.7°C

TCO(2) : 13.2°C TH1(2) : 61.8°C

		TC1 °C	TC2 °C	TH2 °C	TH3 °C	*ALPHA	*BETA
		24.0	40.0	43.4	27.3	0.170	0.156
		23.8	39.4	42.8	27.0	0.168	0.152
		23.6	39.2	42.4	26.8	0.170	0.142
1		23.5	38.9	42.0	26.6	0.167	0.135
		23.3	38.5	41.6	26.3	0.164	0.133
		23.2	38.3	41.3	26.2	0.166	0.271
1		23.1	38.0	41.2	26.0	0.161	0.135
		23.0	37.8	40.9	25.8	0.157	0.129
2		22.9	37.7	40.8	25.8	0.162	0.129
3	00	22.8	37.5	40.8	25.7	0.161	0.136
3		22.7	37.4	40.7	25.6	0.161	0.135
3		22.7	37.3	40.6	25.5	0.156	0.135
3	90	22.6	37.2	40.6	25.5	0.161	0.138
4		22.6	37.0	40.5	25.4	0.157	0.141
4	50	22.6	36.9	40.4	25.3	0.152	0.141
4	80	22.5	36.8	40.4	25.2	0.151	0.144
5	10	22.5	36.7	40.3	25.2	0.152	0.143
5	40	22.4	36.7	40.3	25.2	0.157	0.143
5	70	22.4	36.6	40.3	25.1	0.151	0.147
6	00	22.3	36.5	40.2	25.1	0.157	0.146
6	30	22.3	36.4	40.2	25.1	0.157	0.150
6	60	22.3	36.4	40.2	25.1	0.157	0.149
		22.3	36.4	40.2	25.1	0.157	0.149

 $ALPHA = \frac{TH3 - TC1}{TH2 - TC1}$ 

5

 $BETA = \frac{TH2 - TC2}{TH1 - TC2}$ 

TABLE A.I.9 - Temperature - Time Data - Observed

Set No. : 3 (Two tanks in series)

N - Agitator speed : 200 rpm W - Hot water flowrate : 13.5 ml/sec M - Cold water flowrate:' 28.5 ml/sec TCO(1) : 13.7°C TH1(1) : 83.2°C TCIS TC2S TH2S TH3S 22.8°C 39.8°C 45.2°C 25.9°C

TCO(2) : 13.7°C TH1(2) : 60.9°C

.

Ti se	me :C	TC1 °C	TC2 °C	TH2 °C	TH3 °C	ALPHA	BETA
3 6 9 12 15 18 21 24 27 30 33 36 39 42 45 48 51 54 57 60 63	30         30	22.4 22.1 21.8 21.6 21.4 21.2 21.1 20.8 20.7 20.5 20.4 20.3 20.2 20.1 20.1 20.0 20.0 19.9 19.9 19.8 19.8	38.5 37.2 35.9 35.2 34.7 34.2 33.7 33.4 33.2 33.0 32.8 32.6 32.5 32.3 32.2 32.0 31.9 31.8 31.7 31.7 31.7	41.3 40.1 39.3 38.7 38.1 37.7 37.2 37.1 36.9 36.6 36.4 36.3 36.2 36.1 36.0 35.8 35.7 35.6 35.4 35.3 35.3	25.4 25.1 24.7 24.3 24.1 23.8 23.7 23.2 23.1 22.9 22.9 22.9 22.9 22.9 22.9 22.8 22.7 22.6 22.6 22.6 22.5 22.5 22.4 22.4 22.4 22.3 22.3	0.159 0.167 0.166 0.158 0.162 0.157 0.162 0.148 0.148 0.148 0.149 0.156 0.156 0.156 0.156 0.156 0.156 0.157 0.158 0.159 0.159 0.159 0.161 0.162 0.162	0.125 0.122 0.136 0.136 0.129 0.131 0.128 0.134 0.133 0.129 0.128 0.130 0.131 0.133 0.131 0.131 0.131 0.127 0.123 0.126
66		19.8 19.8	31.6 31.6	35.3 35.3	22.3 22.3	0.162 0.162	0.126 0.126

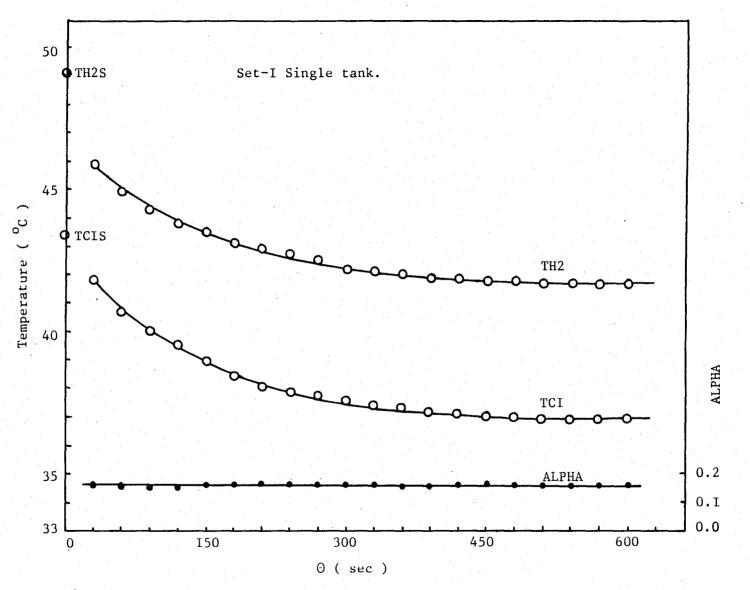


Figure A-I.I. Temperature change with time for unsteady-state heat transfer.

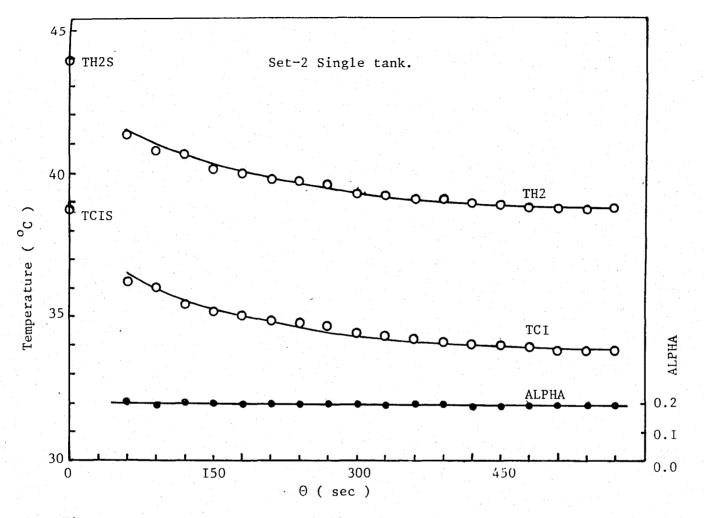


Figure A-I.2. Temperature change with time for unsteady-state heat transfer.

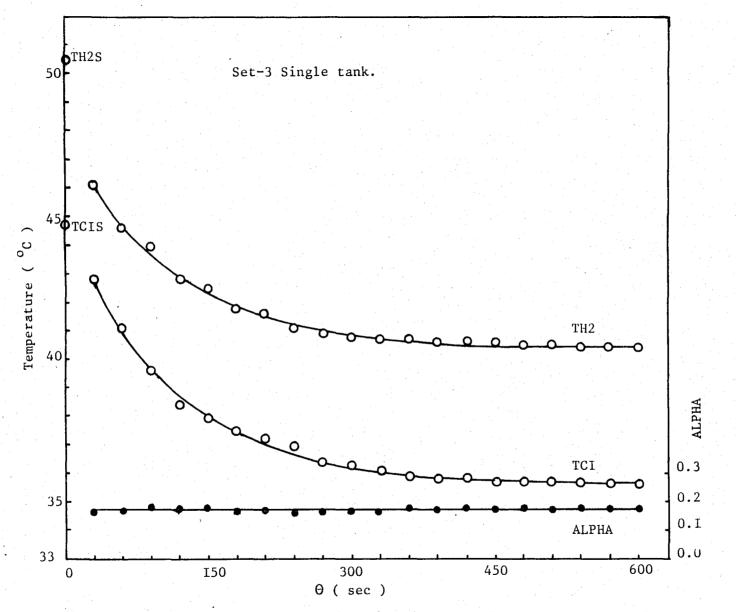
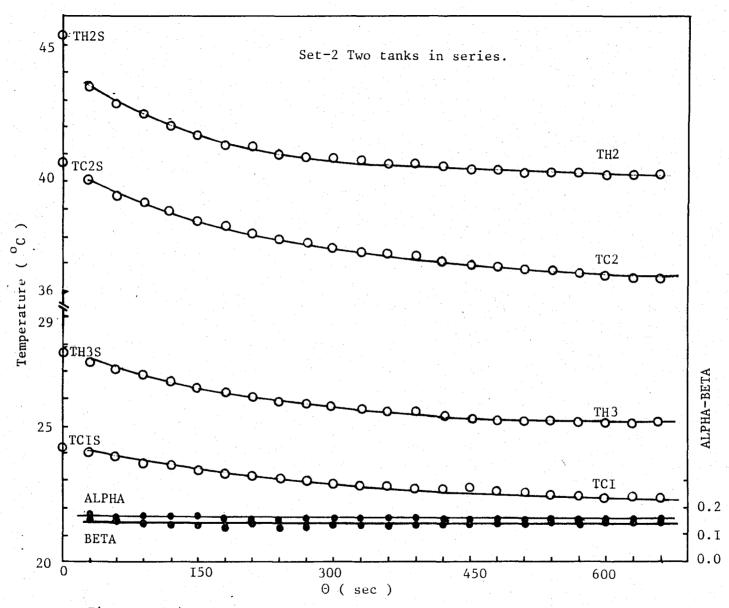
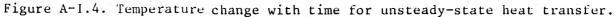


Figure A-I.3. Temperature change with time for unsteady-state heat transfer.





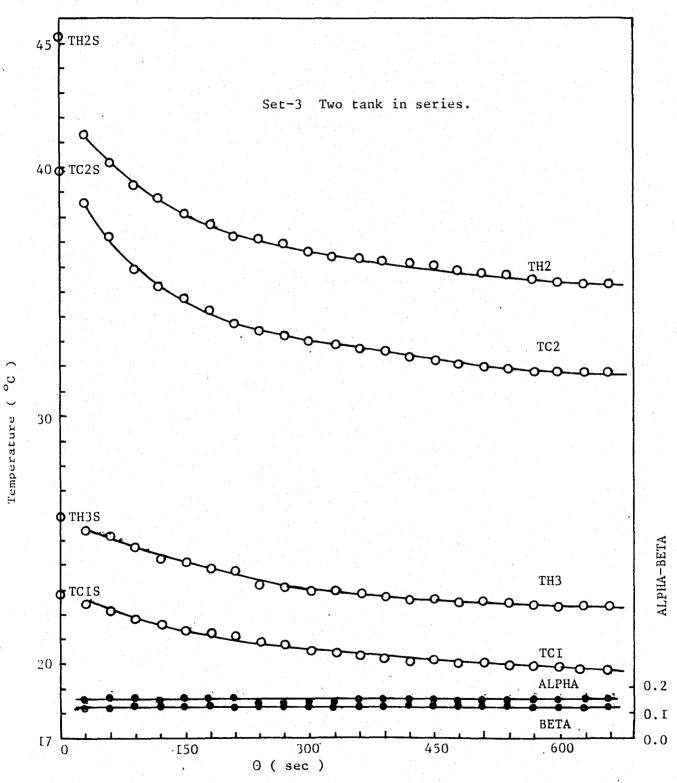


Figure A-I.5. Temperature change with time for unsteady-state heat transfer.

## APPENDIX II MATHEMATICAL MODELLING

### A.II.1 MATHEMATICAL MODELLING FOR SINGLE TANK

As shown in Figure 3.1, flow of cold water was in the tank side and hot water in the coil side. Counter current was performed in this system.

Assuming W = W.C\_h^{\rho}h, M = M.C\_w^{\rho}\rho\_w and V = V.C\_w^{\rho}\rho\_w heat balance is;

$$W T_1 - W T_2 + Mt_0 - Mt_1 = V \frac{dt_1}{d\theta}$$
 (A.2.1)

Assuming proper agitation, (i.e. temperature in the tank,  $t_1$ ) the heat transfer rate equation for the tank is

$$\frac{U_{1}A(T_{2} - T_{1})}{\ln \frac{T_{2} - t_{1}}{T_{1} - t_{1}}} = W(T_{1} - T_{2})$$

$$\ln \frac{T_{2} - t_{1}}{T_{1} - t_{1}} = \exp[-U_{1}A/W] = \alpha$$

$$T_{2} - t_{1} = \alpha(T_{1} - t_{1})$$

or

$$T_{2} = \alpha T_{1} - \alpha t_{1} + t_{1}$$
  

$$T_{2} = \alpha T_{1} + (1 - \alpha) t_{1}$$
(A.2.2)

From Equation (A.2.1)

$$T_{2} = \frac{WT_{1} + Mt_{0} - Mt_{1} - V(dt_{1}/d\theta)}{W}$$

$$WT_{1} + Mt_{0} - Mt_{1} - V(dt_{1}/d\theta) - WT_{1} - W(1 - \alpha)t_{1} = 0$$

First order linear equation with constant coefficients was obtained,

$$[M + W(1 - \alpha)]t_{1} + V(dt_{1}/d\theta) = W(1 - \alpha)T_{1} + Mt_{0}$$
(A.2.3)  
F = M + W(1 - \alpha)  
H = W(1 - \alpha)T\_{1} + Mt\_{0}

Equation (A.2.3) can be written as,

$$Ft_1 + V(dt_1/d\theta) = H$$

then;

$$Vm + F = 0$$
  $m = -F/V$ 

And therefore complementary function is

$$t_1 = Ae^{m\theta}$$
  
 $t_1 = Ae^{-F/V\theta}$ 

The right hand side of Equation (A.2.3) is a constant and therefore the particular integral is a constant and equal to right hand side of Equation (A.2.3) divided by the coefficient  $t_1$ , then particular solution;

$$t_{ln} = H/F = B$$

And the complete solution is;

$$t_1 = Ae^{m_1 \Theta} + B$$

where A is arbitrary constant, using following initial conditions it can be calculated

at 
$$\theta = 0$$
  $t_1 = t_{1s}$   
 $t_1 = [t_{1s} - H/F]e^{-F/V\theta} + \frac{H}{T}$ 

inserting W = WC<sub>h</sub> $_{h}$ , M = MC<sub>W</sub> $_{W}$ , V = VC<sub>W</sub> $_{W}$  into above equation, final form is,

$$t_{1} = [t_{1s} - \frac{WC_{h}\rho_{h}(1-\alpha)T_{1} + MC_{w}\rho_{w}t_{o}}{MC_{w}\rho_{w} + WC_{h}\rho_{h}(1-\alpha)}]exp((MC_{w}\rho_{w}+WC_{h}\rho_{h}(1-\alpha)/VC_{w}\rho_{w})\theta$$
$$- \frac{WC_{h}\rho_{h}(1-\alpha)T_{1} + MC_{w}\rho_{w}t_{o}}{MC_{w}\rho_{w} + WC_{h}\rho_{h}(1-\alpha)}$$
(A.2.4)

Using Equations (A.2.2) and (A.2.4), unknown temperatures can be calculated.

### A.II.2 MATHEMATICAL MODELLING FOR TWO TANKS IN SERIES

As shown in Figure 3.2, heat balances for the two tanks in series are as follows. Again assume that  $M = MC_w \rho_w$ ,  $W = WC_h \rho_h$  and  $V = VC_w \rho_w$ .

Tank I 
$$WT_2 - WT_3 + Mt_0 - Mt_1 = V(dt_1/d\theta)$$
 (A.2.5)

Tank II 
$$WT_1 - WT_2 + Mt_1 - Mt_2 = V(dt_2/d\theta)$$
 (A.2.6)

Heat transfer rate equations for the two tanks

T\_ - t\_

Tank I

Tank I 
$$\frac{3}{T_2 - t_1} = \exp[-U_1 A/W] = \alpha$$
  
Tank II 
$$\frac{T_2 - t_2}{T_1 - t_2} = \exp[-U_2 A/W] = \beta$$
  

$$T_3 - t_1 = \alpha T_2 - \alpha t_1$$
  

$$T_2 - t_2 = \beta T_1 - \beta t_2$$
  

$$T_3 = \alpha T_2 + t_1(1 - \alpha)$$
  

$$T_2 = \beta T_1 + t_2(1 - \beta)$$
  
(A.2.8)

These four exuations (A.2.5), (A.2.6), (A.2.7), (A.2.8) have to be solved simultaneously. Frequency of appearance of variables

т <sub>3</sub>	<sup>T</sup> 2	t <sub>1</sub>	t <sub>2</sub>
1 1	1	1 +	+
	1 1	1 1	1 1
2	4	3,+	2,+

+, shows differential form

 $T_3$ , must be eliminated first, then  $T_2$  because  $t_1$  and  $t_2$  appear in a differentiated form substitution of Equation (A.2.7) into Eq. (A.2.5).

$$WT_{2} - W(\alpha T_{2} + t_{1}(1 - \alpha)) + Mt_{0} - MC_{W} = V(dt_{1}/d\theta)$$
  

$$WT_{2} - W\alpha T_{2} - Wt_{1} + W\alpha t_{1} + Mt_{0} - Mt_{1} = V(dt_{1}/d\theta)$$
  

$$W(1 - \alpha)T_{2} + Mt_{0} + (W\alpha - W - M)t_{1} = V(dt_{1}/d\theta)$$
 (A.2.9)

now eliminate  $T_2$  from the system by substituting from Equation (A.2.8) to Equations (A.2.9) and (A.2.6)

$$W(1 - \alpha)[\beta T_{1} + T_{2}(1 - \beta)] + Mt_{0} + (W\alpha - W - M)t_{1} = V(dt_{1}/d\theta)$$
$$W(1 - \alpha)\beta T_{1} + W(1 - \alpha)(1 - \beta)t_{2} + Mt_{0} + (W\alpha - W - M)t_{1} = V(dt_{1}/d\theta)$$
(A.2.10)

and

$$WT_{1} - W[\beta T_{1} + t_{2}(1 - \beta)] + Mt_{1} - Mt_{2} = V(dt_{2}/d\theta)$$

$$WT_{1} - W\beta T_{1} - W(1 - \beta)t_{2} + Mt_{1} - Mt_{2} = V(dt_{2}/d\theta)$$

$$(W - W\beta)T_{1} + Mt_{1} - [W(1 - \beta) + M]t_{2} = V(dt_{2}/d\theta)$$

$$t_{1} = \frac{1}{M} [V(dt_{2}/d\theta) + ((W/M)(1 - \beta) + 1)t_{2} + (W\beta - W)T_{1}]$$

$$t_{1} = (V/M)(dt_{2}/d\theta) + [(W/M)(1 - \beta) + 1]t_{2} + ((W/M)\beta - (WC_{h}/MC_{w})T_{1})$$

$$(A.2.11)$$

$$(A.2.11)$$

$$(A.2.12)$$

using Equations (A.2.11) and (A.2.12) to eliminate  $t_1$  and its differential coefficient from Equation (A.3.10)

121

$$+ \left[ (W/M)(1 - \beta) + 1 \right] t_{2} + ((W/M)\beta - (W/M)T_{1}] = W[(V/M)(d^{2}t_{2}/d\theta^{2}) + [(W/M)(1 - \beta) + 1](dt_{2}/d\theta)] \\ W(1 - \alpha)\betaT_{1} + W(1 - \alpha)(1 - \beta)t_{2} + Mt_{0} + \frac{VW}{M}\alpha \frac{dt_{2}}{d\theta} - \frac{W}{M} \frac{dt_{2}}{d\theta} \\ - \frac{W}{M} \frac{dt_{2}}{d\theta} - V \frac{dt_{2}}{d\theta} + \left[ -\frac{W^{2}}{M} \alpha(1 - \beta) + W\alpha \right] t_{2} + \left[ \frac{W^{2}}{M} \alpha \beta - \frac{W^{2}}{M} \alpha \right] T_{1} \\ - \left[ -\frac{W^{2}}{M} \beta - \frac{W^{2}}{M} \right] T_{1} - \left[ W\beta - W \right] T_{1} = \frac{V^{2}}{M} \frac{d^{2}t_{2}}{d\theta^{2}} + \left[ -\frac{VW}{M} (1 - \beta) + V \right] - \frac{dt_{2}}{d\theta} \\ \frac{V^{2}}{M} \frac{d^{2}t_{2}}{d\theta} + \left[ -\frac{VW}{M} (1 - \beta) + W - \frac{VW}{M} \alpha + \frac{WV}{M} + V \right] \frac{dt_{2}}{d\theta} = \\ Mt_{0} + \left[ W(1 - \alpha)\beta + \frac{W^{2}}{M} \alpha \beta - \frac{W^{2}}{M} \alpha + \frac{W^{2}}{M} (1 - \beta) + W(1 - \beta) \right] T_{1} \\ + \left[ W(1 - \alpha)(1 - \beta) + W\alpha + \frac{W^{2}}{M} \alpha(1 - \beta) - \frac{W^{2}}{M} (1 - \beta) - W \\ - W(1 - \beta) - M \right] t_{2} \\ \frac{V^{2}}{M} \frac{d^{2}t_{2}}{d\theta^{2}} + \left[ 2 \frac{VW}{M} + 2V + \frac{VM}{M} \beta - \frac{VW}{M} \alpha \right] \frac{dt_{2}}{d\theta} = \\ Mt_{0} + \left[ M\beta - W\alpha\beta + \frac{W^{2}}{M} \alpha\beta - \frac{W^{2}}{M} \alpha + \frac{W^{2}}{M} - \frac{W^{2}}{M} \beta + W - W\beta \right] T_{1} \\ + \left[ W - W\alpha - W\beta + W\alpha\beta + W\alpha - \frac{W^{2}}{M} \alpha - \frac{W^{2}}{M} \alpha\beta - \frac{W^{2}}{M} \alpha\beta - \frac{W^{2}}{M} + \frac{W^{2}C_{h}^{2}}{M} \beta - W \right]$$

+ 
$$[(W/M)(1 - \beta) + 1]t_2 + ((W/M)\beta - (W/M)T_1] =$$

 $W(1 - \alpha)\beta T_1 + W(1 - \alpha)(1 - \beta)t_2 + Mt_0 + (W\alpha - W - M)[(V/M)(dt_2/d\theta)$ 

Second order linear differential equation with constant coefficients is obtained.

$$(V^{2}/M)(d^{2}t_{2}/d\theta^{2}) + [2(VW/M) + 2V - (VW/M)\beta - (VW/M)\alpha](dt_{2}/d\theta) + [M + W - (W^{2}/M)\beta + (W^{2}/M) + (W^{2}/M)\alpha\beta - (W^{2}/M)\alpha - W\alpha\beta]t_{2} = Mt_{0} + [(W^{2}/M)\alpha\beta - W\alpha\beta - (W^{2}/M)\alpha + (W^{2}/M) - (W^{2}/M)\beta + W]T_{1}$$
(A.2.13)

$$F = M + W - (W^{2}/M)\beta + (W^{2}/M) + (W^{2}/M)\alpha\beta - (W^{2}/M)\alpha - W\alpha\beta$$
$$E = Mt_{0} + [(W^{2}/M)\alpha\beta - W\alpha\beta - (W^{2}/M)\alpha + (W^{2}/M) - (W^{2}/M)\beta + W]T.$$

Auxiliary equation

$$(d^{2}t_{2}/d\theta^{2}) + [2(W/V) + W(M/V) - (W/V)\beta - (W/V)\alpha](dt_{2}/d\theta)$$
$$+ [(M^{2}/V^{2}) + (MW/V^{2}) - (W^{2}/V^{2})\beta + (W^{2}/V^{2}) - (W^{2}/V^{2})\alpha\beta$$
$$- (W^{2}/V^{2})\alpha - (WV/V^{2})\alpha\beta]t_{2} = 0$$

if,

$$BP = 2(W/V) + (2M/V) - (W/V)\beta - (W/V)\alpha$$
$$DP = (M^2/V^2) + (MW/V^2) - (W^2/V^2)\beta + (W^2/V^2) - (W^2/V^2)\alpha\beta$$
$$- (W^2/V^2)\alpha - (WM/V^2)\alpha\beta$$

Then above equation can be written as

$$m^2 + BPm + DP = 0$$
 (A.2.14)

The roots of Equation (A.2.14) are

$$m_{1,2} = \frac{-BP \pm \sqrt{BP^2 - 4DP}}{2}$$

and therefore the complementary function is

$$t_{2c} = Ae^{m_1\theta} + Be^{m_2\theta}$$

The right hand side of Equation (A.2.13) is a constant and therefore the particular integral is a constant and equal to the right hand side of Equation (A.2.13) divided by the coefficient  $t_2$ , then particular solution is;

$$t_{2p} = E/F = CP$$
  $F = DP(V^2/M)$ 

And the complete solution is,

$$t_2 = Ae^{m_1\theta} + Be^{m_2\theta} + CP$$
 (A.2.15)

where A and B arbitrary constants to be calculated from the following initial conditions

at 
$$\theta = 0$$
  $t_2 = t_{2s}$  {After reaching steady-state}  
 $t_1 = t_{1s}$ 

$$CP = \frac{Mt_{0} + (W - W\alpha\beta + (W^{2}/M) - (W^{2}/M)\alpha - (W^{2}/M)\beta + (W^{2}/M)\alpha\beta)T_{1}}{M + W - W\alpha\beta - (W^{2}/M)\alpha + (W^{2}/M)\alpha\beta - (W^{2}/M)\beta + (W^{2}/M)}$$

 $\theta = 0$  Equation (A.2.15)

$$t_{2s} = A + B + CP$$

 $A = t_{2s} - B - CP$ 

Substituting Equation (A.2.15) to Equation (A.2.11)

$$t_{1} = (V/M)(m_{1}Ae^{m_{1}\theta} + m_{2}Be^{m_{2}\theta}) + [(W/M)(1 - \beta) + 1](Ae^{m_{1}\theta} + Be^{m_{2}\theta} + CP) + ((W/M)\beta - (W/M)T_{1}$$
(A.2.16)

$$\theta = 0 \rightarrow t_1 = t_{1s}$$

$$t_{1c} = (V/M)(m_1A + m_2B) + [(W/M)(1 - \beta) + 1](A + B + CP)$$

+ 
$$((W/M)\beta - (W/M))T_1$$
  
 $t_{1s} = (V/M)[m_1(t_{2s} - B - CP) + m_2B] + [(W/M)(1 - \beta) + 1]t_{2s}$ 

+ 
$$((W/M)\beta - (W/M))T_1$$
  
 $t_{1s} = (V/M)(m_1t_{2s} - mB + m_2B - m_1CP) + [(W/M)(1 - \beta) + 1]t_{2s}$   
+  $((W/M)\beta - (W/M))T_1$ 

$$t_{1s} = ((W/M)\beta - (W/M))T_1 + ((W/M)(1 - \beta) + 1 + (V/M)m_1)t_{2s}$$

+ 
$$(V/M)(m_2 - m_1)B - (V/M)m_1CP$$

$$B = \frac{t_{1s} + ((W/M) - (W/M)\beta)T_1 - (V/M)m_1CP - [(W/M)(1-\beta) + (V/M)m_1+1]t_{2s}}{(V/M)(m_2 - m_1)}$$

Using Equations (A.2.7), (A.2.8) and (A.2.15), (A.2.16) and substituting  $M = MC_w \rho_w$ ,  $W = WC_h \rho_h$  and  $V = VC_w \rho_w$  unknown temperatures are calculated.

# APPENDIX III program listing

· · · ·	· 000010	DOUCDAN UEAT(NATA DECUNT TADES-NATA TADES-DESUNT)
1	000010 00002 C	PROGRAM HEAT(DATA, RESULT, TAPES=DATA, TAPE6=RESULT)
	60002 C	DETERMINATION OF HEAT TRANSFER COEFFICIENT
•	247000	FOR SINGLE TANK
)	00005 C	
	00005 C	THI THE INLET TEMPERATURE OF HOT STREAM TO TANK (C)
1994 - A.	0 0 0 0 7 C	TH2 THE EXIT TEMPERATURE OF HOT STREAM FROM TANK (C)
)	0 N G 0 3 C	TCO THE INLET TENPERATURE OF COLD STREAM TO TANK (C)
	00009 C	TCI THE EXIT TEMPERATURE OF COLD STREAM FROM TANK (C)
<b>,</b>	00010C	A FLOW RATE OF COLD STREAM (M**3/SEC)
) :	00011 C	W FLOW RATE OF COLD STREAM ("**3/SEC)
	00012°C	DT DIAMETER OF TANK (M) DC DIAMETER OF COIL (M)
)	000150	DC DIAMETER OF COIL (M) DH DIAMETER OF TUBE (M)
	6 PC 15 C	DA DIAMETER OF AGITATOR (M)
	00015 C	N AGITATOR SPEED (REV/SEC)
ì	000.17 C	X THICKNESS OF TUBE (M)
	C PE 13 C	L LENGTH OF TUDE (4)
	30017C	K THERMAL CONDUCTIVITY OF TUBE CATERIAL (W/C*N)
) · · ·	0 00 20 C	VH VELOCITY OF HOT STREAM (M/SEC)
	00021 C	CP HEAT CAPACITY (J/KG*C)
	0.0022 C	R) DENSITY (KG/***3)
a) in the	00023 C	AU VISCOSITY ( <g h*sec)<="" td=""></g>
	000240	V VOLUMEOF LIQUID IN EACH TANK (M**3)
)	37625 C 37025 C	J=1 INDICATES STEADY-STATE CASE J=2 INDICATES UNSTEADY-STATE CASE
1	0.0023C	ICOUNT INDICATES STEPS FOR STEADY-STATE CASE
	00023C	KL INDICATES STEPS FOR UNSTEADY-STATE CASE
3	000270 ····	
	26332C	
	00031	DINENSION THETA(2), TH1(2), TCO(2)
) .	00032	REAL MUHUNUCUKHUKCUNREHUNPRHUNUDHUNUDCULUMUMIUMIANKUMUM
	3 PO 33 C	
	00034	READ(5,+)DC, DH, DT, L, DA, N, X, K, V, N, M, TCO(1), TH1(1), TCO(2), TH1(2)
)	30235	WRITE(6/8) CC/DH/DT/L/DA/N/X/V/W/M/TCO(1)/TH1(1)/TCO(2)/TH1(2)
	00035.3 00037+	FORMAT(20x,*DC =*/F7_4/6x,*(4)*//20x,*DH =*/E11_4/2x,*(H)*// /20x,*DT =*/F7_4/6x,*(4)*//20x,*L =*/F7_4/6x,*(4)*//20x/*DA =
)	50633+	= E11_4/2X/ (A)1//20X/ A = = //F7_4/5X/ (REV/SEC)1//20X/ X = = //
	000:39 +	F7.476X,*(())*//20X,*V =*/E11.472X,*((***3)*//20X,*W =*/F11.47
	2004)+	2x, '(***3/SEC)', /, 20x, '* = ', E11, 4, 2x, '(***3/SEC)', /, 23x, 'TCO(1)=*
	06041+	<pre>&gt;F5_2/6x/*(c)*//20x/*TH1(1)=*/F5_2/6x/*(c)*//20x/*TCC(2)=*/F5_2/</pre>
	20042+	6x,'(C)',/,20x,'TH1(2)=',F5,2,6x,'(C)',//)
	00043 C	
	357744 C	
	0 00 45	TH GT 4(1) =5 00 00.
· · ·	00045 00047	THETA(2)=0. IC 30 MT=0
•	20247 20243	A=1.
	000420	na = 1 ● An and a standard standard standard standard standard standard standard standard standard standard st An an an an an an an an an an an an an an
¥	30353C	STEADY-STATE CASE
1.1.1.1	300510	
	- 00052 C	
1	00053	J=1
	0.0054	$VH = \frac{1}{3} ((3.1416 * DH * DH) / 4.)$
	07255	ROC = FRO(TCO(J))
	0.9955	20H= FR 0(TH 1(J))
	0.0057	SUC=FAU(TCO(J))
)	20253 20057	HUH=F-U(TH1(J)) CPC=FCP(TCJ(J))
	00007	CPH=FCP(TH1(J))
	00261	$\kappa_{C} = F \kappa (TCO(J))$
)	20262	$\kappa_{H} = FK (TH1(J))$
	0.0063	u NUN=1, and a state of the st
	00064	t WC=1. The second second second second second second second second second second second second second second s
.!	00065	uurati anti anti anti anti anti anti anti a
	A A A	
	00365	IF (ICCUNT.E1.3) GO TO 66

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00067
         11 THO=(TH2+TH1(J))/2.
0.00.63
            IF(J_GT_1) GO TO 33
000.67
            x1 = TC1
30373
            X2=TH2
00071 C
00072
        -33
            ROC = FRO(TC1)
00073
            ROH= FRO(THO)
00074
            AU C = FM U(TC1)
00075
            HUH=FHU(THO)
00075
            CPC = FCP(TC1)
00077
            CPH = FCP(THO)
00073
            KC = FK (TC1)
00077
            KH = FK (THO)
20080
            IF (ICOUNT_ED_1) GD TO 65
00081
            HUW=FXU(TW)
            HREH=(DH*(W/((3.1416*DH*DH)/4.))*ROH)/3UH
28000
        65
00083
            NPRH=(CPH+ "10"H)/KH
DODS4C LANINAR FLOW
20285
            IF (NREH. GT. 100) GO TO 83
300.35
            HUDH=1_36* ((HREH*NPRH)/(L/DH)) **0_33
00087
            HI = (KH * NUD H) / DH
0.00.83
            00 TO 55
30087C
D0091C TRANSITION REGION
30391
            IF (NREH_GT_10000) SO TO 88
        77
00092
            E=0.116*((NFEH)**0.67-125.)/NREH
00093
            X=(3_1416*R0H*VH*CPH*DH*DH)/(4_*KH*L)
000.94
            HI=1.75*KH/0H*((3.1416*0H)/(4.*L)*NREH*NPRH)**.33
0.00.95
            GO TO 55
000.95 C
10097 C
        TURBULENT FLOW
00093
            NUDH=C.023*NREH**5.8*NPRH**0.33*(HUH/4UH)**0.14
        83
300.97
            HI=((KH*NUDH)/DH)*(1+3_5*DH/DC)
            H3=(KC/DT) +1 _40+(((D4++2)+N+ROC)/NUC)++3_62
20100
        55
00101+
            *((CPC*MUC)/KC)**0_33*(MUC/MUW)**0_14
00102
            IF (ICOUNT_EA.D) GD TO 36
            DELT=(THO-TC1)/(1,+(HI/HO))
0.01.03
00104
            TW=THO-DELT
0.01.05
            U=1_{*}/(1_{*}/HI+1_{*}/HO+2_{*}6E-5)
        55
00105
            ALPHA=EXP(-U*(3_1415*DH*L)/(W*ROH*CPH))
36167 C
0.71.08
            VI = V + E OC
0.0107
            WI=W*ROH
ē0113
             4I=4*R-0C
50111
            G=VI*CPC
00112
            F= 1I + C P C + W I + C P H + (1 - AL PHA)
            H=WI*CPH*(1_-ALPHA)*TH1(J)+MI*CPC*TCO(J)
00115
00114
            B = H/F
00115
            IF(ICOUNT, 22,3) GO TO 21
00115
            TC1S=TC1
00117
            A=TC1S-B
0.01.13
            TC1=A*ExP(-(F/G)*THETA(J))+3
        21
            TH2= AL PHA * TH1(J)+(1.-ALPHA) * TC1
00119
50123
            IF (J_E 4. 1) GO TO 7
00121
            WRITE(S,1) THETA(2), TC1, TH2
            FORMAT (7 X/ F7 .2/10X/F6.3/10X/F6.3)
00122
         1
65123
            GC TO 101
00124
            IF (ICCUNT, EQ.U) GO TO 3
         7
00125
            DIFR1=X1-TC1
20125
            DIFR2=X2-TH2
20127
            IF(DIFR1.ST.0.0001) GO TO 3
            IF (DIFR2 .ST. 0.0001). GO TO 3
00123
            IF (A . 4 . 0) GD TO 3
00127
00133
            WRITE(6,5) TC1, TH2
         5 FORMAT (1 1x . ' TCIS=' . F6. 3. 5x . ' TH2S=' . F6. 3)
0.01.31
            GO TO 77
00132
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	CC133 3	ICOUNT=ICOUNT +1
,	00134	GO TO 11
	00135 99	
	00135 2	FORMAT (10X, 'TIME', 10X, 'TCOLD1', 10X, 'THOT2',
	00137+	//1UX/4('_')/10X/6('_')/10X/5('_'))
	G 01 33 C	
	00137C	UNSTEADY-STATE CASE
	00140 C	
	00141	<b>j</b> =2
	00142	KL =0.
	00143 101	THETA(J) = THETA(J) + 30.
	00144	IF (THE TA (J). GT. 1800) GO TO 111
	20145	KL=KL+1
	00145	IF (KL. GT.1) GO TO 21
	00147	GO TO 11
	CC143 111	STOP
	00149	END
	00150	FUNCTION FCP(T)
	00151	FCP=_419313E+4744578*T+_100875E-1*T*T
	00152	RETURN
	00153	END
	00154	FUNCTION FUN (T)
	00155	FXU= 148237E-2- 295743C-4*T+ 258156E-6*T*T- 822939E-9*T*T*T
	00155	RETURN
	00157	END
	30153	FUNCTION FRG (T)
	0.01.57	FRO=1./(.997426E-3+.135802E-6*T+.325184E-8*T*T)
	00160	RETURN
	07161	END
	00162	FUNCTION FK(T)
	10163	FK=_ 579671+_178690E=2*T=_684359E=5*T*T
	00164	RETURN
	30165	END
		CLP, AA, PO4 , 0.199KLNS.
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00067-0740 CPC(1) = FCP(TCO(J))CPH(1) = FCP(TH1(J))00063 00069 KC(1) = FK(TCO(1))39673 KH (1) = FK (TH1(J)) 00071 CC = CPC(1)00072 CH = CPH(1)30375 RC = ROC(1)0.0074 RH =ROH(1) 00075 MUW(1)=1. 00075 "UH(1)=1. 51677 ▶'UC(1)=1. 00073 210W(2)=1. 20277 M = IR<u>5 no 33</u> WI = d20281 VI =V 21982 C 3,00,83 IF (ICOUNT\_EQ\_D)GO TO 66 07034 11 I = 1 0.00.85 THO(1)=(TH3+TH2)/2. 00085 THO(2)=(TH2+TH1(J))/2. 20187 TC(1) = TC159085 TC(2) = TC230387 IF(J\_GT\_1) GO TO 33 0.00.90 x1 = TC219991  $x_{2} = t_{01}$ 0 nº 92 X3 =T H2 971.93 X4 = TH3 303940 301.95 33 ROC(I) = FRO(TC(I))00095 ROH(I) = FRO(THO(I))20397 AUC(I) = F AU(TC(I))200.93 HUH(I) = FHU(THO(I)) 30299 CPC(I) = FCP(TC(I))CPH(I) = FCP(THO(I))06100 30101  $KC \quad (I) = FK \quad (TC(I))$ 50102 KH (I) = FK(THO(I)) 00103 IF(ICOUNT\_E1\_1) GO TO 65 50104 HUW(I) = FHU(TW(I))IREH(I)=(DH\*(#/((3.1416\*DH\*DH)/4.))\*ROH(T))/ 10H(I) . 10103 55 0115 NPRH(I)=(CPH(I) \* 50H(I))/KH(I) 36157C LAMINAR FLOW 001.03 IF (NREH(I) .ST. 107) 50 TO 88 HUDH(I)=1, 30\*((MTEH(I)\*HPRH(I))/(L/DH))\*\*0.33 0109 (111) H1(I)=(KH(I)\*NUDH(I))/OH : 0111 GO TO 55 00112 C C 71 1 3 C TRANSITION REGION 00114 IF (MREH(I), GT.10000) (60 TO 38 77 H=D\_116\*((NRTH(I))\*\*D.67-125.)/AREH(I) 0115 1115 x=(3.1416\*R0H(I)\*VH\*CPH(I)\*DH\*OH)/(4.\*KH(I)\*L) HI(I)=1,75\*KH(I)/DH\*((3,1415\*DH)/(4,\*L)\*NREH(I)\*NPPH(I))\*\*.33 0117 0113 01170 GO TO 55 00120C TURBULENT FLOW NUOH(I)=0\_023+NREH(I) ++3\_3+NPRH(I) ++0\_33 10121 33 HI(I)=((KH(I)\*XUOH(I))/DH)\*(1+3,5\*0H/DC)\*(MUH(I)/MU/(I))\*\*C,14 20122 HO(I)=(<C(I)/DT)\*1\_40\*(((DA\*\*2)\*4\*?)C(I))/"UC(I))\*\*0+62 30123 55 5 124+ \* ((CPC(I)\*RUC(I))/<C(I))\*\*0\_33\*(MUC(I)/UUV(I))\*\*0\_14 50125 IF (ICOUNT\_E2.3) GO TO 56 20125 belt(I)=(THD(I)-TC(I))/(1.+(HI(I)/HO(I))) 01127 01123  $T_{W}(I) = THO(I) - OELT(I)$ U(I)=1=/(1\_/HI(I)+1=/HO(I)+2=65-5) 55 0.01.29 IF (ICOUNT\_EA\_U) GO TO 43 CP130 I = I + 1IF(1.GT.2) GO TU 44 60131 GO TO 33 0.0132

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00133	43	U(2)=U(1)
00134		CPH(2) = CPH(1)
00135		ROH(2) = ROH(1)
00135	44	ALPHA=ExP(-U(1)*(3,1416*DH*L)/(W*ROH(1)*CPH(1)))
60137		BETA = EXP(-U(2)*(3,1416*DH*L)/(W*RDH(2)*CPH(2)))
0 C 1 33 C		
0132		IF (ICOUNT, EQ.U) GO TO 72
0014)C	1.2.2.2.2	
00141	22	TCL=(1C1+TC2)/2.
00142		THL=(TH1(J)+TH3)/2
00143		CC=FCP(TCL)
00144		CH=FCP(THL)
00145		RC=FRO(TCL)
00145		RH = FRO(THL)
0.0147		
		$hI = h \star CC \star RC$
.301.43		WI=+*CH*RH
0.0149		VI=V*CC*RC
0.0 <b>0153</b>	72	BP=2.*(11/VI)+(W1/VI)*(2BETA-ALPHA)
	16	
20151		DP=({IT+MI}/(VI*VI)+((NI*WI)/(VI*VI))*(1-ALPHA*BETA)+
0.0152+		((VI*WI)/(VI*VI))*(1+ALPHA*3ETA-ALPHA-9ETA)
0 71 53		1=(-3P+((8P*3P)-(4*0P))** <sup>6</sup> ,5)/2,
0154		
		12=(-9P-((BP*3P)-(4*0P))**0.5)/2.
0.0155		CP=(MI+TCO(J)+TH1(J)+(JI+ALPHA+GETA+((WI+WI)/MI)
31155+		* (1-ALPHA-SETA+ALPHA*BETA)))/(#I+VI-VI*ALPHA*BETA+
20157+		((WI*WI)/HI)*(1-ALPHA~BETA+ALPHA*BETA))
0 1 53		IF(100UNT,E4.0) GO TO 21
J0157		TC1S=TC1
00160		TC 25 =T C2
00161		K = (KI/(1) * (BETA-1))
20162		B1=TC1S-K*TH1(J)-(1-K)*TC2S+(VI/1I)*11*(CP-TC2S)
56163		B2=(VI/4I)*(M2-M1)
20164		3=31/32
3-71-65		A=TC2S-B-CP
E 11 65	21	TC2= 4* EXP(11 *THETA(J))+3*EXP(12*THETA(J))+CP
0.21.67		TC1=(VI/HI)*(M1*A*EXP(H1*THETA(J))+H2*B*EXP(H2*THETA(J)))
0163+		+TC2+(41/11)*(1-BETA)*(TC2-TH1(J))
01167		TH2=3ETA*TH1(J)+(1-3ETA)*TC2
00170		TH3=ALPH≜★TH2+(1→ALPHA)★TC1
00171		
		- IF ( 1 - F 9 - T ) 63 - IP - Z * - * * * * * * * * * * * * * * * *
		IF(J_EQ.1)GO TO 7
01172		WRITE(6,1) THETA(2), TC1, TC2, TH2, TH3
	1	
00172	1	WRITE(6,1) THETA(2), TC1, TC2, TH2, TH3 FORMAT(7X, F7,2,10X, F6,3,10X, F6,3,2(9X, F6,3))
00172 00173 00174		WRITE(6,1) THETA(2), TC1, TC2, TH2, TH3 FORMAT(7, x, F7, 2, 10, x, F6, 3, 10, x, F6, 3, 2(9, x, F6, 3)) GO TO 191
00172 00173 00174 00174	7	URITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7,X,F7,2,10X,F6,3,10X,F6,3,2(9,X,F6,3)) GD TD 191 IF(1CCUNT_EU,0) GD TO 3
00172 00173 00174 00175 00175		URITE(6/1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 151 IF(1CCUNT_EU,0) GD TO 3 DIFR1=x1-TC2
00172 00173 00174 00174	7	URITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7,X,F7,2,10X,F6,3,10X,F6,3,2(9,X,F6,3)) GD TD 191 IF(1CCUNT_EU,0) GD TO 3
00172 00173 00173 00175 00175 00175 00175	7	URITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GO TO 191 IF(ICCUNT,EU,0) GO TO 3 DIFR1=x1-TC2 DIFR2=X2-TC1
0 1 72 0 1 73 0 1 74 0 1 74 0 1 75 0 1 75 0 1 77 0 1 73	7	URITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT,EU,0) GD TO 3 DIFR1=x1-TC2 D1FR2=x2-TC1 D1FR3=x3-TH2
0 1 72 0 1 73 0 1 74 0 1 74 0 1 75 0 1 75 0 1 75 0 1 73 0 1 73 0 1 79	7	URITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.EU,0) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=x4-TH3
0 1 72 0 1 73 0 01 74 0 1 75 0 1 75 0 1 75 0 1 77 0 01 73 0 01 79 0 1 20	7	$ \begin{array}{l} $$ $ $ $ $ $ $ $ $ $ $ $ $ $ $ $ $ $ $$
0 1 72 0 1 73 0 01 74 0 1 75 0 1 75 0 1 75 0 1 77 0 01 73 0 01 79 0 1 20	7	URITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.EU,0) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=x4-TH3 IF(DIFP1.GT_7,DU01) GG TO 3
0 1 72 0 1 73 0 01 74 0 1 75 0 1 75 0 1 75 0 1 75 0 01 73 0 01 73 0 01 79 0 1 20 0 1 21	7	$ \begin{array}{l} \label{eq:relation} & \text{URITE}(6,1)  \text{THETA}(2), \text{TC1}, \text{TC2}, \text{TH3} \\ & \text{FORMAT}(7,x,F7,2,10) \\ & \text{FORMAT}(7,x,F7,2,10) \\ & \text{GOTO}(10,10) \\ & GOTO$
0 1 72 0 1 73 0 1 74 0 1 75 0 1 75 0 1 75 0 1 75 0 1 77 0 1 73 0 1 79 0 1 82	7	$ \begin{array}{l} \label{eq:relation} & \text{WRITE}(6,1)  \text{THETA}(2), \text{TC1}, \text{TC2}, \text{TH3} \\ & \text{FORMAT}(7,x,F7,2,10) \\ & \text{GOTO}(10,10)
0 1 72 0 1 73 0 1 74 0 1 75 0 1 75 0 1 75 0 1 75 0 1 77 0 1 73 0 1 79 0 1 80 0 1 80 0 1 83	7	$ \begin{array}{l} \label{eq:relation} & \text{WRITE}(6,1)  T \; (\text{ETA}(2), \text{TC1}, \text{TC2}, \text{TH3}, \text{FORMAT}(7,x,\text{F7},2,1)(x,\text{F6},3,1)(x,\text{F6},3,2)(9,x,\text{F6},3)) \\ & \text{GO} \; \text{TO} \; 191 \\ & \text{IF}(1) \text{COUNT}, \text{EU}_{9}(9) \; \text{GO} \; \text{TO} \; 3 \\ & \text{DIFR} 1 = x1 - \text{TC2} \\ & \text{DIFR} 2 = x2 - \text{TC1} \\ & \text{DIFR} 3 = x3 - \text{TH2} \\ & \text{DIFR} 3 = x3 - \text{TH2} \\ & \text{DIFR} 4 = x4 - \text{TH3} \\ & \text{IF}(9) \text{IF}(91, \text{GT}, 3, \text{DUB1}) \; \text{GG} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GG} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(9) \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(2, \text{IF}(2, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(2, \text{IF}(2, \text{ST}, 3, \text{ST}, 3, \text{CUB1}) \; \text{GO} \; \text{TO} \; 3 \\ & \text{IF}(2, \text{IF}(2, \text{ST}, 3, \text{ST}, 3, \text{ST}, 3, \text{ST}, 3, \text{ST}) \\ & \text{IF}(2, \text{IF}(2, \text{ST}, 3, \text{ST}) \; \text{IF}(2, \text{ST}, 3, \text{ST})  \text{IF}(2, \text{ST}) \; \text{IF}(2, \text{ST}) \; \text{IF}(2, \text{ST}) \; \text{IF}(2, \text{ST}) \; \text{IF}(2, \text{ST}) \; \text{IF}(2, \text{ST}) \; \text{IF}(2, \text{ST})$
0 1 72 0 1 73 0 1 74 0 1 75 0 1 75 0 1 75 0 1 75 0 1 77 0 1 73 0 1 79 0 1 82	7	WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FORMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GO TO 191 IF(ICCUNT.E0,9) GO TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=x4-TH3 IF(0IFP1.GT.P.DUB1) GG TO 3 IF(0IFR2.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CUE1) GC TO 3 IF(0IFR3.ST.P.CUE1) GC TO 3 IF(0IFR4.GT.P.CUE1) GC TO 3 WRITE(6,5) TC1,TC2,TH2,TH3
0 n1 72 0 n1 73 0 01 74 0 n1 75 0 n1 75 0 n1 75 0 n1 77 0 n1 77 0 n1 83 0 n1 83 0 n1 84	7	WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FORMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GO TO 191 IF(ICCUNT.E0,9) GO TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=x4-TH3 IF(0IFP1.GT.P.DUB1) GG TO 3 IF(0IFR2.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CUE1) GC TO 3 IF(0IFR3.ST.P.CUE1) GC TO 3 IF(0IFR4.GT.P.CUE1) GC TO 3 WRITE(6,5) TC1,TC2,TH2,TH3
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un183 Un183 Un183	7	WRITE(6,1) THETA(2), TC1, TC2, TH2, TH3 FORMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GO TO 191 IF(1CCUNT.TU,9) GO TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=x4-TH3 IF(0IFP1.GT.P.DUB1) GG TO 3 IF(0IFR2.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CUB1) GG TO 3 IF(0IFR3.ST.P.CUB1) GG TO 3 IF(0IFR4.GT_P.CUB1) GG TO 3 WRITE(6,5) TC1,TC2,TH2,TH3 FORMAT(1Px,TCIS=',F6,3,5x,'TC2S=',F6,3,5x,'TH2S=',F6,3,
0 n1 72 0 n1 73 0 01 74 0 n1 75 0 n1 75 0 n1 75 0 n1 77 0 n1 73 0 n1 73 0 n1 83 0 n1 83 0 n1 84 0 n1 85 0 n1 85	7	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.E0,0) GD TO 3 DIFR1=x1-TC2 DIFR2=X2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.P.RUP1) GG TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR4.GT_P.CC1) GC TO 3 NRITE(6,6) TC1,TC2,TH2,TH3 FURMAT(1Px,*TCIS=*,F6,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6,3, 5x,*TH3S=*,F6,3)</pre>
0 n1 72 0 n1 73 0 01 74 0 n1 75 0 n1 75 0 n1 75 0 n1 77 0 n1 73 0 n1 73 0 n1 83 0 n1 83 0 n1 84 0 n1 85 0 n1 85 0 n1 85 0 n1 85 0 n1 85	7	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.TU,0) GD TO 3 DIFR1=x1-TC2 DIFR2=X2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.P.BUB1) GG TO 3 IF(0IFR2.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR4.GT_P.CC1) GC TO 3 NRITE(6,6) TC1,TC2,TH2,TH3 FURMAT(1Px,*TCIS=*,F6,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6,3, Sx,*TH3S=*,F6,3) GD TO PP</pre>
0 n1 72 0 n1 73 0 01 74 0 n1 75 0 n1 75 0 n1 75 0 n1 77 0 n1 73 0 n1 73 0 n1 83 0 n1 83 0 n1 84 0 n1 85 0 n1 85 0 n1 85 0 n1 85 0 n1 85	7	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.E0,0) GD TO 3 DIFR1=x1-TC2 DIFR2=X2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.P.RUP1) GG TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR4.GT_P.CC1) GC TO 3 NRITE(6,6) TC1,TC2,TH2,TH3 FURMAT(1Px,*TCIS=*,F6,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6,3, 5x,*TH3S=*,F6,3)</pre>
0 n1 72 0 n1 73 0 c1 74 0 n1 75 0 n1 75 0 n1 75 0 n1 77 0 n1 73 0 n1 73 0 n1 83 0 n1 83 0 n1 84 0 n1 85 0 n1 85 0 n1 85 0 n1 85 0 n1 85 0 n1 85	7 5	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FURMAT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 151 IF(ICCUNT.E0,0) GD TO 3 DIFR1=x1-TC2 DIFR2=X2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.P.RUB1) GG TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 NRITE(5,5) TC1,TC2,TH2,TH3 FURMAT(17x,*TCIS=*,F5,3,5x,*TC2S=*,F5,3,5x,*TH2S=*,F6,3, Sx,*TH3S=*,F5,3) GD TD 29 IC DUNT=ICDUNT.+1</pre>
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un185	7 5 3	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FUPMAT(7x,F7,2,10x,F6,3,10x,Fc,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.TU,0) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.P.RUB1) GG TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 IF(0IFR3.ST.P.CC1) GC TO 3 NRITE(6,6) TC1,TC2,TH2,TH3 FORMAT(17x,*TCIS=*,F6,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6.3,</pre>
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un187 Un183 Un189 Un190	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FUPMAT(7x,F7,2,10x,F6,3,10x,Fc,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.TU,D) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT_7.DUB1) GG TO 3 IF(0IFR3.ST_7.C(C1) GC TO 3 IF(0IFR3.ST_7.C(C1) GC TO 3 IF(0IFR3.ST_7.C(C1) GC TO 3 NRITE(6,5) TC1,TC2,FH2,TH3 FDRMAT(17x,'TCIS=',F5,3,5x,'TC2S=',F6,3,5x,'TH2S=',F6,3, SX,'TH3S=',F6,3) GD TD 79 IC DUNT=ICOUNT.+1 GD TO 11 NRITE(6,2)</pre>
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un185	7 5 3	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FUPMAT(7x,F7,2,10x,F6,3,10x,Fc,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.TU,D) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT_1.DUB1) GG TO 3 IF(0IFR3.ST_0.CC1) GC TO 3 IF(0IFR3.ST_0.CC1) GC TO 3 IF(0IFR3.ST_0.CC1) GC TO 3 WRITE(6,5) TC1,TC2,TH2,TH3 FDRMAT(17x,*TCIS=*,F5,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6,3, 5x,*TH3S=*,F6.3) GD TD 09 ICDUNT=ICDUNT.+1 GD TO 11 WRITE(6,2) FDRMAT(17x,*TINC*,1UX,*TC0LD1*,10x,*TC0LD2*,10x,*TH0T2*,</pre>
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un185 Un183 Un183 Un183 Un183 Un183	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FUPMAT(7x,F7,2,10x,F6,3,10x,Fc,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.TU,D) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT_1.DUB1) GG TO 3 IF(0IFR3.ST_0.CC1) GC TO 3 IF(0IFR3.ST_0.CC1) GC TO 3 IF(0IFR3.ST_0.CC1) GC TO 3 WRITE(6,5) TC1,TC2,TH2,TH3 FDRMAT(17x,*TCIS=*,F5,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6,3, 5x,*TH3S=*,F6.3) GD TD 09 ICDUNT=ICDUNT.+1 GD TO 11 WRITE(6,2) FDRHAT(10x,*TINC*,1UX,*TC0LD1*,10x,*TC0LD2*,10x,*TH0T2*,</pre>
Un172 Un172 Un173 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un183 Un183 Un191 Un192 +	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 F0 %AT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) G0 T0 101 IF(1CCUNT_E0,0) G0 T0 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 0IFR4=x4-TH3 IF(0IFR1.GT_7.AUG1) G0 T0 3 IF(0IFR2.ST_7.CCF1) G0 T0 3 IF(0IFR3.ST_7.CCF1) G0 T0 3 IF(0IFR3.ST_7.CUG1) G0 T0 3 IF(0IFR4.GT_7.CUG1) G0 T0 3 NRITE(6,5) TC1,TC2,TH2,TH3 F0 %AT(17x,'TCIS=',F0,3,5x,'TC2S=',F6,3,5x,'TH2S=',F6,3,5x,</pre>
Un172 Un172 Un173 Un175 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un183 Un183 Un191 Un192 + Un195 +	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FUPMAT(7x,F7,2,10x,F6,3,10x,Fc,3,2(9x,F6,3)) GD TD 191 IF(ICCUNT.TU,D) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=X3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT_7.DUB1) GG TO 3 IF(0IFR3.3T_7.CC1) GC TO 3 IF(0IFR3.3T_7.CC1) GC TO 3 IF(0IFR3.5T_7.CC1) GC TO 3 WRITE(6,5) TC1,TC2,TH2,TH3 FDRMAT(17x,*TCIS=*,F5,3,5x,*TC2S=*,F6,3,5x,*TH2S=*,F6,3, 5x,*TH3S=*,F6,3) GD TD 79 ICDUNT=ICDUNT.+1 GD TO 11 WRITE(6,2) FDRHAT(17x,*TINC*,1UX,*TC0LD1*,10x,*TC0LD2*,10x,*TH0T2*,</pre>
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un177 Un177 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un183 Un183 Un192 Un192 + Un193 + Un194 C	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FD9MAT(7x,F7.2/10x,F6.3/10x,F6.3/2(9x,F6.3)) GD TD 151 IF(1CCUNT.TU,D) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 OIFR4=x4-TH3 IF(0IF91.6T. 7.0001) G0 TO 3 IF(0IF82.ST. 7.0001) G0 TO 3 IF(0IF83.ST. 7.0001) G0 TO 3 REITE(6,5) TC1,TC2,FH2,TH3 FD9MAT(17x,'TCIS=',F6.3,5x,'TC2S=',F6.3,5x,'TH2S=',F6.3, Sx,'TH3S=',F6.3) GD TD 09 ICOUNT=ICOUNT.+1 GD TO 11 WRITE(6,2) FDFMAT(17x,'TINC',10x,'TC0LD1',10x,'TC0LD2',10x,'TH0T2', 10x,'TH0F3',//10x,4('_'),10x,6('_'),10x,6('_'),10x,5('_'), 10x,5('_'))</pre>
Un172 Un173 UC174 Un175 Un175 Un175 Un175 Un177 Un177 Un177 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un185 Un183 Un183 Un192 Un192 + Un193 + Un194 C	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 F0 %AT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) G0 T0 101 IF(1CCUNT_E0,0) G0 T0 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 0IFR4=x4-TH3 IF(0IFR1.GT_1.RU01) G0 T0 3 IF(0IFR2.ST_1.CL01) G0 T0 3 IF(0IFR3.ST_0.CU01) G0 T0 3 IF(0IFR4.GT_0.CU01) G0 T0 3 NRITE(6,6) TC1,TC2,TH2,TH3 F0 %AT(17x,'TCIS=',F0,3,5x,'TC2S=',F6,3,5x,'TH2S=',F6,3, 5x,'TH3S=',F6,3) G0 T0 49 IC 0UNT=IC0UNT_+1 G0 T0 41 NRITE(6,2) F0 RHAT(10x,'TINC',10x,'TC0L01',10x,'TC0L02',10x,'TH0T2', 10x,'TH0F3',7.10x,4('_'),10x,6('_'),10x,6('_'),10x,5('_'),</pre>
Un172 Un172 Un173 Un175 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un183 Un192 Un192 + Un195 + Un195 + Un195 C	7 5 3 99	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FD9MAT(7x,F7.2/10x,F6.3/10x,F6.3/2(9x,F6.3)) GD TD 151 IF(1CCUNT.TU,D) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 OIFR4=x4-TH3 IF(0IF91.6T. 7.0001) G0 TO 3 IF(0IF82.ST. 7.0001) G0 TO 3 IF(0IF83.ST. 7.0001) G0 TO 3 REITE(6,5) TC1,TC2,FH2,TH3 FD9MAT(17x,'TCIS=',F6.3,5x,'TC2S=',F6.3,5x,'TH2S=',F6.3, Sx,'TH3S=',F6.3) GD TD 09 ICOUNT=ICOUNT.+1 GD TO 11 WRITE(6,2) FDFMAT(17x,'TINC',10x,'TC0LD1',10x,'TC0LD2',10x,'TH0T2', 10x,'TH0F3',//10x,4('_'),10x,6('_'),10x,6('_'),10x,5('_'), 10x,5('_'))</pre>
Un172 Un172 Un173 Un175 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un183 Un192 Un192 + Un192 + Un193 + Un195 C	7 5 3 99 2	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FDPMAT(7x,F7.2,10x,F6.3,10x,F6.3,2(9x,F6.3)) GD TD 101 IF(ICCUNT.TU,U) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.1.DUP1) GG TO 3 IF(0IFP1.GT.1.DUP1) GG TO 3 IF(0IFP2.ST.1.DUP1) GG TO 3 IF(0IFP2.ST.1.DUP1) GG TO 3 REITE(6,5) TC1,TC2,TH2,TH3 FDPMAT(17x,'TC1S=',F6.3,5x,'TC2S=',F6.3,5x,'TH2S=',F6.3, GD TD 29 ICDUNT=ICDUNT.+1 GD TO 11 WRITE(6,2) FDFHAT(17x,'TINC',10x,'TC0LD1',10x,'TC0LD2',10x,'TH0T2', 10x,'TH0T3',/10x,4('_'),10X,6('_'),10X,6('_'),10X,5('_'), UNSTEADY-STATE CASE</pre>
Un172 Un172 Un173 Un175 Un175 Un175 Un175 Un175 Un177 Un177 Un177 Un183 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un185 Un187 Un192 + Un192 + Un195 C Un197	7 5 3 99 2	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FD @ AT(7x,F7,2,10x,F6,3,10x,F6,3,2(9x,F6,3)) GD TD 101 IF(ICCUNT_TU,D) GD TO 3 DIFR1=x1-TC2 D1FR2=x2-TC1 D1FR3=x3-TH2 UFR4=K4-TH3 IF(D1FP1,GT_P,DUP1) GG TO 3 IF(D1FP2,GT_P,CUP1) GC TO 3 IF(D1FP2,GT_P,CUP1) GC TO 3 IF(D1FP2,GT_P,CUP1) GC TO 3 WRITE(6,5) TC1,TC2,TH2,TH3 FD @ AT(17x,TC1S=',F0,3,5x,'TC2S=',F6,3,5x,'TH2S=',F6,3, Sx,'TH3S=',F6,3) GD TD 79 IC DURT=ICDURT_+1 BD TO 11 WRITE(6,2) FD FHAT(10x,'TIM2',10x,'TC0LD1',10x,'TC0LD2',10x,'TH0T2', 10x,'TH0F3',/,10x,4('_'),10x,6('_'),10x,6('_'),10x,5('_'), 10x,S('_')) UNS TEADY-STATE CASE 'J=2</pre>
Un172 Un172 Un173 Un175 Un175 Un175 Un175 Un175 Un177 Un179 Un183 Un183 Un183 Un185 Un185 Un185 Un185 Un183 Un192 Un192 + Un192 + Un193 + Un195 C	7 5 3 99 2	<pre>WRITE(6,1) THETA(2),TC1,TC2,TH2,TH3 FDPMAT(7x,F7.2,10x,F6.3,10x,F6.3,2(9x,F6.3)) GD TD 101 IF(ICCUNT.TU,U) GD TO 3 DIFR1=x1-TC2 DIFR2=x2-TC1 DIFR3=x3-TH2 DIFR4=X4-TH3 IF(0IFP1.GT.1.DUP1) GG TO 3 IF(0IFP1.GT.1.DUP1) GG TO 3 IF(0IFP2.ST.1.DUP1) GG TO 3 IF(0IFP2.ST.1.DUP1) GG TO 3 REITE(6,5) TC1,TC2,TH2,TH3 FDPMAT(17x,'TC1S=',F6.3,5x,'TC2S=',F6.3,5x,'TH2S=',F6.3, GD TD 29 ICDUNT=ICDUNT.+1 GD TO 11 WRITE(6,2) FDFHAT(17x,'TINC',10x,'TC0LD1',10x,'TC0LD2',10x,'TH0T2', 10x,'TH0T3',/10x,4('_'),10X,6('_'),10X,6('_'),10X,5('_'), UNSTEADY-STATE CASE</pre>

00197	1.01	THETA(J) = THETA(J) + 3D.
0.020.0		IF (THE TA (J). GT. 2400) GO TO 111
30261		KI = KI + 1
00202		IF(KL, GT. 1) GO TO 21
00203		GO TO 11
00204		
002.05		END
50205		FUNCTION FCP(T)
00207		FCP=_41731SE+4744673*T+_100875E-1*T**2=
30263		RETURN
00209		END •••
00210		FUNCTION FILL(T)
00211		F4U= 148237E-2- 295743E-4*T+ 258156E-6*T*T- 322939E-9*T*T*T
00212		RETURN
30213		END Contraction of the second s
00214		FUNCTION FRU(T)
20215		FRO=1./(.797426E-3+.135802E-6*T+.325184E-8*T*T)
00215		RETURN
00217		END
00213		FUNCTION FR(T)
30219		Fx = _ 57 7571 + _ 17 8690 E - 2 * T 684359 E - 5 * T * T
0,02,20		RETURN
00221		END a strand the factor of the second strategies of the second strategi
39.51	_4?_U	CLP, AA, PJ4 , 9.255KLNS.

)

# APPENDIX IV PHYSICAL PROPERTIES OF WATER

In this section physical properties of water as a function of temperature are given [35].

A.4.A FUNCTION FK(T)

FK(T) subprogram, using following equation

 $K = 0.570671 + 0.178690 \times 10^{-2} T - 0.684359 \times 10^{-5} T^{2}$ 

calculates thermal conductivity of water at a given temperature.

#### A.4.B FUNCTION FMU(T)

FMU(T) subprogram calculates viscosity of water at a given temperature, using equation given below.

 $\mu = 0.148237 \times 10^{-2} - 0.295743 \times 10^{-4} T + 0.258156 \times 10^{-6} T^2 - 0.822939 \times 10^{-9} T^3$ 

#### A.4.C FUNCTION FCP(T)

FCP(T) subprogram calculates specific heat of water at a given temperature, using following equation.

 $CP = 0.419318 \times 10^4 - 0.744678T + 0.100875 \times 10^{-1}T^2$ 

#### A.4.D FUNCTION FRO(T)

FRO(T) subprogram calculates density of water at a given temperature, using following equation

 $\rho = 1/(0.997426 \times 10^{-3} + 0.135802 \times 10^{-6} T + 0.325184 \times 10^{-8} T^2)$ 

## APPENDIX V

### DERIVATION OF ERROR EQUATIONS

Empirical equation used to calculate film heat transfer coefficient in agitated side of vessel is,

H0 = 1.40(
$$D_{T}^{-1}$$
)k( $D_{A}^{2}N\rho/\mu$ )<sup>0</sup>·<sup>62</sup>( $C_{p}\mu/k$ )<sup>1/3</sup>( $\mu/\mu_{w}$ )<sup>0</sup>·<sup>14</sup>

where

$$\rho = f(T), \quad \mu = f(T), \quad C_p = f(T), \quad \mu_w = w(T_w)$$

and

$$HO = f(D_{T}, D_{A}, N, T, T_{W})$$
  
$$HO = 1.40D_{T}^{-1}D_{A}^{1 \cdot 2^{4}}N^{0 \cdot 6^{2}}k^{0 \cdot 7^{7}}\rho^{0 \cdot 6^{2}}C_{p}^{0 \cdot 3^{3}}\mu^{-0 \cdot 1^{5}}\mu_{W}^{-0 \cdot 1^{4}}$$
(A.5.1)

Taking the partial derivatives of Eq. (A.5.1)

$$dHO = \frac{\partial f}{\partial D_{T}} dD_{T} + \frac{\partial f}{\partial D_{A}} dD_{A} + \frac{\partial f}{\partial N} dN + \frac{\partial f}{\partial T} dT + \frac{\partial f}{\partial T_{W}} dT_{W}$$
(A.5.2)

$$\frac{\partial f}{\partial D_{c}} = 1.40 \left[ -D_{T}^{-2} \right] k^{0} \cdot {}^{77} D_{A}^{1} \cdot {}^{24} N^{0} \cdot {}^{62} \rho^{0} \cdot {}^{62} C_{p}^{0} \cdot {}^{33} \mu_{W}^{-0} \cdot {}^{14} \mu^{-0} \cdot {}^{15}$$

$$\frac{\partial f}{\partial D_{A}} = 1.40[1.24D_{A}^{0} \cdot {}^{24}]D_{T}^{-1}N^{0} \cdot {}^{62}k^{0} \cdot {}^{77}\rho^{0} \cdot {}^{62}C_{p}^{0} \cdot {}^{33}\mu_{w}^{-0} \cdot {}^{14}\mu^{-0} \cdot {}^{15}$$

$$\begin{aligned} \frac{\partial f}{\partial N} &= 1.40[0.62N^{-0} \cdot {}^{38}]D_{T}^{-1}D_{A}^{1} \cdot {}^{24}N^{0} \cdot {}^{62}k^{0} \cdot {}^{77}\rho^{0} \cdot {}^{62}C_{p}^{0} \cdot {}^{33}\mu_{w}^{-0} \cdot {}^{14}\mu^{-0} \cdot {}^{15} \\ \frac{\partial f}{\partial T_{w}} &= 1.40[-0.14(\mu_{w})^{+}\mu_{w}^{-1} \cdot {}^{14}]D_{T}^{-1}D_{A}^{1} \cdot {}^{24}N^{0} \cdot {}^{62}k^{0} \cdot {}^{77}\rho^{0} \cdot {}^{62}C_{p}^{0} \cdot {}^{33}\mu_{w}^{-0} \cdot {}^{14}\mu^{-0} \cdot {}^{15} \\ \frac{\partial f}{\partial T} &= 1.40[0.77(k)^{+}k^{-0} \cdot {}^{33}\rho^{0} \cdot {}^{62}C_{p}^{0} \cdot {}^{33}\mu^{-0} \cdot {}^{15} + 0.62(\rho)^{+}\rho^{-0} \cdot {}^{38}k^{0} \cdot {}^{77} \\ &\cdot C_{p}^{0} \cdot {}^{33}\mu^{-0} \cdot {}^{15} + 0.33(C_{p})^{+}C_{p}^{-0} \cdot {}^{77}k^{0} \cdot {}^{33}\rho^{0} \cdot {}^{62}\mu^{-0} \cdot {}^{15} \\ &- 0.15(\mu)^{+}\mu^{-1} \cdot {}^{15}k^{0} \cdot {}^{77}\rho^{0} \cdot {}^{77}C_{p}^{0} \cdot {}^{33}]D_{T}^{-1}D_{A}^{1} \cdot {}^{24}N^{0} \cdot {}^{62} \end{aligned}$$

where,

$$(\mu_w)'$$
,  $(\mu)'$ ,  $(k)'$ ,  $(C_p)'$ ,  $(\rho)'$  are derivative values.

Provided the errors  $\delta HO$ ,  $\delta D_C$ ,  $\delta D_A$ ,  $\delta N$ ,  $\delta T$  and  $\delta T_W$  are small, dHO,  $dD_C$ ,  $dD_A$ , dN, dT,  $dT_W$  can be replaced by them in Equation (A.5.2). Thus

$$\delta H0 = 1.40 D_{T}^{-1} D_{A}^{1} \cdot {}^{24} N^{0} \cdot {}^{62} k^{0} \cdot {}^{77} C_{p}^{0} \cdot {}^{33} \rho^{0} \cdot {}^{62} \mu_{W}^{-0} \cdot {}^{14} \mu^{-0} \cdot {}^{15} [-(\delta D_{T} / D_{T})]$$

$$+ 1.24 (\delta D_{A} / D_{A}) + 0.62 (\delta N / N) - 0.14 (\mu_{W})' (\delta T_{W} / \mu_{W})$$

$$+ (0.77 ((k)' / k) + 0.62 ((\rho)' / \rho) + 0.33 ((C_{p})' / C_{p})]$$

$$- 0.15 ((\mu)' / \mu) \delta T]$$

## finally;

$$\begin{split} \delta HO/HO &= \left| -\delta D_{T}/D_{T} \right| + \left| 1.24\delta D_{A}/D_{A} \right| + \left| 0.62\delta N/N \right| + \left| -0.14(\mu_{w})'(\delta T_{w}/\mu_{w}) \right| \\ &+ \left| 0.77(k)'(\delta T/k) \right| + \left| 0.62(\rho)'(\delta T/\rho) \right| + \left| 0.33(C_{p})'(\delta T/C_{p}) \right| \\ &+ \left| -0.15(\mu)'(\delta T/\mu) \right| \end{split}$$

is found. Knowing the all other variables HO and so, unknown true value of HO' can be calculated.

Empirical equation used to calculate film heat transfer coefficient in the coil side is;

$$HI = 0.023k \ D_{H}^{-1}(C_{p}\mu/k)^{0} \cdot {}^{33}(D_{H}V_{H}/\mu)^{0} \cdot {}^{8}(\mu/\mu_{W})^{0} \cdot {}^{14}(1 + 3.5(D_{H}/D_{C})^{0})^{0}$$

$$V_{H} = \frac{W}{(3.1416/4)D_{H}^{2}}$$

$$HI = 0.028k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_{p}^{0} \cdot {}^{33} \ \mu_{W}^{-0} \cdot {}^{14}D_{H}^{-1} \cdot {}^{8}W^{0} \cdot {}^{8}$$

$$+ 0.098k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_{p}^{0} \cdot {}^{33}\mu_{W}^{-0} \cdot {}^{14}W^{0} \cdot {}^{8}D_{C}^{-1}D_{H}^{-0} \cdot {}^{8} \qquad (A.5.3)$$

It can be shown that

$$HI = f_1(D_H, W, T, T_w) + f_2(W, D_H, D_c, T, T_w)$$

Taking the partial derivative of Equation (A.5.3)

$$dHI = (\partial f_{1}/\partial D_{H})dD_{H} + (\partial f_{1}/\partial W)dW + (\partial f_{1}/\partial T_{W})dT_{W} + (\partial f_{1}/\partial T)dT + (\partial f_{2}/\partial D_{H}) + (\partial f_{2}/\partial W)dW + (\partial f_{2}/\partial D_{C})dD_{C} + (\partial f_{2}/\partial T_{W})dT_{W} + (\partial f_{2}/\partial T)dT (A.5.4) 
$$\frac{\partial f_{1}}{\partial D_{H}} = 0.028[-1.8D_{H}^{-2} \cdot {}^{8}]k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_{p}^{0} \cdot {}^{33}\mu_{W}^{-0} \cdot {}^{14}W^{0} \cdot {}^{8} \frac{\partial f_{1}}{\partial V_{H}} = 0.023[0.8W^{0} \cdot {}^{2}]k^{0} \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_{p}^{0} \cdot {}^{33}\mu_{W}^{-0} \cdot {}^{14}D_{H}^{-1} \cdot {}^{8}$$$$

$$\frac{\partial f_2}{\partial D_H} = 0.098[-0.8D_H^{-1} \cdot {}^8]k^0 \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33}\mu_W^{-0} \cdot {}^{14}W^0 \cdot {}^8D_c^{-1}$$

$$\frac{\partial f_2}{\partial D_c} = 0.098[-D_c^{-2}]k^0 \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33}\mu_W^{-0} \cdot {}^{14}W^0 \cdot {}^8D_H^{-0} \cdot {}^8$$

$$\frac{\partial f_2}{\partial W} = 0.098[0.8W^{-0} \cdot {}^2]k^0 \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33}\mu_W^{-0} \cdot {}^{14}D_c^{-1}D_H^{-0} \cdot {}^8$$

$$\frac{\partial f_2}{\partial W} = 0.098[-0.14(\mu_W) \cdot {}^{-1}\mu_W^{-1} \cdot {}^{14}]k^0 \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33}W^0 \cdot {}^8D_c^{-1}D_H^{-0} \cdot {}^8$$

$$\frac{\partial f_2}{\partial T_W} = 0.098[-0.14(\mu_W) \cdot {}^{-1}\mu_W^{-1} \cdot {}^{14}]k^0 \cdot {}^{77}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33}W^0 \cdot {}^8D_c^{-1}D_H^{-0} \cdot {}^8$$

 $\frac{\partial f_1}{\partial T_W} = 0.023[-0.14(\mu_W)^* \mu_W^{-1} \cdot {}^{14}] k^0 \cdot {}^{77} \mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33}D_H^{-1} \cdot {}^{8}W^0 \cdot {}^{8}$ 

+ 0.33(C<sub>p</sub>)'C<sub>p</sub><sup>-0</sup>·<sup>77</sup>k<sup>0</sup>·<sup>77</sup> $\mu$ <sup>-0</sup>·<sup>33</sup>]D<sub>H</sub><sup>-1</sup>·<sup>8</sup>W<sup>0</sup>·<sup>8</sup> $\mu_W$ <sup>-0</sup>·<sup>14</sup>

 $\frac{\partial f_1}{\partial T} = 0.023[(0.77(k)'k^{-0} \cdot {}^{33}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33} - 0.33(\mu)'\mu^{-1} \cdot {}^{33}k^0 \cdot {}^{77}C_p^0 \cdot {}^{33}$ 

$$\frac{\partial f_2}{\partial T} = 0.098[0.77(k)'k^{-0} \cdot {}^{33}\mu^{-0} \cdot {}^{33}C_p^0 \cdot {}^{33} - 0.33(\mu)'\mu^{-1} \cdot {}^{33}k^0 \cdot {}^{77}C_p^0 \cdot {}^{33}$$
$$+ 0.33(C_p)'C_p^{-0} \cdot {}^{77}k^0 \div {}^{77}\mu^{-0} \cdot {}^{33}]D_H^{-0} \cdot {}^{8}D_C^{-1}W^0 \cdot {}^{8}\mu_w^{-0} \cdot {}^{14}$$

Provided the errors  $\delta HI$ ,  $\delta D_c$ ,  $\delta D_H$ ,  $\delta W$ ,  $\delta T$ ,  $\delta T_W$  are small enough dHI,  $dD_c$ ,  $dD_H$ , dW, dT,  $dT_W$  can be replaced by them in Equation (A.5.4). Thus

$$\begin{split} \delta HI &= 0.028 k^{0} \cdot {}^{77} \mu^{-0} \cdot {}^{33} C_p^{0} \cdot {}^{33} \mu_W^{-0} \cdot {}^{14} W^{0} \cdot {}^{8} D_H^{-1} \cdot {}^{8} [-1.8 (\delta D_H / D_H) \\ &+ 0.8 (\delta W / W) - 0.14 (\delta_W) ' (\delta T_W / \mu_W) + (0.77 ((k)' / k) \\ &- 0.33 ((\mu)' / \mu) + 0.33 ((C_p)' / C_p)) \delta T \\ &+ 0.098 k^{0} \cdot {}^{77} \mu^{-0} \cdot {}^{33} C_p^{0} \cdot {}^{33} \mu_W^{-0} \cdot {}^{14} D_H^{-0} \cdot {}^{8} W^{0} \cdot {}^{8} D_c^{-1} [0.8 (\delta W / W) \\ &- (\delta D_c / D_c) - 0.8 (\delta D_H / D_H) - 0.14 (\mu_W)' (\delta T_W / \mu_W) \\ &+ 0.77 (k)' (\delta T / k) - 0.33 (\mu)' (\delta T / \mu)] + 0.33 ((C_p)' / C_p) \delta T \end{split}$$

and final form of the equation is,

$$\begin{split} \delta HI &= HI[|-0.8(\delta D_{H}/D_{H})| + |0.8(\delta W/W)| + |-0.14((\mu_{W})'/\mu_{W})(\delta T_{W})| \\ &+ |0.77((k)'/k)| + |0.33((C_{p})'/C_{p})| - |0.33((\mu)'/\mu)|\delta T] \\ &+ |0.028k^{0} \cdot {}^{77}C_{p}^{0} \cdot {}^{33}\mu^{-0} \cdot {}^{33}\mu_{W}^{-0} \cdot {}^{14}W^{0} \cdot {}^{8}D_{H}^{-1} \cdot {}^{8}(-\delta D_{H}/D_{H}) \\ &+ |0.098k^{0} \cdot {}^{77}C_{p}^{0} \cdot {}^{33}\mu^{-0} \cdot {}^{33}\mu_{W}^{-0} \cdot {}^{14}W^{0} \cdot {}^{8}D_{H}^{-1} \cdot {}^{8}D_{L}^{-1}(-\delta D_{L}/D_{L})| \end{split}$$