## DOKUZ EYLÜL UNIVERSITY

GRADUATE SCHOOL OF NATURAL AND APPLIED SCIENCES

# THEORETICAL AND EXPERIMENTAL STUDY OF TRANSIENT FLOWS IN PRESSURISED PIPELINES 

by
Nuri Seçkin KAYIKÇI

September, 2007
İZMİR

# THEORETICAL AND EXPERIMENTAL STUDY OF TRANSIENT FLOWS IN PRESSURISED PIPELINES 

A Thesis Submitted to the<br>Graduate School of Natural and Applied Sciences of Dokuz Eylül University<br>In Partial Fulfillment of the Requirements for the Degree of Doctor of Philosophy in Civil Engineering, Hydraulics, Hydrology and Water Resources<br>Engineering program

by
Nuri Seçkin KAYIKÇI

September, 2007
İZMİR

## Ph.D. THESIS EXAMINATION RESULT FORM

We have read the thesis entitled "THEORETICAL AND EXPERIMENTAL STUDY OF TRANSIENT FLOWS IN PRESSURISED PIPELINES" completed by NURİ SEÇKİN KAYIKÇI under supervision of PROF. DR. M. ŞÜKRÜ GÜNEY and we certify that in our opinion it is fully adequate, in scope and in quality, as a thesis for the degree of Doctor of Philosophy.

Supervisor

Thesis Committee Member
$\qquad$
$\qquad$

Examining Committee Member

Thesis Committee Member

Examining Committee Member

## Prof.Dr. Cahit HELVACI

Director
Graduate School of Natural and Applied Sciences

## ACKNOWLEDGMENTS

Author thank to Prof. Dr. M. Şükrü GÜNEY for his helps and directions during the setting up of the pipelines in the laboratory and due to financial supports of project department of Dokuz Eylül University.

Nuri Seçkin KAYIKÇI

# THEORETICAL AND EXPERIMENTAL STUDY OF TRANSIENT FLOWS IN PRESSURISED PIPELINES 


#### Abstract

The aim of the study is the investigation of unsteady flows due to pump rundown in pressurised steel pipeline systems. The numerical results are found by means of a computer program using characteristics method. Theoretical computations realised by Fortran computer programs and experimental results obtained from pressure transient data logger were compared to each other and all results were evaluated. Centrifugal pump was shut down manually and developed pressure heads were drawn on the same graphics. Experiments are carried out for different steady state discharges. Results are observed and examined about the response behaviour of the computer programs against to more realistic results provided by the pressure transient data logger. It is observed that the theoretical and experimental results are in an acceptable accord in the first system. This accordance is better for lower and moderate discharges. The accord between theoretical and experimental results may be improved by translating the behaviour of the check valve more realistically in second system.


Keywords : Unsteady Flows, Centrifugal Pump, Pump rundown, Method of Characteristics, Steel Pipeline, Reduced Bore Ball Valve, Disc Type Check Valve

# BASINÇLI BORULARDA KARARSIZ AKIMLARIN TEORİK VE DENEYSEL OLARAK ÇALIŞILMASI 


#### Abstract

ÖZ

Çalışmanın amacı basınçlı çelik boru sistemlerinde pompanın anlık olarak durmasından dolayı oluşan kararsız akımların araştırılmasıdır. Karakteristikler metodunu kullanarak bilgisayar programı yardımıyla sayısal sonuçlar bulunmuştur. Fortran bilgisayar programı ile gerçekleştirilmiş teorik hesaplamalar ve kararsız akım basınç ölçer cihazından elde edilmiş deneysel sonuçlar birbirleriyle karşlaştırılmış ve değerlendirilmiştir. Santrifüj pompası elle kapatılmış ve ölçülen basınç yükseklikleri hesaplanan teorik değerlerle beraber aynı şekil üzerinde gösterilmiştir. Deneyler farklı kararlı akım debi değerlerinde gerçekleştirilmiştir. Birinci sistemde teorik ve deneysel neticelerin kabul edilebilir mertebede uyumlu oldukları gözlenmiştir. Bu uyum düşük ve orta kararlı akım debileri için daha iyi olmuştur. İkinci sistemde ise çek-valfin davranışı çok iyi yansıtılamadığından dolayı teorik ve deneysel neticeler arasında uyumsuzluklar gözlenmiştir. Kabul edilebilir bir uyum çek-valfin davranışının daha kapsamlı bir şekilde araştırılıp belirlenmesi ile mümkün olabilecektir.


Anahtar kelimeler : Kararsız akımlar, Santrifüj pompası, Pompanın durması, Karaktaristikler Metodu, Çelik boru hattı, Dar geçişli küresel vana, Disk tipi çekvalf

## CONTENTS

Page
THESIS EXAMINATION RESULT FORM ..... ii
ACKNOWLEDGEMENTS ..... iii
ABSTRACT ..... iv
ÖZ ..... v
CHAPTER ONE - INTRODUCTION ..... 1
1.1 Definitions ..... 1
1.2 History of water hammer analysis .....  1
1.3 The scope of the study ..... 9
CHAPTER TWO - BASIC EQUATIONS OF UNSTEADY FLOW ..... 10
2.1 Equation of Motion ..... 10
2.2 Continuity Equation ..... 13
2.3 Celerity in pipelines ..... 17
CHAPTER THREE -NUMERICAL SOLUTION WITH THE USE OF METHOD OF CHARACTERISTICS ..... 19
3.1 Characteristic Equations ..... 19
3.2 Equations in Finite Difference Form ..... 22
3.3 Boundary Conditions ..... 25
3.4 Courant Criterion ..... 27
3.5 Losses at regulating valve and relevant equations ..... 28
3.5.1 Positive Flow ..... 28
3.5.2 Negative (Reverse) Flow ..... 30
3.6 Calculation of Darcy Weisbach Friction Coefficient ..... 31
4.1 Introduction ..... 32
4.2 Similarity laws and pump characteristics ..... 33
4.3 Head balance and torque angular reducing equations for the $1^{\text {st }}$ system ..... 48
4.3.1 Head balance equation for the $1^{\text {st }}$ system ..... 48
4.3.2 Speed change equation for the first system ..... 51
4.3.3 Newton Raphson numerical method for first system. ..... 53
4.4 Head balance and torque angular reducing equations for the $2^{\text {nd }}$ system. ..... 56
4.4.1 Head balance equation for the $2^{\text {nd }}$ system ..... 56
4.4.2 Torque angular deceleration equation (calculation of speed change equation) for the second system ..... 58
4.4.3 Newton Raphson method for the second system ..... 58
4.5 Use of the Newton Raphson numerical method for determination of steady state parameters ..... 59
4.6 Presence of disc type check valve in second system ..... 61
CHAPTER FIVE - EXPERIMENTAL SET UP ..... 62
5.1 The first experimental set up ..... 62
5.2 The second experimental set up ..... 66
5.3 Pump characteristic curves ..... 70
5.4 Pressure transient data logger ..... 81
5.5 Computation of valve and elbow loss coefficients ..... 82
5.5.1 The use of the differential manometer ..... 82
5.5.2 The loss coefficients of reduced bore ball valve ..... 83
5.5.3 The loss coefficients of elbows. ..... 85
5.6 The loss coefficients of disc type check valve and valves for the second system ..... 86
5.7 Calibration of triangular weir ..... 88
CHAPTER SIX - DESCRIPTION OF COMPUTER PROGRAMS ..... 90
6.1 The flow chart of the computer program. ..... 90
6.2 Description of parameters used in the computer programs ..... 91
6.2.1 Fortran computer program used for the first system ..... 91
6.2.2 Fortran computer program used for the second system ..... 94
CHAPTER SEVEN - EXPERIMENTAL RESULTS ..... 95
7.1 Results of experiments performed in the first system ..... 95
7.1.1 Results of experiments performed for lower steady state discharge ..... 95
7.1.2 Results of experiments performed for moderate steady state discharge.. 98 ..... 98
7.1.3 Results of experiments performed for upper steady state discharge. ..... 100
7.2 Results of experiments performed in the second system ..... 100
7.2.1 Results of experiment performed for lower steady state discharges ..... 101
7.2.2 Results of experiment performed for moderate steady state discharge.. 106
7.2.3 Results of experiment performed for upper steady state discharge ..... 109
CHAPTER EIGHT - THEORETICAL RESULTS ..... 112
8.1 Results of computations performed for the first system ..... 112
8.1.1 Results of computation performed for lower steady state discharge ..... 112
8.1.2 Results of computation performed for moderate steady state discharge ..... 16
8.1.3 Results of computation performed for upper steady state discharge ..... 121
8.2 Results of computations performed for the second system ..... 124
8.2.1 Results of computation performed for lower steady state discharges ..... 124
8.2.2 Results of computation performed for moderate steady state discharge 129 ..... 129
8.2.3 Results of computation performed for upper steady state discharges. ..... 135
CHAPTER NINE - COMPARISON OF EXPERIMENTAL AND THEORETICAL RESULTS ..... 140
9.1 Comparison of experimental results with those obtained from computations performed for the first system ..... 140
9.1.1 Comparison of experimental results with theoretical results performed for lower steady state discharges ..... 140
9.1.2 Comparison of experimental results with theoretical results performed for moderate steady state discharges ..... 141
9.1.3 Comparison of experimental results with theoretical results performed for upper steady state discharge ..... 143
9.2 Comparison of experimental results with those obtained from computations performed for the second system ..... 143
9.2.1 Comparison of experimental results with theoretical results performed for lower steady state discharge ..... 144
9.2.2 Comparison of experimental results with theoretical results performed for moderate steady state discharges ..... 146
9.2.3 Comparison of experimental results with theoretical results performed for upper steady state discharges ..... 150
CHAPTER TEN - CONCLUSIONS ..... 152
REFERENCES ..... 154
APPENDIX A ..... 159
APPENDIX B ..... 172
APPENDIX C. ..... 178

## CHAPTER ONE

## INTRODUCTION

### 1.1 Definitions

The flow processes are governed by equation of motion (or momentum, dynamic) and equation of continuity (conservation of mass). In steady flow, there is no change in conditions at a point with time. In unsteady flow, conditions at a point may change with time. Steady flow is a special case of unsteady flow in which the steady flow equations must satisfy. The terms water hammer and transient flow are used synonymously to describe unsteady flow of fluids in pipelines, although use of the former is customarily restricted to water. The term surge refers to those unsteady flow situations that can be analysed by considering the fluid to be incompressible and the conduit wall rigid. Liquid column separation refers to the situation in a pipeline in which gas and (or) vapour is collected at some section (Wylie \& Streeter, 1993).

### 1.2 History of the water hammer analysis

The study of transient flows has been started with the study of sound waves in the air. Separately, studies related to wave propagation inside the shallow water and blood flow in the artery have been also performed.

Newton has been studied the propagation of the sound waves in the air and water waves into the channels. Lagrange has been examined sound velocity in the air theoretically as well. Euler derived partial differential equation for propagated wave. Lagrange analysed flows of compressible and incompressible fluids. For this objective, he developed concept of velocity potential (Chaudhry, 1987).

In 1789, Monge developed a graphical method integrating partial differential equations. Method of characteristics was described. Young studied the propagation of pressure wave inside the pipes. Helmholtz has attracted attention to pressure wave velocity in the water inside the pipes, which was less than in unconfined water. He
interpreted this difference to elasticity of pipe wall accurately. In 1850's Weber studied the incompressible fluid flow in the elastic pipes. He conducted the experiments and specified the velocity of pressure waves. He developed dynamic and continuity equations. Resal has developed second order wave equation together with continuity and dynamic equations. In 1878 Korteweg firstly determined wave velocity considering both elasticity of pipe wall and fluid. Michaud has studied water hammer problem. In 1883 Gromeka included friction losses in the analysis of water hammer for first stage. He has assumed that liquid is incompressible and friction losses are directly proportional to flow velocity (Koç, 2001). Weston and Carpenter have realised certain experiments to develop empirical relation between pressure rise against to reduced flow velocity inside the pipes. These experiments were unsuccessful because of the inadequate pipe length. Frizell has developed expressions for rising pressure value due to sudden pause of flow and velocity of water hammer pressures. He also studied branching pipes (Chaudhry, 1987).

In 1897, Joukowski has conducted extensive experiments using the pipes in Moscow, Russia. He published his classical report upon water hammer basic theory. He developed a formula for wave velocity considering elasticity for both water and pipe wall, and relation between pressure rising resulting from reduction of flow velocity. Developed methods were based on conservation of energy and continuity equation. He discussed propagation and reflection of pressure wave throughout the pipe. He also studied the effects of air chambers, surge tanks, security valves damping water hammer pressures. He has founded that if valve is closed before $2 \mathrm{~L} / \mathrm{a}$ second, maximum pressure will be occurred where $L$ is the pipe length and $a$ is the wave celerity (Chaudhry, 1987).

Allievi published general water hammer theory in 1902. His produced dynamic equation was more accurate than that of Korteweg. He constituted and presented graphics for pressure drop as a result of systematic opening or closing valves. Allievi became creative person of basic water hammer theory (Chaudhry, 1987).

Between 1940 and 1960, Gray realised water hammer analysis using computer application and method of characteristics. Lai used Gray's studies in his doctorate study and together with Streeter, they presented their publications using computers and method of characteristics.

Since 1960 to 1970, method of characteristics was developed by Streeter and he studied column separation and boundary conditions for pumps and air chambers. Graphical methods were presented by John Parmakian (1963). Velocity of water hammer wave in an elastic pipe was studied by Halliwell in 1963. In 1965, M. Marchal, G. Flesch, P. Suter examined the calculation of water hammer problems by means of a digital computers. In 1968, the paper about water hammer control into the pipeline systems has been presented by Kinno (Koç, 2001). H. Kinno in 1968 realised his study upon water hammer control in centrifugal pump systems. Wood realised the research on calculation of water hammer pressure due to valve closure in 1968.

A lot of studies about water hammer and transient flows in pressurised pipelines, have been realised since 1970. Some of them investigated the effects of valve closing upon transient flows, the high and low pressures as a result of sudden pause or sudden starting of operation of pump, power failures of turbines, the water hammer using different methods, transient flows in branching pipes, the prevention of water hammer pressures. After 1970, water hammer calculations are performed by both computers and graphical methods. In 1973, Wood and Jhones clarified water hammer pressures for various types of valves. In July 1973, computer analysis of water hammer in pipeline systems was performed by Sheer, Baasch and Gibbs. In 1975, Streeter and Wylie realised their study as 'Transient analysis of offshore loading systems'. In 1977, Benjamin Donsky presented upsurge and speed rise charts due to pump shut down (Donsky, 1961). Vardy studied the method of characteristics for the solution of unsteady flow networks in 1977. In 1980, Provoost realised the study upon the dynamic behaviour of non return valves. One dimensional model for transient gas-liquid flows in ducts was examined in 1980 by Hancox, Ferch, Liu and Nieman. Wave propagation in plastically deforming ducts was studied in 1980 by

Twyman, Thorley and Hewavitarne. In 1987, Wiggert, Hatfield and Stuckenbruch realised water hammer analysis in pipes using characteristic methods (Koç, 2001). In 1989, Thorley presented his study as 'Check valve behaviour under transient flow conditions'. Stittgen and Zielke performed a study in 1990 as fluid structure interaction in the flexible curved pipes. Besides these studies, various scientific books and reports upon water hammer and unsteady flows are prepared and published by Watters, 1979, Chaudhry, 1987, Tullis, 1989, Thorley, 1991, International Association of Hydraulic Research, IAHR, 1994, Larock and Jeppson, 2000, Streeter and Wylie, 1993.

Zaruba also published his studies upon water hammer in 1993. Popescu, Arsenie and Vlase presented their book in 2003 which contains water hammer, unsteady flow computations and experimental results conducted in Romania and they explained the applications and practices upon the water structures placed on Danube River and some pipelines. Some computer programs are developed by hydraulic departments of several universities and engineering firms related to water hammer, transient flows and unsteady flow calculations in pipelines such as, Washington State University prepared by Chaudhry, some other universities in USA and Europe, for instance, engineering company, Haestad Inc in USA, Denmark, Delft Hydraulics etc.

Werner Burmann (1975) investigated the behaviour of water hammer phenomena in pipe systems of several different flow sections. In this report, the solution of the differential equations covering non-stationary fluid flow in pipe systems of several different flow sections by means of the method of characteristics is sketched.

Karney Bryan W. and Duncan McInnis (1990) studied transient flow in water distribution systems. They emphasize that the details of how a hydraulic system is modelled or represented can have a critical impact on the predicted transient conditions.

William Rahmeyer (1996) published a paper titled as 'Dynamic Flow Testing of Check Valves'. The two objectives of this paper are to present a test method by
which check valves can be dynamically tested for sudden closure due to reverse flow, and to discuss the valve and pipe characteristics which affect the reverse velocities and pressure surges at the check valves.

Ezzeddine Hadj-Taieb and Taieb Lili (1998) presented their study about the transient flow in homogeneous gas-liquid mixtures in rigid and quasi-rigid pipes. Two mathematical models based on the gas-fluid mass ratio are presented. The fluid pressure and velocity are considered as two principal dependent variables and the gas-fluid mass ratio is assumed to be constant. By application of the conservation of mass and momentum laws, non-linear hyperbolic systems of two differential equations are obtained and integrated numerically by a finite difference conservative scheme. Numerical solutions are compared with numerical results available in literature and experiment developed in the laboratory. The results show that the pressure wave propagation is significantly influenced by the gas-fluid mass ratio and the elasticity of pipe wall. They indicate that the pipe elasticity and liquid compressibility may be neglected for great values of gas-liquid mass ratio but not for the smaller ones.

Colin Kirkland (1998) presented a paper related to the controlled release and intake of air into the pipelines to maximize their performance. This paper describe some of the roles that the air release valve plays in various pipelines, and how it can be safely and effectively used in order to achieve greater results.

Yukio Kono, Masaji Watanabe and Tomonoki Ito (1998) introduced a method in which the upstream finite different approximation is applied to the systems governing a liquid and a two phase flow of the mixture of the liquid and the vapour. They also introduced some results of analysis using this method and compare those with experimental results and concluded that this method is capable for solving the non-linear hyperbolic type partial differential equation and parabolic equation simultaneously.

Kameswara Rao C.V. and Eswaran K. (1999) performed their study titled as 'Pressure Transients in Incompressible Fluid Pipeline Networks'. They explained that pressure surges in pipeline are caused due to different events either planned or accidental. It is essential to determine the magnitude and frequency of pressures and forces triggered due to these transients to estimate the stresses and vibration levels in the pipeline networks. In their paper, an effort is made to study these transients in incompressible fluid flow systems and the development of a computer program HYTRAN is described. This computer program comprehensively incorporates the method of characteristics for the calculation of the time dependent head and velocity of the fluid at any point in a complex fluid/water pipeline network upon the beginning of any event such as pump failure, load reduction on a turbine, etc. The time history of the machine parameters in the case of pumps and turbines during such events can also be obtained as output. Two case studies have been taken up and the results are discussed.

Computer program named 'Pipenet-Transient Module' was produced in 1999 by Sunrise Systems Ltd. The PIPENET Transient Module provides a speedy and cost effective means of in-house rigorous transient analysis. The transient module can be used for predicting pressure surges, calculating hydraulic transient forces or even modelling control systems in flow networks.

A German technical/scientific institute has developed a new passive security system in 2003 for pipelines to avoid pressure surges in both down- and upstream sections of the pipe system that contains fast closing valves. Innovative aspects are system operates without additional energy support, water hammer is strongly reduced, cavitation hammer due to vapour bubble collapse is totally avoided, valve closing process performed is always the fastest without risk of pipe damage, optimal adaptation of closing process to operation parameter and switching state of the respective plant.

George E. Alves (2004) presented his study titled 'Hydraulic Analysis of Sudden Flow Changes in a Complex Piping Circuit'. He explained that a problem arose if
there were sudden failures of power to one or more pumps of a larger scale complex piping circuit composed of several individual pumping systems. His study describes the application of several published methods of hydraulic transient analysis to the problem. The performance of the system computed under several assumptions is discussed, and a comparison is made with experimentally determined values.

Daniel Ward (2004) realised the study titled 'Automatic and Remotely Controlled Shutoff for Direct Flow Liquid Manure Application Systems'.
G.A. Clark, A.G. Smajstrla and D.Z. Haman (2004) studied water hammer in Irrigation Systems' in University of Florida, Gainesville. This publication discusses the causes of water hammer and the importance of proper system design and management to ensure a cost effective, long-lasting irrigation system.
A. Bergant, A.R. Simpson and A.S. Tijsseling (2005) realised a study titled 'Water Hammer with Column Separation: A Historical Review'. This study reviews water hammer with column separation from the discovery of the phenomenon in the late $19^{\text {th }}$ century, the recognition of its danger in the 1930s, the development of numerical methods in the 1960s and 1970s, to the standard models used in commercial software packages in the late $20^{\text {th }}$ century.

Tilman Diesselhorst and Ulrich Neumann (2005) performed the study of 'Optimization of Loads in Piping Systems by the Realistic Calculation Method: Applying Fluid-Structure Interaction (FSI) and Dynamic Friction'. They demonstrated that to reduce costs and to extend the life time of piping systems their design loads due to valve action have to be optimized. To get the best effect, the results of the fluid dynamic and structural calculations should be realistic as far as possible. Therefore the calculation programs were coupled to consider the fluid structure interaction and the effect of dynamic fluid friction was introduced to get realistic results of oscillations due to pressure surges. Detailed modelling of check valve behaviour allows minimizing the pressure surge loading by improving the valve function and adapting it to the system behaviour. The method was validated at
measurements of load cases in power plant piping systems. Results with different load cases show the effectiveness of reducing the fluid forces on piping. Examples are presented to prove the reduction of supports.

Algirdas Kaliatka, Eugenijus Uspuras and Mindaugas Vaisnoras (2005) presented water hammer phenomenon simulations employing the RELAP5 code, a comparison of RELAP5 calculated and measured at CWHTF and AEKI test facilities pressure transient values after a fast opening of the valve and at the appearance of condensation - induced water hammer. An analysis of rarefaction wave travels inside the pipe and the condensation of vapour bubbles in the liquid column for CWHTF experiment is presented. The dependence of the pressure peaks on the evacuation height and the length of the pipeline were investigated. A comparison of RELAP5 code CWHTF experiment simulation by using similar equilibrium options (HEM) and without these options is also presented. The capability of RELAP5 computer code to simulate condensation induced water hammer was investigated.

Kala K. Fleming, Joseph P. Dugandzic, Mark W. LeChevallier and Rich W. Gullick (2006) prepared the research project titled as 'Susceptibility of Potable Water Distribution Systems to Negative Pressure Transients'. They stated that investigating pressure transients improves understanding of how a system will behave in response to a variety of events such as power cut, routine pump shut downs, valve operations, flushing, fire fighting, main breaks and other events that can create significant, rapid, temporary drops in system pressure. This project built upon the work done in previous projects. The purpose of investigating pressure transients is to improve the operator's understanding of how the system will behave in response to a variety of events such as power cut, pump shut downs, valve operations, flushing, fire fighting, main breaks and other events that can create significant rapid drops in system pressure and/or low pressure waves. A review of system conditions and utility procedures are recommended to effectively minimize a system's effects to pressure transients.

Robert A. Leishear (2007) performed his study named 'Dynamic Pipe Stresses During Water Hammer: A Finite Element Approach'. According to his study, in the wake of the pressure wave, dynamic stresses are created in the pipe wall, which contribute to pipe failures. A finite element analysis computer program was used to determine the three dimensional dynamic stresses that result from pipe wall vibration at a distance from the end of a pipe, during a water hammer event. The analysis was used to model a moving shock wave in a pipe, using a step pressure wave. Both aluminum and steel were modelled for an 8 NPS pipe, using ABAQUS. For either material, the maximum stress was seen to be equal when damping was neglected. At the time the maximum stress occured, the hoop stress was equivalent to twice the stress that would be expected if an equivalent static stress was applied to the inner wall of the pipe. Also the radial stress doubled the magnitude of the applied pressure.

### 1.3 The scope of the study

The objective of this study is to investigate transient flow due to pump failure. It is aimed to investigate the agreement between results obtained from measurements and those obtained from numerical calculations. The appeared upper and lower pressures in the system as a result of sudden pump failure or shutdown at steady state in the reservoir-pump-pressurised steel pipeline systems are examined.

This study is performed in Şahabettin Demirağ Hydraulic Research Laboratory in Dokuz Eylül University of Civil Engineering Department in İzmir, Turkey. Two different experimental set up are constructed. The first system includes a pipe of 28 meters without a check valve. In the other system, the pipe has a length of 108 meters and a check valve is placed after the pump. In the former system, unsteady flow pressures are measured at downstream of the pump. The unsteady flow pressures are recorded by means of measuring data logger in the latter system at downstream end of the pump and nearly at mid-point of the pipe. Pressures and discharges are calculated at nodal points by using Fortran computer programs.

## CHAPTER TWO

## BASIC EQUATIONS OF UNSTEADY FLOW

One dimensional differential equations of transient flow may be obtained by Newton's second law of motion and conservation of mass. In these expressions, the dependent variables are H , hydraulic grade line (HGL, piezometric head) and V, average velocity (or Q , discharge). Independent variables, the distance along measured pipe length from upstream end and the time are denoted by x and t .

### 2.1 Equation of motion

By referring to Figure 2.1 , the Newton's second law of motion is applied ( $\Sigma \mathrm{F}=\mathrm{m} . \mathrm{a}$ ). The subscript x denotes the space derivative.

HGL


Figure 2.1: Free body diagram of cylindrical tube for application of equation of motion.

The forces on the free body in the x direction are the pressure forces, shear force and gravity force.

$$
\begin{equation*}
p A-\left[p A+\frac{\partial(p)}{\partial x} A \delta x\right]-\tau_{0} \pi D \delta x-\gamma A \delta x \sin \alpha=\rho A \delta x \frac{d V}{d t} \tag{2-1}
\end{equation*}
$$

where p is the centreline pressure, A is the cross sectional area of the pipe, $\delta \mathrm{x}$ is the length of cylinder tube, $\alpha$ is the pipe inclination angle with the horizontal, $\tau_{0}$ is the wall shear stress, $D$ is the pipe diameter, $\gamma$ is the specific weight of fluid, $\rho$ is the mass density of fluid, V is the average velocity and g is the acceleration due to gravity.

After some algebra, we obtain the Equation (2-2),

$$
\begin{equation*}
\frac{\partial p}{\partial x} A+\tau_{0} \pi D+\rho g A \sin \alpha+\rho A \frac{d V}{d t}=0 \tag{2-2}
\end{equation*}
$$

Wall shear stress $\tau_{0}$ is assumed to be the same as steady turbulent flow.

$$
\begin{equation*}
\tau_{0}=\frac{\rho f V|V|}{8} \tag{2-3}
\end{equation*}
$$

where f is Darcy Weisbach friction coefficient. If $\mathrm{J}_{\mathrm{e}}$ denotes energy grade line slope,

$$
\begin{equation*}
J_{e}=\frac{f \cdot V \cdot|V|}{D .2 g} \tag{2-4}
\end{equation*}
$$

In order to take into consideration the negative flow, absolute value of velocity term in Equation (2-3) is taken. Because shear stress must be surely reverse to the velocity direction. $\mathrm{dV} / \mathrm{dt}$ is total derivative and from the chain rule,

$$
\begin{equation*}
\frac{d V}{d t}=V \frac{\partial V}{\partial x}+\frac{\partial V}{\partial t} \tag{2-5}
\end{equation*}
$$

Velocity V of fluid slice changes both with distance and time. Substituting Equations (2-3) and (2-5) into Equation (2-2) and dividing all terms by $\rho \mathrm{A}$, one obtains
$\frac{1}{\rho} \frac{\partial p}{\partial x}+\frac{f V|V|}{2 D}+g \sin \alpha+V \frac{\partial V}{\partial x}+\frac{\partial V}{\partial t}=0$

Piezometric head H (or HGL elevation above reference frame) may be introduced by using the common relationship,

$$
\begin{equation*}
p=\rho g(H-z) \tag{2-7}
\end{equation*}
$$

For fluid slice,

$$
\begin{equation*}
\sin \alpha=\frac{\partial z}{\partial x} \tag{2-8}
\end{equation*}
$$

The following Equation is obtained,

$$
\begin{equation*}
\frac{\partial p}{\partial x}=\rho g\left(\frac{\partial H}{\partial x}-\sin \alpha\right) \tag{2-9}
\end{equation*}
$$

In this expression, the density, $\rho$ is assumed to be constant. Equation (2-6) is valid for both liquids and gases, but Equation (2-9) is restricted with liquids. Substituting Equation (2-9) into Equation (2-6),
$g \frac{\partial H}{\partial x}+V \frac{\partial V}{\partial x}+\frac{\partial V}{\partial t}+\frac{f V|V|}{2 D}=0$

Equation (2-10) is for unsteady flows and called equation of motion. For the special case of steady flows in pipes with constant diameter, $\partial \mathrm{V} / \partial \mathrm{x}$ and $\partial \mathrm{V} / \partial \mathrm{t}$ are zero. In this case,

$$
\begin{equation*}
g \frac{\partial H}{\partial x}=\frac{-f V|V|}{2 D} \tag{2-11}
\end{equation*}
$$

$$
\begin{equation*}
\Delta H=\frac{-f \Delta x V|V|}{2 g D} \tag{2-12}
\end{equation*}
$$

The common Darcy-Weisbach equation is obtained.

### 2.2 Continuity Equation

Elastic deformations are allowed. The conservation of mass law expresses that rate of mass inflow into the control volume equals time rate of mass increase within the control volume shown in Figure 2.2.
$\frac{-\partial(\rho A(V))}{\partial x} \delta x=\frac{\partial}{\partial t}(\rho A \delta x)$

HGL


Figure 2.2: Control volume for continuity equation.

$$
\begin{equation*}
\frac{\partial(\rho A V)}{\partial x}+\frac{\partial(\rho A)}{\partial t}=0 \tag{2-14}
\end{equation*}
$$

By expanding the first term in the left side of Equation (2-14),
$\rho A \frac{\partial V}{\partial x}+V \frac{\partial(\rho A)}{\partial x}+\frac{\partial(\rho A)}{\partial t}=0$

In Equation (2-15), last two terms indicate the total derivative of $\rho \mathrm{A}$. From Chain rule,
$\frac{d(\rho A)}{d t}=V \frac{\partial(\rho A)}{\partial x}+\frac{\partial(\rho A)}{\partial t}$

If we divide the all terms by $(\rho \mathrm{A})$ in Equation (2-15) and place the Equation (2-16) into Equation (2-15),

$$
\begin{equation*}
\frac{1}{\rho A}\left[\frac{d(\rho A)}{d t}\right]+\frac{\partial V}{\partial x}=0 \tag{2-17}
\end{equation*}
$$

We can write the Equation (2-17) such that,
$\frac{1}{\rho A}\left(\rho \frac{d A}{d t}+\frac{d \rho}{d t} A\right)+\frac{\partial V}{\partial x}=0$

Or ,
$\frac{1}{A} \frac{d A}{d t}+\frac{1}{\rho} \frac{d \rho}{d t}+\frac{\partial V}{\partial x}=0$

Equation (2-19) may be used in both cylindrical pipes and tubes having variable diameter. This is also valid for very flexible tubes and gaseous flows (Wylie \& Streeter, 1993).

The first term of Equation (2-19) relates to elasticity of pipe wall and deformation rate (strain) due to pressure change. Second term takes into account the compressibility of the liquid. From the expression of the bulk modulus,

$$
\begin{equation*}
\frac{d \rho}{d t} \frac{1}{\rho}=\frac{1}{K} \frac{d p}{d t} \tag{2-20}
\end{equation*}
$$

Where, K is the bulk modulus of elasticity of fluid, e is the thickness of the pipe wall, E is the Young modulus of elasticity of pipe material. The variation of axial tensile stress $\sigma_{1}$ with time may be designated for three conditions as follow:
a) pipeline is anchored only at upstream end,

$$
\begin{equation*}
\frac{d \sigma_{1}}{d t}=\frac{A}{\pi D e} \frac{d p}{d t}=\frac{D}{4 e} \frac{d p}{d t} \tag{2-21}
\end{equation*}
$$

b) Pipeline is anchored against longitudinal motions.

$$
\begin{equation*}
\frac{d \sigma_{1}}{d t}=\mu \frac{d \sigma_{2}}{d t} \tag{2-22}
\end{equation*}
$$

c) Pipeline has expansion joints along its length.
$\frac{d \sigma_{1}}{d t}=0$

After some algebra Equation (2-19) becomes

$$
\begin{equation*}
\frac{1}{\rho} \frac{d p}{d t}+a^{2} \frac{\partial V}{\partial x}=0 \tag{2-24}
\end{equation*}
$$

Where a is the wave velocity (celerity).

$$
\begin{equation*}
a^{2}=\frac{\frac{K}{\rho}}{1+\left[\left(\frac{K}{E}\right)\left(\frac{D}{e}\right) C_{1}\right]} \tag{2-25}
\end{equation*}
$$

The coefficient, $\mathrm{C}_{1}$ is described for each condition separately,
a) pipeline is anchored only at upstream end,
$c_{1}=1-\frac{\mu}{2}$
b) Pipeline is anchored against longitudinal motions.
$c_{1}=1-\mu^{2}$
c) Pipeline has expansion joints along its length.
$c_{1}=1$

Where, $\mu$ is the Poisson's ratio. From the definition of piezometric head,
$\frac{d p}{d t}=\rho g\left(\frac{d H}{d t}-\frac{d z}{d t}\right)$
$\frac{d p}{d t}=\rho g\left(V \frac{\partial H}{\partial x}+\frac{\partial H}{\partial t}-V \frac{\partial z}{\partial x}-\frac{\partial z}{\partial t}\right)$

If the pipeline has not transverse motion, $\partial \mathrm{z} / \partial \mathrm{t}=0$ and $\partial \mathrm{z} / \partial \mathrm{x}$-Sin $\alpha=0$. Equation (2-24) becomes :

$$
\begin{equation*}
V \frac{\partial H}{\partial x}+\frac{\partial H}{\partial t}-V \operatorname{Sin} \alpha+\frac{a^{2}}{g} \frac{\partial V}{\partial x}=0 \tag{2-31}
\end{equation*}
$$

This equation is called one dimensional continuity equation (Wylie \& Streeter, 1993).

### 2.3 Celerity in pipelines

In the wave propagation speed formula, Equation (2-25), K is the bulk modulus of elasticity of water and E is the Young modulus of elasticity of steel pipe in Pa. The value of K for water is $2 \cdot 24.10^{9} \mathrm{~Pa}$ and that of E for steel pipes is $2.07 .10^{11} \mathrm{~Pa}$.

If the D/e ratio is less than 40 , thick walled pipe formulas are used. If this ratio is greater than 40, one uses thin walled pipe formulas (Watters, 1979). In pipeline system used in this research, this ratio $\mathrm{D} / \mathrm{e}$ is equal to $0.125 / 0.005=25$. This value is less than 40. Consequently, we will use proposed formulas for thick walled pipes in the analysis. Conditions are such that
a) pipeline is anchored only at upstream end,

$$
\begin{equation*}
C_{1}=\frac{2 e}{D}(1+\mu)+\frac{D}{D+e}\left(1-\frac{\mu}{2}\right) \tag{2-32}
\end{equation*}
$$

b) Pipeline is anchored against longitudinal motions.
$C_{1}=\frac{2 e}{D}(1+\mu)+\frac{D\left(1-\mu^{2}\right)}{D+e}$
c) Pipeline has expansion joints along its length.

$$
\begin{equation*}
C_{1}=\frac{2 e}{D}(1+\mu)+\frac{D}{D+e} \tag{2-34}
\end{equation*}
$$

In our experimental set up, pipeline is anchored against longitudinal motions. Therefore, we will use condition (b), Equation (2-33) and Equation (2-25) in the calculation of celerity. In these formulas, $\mu$ is the Poisson's ratio and taken as 0.3 for
steel pipes. Further $\rho$, mass density is taken as $1000 \mathrm{~kg} / \mathrm{m}^{3}$ for water. C 1 is calculated as 0.979 . However according to realised computation using Equation (2-25), we will take the celerity value in our transient flow calculations as $1331 \mathrm{~m} / \mathrm{s}$.

For the case a, C 1 is 0.921 and the celerity is $1339 \mathrm{~m} / \mathrm{s}$. For the case $\mathrm{c}, \mathrm{C} 1$ is 1.065 and celerity is $1318 \mathrm{~m} / \mathrm{s}$. C 1 and celerity value for our system according to case b , were realised between the case a and case c .

## CHAPTER THREE

## NUMERICAL SOLUTION WITH THE USE OF METHOD OF CHARACTERISTICS

Equation of motion (dynamic or momentum equation) and continuity equation were developed in Chapter Two, are first order hyperbolic partial differential equations. Analytical solution of these equations is not possible. As a first step, partial differential equations are transformed to ordinary differential equations by using the characteristic method, and then, they are solved numerically by explicit finite difference scheme (Wylie \& Streeter, 1978, 1993).

### 3.1 Characteristic equations

Equation of motion and Equation of continuity are transformed to two ordinary differential equations by the method of characteristics. Less important terms, V. $\partial \mathrm{V} / \partial \mathrm{x}$ in Equation (2-10), V. $\partial \mathrm{H} / \partial \mathrm{x}$ and V.sin $\alpha$ in Equation (2-31) are neglected for the sake of simplicity (Wylie \& Streeter, 1978, 1993).

By neglecting the above mentioned terms, Equation (2-10) and (2-31) become

$$
\begin{align*}
& L_{1}=g \frac{\partial H}{\partial x}+\frac{\partial V}{\partial t}+\frac{f V|V|}{2 D}=0  \tag{3-1}\\
& L_{2}=\frac{\partial H}{\partial t}+\frac{a^{2}}{g} \frac{\partial V}{\partial x}=0 \tag{3-2}
\end{align*}
$$

These equations are combined linearly by using unknown multiplier $\lambda$.
$L=L_{1}+\lambda L_{2}=\lambda\left[\frac{\partial H}{\partial x} \frac{g}{\lambda}+\frac{\partial H}{\partial t}\right]+\left[\frac{\partial V}{\partial x} \lambda \frac{a^{2}}{g}+\frac{\partial V}{\partial t}\right]+\frac{f V|V|}{2 D}=0$

Considering the chain rule,
$\frac{d H}{d t}=\frac{\partial H}{\partial x} \frac{d x}{d t}+\frac{\partial H}{\partial t} \quad ; \quad \frac{d V}{d t}=V \frac{\partial V}{\partial x}+\frac{\partial V}{\partial t}$
$\frac{d V}{d t}=\frac{\partial V}{\partial x} \frac{d x}{d t}+\frac{\partial V}{\partial t} \quad ; \quad \frac{d H}{d t}=V \frac{\partial H}{\partial x}+\frac{\partial H}{\partial t}$

From examination of Equation (3-3), it is seen that, if
$\frac{d x}{d t}=\frac{g}{\lambda}=\frac{\lambda a^{2}}{g}$

The Equation (3-3) becomes ordinary differential equation. If we substitute Equations (3-4) and (3-5) into Equation (3-3),
$\lambda \frac{d H}{d t}+\frac{d V}{d t}+\frac{f V|V|}{2 D}=0$

If we solve Equation (3-5) for $\lambda$, we obtain particular values of $\lambda$,
$\lambda=\mp \frac{g}{a}$

If we substitute these $\lambda$ values into Equation (3-5),
$\frac{d x}{d t}=\mp a$

The substitution of positive and negative values of $\lambda$ in Equation (3-7) into Equation (3-6) yields two pairs of equations. These equations are named as $\mathrm{C}^{+}$and $\mathrm{C}^{-}$ characteristics equations. If the slope is positive then $\mathrm{C}^{+}$, if the slope is negative then $\mathrm{C}^{-}$equations are formed.

Equation (3-9) is valid along the $\mathrm{C}^{+}$characteristics defined by Equation (3-10).

$$
\begin{align*}
& \frac{g}{a} \frac{d H}{d t}+\frac{d V}{d t}+\frac{f V|V|}{2 D}=0  \tag{3-9}\\
& \frac{d x}{d t}=a \tag{3-10}
\end{align*}
$$

Equation (3-11) is valid along the $\mathrm{C}^{-}$characteristics defined by Equation (3-12).
$\frac{-g}{a} \frac{d H}{d t}+\frac{d V}{d t}+\frac{f V|V|}{2 D}=0$
$\frac{d x}{d t}=-a$


Figure 3.1 : Characteristic lines in the x-t plane.

The wave celerity, a is considered as a constant in the pipeline. Hence, Equations (3-10) and (3-12) correspond to straight lines on the x-t plane. These lines are called
characteristics. Equations (3-9) and (3-11) are called compatibility equations. Each of these equations is valid only along suitable characteristic line. (Wylie \& Streeter, 1978, 1993).

### 3.2 Equations in finite difference form

Length of the pipeline, L is divided into N equal reaches. Each of these reaches is $\Delta x=L / N$ long. Time increment is $\Delta t=\Delta x / a$. If dependent variables $V$ and $H$ in Equation (3-10) are known at point A , this equation can be integrated between points $A$ and $P$, and the obtained equation contains the unknowns at point $P$, namely $V_{P}$ and $\mathrm{H}_{\mathrm{p}}$.

Equation (3-12) is valid along the negative sloped characteristic as shown by BP in Figure 3.2. Integration of the $\mathrm{C}^{-}$compatibility equation along the line BP , with conditions known at B and unknown at P , leads to a second equation in terms of the same two unknown variables at $P$, namely $V_{P}$ and $H_{P}$. The simultaneous solution yields conditions at the particular time and position in the xt plane designated by point $P$.

By multiplying Equation (3-10) with $(\mathrm{a} / \mathrm{g}) \mathrm{dt}=\mathrm{dx} / \mathrm{g}$, writing discharge term in place of velocity and introducing the pipeline area, Equation becomes,

$$
\begin{align*}
& \int_{H_{A}}^{H_{p}} d H+\frac{a}{g A} \int_{Q_{A}}^{Q_{p}} d Q+\frac{f}{2 g D A^{2}} \int_{x_{A}}^{x_{p}} Q|Q| d x=0  \tag{3-13.a}\\
& \int_{H_{B}}^{H_{P}} d H+\frac{a}{g A} \int_{Q_{B}}^{Q_{p}} d Q+\frac{f}{2 g D A^{2}} \int_{x_{B}}^{x_{p}} Q|Q| d x=0 \tag{3-13.b}
\end{align*}
$$

Expressions below are obtained after the integration of Equations (3-13) along the two characteristics.


Figure 3.2: x-t grid for solving single pipe problems.
$H_{P}-H_{A}+\frac{a}{g A}\left(Q_{P}-Q_{A}\right)+\frac{f \Delta x}{2 g D A^{2}} Q_{A}\left|Q_{A}\right|=0$
$H_{P}-H_{B}-\frac{a}{g A}\left(Q_{P}-Q_{B}\right)-\frac{f \Delta x}{2 g D A^{2}} Q_{B}\left|Q_{B}\right|=0$

These two compatibility equations introduce the piezometric head along pipeline and discharge during transient flow. Solving these equations for $\mathrm{H}_{\mathrm{P}}$,
$C^{+}: H_{P}=H_{A}-B\left(Q_{P}-Q_{A}\right)-R Q_{A}\left|Q_{A}\right|$
$C^{-}: H_{P}=H_{B}+B\left(Q_{P}-Q_{B}\right)+R Q_{B}\left|Q_{B}\right|$

Where $\mathrm{B}=\mathrm{a} /(\mathrm{gA})$ which is pipe impedance and $\mathrm{R}=(\mathrm{f} \Delta \mathrm{x}) /\left(2 \mathrm{gDA}^{2}\right)$ which is pipeline resistance coefficient.

In steady state conditions, $\mathrm{Q}_{\mathrm{A}}=\mathrm{Q}_{\mathrm{B}}=\mathrm{Q}_{\mathrm{P}}$ and $\left(\mathrm{RQ}_{\mathrm{A}} \mathrm{Q}_{\mathrm{A}}\right)$ is the steady state friction loss along the reach $\Delta \mathrm{x}$.

The solution of liquid transient problem is generally started with steady state conditions at time zero. For $\mathrm{t}=0, \mathrm{H}$ and Q are known as initial values at each calculating section. $H$ and $Q$ are calculated for each grid point over $t=\Delta t$. Then procedure is repeated for $t=2 \Delta t$. Procedure is continued until the total number of time step is covered.

Equations (3-16) and (3-17) can be written in simplified form and compatible to computer applications.
$C^{+}: H_{P i}=C_{P}-B Q_{P i}$
$C^{-}: H_{P i}=C_{M}+B Q_{P i}$

Where $C_{P}$ and $C_{M}$ are known constants.
$C_{P}=H_{i-1}+B Q_{i-1}-R Q_{i-1}\left|Q_{i-1}\right|$
$C_{M}=H_{i+1}-B Q_{i+1}+R Q_{i+1}\left|Q_{i+1}\right|$

Eliminating of Qpi in Equations (3-18) and (3-19),

$$
\begin{equation*}
H_{P i}=\frac{C_{P}+C_{M}}{2} \tag{3-22}
\end{equation*}
$$

After the calculation of $\mathrm{H}_{\mathrm{Pi}}$, $\mathrm{Q}_{\mathrm{Pi}}$ can be calculated either from Equation (3-18) or Equation (3-19).

H and Q values having subscripts such as (i-1) and (i+1) at each section are always available from previous time step. They are given either initial conditions or appear as a consequence of previous stage calculations.

### 3.3 Boundary conditions

At end points, there exists only one compatibility equation but two unknowns. Because of that a second equation corresponding to boundary condition is required. The equation along negative characteristic for upstream end and the equation along positive characteristic for downstream end are combined with the relevant boundary condition.


b: Downstream end

Figure 3.3: Characteristic at boundaries

If we consider a reservoir at upstream end, water surface elevation of upstream reservoir can be assumed as a constant. Water surface elevation defines upstream boundary condition. $\mathrm{H}_{\mathrm{P}, 1}=\mathrm{H}_{\mathrm{R}} . \mathrm{H}_{\mathrm{R}}$ is the water surface elevation of upstream reservoir upon reference level. If the reservoir level is changed as a known pattern (for example, as a sinus wave), boundary condition will be such a manner that,

$$
\begin{equation*}
H_{P 1}=H_{R}+\Delta H \operatorname{Sin} \omega t \tag{3-23}
\end{equation*}
$$

Where $\omega$ is the angular velocity in rad/s and $\Delta \mathrm{H}$ is the wave amplitude in $\mathrm{m} . \mathrm{H}_{\mathrm{P} 1}$ is known at each time step and $\mathrm{Q}_{\mathrm{p} 1}$ is determined from Equation (3-24) directly.

$$
\begin{equation*}
Q_{P 1}=\frac{H_{P 1}-C_{M}}{B} \tag{3-24}
\end{equation*}
$$

If the centrifugal pump exists at upstream end, the head discharge characteristic curve of centrifugal pump operating at constant speed may be included in the analysis. If the pump provides the flow from suction reservoir, equation below may be used.

$$
\begin{equation*}
H_{P 1}=H_{S}+Q_{P 1}\left(a_{1}+a_{2} Q_{P 1}\right) \tag{3-25}
\end{equation*}
$$

Where $H_{S}$ is the shutoff head, $a_{1}$ and $a_{2}$ are the constants describing characteristic curve. If we solve Equation (3-19) and Equation (3-25) simultaneously,

$$
\begin{equation*}
Q_{P 1}=\frac{1}{2 a_{2}}\left[B-a_{1}-\sqrt{\left(B-a_{1}^{2}\right)+4 a_{2}\left(C_{M}-H_{S}\right)}\right] \tag{3-26}
\end{equation*}
$$

is obtained. With known $\mathrm{Q}_{1}, \mathrm{H}_{1}$ from either Equation (3-19) or Equation (3-25) is determined.

If there is a valve at downstream end of the pipe whose index is $\mathrm{NS}=\mathrm{N}+1$, the valve closure law may be written as,

$$
\begin{equation*}
Q_{P N S}=\frac{Q_{0}}{\sqrt{H_{0}}} \tau \sqrt{H_{P N S}} \tag{3-27}
\end{equation*}
$$

Where $\mathrm{Q}_{0}$ is the steady state discharge, $\Delta \mathrm{H}=\mathrm{H}_{\text {PNS }}$ is the sudden drop of HGL (of piezometric head) at valve, $\mathrm{H}_{0}$ is the steady state head loss throughout the valve. For steady flow, $\tau=1 . \tau$ is called the dimensionless valve opening. When there is no flow, $\tau=0$. Solving Equations (3-24) and (3-36) simultaneously,

$$
\begin{equation*}
Q_{P N S}=-B . C_{V}+\sqrt{\left(B . C_{V}\right)^{2}+2 C_{V} C_{P}} \tag{3-28}
\end{equation*}
$$

Where $\mathrm{C}_{\mathrm{V}}=\left(\mathrm{Q}_{0} \tau\right)^{2} /\left(2 \mathrm{H}_{0}\right)$. Values corresponding to $\mathrm{H}_{\mathrm{PNS}}$ can be determined from Equation (3-18) or Equation (3-27).

In our first system, there is a pump at the beginning of the pipeline and a triangular weir at the downstream end. Discharge adjustment valve is placed at an interior point of the pipe. During the measurements, valves openings are kept constant and therefore in the calculations, $\tau=1$.

In the second system, there is a pump with valve and check valve at the upstream end, a reservoir at the downstream end.

### 3.4 Courant Criterion

According to Courant criterion,

$$
\begin{equation*}
\Delta t \leq \frac{\Delta x}{V \pm a} \tag{3-29}
\end{equation*}
$$

In which $\Delta \mathrm{t}$ is time step or time increment in seconds, $\Delta \mathrm{x}$ is the length of one reach in meters, V is the flow velocity in $\mathrm{m} / \mathrm{s}$ and a is the pressure wave propagation celerity in $\mathrm{m} / \mathrm{s}$. If the flow velocity V is neglected before the celerity, then the common form $\Delta \mathrm{t}=\Delta \mathrm{x} / \mathrm{a}$ is obtained.

Wylie \& Streeter, 1993 provided the following criterion to obtain stable results and to prevent discrepancy in the results of numerical solution.

$$
\begin{equation*}
\Delta t \leq \frac{\left(I \cdot \pi \cdot N_{R}\right)}{100 \cdot T_{R}} \tag{3-30}
\end{equation*}
$$

In which I is the moment of inertia of the rotating parts which is equal to $\mathrm{WR}_{\mathrm{g}}{ }^{2} / \mathrm{g}$.
$\Delta t$ value must be lower or equal to the right side of the Equations (3-29) and (330). This is the provision to obtain stable results in the Fortran computer program calculations. If we select the $\Delta t$ value greater than the value calculated from the right sides of the Equations (3-29) and (3-30), the results will be unstable.

### 3.5 Losses at regulating valve and relevant equations

### 3.5.1 Positive Flow

Let us denote by HPU the piezometric head before the valve (point $U$ ) and HPD the piezometric head after the valve (point D).

There are four unknowns which are HPU, HPD, Qpu and QpD. We need four equations. These equations are $\mathrm{C}^{+}, \mathrm{C}^{-}$, continuity and energy equations. The continuity equation is,
$Q_{P U}=Q_{P D}=Q_{P i}$

If CK denotes the valve head loss coefficient, the energy equation will be as :

$$
\begin{equation*}
H P U=H P D+\frac{C K \cdot Q_{P i}{ }^{2}}{2 g \cdot A^{2}} \tag{3-32}
\end{equation*}
$$

HGL

(i)

Figure 3.4: Head loss through the reduced bore ball valve.

The characteristic equations :
$C^{+}: H P U=C P-B \cdot Q_{P i}$
$C^{-}: H P D=C M+B \cdot Q_{P i}$

Where

$$
\begin{equation*}
C P=H_{i-1}+B Q_{i}-R Q_{i-1}\left|Q_{i-1}\right| \tag{3-35}
\end{equation*}
$$

$C M=H_{i+1}-B Q_{i+1}+R Q_{i+1}\left|Q_{i+1}\right|$

After the simultaneous solution, one obtains
$Q_{P_{i}}=0.5\left(-C 4+\sqrt{C 4^{2}-4 C 5}\right)$

Where,
$C 4=\frac{2 B}{C 3}$
$C 5=\frac{C M-C P}{C 3}$
$C 3=\frac{C K}{2 g A^{2}}$

HPU and HPD are determined by using Equations (3-33) and (3-34), after the calculation of $\mathrm{Q}_{\mathrm{Pi}}$ value from Equation (3-37).

C 4 and C 5 are the constants.

### 3.5.2 Negative (Reverse) Flow



Figure 3.5: Head loss through the reduced bore ball valve for reverse flow.

The continuity equation requires

$$
\begin{equation*}
Q_{P U}=Q_{P D}=Q_{P i} \tag{3-41}
\end{equation*}
$$

The energy equation is

$$
\begin{equation*}
H P D=H P U+\frac{C K \cdot Q_{P i}{ }^{2}}{2 g \cdot A^{2}} \tag{3-42}
\end{equation*}
$$

The characteristic equations may be written as

$$
\begin{align*}
& C^{+}: H P U=C P-B \cdot Q_{P i}  \tag{3-43}\\
& C^{-}: H P D=C M+B \cdot Q_{P i} \tag{3-44}
\end{align*}
$$

Where

$$
\begin{equation*}
C P=H_{i-1}+B Q_{i}-R Q_{i-1}\left|Q_{i-1}\right| \tag{3-45}
\end{equation*}
$$

$$
\begin{equation*}
C M=H_{i+1}-B Q_{i+1}+R Q_{i+1}\left|Q_{i+1}\right| \tag{3-46}
\end{equation*}
$$

After the simultaneous solution, one obtains
$Q_{P_{i}}=0.5\left(C 4-\sqrt{C 4^{2}+4 C 5}\right)=X_{2}$

Similarly HPU and HPD are obtained from the relevant characteristic equations.

### 3.6 Calculation of Darcy-Weisbach friction coefficient

Darcy Weisbach friction coefficient is assumed to be equal to that corresponding to the steady state given by Equation (3-48) (Yanmaz, 2001).

$$
\begin{equation*}
\frac{1}{\sqrt{f}}=1.14-2 \log \left(\frac{k_{s}}{D}+\frac{21.25}{R_{e}^{0,9}}\right) ; \quad \text { for } \mathrm{R}_{\mathrm{e}}>3000 \tag{3-48}
\end{equation*}
$$

Where f is the Darcy-Weisbach friction coefficient, $\mathrm{R}_{\mathrm{e}}=\mathrm{VD} / v$ is Reynold's number, $k_{s} / D$ is relative roughness. Roughness height, $k_{s}$ is read from the table as 0.00015 m for steel pipes.

In the Fortran computer programs, Darcy Weisbach friction coefficient is calculated by using Equation (3-48) for steady state discharges.

## CHAPTER FOUR

## TRANSIENTS CAUSED BY CENTRIFUGAL PUMPS

### 4.1 Introduction

Many transient situations caused by pumps are due to sudden starting up of pump, sudden pause or sudden power failure of pump, associated with opening or closing of valve. In this research, we examine the sudden power failure of pump excluding valve opening or closing.

Changes in operating condition of turbo machine result to develope transient flow in hydraulic system. Method of characteristics discussed in chapter three is used for analysing this transient. Special boundary conditions at the upstream and downstream ends of the pipeline are developed.

Transients caused by pump operation are generally violent and pipeline should be designed to withstand the negative and positive pressures which will be developed in the pipeline. After the electricity connection is cut off, pump speed reduces. Flow inside the discharge line reduces to zero rapidly and then returns to the pump. While the impeller of pump is rotated in respective normal direction, in the situation of deceleration to the zero velocity, to zero rotational speed, pump is said to operate in 'zone of energy dissipation'. When the pump returns reverse, pump is said to operate in 'turbine zone'. Because of the reverse flow, pump velocity decrease to zero rapidly and then pump starts to operate in reverse direction. Pump velocity increases in reverse direction until the pump velocity reaches normal run away speed (Chaudhry, 1987). In the second pipeline system, a disc type check valve is used in order to prevent reverse flow after the power failure.

If HGL (piezometric head line) drops to below of pipeline elevation at any point, pressure will be negative and if the decrease in pressure is severe, cavitation may occur and water column in the pipeline may be separated in this point. Excessive pressure will be produced when the two columns are joined again.

### 4.2 Similarity laws and pump characteristics

There are four quantities describing the pump characteristics. Total dynamic head H , discharge Q , shaft torque T and rotational speed N . Two of these four quantities are considered independently, i.e. for specific Q and $\mathrm{N}, \mathrm{H}$ and T are designated from characteristics. Two basic assumptions are made. 1) Steady state characteristics are hold for unsteady state conditions. 2) Similarity relations are valid (Wylie \& Streeter, 1978, 1993). Similarity equations are designated as below.

$$
\begin{align*}
& \frac{Q_{1}}{\left(N_{1} D_{1}{ }^{3}\right)}=\frac{Q_{2}}{\left(N_{2} D_{2}{ }^{3}\right)}  \tag{4-1.a}\\
& \frac{H_{1}}{\left(N_{1} D_{1}\right)^{2}}=\frac{H_{2}}{\left(N_{2} D_{2}\right)^{2}} \tag{4-1.b}
\end{align*}
$$

Where, 1 and 2 subscripts are referred to two different dimensional units of similarity series. For a given unit, Equations (4-1) takes a form such that,
$\frac{H_{1}}{N_{1}{ }^{2}}=\frac{H_{2}}{N_{2}{ }^{2}}$
$\frac{Q_{1}}{N_{1}}=\frac{Q_{2}}{N_{2}}$

Similarity theory assumes that the efficiency does not change with pump's dimension. Hence
$\frac{T_{1} N_{1}}{Q_{1} H_{1}}=\frac{T_{2} N_{2}}{Q_{2} H_{2}}$

Other combinations of Equations (4-2) and (4-3) produce the expressions below.

$$
\begin{equation*}
\frac{T_{1}}{N_{1}{ }^{2}}=\frac{T_{2}}{N_{2}{ }^{2}} \quad \frac{H_{1}}{Q_{1}{ }^{2}}=\frac{H_{2}}{Q_{2}{ }^{2}} \quad \frac{T_{1}}{Q_{1}{ }^{2}}=\frac{T_{2}}{Q_{2}{ }^{2}} \tag{4-4}
\end{equation*}
$$

If we study with dimensionless characteristics,

$$
\begin{equation*}
h=\frac{H}{H_{R}} \quad \beta=\frac{T}{T_{R}} \quad v=\frac{Q}{Q_{R}} \quad \alpha=\frac{N}{N_{R}} \tag{4-5}
\end{equation*}
$$

Where, the subscript R denotes the rated quantities which correspond to values of H , $\mathrm{T}, \mathrm{Q}$ and N at the best efficiency point on the pump characteristic curve. Dimensionless similarity relations can be expressed as follows,

$$
\begin{equation*}
\frac{h}{\alpha^{2}} v s . \frac{v}{\alpha} \quad \frac{\beta}{\alpha^{2}} v s . \frac{v}{\alpha} \quad \frac{h}{v^{2}} v s . \frac{\alpha}{v} \quad \frac{\beta}{v^{2}} \text { vs. } \frac{\alpha}{v} \tag{4-6}
\end{equation*}
$$

From the computational aspect, these relations are not convenient. Because the signs of $h, \beta, v$ and $\alpha$ might be change due to different operating zones and during the transients, the values of all may go to zero. Marchal, Flesch, Suter have been coped with this difficulty using the expressions below (Wylie \& Streeter, 1978, 1993).
$\frac{h}{\alpha^{2}+v^{2}} v s \cdot \tan ^{-1}\left(\frac{v}{\alpha}\right)$
$\frac{\beta}{\alpha^{2}+v^{2}} v s \cdot \tan ^{-1}\left(\frac{v}{\alpha}\right)$

The angle $\theta=x=\pi+\tan ^{-1}(\mathrm{v} / \alpha)$ may be drawn as absisca against $\mathrm{WH}(\mathrm{x})$ or $\mathrm{WB}(\mathrm{x})$ as shown in Figure 4.1. Where
$W H(x)=\frac{h}{\alpha^{2}+v^{2}}$

$$
\begin{align*}
& W B(x)=\frac{\beta}{\alpha^{2}+v^{2}}  \tag{4-8.b}\\
& x=\pi+\tan ^{-1}\left(\frac{v}{\alpha}\right) \tag{4-8.c}
\end{align*}
$$

H, Q, efficiency $(\eta)$, and power $(P)$ values have been provided from the pump manufacturer for a given N value as pump characteristic curve. Torque values are calculated using these data. These four values produce $h, v, \beta$ and $\alpha$ at the time in which rated values are given.

Angle, x, WH, WB may be determined for each operation point by the use of Equation (4-8). Producing 89 values of WH and WB with $\Delta x=\pi / 44$ radians are demonstrated as reasonable number (Wylie \& Streeter, 1978). With specified $\mathrm{N}_{\mathrm{S}}$ (specific speed), WH is called the dimensionless head data and WB is called the dimensionless torque data (Wylie \& Streeter, 1978, 1993).

In many design conditions, complete pump characteristics (WH and WB) are not available in the manufacturer. From other available test data, WH and WB values can be generated. Curves tend to similar shapes for same specific speeds.

$$
\begin{equation*}
N_{S}=\frac{N_{R} \sqrt{Q_{R}}}{H_{R^{\frac{3}{4}}}} \tag{4-9}
\end{equation*}
$$

Where $\mathrm{N}_{\mathrm{S}}$ is the dimensionless specific speed, $\mathrm{N}_{\mathrm{R}}$ is the rated rotational velocity in rpm, $Q_{R}$ is the rated discharge in $\mathrm{m}^{3} / \mathrm{s}, \mathrm{H}_{\mathrm{R}}$ is the rated head in meters. If such data are not available, data can be selected by comparison with other curves which are valid for various specific speeds available in literature. These data are dimensionless, therefore they can be used in both English Gravitational System of Units (EGSU) and International System of Units (SI) (Wylie \& Streeter, 1978, 1993).


Figure 4.1: Complete pump characteristics ( 89 values of WH and 89 values of WB vs. $\theta$, theta angles) for $\mathrm{Ns}=25$ (SI) (Mays, 1999).

Our pump is operated in normal zone in steady state in the first quadrant, zone A excluding zones B and C .

After the pump is run down, v will be decreased to zero, dissipation zone (Quadrant 4, Zone H) will be dominant. In this zone, $\alpha$ is already greater than zero but there are deceleration in both v and $\alpha$. Then pump will operate in turbine zone in the third quadrant. In turbine zone, pump impeller rotates with reverse speed due to backward flow, i.e. $\mathrm{v} \leq 0$ and $\alpha<0$ (Zone G). Quadrant 2 corresponds to reverse speed dissipation zone.

To obtain graphs of WH and WB related to our pump shown the steps given in Table 4-1 are followed.

Table 4.1: Calculation of WH and WB values of our pump for normal pump operation zone.

| Theta | $\alpha$ | V | $\mathrm{Q}, \mathrm{m}^{3} / \mathrm{h}$ | $\mathrm{H}, \mathrm{m}$ | $\mathrm{P}, \mathrm{kw}$ | h | Beta, $\beta$ | WH | WB |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 184.09 | 1 | 0.01 | 2.15 | 14.00 | 3.10 | 1.27 | 0.63 | 1.27 | 0.63 |
| 188.18 | 1 | 0.08 | 11.43 | 14.00 | 3.17 | 1.27 | 0.64 | 1.26 | 0.64 |
| 192.27 | 1 | 0.16 | 20.83 | 14.00 | 3.50 | 1.27 | 0.71 | 1.24 | 0.69 |
| 196.36 | 1 | 0.23 | 30.44 | 14.00 | 3.60 | 1.27 | 0.73 | 1.20 | 0.69 |
| 200.45 | 1 | 0.31 | 40.40 | 14.00 | 3.70 | 1.27 | 0.75 | 1.15 | 0.68 |
| 204.54 | 1 | 0.39 | 50.80 | 13.90 | 3.80 | 1.26 | 0.77 | 1.09 | 0.67 |
| 208.63 | 1 | 0.47 | 61.80 | 13.75 | 4.00 | 1.25 | 0.81 | 1.01 | 0.66 |
| 212.72 | 1 | 0.56 | 73.56 | 13.70 | 4.10 | 1.24 | 0.83 | 0.94 | 0.63 |
| 216.81 | 1 | 0.66 | 86.33 | 13.40 | 4.20 | 1.21 | 0.85 | 0.84 | 0.59 |
| 220.90 | 1 | 0.77 | 100.38 | 12.80 | 4.50 | 1.16 | 0.91 | 0.72 | 0.57 |
| 224.99 | 1 | 0.89 | 116.08 | 11.90 | 4.70 | 1.08 | 0.95 | 0.60 | 0.53 |
| 229.09 | 1 | 1.03 | 133.96 | 10.50 | 4.80 | 0.95 | 0.97 | 0.46 | 0.47 |
| 233.18 | 1 | 1.19 | 154.67 | 9.00 | 5.00 | 0.81 | 1.02 | 0.33 | 0.42 |

Graphs given in Figure 4.2 and 4.3 are drawn with respect to WH and WB values obtained and specified and calculated in Table 4.1 for normal pump operation zone.

For instance, if we calculate the first row of Table 4.1, we first determine the column 1 about $\pi / 44$ radians. To find angle Theta, $\theta$, (180/44)*45=184.09. We calculate the v from Equation (4-8.c), that is, $\alpha \cdot \tan (\operatorname{theta}-\pi)=\mathrm{v}$. The value of $\alpha$ is known as 1 for steady state normal pump operation zone. The value of Q is determined from that $\mathrm{Q}=\mathrm{v} \cdot \mathrm{Q}_{\mathrm{R}}$ read from the pump characteristic curve of values of $H$, and $P$ against to discharge, $Q$ given in Figure 5.7. Once h, $\beta$ are determined from Equation (4-5), $\mathrm{h}=\mathrm{H} / \mathrm{H}_{\mathrm{R}}$ and $\beta=\mathrm{P} / \mathrm{P}_{\mathrm{R}}$ in which $H_{R}, \mathrm{Q}_{\mathrm{R}}$ and $\mathrm{P}_{\mathrm{R}}$ correspond to $11 \mathrm{~m}, 130$ $\mathrm{m}^{3} / \mathrm{h}$ and 4.9 kw respectively for rated conditions. From Equations (4-8.a) and (48.b), we find the values of WH and WB for normal pump operation zone and we drawn the graphs in Figure 4.2 and 4.3 representing the WH and WB values for normal pump operation zone with dark blue lines in the figures as also specified in Table 4.1.


Figure 4.2: Dimensionless head data, WH, drawn for normal pump operation zone. X axis corresponds to theta $(\theta)$ and y axis corresponds to WH values (Wylie \& Streeter, 1993) (between 184.09 degrees Theta angle with 233.18 degrees Theta angle).

As seen form the graph, lines for $\mathrm{N}_{\mathrm{S}}=25$ and 45.54 are in good agreement with each other. Because of this agreement, we can consider the dimensionless head data in our computations in system one and system two corresponding to specific speed 25 (SI). But we also realised interpolations between $\mathrm{N}_{\mathrm{S}}=25$ (SI) and $\mathrm{N}_{\mathrm{S}}=147$ (SI) to find values of WH and WB, and in the computer program we used the interpolated values corresponding to $\mathrm{N}_{\mathrm{S}}=45.55$ (SI) (for our pump).

In Figures 4.2 and 4.3, values of WH and WB against to three different specific speeds have no dimensions. Therefore WH and WB values presented by Wylie \& Streeter, 1993 are valid both in English Gravitational System of Units (EGSU) and International System of Units, SI. WH and WB values against to three different specific speeds, Ns presented in Wylie \& Streeter $(1978,1993)$ were developed from

Hollander's experiments. 89 values of WH and WB have proved to be a reasonable number (Wylie \& Streeter, 1978, 1993).


Figure 4.3: Dimensionless torque data, WB, drawn for normal pump operation zone. X axis represents the theta $(\theta)$ and y axis represents the WB values (Wylie \& Streeter, 1993). (Note that between 184.09 degrees with 233.18 degrees Theta angle).

WH and WB values corresponding to three different values of Ns (25, 147 and 261) are presented in Table 4.2. Table 4.3 contains the WH and WB values corresponding to our pump ( $\mathrm{Ns}=45.5$ ) obtained from interpolation between $\mathrm{Ns}=25$ and $\mathrm{Ns}=147$.

Table 4.2: Dimensionless pump characteristics for various specific speeds against to $\pi / 44$ radians equal angles.

|  | $\mathrm{N}_{\mathrm{S}}=25(\mathrm{SI})$ |  | $\mathrm{N}_{\mathrm{S}}=147$ (SI) |  | $\mathrm{N}_{\mathrm{S}}=261(\mathrm{SI})$ |  |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| Theta | $\mathrm{h} /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\beta /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\mathrm{h} /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\beta /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\mathrm{h} /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\beta /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ |
| 0.000 | 0.634 | -0.684 | -0.690 | -1.420 | -2.230 | -2.260 |
| 4.090 | 0.643 | -0.547 | -0.599 | -1.328 | -2.000 | -2.061 |

Table 4.2: (Continued)

| 8.181 | 0.646 | -0.414 | -0.512 | -1.211 | -1.662 | -1.772 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 12.272 | 0.640 | -0.292 | -0.418 | -1.056 | -1.314 | -1.465 |
| 16.363 | 0.629 | -0.167 | -0.304 | -0.870 | -1.089 | -1.253 |
| 20.454 | 0.613 | -0.105 | -0.181 | -0.677 | -0.914 | -1.088 |
| 24.545 | 0.595 | -0.053 | -0.078 | -0.573 | -0.750 | -0.921 |
| 28.636 | 0.575 | -0.012 | -0.011 | -0.518 | -0.601 | -0.789 |
| 32.727 | 0.552 | 0.042 | 0.032 | -0.380 | -0.440 | -0.632 |
| 36.818 | 0.533 | 0.097 | 0.074 | -0.232 | -0.284 | -0.457 |
| 40.909 | 0.516 | 0.156 | 0.130 | -0.160 | -0.130 | -0.300 |
| 44.999 | 0.505 | 0.227 | 0.190 | 0.000 | 0.055 | -0.075 |
| 49.090 | 0.504 | 0.300 | 0.265 | 0.118 | 0.222 | 0.052 |
| 53.181 | 0.510 | 0.371 | 0.363 | 0.308 | 0.357 | 0.234 |
| 57.272 | 0.512 | 0.444 | 0.461 | 0.442 | 0.493 | 0.425 |
| 61.363 | 0.522 | 0.522 | 0.553 | 0.574 | 0.616 | 0.558 |
| 65.454 | 0.539 | 0.596 | 0.674 | 0.739 | 0.675 | 0.630 |
| 69.545 | 0.559 | 0.672 | 0.848 | 0.929 | 0.580 | 0.621 |
| 73.636 | 0.580 | 0.738 | 1.075 | 1.147 | 0.691 | 0.546 |
| 77.727 | 0.601 | 0.763 | 1.337 | 1.370 | 0.752 | 0.525 |
| 81.818 | 0.630 | 0.797 | 1.629 | 1.599 | 0.825 | 0.488 |
| 85.909 | 0.662 | 0.837 | 1.929 | 1.839 | 0.930 | 0.512 |
| 89.999 | 0.692 | 0.865 | 2.180 | 2.080 | 1.080 | 0.660 |
| 94.090 | 0.722 | 0.883 | 2.334 | 2.300 | 1.236 | 0.850 |
| 98.181 | 0.753 | 0.886 | 2.518 | 2.480 | 1.389 | 1.014 |
| 102.272 | 0.782 | 0.877 | 2.726 | 2.630 | 1.548 | 1.162 |
| 106.363 | 0.808 | 0.859 | 2.863 | 2.724 | 1.727 | 1.334 |
| 110.425 | 0.832 | 0.838 | 2.948 | 2.687 | 1.919 | 1.512 |
| 114.545 | 0.857 | 0.804 | 3.026 | 2.715 | 2.066 | 1.683 |
| 118.636 | 0.879 | 0.756 | 3.015 | 2.688 | 2.252 | 1.886 |
| 122.727 | 0.904 | 0.703 | 2.927 | 2.555 | 2.490 | 2.105 |
| 126.818 | 0.930 | 0.645 | 2.873 | 2.434 | 2.727 | 2.325 |
| 130.909 | 0.959 | 0.583 | 2.771 | 2.288 | 3.002 | 2.580 |

Table 4.2 : (Continued)

| 134.999 | 0.996 | 0.520 | 2.640 | 2.110 | 3.225 | 2.770 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 139.090 | 1.027 | 0.454 | 2.497 | 1.948 | 3.355 | 2.886 |
| 143.181 | 1.060 | 0.408 | 2.441 | 1.825 | 3.475 | 2.959 |
| 147.272 | 1.090 | 0.370 | 2.378 | 1.732 | 3.562 | 2.979 |
| 151.363 | 1.124 | 0.343 | 2.336 | 1.644 | 3.604 | 2.962 |
| 155.454 | 1.165 | 0.331 | 2.288 | 1.576 | 3.582 | 2.877 |
| 159.545 | 1.204 | 0.329 | 2.209 | 1.533 | 3.540 | 2.713 |
| 163.636 | 1.238 | 0.338 | 2.162 | 1.522 | 3.477 | 2.556 |
| 167.727 | 1.258 | 0.354 | 2.140 | 1.519 | 3.321 | 2.403 |
| 171.818 | 1.271 | 0.372 | 2.109 | 1.523 | 3.148 | 2.237 |
| 175.909 | 1.282 | 0.405 | 2.054 | 1.523 | 2.962 | 2.080 |
| 179.999 | 1.288 | 0.450 | 1.970 | 1.490 | 2.750 | 1.950 |
| 184.090 | 1.281 | 0.486 | 1.860 | 1.386 | 2.542 | 1.826 |
| 188.181 | 1.260 | 0.520 | 1.735 | 1.223 | 2.354 | 1.681 |
| 192.272 | 1.225 | 0.552 | 1.571 | 1.048 | 2.149 | 1.503 |
| 196.363 | 1.172 | 0.579 | 1.357 | 0.909 | 1.909 | 1.301 |
| 200.454 | 1.107 | 0.603 | 1.157 | 0.914 | 1.702 | 1.115 |
| 204.545 | 1.031 | 0.616 | 1.016 | 0.766 | 1.506 | 0.960 |
| 208.636 | 0.942 | 0.617 | 0.927 | 0.734 | 1.310 | 0.840 |
| 212.727 | 0.842 | 0.606 | 0.846 | 0.678 | 1.131 | 0.750 |
| 216.818 | 0.733 | 0.582 | 0.744 | 0.624 | 0.947 | 0.677 |
| 220.909 | 0.617 | 0.546 | 0.640 | 0.570 | 0.737 | 0.604 |
| 224.999 | 0.500 | 0.500 | 0.500 | 0.500 | 0.500 | 0.500 |
| 229.090 | 0.368 | 0.432 | 0.374 | 0.407 | 0.279 | 0.352 |
| 233.181 | 0.240 | 0.360 | 0.191 | 0.278 | 0.082 | 0.161 |
| 237.272 | 0.125 | 0.288 | 0.001 | 0.146 | -0.112 | -0.040 |
| 241.363 | 0.011 | 0.214 | -0.190 | 0.023 | -0.300 | -0.225 |
| 245.454 | -0.102 | 0.123 | -0.384 | -0.175 | -0.505 | -0.403 |
| 249.545 | -0.168 | 0.037 | -0.585 | -0.379 | -0.672 | -0.545 |
| 253.636 | -0.255 | -0.053 | -0.786 | -0.585 | -0.797 | -0.610 |
| 257.727 | -0.342 | -0.161 | -0.972 | -0.778 | -0.872 | -0.662 |

Table 4.2 : (Continued)

| 261.818 | -0.423 | -0.248 | -1.185 | -1.008 | -0.920 | -0.699 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 265.909 | -0.494 | -0.314 | -1.372 | -1.277 | -0.949 | -0.719 |
| 269.999 | -0.556 | -0.372 | -1.500 | -1.560 | -0.960 | -0.730 |
| 274.090 | -0.620 | -0.580 | -1.940 | -2.070 | -1.080 | -0.810 |
| 278.181 | -0.655 | -0.740 | -2.160 | -2.480 | -1.300 | -1.070 |
| 282.272 | -0.670 | -0.880 | -2.290 | -2.700 | -1.500 | -1.360 |
| 286.363 | -0.670 | -1.000 | -2.350 | -2.770 | -1.700 | -1.640 |
| 290.454 | -0.660 | -1.120 | -2.350 | -2.800 | -1.890 | -1.880 |
| 294.545 | -0.655 | -1.250 | -2.230 | -2.800 | -2.080 | -2.080 |
| 298.636 | $-0,640$ | -1.370 | -2.200 | -2.760 | -2.270 | -2.270 |
| 302.727 | -0.600 | -1.490 | -2.130 | -2.710 | -2.470 | -2.470 |
| 306.818 | -0.570 | -1.590 | -2.050 | -2.640 | -2.650 | -2.650 |
| 310.909 | -0.520 | -1.660 | -1.970 | -2.540 | -2.810 | -2.810 |
| 314.999 | -0.470 | -1.690 | -1.895 | -2.440 | -2.950 | -2.950 |
| 319.090 | -0.430 | -1.770 | -1.810 | -2.340 | -3.040 | -3.040 |
| 323.181 | -0.360 | -1.650 | -1.730 | -2.240 | -3.100 | -3.100 |
| 327.272 | -0.275 | -1.590 | -1.600 | -2.120 | -3.150 | -3.150 |
| 331.363 | -0.160 | -1.520 | -1.420 | -2.000 | -3.170 | -3.170 |
| 335.454 | -0.040 | -1.420 | -1.130 | -1.940 | -3.170 | -3.200 |
| 339.545 | 0.130 | -1.320 | -0.950 | -1.900 | -3.130 | -3.160 |
| 343.636 | 0.295 | -1.230 | -0.930 | -1.900 | -3.070 | -3.090 |
| 347.727 | 0.430 | -1.100 | -0.950 | -1.850 | -2.960 | -2.990 |
| 351.818 | 0.550 | -0.980 | -1.000 | -1.750 | -2.820 | -2.860 |
| 355.909 | 0.620 | -0.820 | -0.920 | -1.630 | -2.590 | -2.660 |
| 360.000 | 0.634 | -0.684 | -0.690 | -1.420 | -2.230 | -2.260 |

Table 4.3: Dimensionless pump characteristics for $\mathrm{N}_{\mathrm{s}}=45.5$ (SI) (for our pump) interpolated between $\mathrm{N}_{\mathrm{s}}=25$ (SI) and $\mathrm{N}_{\mathrm{s}}=147$ (SI) against to $\pi / 44$ radians equal angles.

|  | $\mathrm{N}_{\mathrm{S}}=25(\mathrm{SI})$ |  | $\mathrm{N}_{\mathrm{S}}=45.5(\mathrm{SI})$ |  | $\mathrm{N}_{\mathrm{S}}=147$ (SI) |  |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| Theta | $\mathrm{h} /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\beta /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\mathrm{h} /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\beta /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\mathrm{h} /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ | $\beta /\left(\alpha^{2}+\mathrm{v}^{2}\right)$ |

Table 4.3 : (Continued)

| 0.000 | 0.634 | -0.684 | 0.411 | -0.808 | -0.690 | -1.420 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4.090 | 0.643 | -0.547 | 0.4338 | -0.6785 | -0.599 | -1.328 |
| 8.181 | 0.646 | -0.414 | 0.451 | -0.5482 | -0.512 | -1.211 |
| 12.272 | 0.640 | -0.292 | 0.4618 | -0.4207 | -0.418 | -1.056 |
| 16.363 | 0.629 | -0.167 | 0.4719 | -0.302 | -0.304 | -0.870 |
| 20.454 | 0.613 | -0.105 | 0.4793 | -0.2013 | -0.181 | -0.677 |
| 24.545 | 0.595 | -0.053 | 0.4816 | -0.1406 | -0.078 | -0.573 |
| 28.636 | 0.575 | -0.012 | 0.4763 | -0.0972 | -0.011 | -0.518 |
| 32.727 | 0.552 | 0.042 | 0.4644 | -0.0291 | 0.032 | -0.380 |
| 36.818 | 0.533 | 0.097 | 0.4557 | 0.0416 | 0.074 | -0.232 |
| 40.909 | 0.516 | 0.156 | 0.451 | 0.1028 | 0.130 | -0.160 |
| 44.999 | 0.505 | 0.227 | 0.4519 | 0.1888 | 0.190 | 0.000 |
| 49.090 | 0.504 | 0.300 | 0.4637 | 0.2693 | 0.265 | 0.118 |
| 53.181 | 0.510 | 0.371 | 0.4852 | 0.3604 | 0.363 | 0.308 |
| 57.272 | 0.512 | 0.444 | 0.5034 | 0.4437 | 0.461 | 0.442 |
| 61.363 | 0.522 | 0.522 | 0.5272 | 0.5308 | 0.553 | 0.574 |
| 65.454 | 0.539 | 0.596 | 0.5617 | 0.6201 | 0.674 | 0.739 |
| 69.545 | 0.559 | 0.672 | 0.6077 | 0.7153 | 0.848 | 0.929 |
| 73.636 | 0.580 | 0.738 | 0.6634 | 0.8096 | 1.075 | 1.147 |
| 77.727 | 0.601 | 0.763 | 0.725 | 0.8652 | 1.337 | 1.370 |
| 81.818 | 0.630 | 0.797 | 0.7983 | 0.9321 | 1.629 | 1.599 |
| 85.909 | 0.662 | 0.837 | 0.8754 | 1.0058 | 1.929 | 1.839 |
| 89.999 | 0.692 | 0.865 | 0.9426 | 1.0696 | 2.180 | 2.080 |
| 94.090 | 0.722 | 0.883 | 0.9935 | 1.1217 | 2.334 | 2.300 |
| 98.181 | 0.753 | 0.886 | 1.0503 | 1.1545 | 2.518 | 2.480 |
| 102.272 | 0.782 | 0.877 | 1.1094 | 1.1723 | 2.726 | 2.630 |
| 106.363 | 0.808 | 0.859 | 1.1541 | 1.1731 | 2.863 | 2.724 |
| 110.425 | 0.832 | 0.838 | 1.1884 | 1.1494 | 2.948 | 2.687 |
| 114.545 | 0.857 | 0.804 | 1.2223 | 1.1259 | 3.026 | 2.715 |
| 118.636 | 0.879 | 0.756 | 1.2388 | 1.0831 | 3.015 | 2.688 |
| 122.727 | 0.904 | 0.703 | 1.2447 | 1.0149 | 2.927 | 2.555 |

Table 4.3 : (Continued)

| 126.818 | 0.930 | 0.645 | 1.2573 | 0.9463 | 2.873 | 2.434 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 130.909 | 0.959 | 0.583 | 1.2642 | 0.8702 | 2.771 | 2.288 |
| 134.999 | 0.996 | 0.520 | 1.2729 | 0.7878 | 2.640 | 2.110 |
| 139.090 | 1.027 | 0.454 | 1.2746 | 0.7056 | 2.497 | 1.948 |
| 143.181 | 1.060 | 0.408 | 1.2926 | 0.6467 | 2.441 | 1.825 |
| 147.272 | 1.090 | 0.370 | 1.3069 | 0.5994 | 2.378 | 1.732 |
| 151.363 | 1.124 | 0.343 | 1.3281 | 0.5621 | 2.336 | 1.644 |
| 155.454 | 1.165 | 0.331 | 1.3541 | 0.5407 | 2.288 | 1.576 |
| 159.545 | 1.204 | 0.329 | 1.3733 | 0.5318 | 2.209 | 1.533 |
| 163.636 | 1.238 | 0.338 | 1.3936 | 0.5374 | 2.162 | 1.522 |
| 167.727 | 1.258 | 0.354 | 1.4066 | 0.5502 | 2.140 | 1.519 |
| 171.818 | 1.271 | 0.372 | 1.4121 | 0.5659 | 2.109 | 1.523 |
| 175.909 | 1.282 | 0.405 | 1.4120 | 0.5933 | 2.054 | 1.523 |
| 179.999 | 1.288 | 0.450 | 1.4029 | 0.6252 | 1.970 | 1.490 |
| 184.090 | 1.281 | 0.486 | 1.3785 | 0.6376 | 1.860 | 1.386 |
| 188.181 | 1.260 | 0.520 | 1.3400 | 0.6384 | 1.735 | 1.223 |
| 192.272 | 1.225 | 0.552 | 1.2833 | 0.6355 | 1.571 | 1.048 |
| 196.363 | 1.172 | 0.579 | 1.2032 | 0.6346 | 1.357 | 0.909 |
| 200.454 | 1.107 | 0.603 | 1.1154 | 0.6385 | 1.157 | 0.914 |
| 204.545 | 1.031 | 0.616 | 1.0436 | 0.6413 | 1.016 | 0.766 |
| 208.636 | 0.942 | 0.617 | 0.9395 | 0.6367 | 0.927 | 0.734 |
| 212.727 | 0.842 | 0.606 | 0.8427 | 0.6181 | 0.846 | 0.678 |
| 216.818 | 0.733 | 0.582 | 0.7349 | 0.5891 | 0.744 | 0.624 |
| 220.909 | 0.617 | 0.546 | 0.6209 | 0.5500 | 0.640 | 0.570 |
| 224.999 | 0.500 | 0.500 | 0.5000 | 0.5000 | 0.500 | 0.500 |
| 229.090 | 0.368 | 0.432 | 0.3690 | 0.4278 | 0.374 | 0.407 |
| 233.181 | 0.240 | 0.360 | 0.2317 | 0.3462 | 0.191 | 0.278 |
| 237.272 | 0.125 | 0.288 | 0.1041 | 0.2641 | 0.001 | 0.146 |
| 241.363 | 0.011 | 0.214 | -0.0229 | 0.1818 | -0.190 | 0.023 |
| 245.454 | -0.102 | 0.123 | -0.1495 | 0.0728 | -0.384 | -0.175 |
| 249.545 | -0.168 | 0.037 | -0.2382 | -0.0331 | -0.585 | -0.379 |

Table 4.3 : (Continued)

| 253.636 | -0.255 | -0.053 | -0.3444 | -0.1426 | -0.786 | -0.585 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 257.727 | -0.342 | -0.161 | -0.4481 | -0.2649 | -0.972 | -0.778 |
| 261.818 | -0.423 | -0.248 | -0.5513 | -0.3760 | -1.185 | -1.008 |
| 265.909 | -0.494 | -0.314 | -0.6419 | -0.4762 | -1.372 | -1.277 |
| 269.999 | -0.556 | -0.372 | -0.7150 | -0.5721 | -1.500 | -1.560 |
| 274.090 | -0.620 | -0.580 | -0.8423 | -0.8310 | -1.940 | -2.070 |
| 278.181 | -0.655 | -0.740 | -0.9085 | -1.0331 | -2.160 | -2.480 |
| 282.272 | -0.670 | -0.880 | -0.9429 | -1.1865 | -2.290 | -2.700 |
| 286.363 | -0.670 | -1.000 | -0.9530 | -1.2981 | -2.350 | -2.770 |
| 290.454 | -0.660 | -1.120 | -0.9446 | -1.4030 | -2.350 | -2.800 |
| 294.545 | -0.655 | -1.250 | -0.9203 | -1.5110 | -2.230 | -2.800 |
| 298.636 | -0.640 | -1.370 | -0.9027 | -1.6041 | -2.200 | -2.760 |
| 302.727 | -0.600 | -1.490 | -0.8577 | -1.6955 | -2.130 | -2.710 |
| 306.818 | -0.570 | -1.590 | -0.8193 | -1.7668 | -2.050 | -2.640 |
| 310.909 | -0.520 | -1.660 | -0.7642 | -1.8082 | -1.970 | -2.540 |
| 314.999 | -0.470 | -1.690 | -0.7100 | -1.8163 | -1.895 | -2.440 |
| 319.090 | -0.430 | -1.770 | -0.6624 | -1.8660 | -1.810 | -2.340 |
| 323.181 | -0.360 | -1.650 | -0.5907 | -1.7494 | -1.730 | -2.240 |
| 327.272 | -0.275 | -1.590 | -0.4982 | -1.6793 | -1.600 | -2.120 |
| 331.363 | -0.160 | -1.520 | -0.3722 | -1.6008 | -1.420 | -2.000 |
| 335.454 | -0.040 | -1.420 | -0.2236 | -1.5076 | -1.130 | -1.940 |
| 339.545 | 0.130 | -1.320 | -0.0519 | -1.4177 | -0.950 | -1.900 |
| 343.636 | 0.295 | -1.230 | 0.0887 | -1.3428 | -0.930 | -1.900 |
| 347.727 | 0.430 | -1.100 | 0.1976 | -1.2263 | -0.950 | -1.850 |
| 351.818 | 0.550 | -0.980 | 0.2889 | -1.1097 | -1.000 | -1.750 |
| 355.909 | 0.620 | -0.820 | 0.3606 | -0.9564 | -0.920 | -1.630 |
| 360.000 | 0.634 | -0.684 | 0.4110 | -0.8080 | -0.690 | -1.420 |

Interpolated dimensionless head and torque values for the whole range of $\theta$ are given in Figures 4.4 and 4.5 respectively with those corresponding to $\mathrm{Ns}=25$ and $\mathrm{Ns}=147$.


Figure 4.4: Dimensionless head data, WH corresponding to Theta, $\theta$

Dimensionless torque data for $\mathbf{N s}=\mathbf{2 5}, 147$ and 45.5 (SI)


Figure 4.5: Dimensionless torque data, WB corresponding to Theta, $\theta$

The moment of inertia (WRR) of the rotating parts is also needed. This information will be reliable if it is provided from the manufacturer. If not available, some
empirical relations given in literature should be used. Donsky presents a graphic of WRR (the weight of motor rotor times square of radius of gyration) for most of motors. Estimated equation is

$$
\begin{equation*}
W R R=3550\left(\frac{H P}{N}\right)^{1.435} \tag{4-10}
\end{equation*}
$$

Where HP is the power in horse power unit ( $1 \mathrm{KW}=1.338 \mathrm{HP}$ ), N is the steady state rotational speed in rpm. The unit of WRR is $1 \mathrm{~b} . \mathrm{ft}^{2}$. We must convert the unit of WRR to $\mathrm{Nt} . \mathrm{m}^{2}$ for the convenience to International System of Units (SI).

The value of WRR can be calculated from Equation (4-10). But this formula may produce acceptable results for pumps manufactured before 1960. Calculated WRR value may be upper for pumps manufactured after 1960 (Wylie \& Streeter, 1978). Other types of formulas have been presented by Thorley (1991) as an alternative to Equation (4-10) presented by Donsky. According to Thorley (1991),

$$
\begin{equation*}
I=I_{P}+I_{M} \tag{4-11}
\end{equation*}
$$

Where I is the total rotational inertia in kg. ${ }^{2}$, $I_{P}$ is the inertia of the pump impellers including the entrained water and shaft in kg. $\mathrm{m}^{2}$ and $\mathrm{I}_{\mathrm{M}}$ is the rotational inertia of motor in kg.m ${ }^{2}$.

$$
\begin{equation*}
I_{P}=1.5 .10^{7}\left(\frac{P_{R}}{N_{R}{ }^{3}}\right)^{0.9556} \tag{4-12}
\end{equation*}
$$

$I_{M}=118\left(\frac{P_{R}}{N_{R}}\right)^{1.48}$

Where $P_{R}$ is the rated shaft power in kW and $\mathrm{N}_{\mathrm{R}}$ is the rated rotational (radial) speed in rpm. Considered WRR value in our computations is summation of the results from

Equations (4-11), WRR is $0.83 \mathrm{Nt}. \mathrm{~m}^{2}$ which corresponds to latter specified value above suitable for SI. We considered in our computations the formula proposed by Thorley (1991), Equation (4-11). Thorley argue that for preliminary studies this value has sufficient accuracy but for final computations Thorley advice that WRR value should be taken as doubled.

Rated torque value, $\mathrm{T}_{\mathrm{R}}$ is needed for our computations as an input value to the computer programs, which can be determined from the formula below.

$$
\begin{equation*}
T_{R}=\frac{30 P_{R}}{\pi N_{R}} \tag{4-14}
\end{equation*}
$$

Where $P_{R}$ is the rated power in Watts and $N_{R}$ is the rotational speed in rpm. According to Equation (4-14), $\mathrm{T}_{\mathrm{R}}$ value in the computations is considered as 32.18 Nt.m. $\alpha, \beta, \mathrm{v}$ and h are all positive over the normal pump operation. They can be negative individually or as a group during the transient flows.

### 4.3 Head balance and Torque angular reducing equations for the $1^{\text {st }}$ system

During computations at the upstream end, the required equations are torqueangular reducing equations describing velocity change against torque as differential equation for pump and head balance equation along pump.

### 4.3.1 Head balance equation for the $1^{\text {st }}$ system

At the upstream end there exists a pump which delivers water to the system without any valve.

Since suction flange is not long, it is not considered during transient analysis. Therefore the pump is assumed to be located at the upstream end of the system. $\mathrm{H}_{\mathrm{s}}$ represent the piezometric head at suction sump.


Figure 4.6: Sketch of pump in our first pipeline system

If $H_{s}$ represent the piezometric head at suction sump, the head balance relationship will be

$$
\begin{equation*}
H_{S}+H=H_{P 1} \tag{4-15}
\end{equation*}
$$

In which, H is the total dynamic head added by pump. Total dynamic head, H is translated by assuming that WH varies linearly,

$$
\begin{equation*}
t d h=H=H_{R} h=H_{R}\left(\alpha^{2}+v^{2}\right)\left(A_{0}+A 1\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right) \tag{4-16}
\end{equation*}
$$

A linear relationship is assumed to be valid for the segment of WH corresponding to considered time interval. The total dynamic head, tdh is given from the dimensionless similarity relationships in Equation (4-16).

The curve of WH as a function of $x=\pi+\tan ^{-1}(\mathrm{v} / \alpha)$ is replaced by a straight line representing the characteristic pump head for the proper vicinity of $x$. In Figure 4.7, the approximate location of $\pi+\tan ^{-1}(\mathrm{v} / \alpha)$ obtained by interpolating v and $\alpha$ from the previous calculations, a straight line is defined through the two adjoining data points, as follows,

$$
\begin{equation*}
I=x / \Delta x+1 \tag{4-17}
\end{equation*}
$$

In Figure 4.7, $(I-1) \Delta x, W H(I)$ and $I \Delta x, W H(I+1)$ are the Cartesian coordinates.


Figure 4.7: Replacement of pump curve with a straight line.

The Cartesian coordinates of two data points are substituted into $\mathrm{WH}=\mathrm{A}_{0}+\mathrm{A}_{1} \mathrm{x}$,
$A_{1}=[W H(I+1)-W H(I)] / \Delta x$
$A_{0}=W H(I+1)-I A_{1} \Delta x$
$C^{-}: H P_{1}=C M+B M \cdot Q_{P 1} \quad ; \quad C M=H_{2}-B Q_{2} \quad ; B M=B+R\left|Q_{2}\right|$

From Equation (4-15),

During the computations, suction pipe has not been considered. If the suction reservoir is taken as datum,

$$
\begin{equation*}
H=H_{P 1} \tag{4-20}
\end{equation*}
$$

$H_{R}\left(\alpha^{2}+v^{2}\right)\left(A_{0}+A 1 \cdot\left(\pi+\tan ^{-1}\left(\frac{v}{\alpha}\right)\right)\right)-C M-B M \cdot Q_{1}=0$

By substituting Equation (4-5),
$Q_{1}=v \cdot Q_{R}$

Resultant head balance equation for system one will be such that,

$$
\begin{equation*}
F_{H}=H_{R}\left(\alpha^{2}+v^{2}\right)\left(A_{0}+A 1 \cdot\left(\pi+\tan ^{-1}\left(\frac{v}{\alpha}\right)\right)\right)-C M-B M \cdot v \cdot Q_{R}=0 \tag{4-22}
\end{equation*}
$$

### 4.3.2 Speed change equation for first system

The change in rotational speed of the pump depends upon the unbalanced torque applied.

$$
\begin{equation*}
T=-\frac{W R_{g}{ }^{2}}{g} \frac{d w}{d t} \tag{4-23}
\end{equation*}
$$

In which W is the weight of rotating parts plus entrained liquid and $\mathrm{R}_{\mathrm{g}}$ is the radius of gyration of the rotating mass, $w$ is the angular velocity in radians per second and $\mathrm{dw} / \mathrm{dt}$ is the angular acceleration in $\mathrm{rad} / \mathrm{s}^{2}$. The unbalanced torque is represented as the average of $T_{0}$, the known torque at beginning of $\Delta t$ and $T_{P}$ the unknown torque at the end of $\Delta t$. Since

$$
\begin{equation*}
w=N_{R} \frac{2 \pi}{60} \alpha \quad \beta_{0}=\frac{T_{0}}{T_{R}} \quad \beta=\frac{T_{P}}{T_{R}} \tag{4-24}
\end{equation*}
$$

Equation (4-25) takes the form that for two $\Delta t$,

$$
\begin{equation*}
\beta=\frac{W R_{g}^{2}}{g} \frac{N_{R}}{T_{R}} \frac{\pi}{30} \frac{\left(\alpha_{0}-\alpha\right)}{\Delta t}-\beta_{0} \tag{4-25}
\end{equation*}
$$

$\alpha_{0}$ is the dimensionless speed at beginning of $\Delta t$. By defining

$$
\begin{equation*}
C T O R Q=\frac{W R_{g}{ }^{2}}{g} \frac{N_{R}}{T_{R}} \frac{\pi}{30 . \Delta t} \tag{4-26}
\end{equation*}
$$

Equation (4-27) becomes

$$
\begin{equation*}
\beta+\beta_{0}-\operatorname{CTORQ}\left(\alpha_{0}-\alpha\right)=0 \tag{4-27}
\end{equation*}
$$

By assuming linear variation of WB through time interval $\Delta t$

$$
\begin{equation*}
\frac{\beta}{\alpha^{2}+v^{2}}=W B(x)=B_{0}+B_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right) \tag{4-28}
\end{equation*}
$$

In which $B_{0}$ and $B_{1}$ are found in the same manner as $A_{0}$ and $A_{1}$. By combining Equations (4-27) and (4-28),
$F_{T}=\left(\alpha^{2}+v^{2}\right)\left[B_{0}+B_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]+\beta_{0}-\operatorname{CTORQ} Q\left(\alpha_{0}-\alpha\right)=0$
which is the speed change equation in v and $\alpha$.

The zero subscripts on the dimensionless speed and torque refer to the values two time steps earlier. We calculate the heads at each $2 \Delta \mathrm{t}$.

Equations (4-22) and (4-29) are solved simultaneously for $\alpha$ and v using the Newton Raphson numerical procedure.

By using founded values of v and $\alpha$ after the simultaneous solution, we correct the v and $\alpha$ values and hence we found the new values of $v$ and $\alpha$. With the new values of v and $\alpha$, we find the transient discharge and transient piezometric head values.

### 4.3.3 Newton Raphson numerical method for first system

If we refer to single pump boundary condition for system one calculations, the solutions of Equations (4-22) and (4-29) is carried out numerically by the NewtonRaphson method. In here, there are two independent variables as $\alpha$ and v in the expressions of Equations (4-22) and (4-29).

In the case of function with one variable, $\mathrm{x}_{\mathrm{n}+1}$ may be found by using the following Newton Raphson formula.


Figure 4.8: Derivation of Newton Raphson formula for one variable.

From Figure 4.8, we define the $\tan (\theta)$ at point A .

$$
\begin{align*}
& \tan \theta=\text { slope }=\frac{|B C|}{|A C|}=\frac{F\left(x_{n}\right)}{x_{n}-x_{n+1}}=\frac{d F\left(x_{n}\right)}{d x} \\
& x_{n+1}=x_{n}-\frac{F\left(x_{n}\right)}{\frac{d F\left(x_{n}\right)}{d x}} \tag{4-30}
\end{align*}
$$

For functions with two variables, say $\mathrm{P}(\mathrm{x}, \mathrm{y})$ and $\mathrm{Q}(\mathrm{x}, \mathrm{y})$, from truncated approximation of Taylor series,
$P\left(x_{n}, y_{n}\right)=P_{n}+\left(x_{n+1}-x_{n}\right) P_{n x}+\left(y_{n+1}-y_{n}\right) P_{n y}=0$
$Q\left(x_{n}, y_{n}\right)=Q_{n}+\left(x_{n+1}-x_{n}\right) Q_{n x}+\left(y_{n+1}-y_{n}\right) Q_{n y}=0$
may be written, where subscripts denote partial derivatives. If we substitute $\mathrm{P}(\mathrm{x}, \mathrm{y})=\mathrm{F}_{\mathrm{H}}(\mathrm{v}, \alpha)$ and $\mathrm{Q}(\mathrm{x}, \mathrm{y})=\mathrm{F}_{\mathrm{T}}(\mathrm{v}, \alpha)$
$F_{H}+F_{H v} \Delta v+F_{H \alpha} \Delta \alpha=0$
$F_{T}+F_{T v} \Delta v+F_{T \alpha} \Delta \alpha=0$
v and $\alpha$ are unknown to be determined. As a first step in the solution v and $\alpha$ may be approximated at the end of $\Delta \mathrm{t}$ by an extrapolation of the two next earlier values.

If $\mathrm{v}_{00}$ denotes one time step early from $\mathrm{v}_{0}$,
$v=2 v_{0}-v_{00}$
$\alpha=2 \alpha_{0}-\alpha_{00}$
$\mathrm{F}_{\mathrm{H}}, \mathrm{F}_{\mathrm{Hv}}, \mathrm{F}_{\mathrm{H} \alpha}, \mathrm{F}_{\mathrm{T}}, \mathrm{F}_{\mathrm{Tv}}, \mathrm{F}_{\mathrm{T} \alpha}$ are evaluated for this v and $\alpha$. The derivatives are

$$
\begin{align*}
& F_{H v}=-Q_{R}(B P+B M)+H_{R}\left\{2 v\left[A_{0}+A_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]+A_{1} \alpha\right\}  \tag{4-34}\\
& F_{H \alpha}=H_{R}\left\{2 \alpha\left[A_{0}+A_{1}\left(\alpha+\tan ^{-1} \frac{v}{\alpha}\right)\right]-v A_{1}\right\}  \tag{4-35}\\
& F_{T_{v}}=2 v\left[B_{0}+B_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]+\alpha B_{1} \tag{4-36}
\end{align*}
$$

$F_{T_{\alpha}}=2 \alpha\left[B_{0}+B_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]-v B_{1}+C_{\text {TORQ }}$

Equation (4-32) may be solved for $\Delta v, \Delta \alpha$.

$$
\begin{gather*}
\Delta \alpha=\frac{\frac{F_{T}}{F_{T v}}-\frac{F_{H}}{F_{H v}}}{\frac{F_{H \alpha}}{F_{H v}}-\frac{F_{T \alpha}}{F_{T v}}}  \tag{4-38}\\
\Delta v=-\frac{F_{H}}{F_{H v}}-\Delta \alpha \frac{F_{H \alpha}}{F_{H v}} \tag{4-39}
\end{gather*}
$$

Then the new $\alpha$ and $v$ values are evaluated. These new values are the improved values and this procedure is repeated either a fixed number of times or until the tolerance TOL is met. The tolerance represents difference between estimated and calculated values and may be around 0.0002 .

$$
\begin{align*}
& \alpha=\alpha+\Delta \alpha \\
& v=v+\Delta v \\
& |\Delta v|+|\Delta \alpha|\langle T O L \tag{4-41}
\end{align*}
$$

After these equations are solved, the values $\mathrm{A}_{0}, \mathrm{~A}_{1}, \mathrm{~B}_{0}, \mathrm{~B}_{1}$ must be verified. First we calculate the integer

$$
\begin{equation*}
I I=\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right) /(\Delta x+1) \tag{4-42}
\end{equation*}
$$

If this equals I in the Equation (4-17) then the solution is represented by the proper straight line segments of WH and WB. If II does not equal I, then the procedure should be repeated by replacing I by II. This may be tried 3 or 4 times, if the equality is not obtained then the program should be stopped with a comment that the boundary condition is in trouble.

### 4.4 Head balance and Torque angular reducing equations for the $2^{\text {nd }}$ system

### 4.4.1 Head balance equation for the $2^{\text {nd }}$ system

As shown in Figure 4.9, there exist a check valve and a regulating valve placed nearby the pump at the upstream end. If H and $\mathrm{h}_{\mathrm{L}}$ denote the total dynamic head of the pump and local losses respectively
$H=H_{P 1}+h_{L}$
may be written. The local losses, $\mathrm{h}_{\mathrm{L}}$ are due to presence of check valve and valve. The $\mathrm{C}^{-}$equation between points (1) and (2),
$H P_{1}=C M+B M \cdot Q_{P 1}$


Figure 4.9: Sketch of the pump in a pipeline for system two including check valve and valve.

If one writes the $\mathrm{C}^{-}$equation for the first reach of the discharge pipe,

$$
\begin{equation*}
H_{P 1}=H_{2}+B\left[Q_{P 1}-Q_{2}\right]+R \cdot Q_{2} \cdot\left|Q_{2}\right| \tag{4-45}
\end{equation*}
$$

Since there is no change in liquid storage between A and B, according to continuity equation, $\mathrm{Q}_{1}=\mathrm{Q}_{2}$. In Equation (4-43), H is obtained from Equation (4-16).

Head losses terms in Equation (4-43) may be written as

Valve head losses $=\mathrm{h}_{\mathrm{L}}=\frac{D H 01 \cdot v \cdot|v|}{\tau_{1}{ }^{2}}+\frac{C V \cdot v \cdot|v|}{\tau_{2}{ }^{2}} ; \quad$ where
$D H 01=\frac{A K V 1 \cdot Q_{P 1}{ }^{2}}{2 g \cdot A^{2}} ; C V=\frac{C K \cdot Q_{P 1}{ }^{2}}{2 g \cdot A^{2}}$

In Equation (4-47), AKV1 and CK correspond to local minor loss coefficients of valve and check valve respectively. $\tau_{1}$ and $\tau_{2}$ are the opening of valve and that of check valve.

If one writes
$Q_{P 1}=v Q_{R}$
and by substitution of Equations (4-44), (4-45), (4-46), (4-47) and (4-48) into Equation (4-43),

$$
F_{H}=-C M-B M \cdot Q_{R} \cdot v+H_{R}\left(\alpha^{2}+v^{2}\left[A_{0}+A_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]-\frac{D H 01 \cdot v \cdot|v|-C V \cdot v \cdot|v|}{\tau^{2}}=0\right.
$$

There are two unknowns $\alpha$ and v only. This equation is to be solved together with the speed change equation (Wylie \& Streeter, 1978).

### 4.4.2 Torque angular deceleration equation (calculation of speed change equation) for the second system

Our equations specified in Section 4.3.1 for first system which are Equations (4-$23),(4-24),(4-25),(4-26),(4-27),(4-28)$ and (4-29) are exactly valid for this section, i.e. for second system.

### 4.4.3 Newton Raphson Method for the second system

$\mathrm{F}_{\mathrm{H}}, \mathrm{F}_{\mathrm{Hv}}, \mathrm{F}_{\mathrm{H} \alpha}, \mathrm{F}_{\mathrm{T}}, \mathrm{F}_{\mathrm{Tv}}, \mathrm{F}_{\mathrm{T} \alpha}$ are evaluated for instantaneous v and $\alpha$. The derivatives are

$$
\begin{equation*}
F_{H v}=T A^{2}\left(H_{R}\left\{2 v\left[A_{0}+A_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]+A_{1} \alpha\right\}-B M \cdot Q_{R}\right)-2 D H 01 \cdot|v|-2 C V \cdot|v| \tag{4-50}
\end{equation*}
$$

$F_{H_{\alpha}}=T A^{2}\left(H_{R}\left\{2 \alpha\left[A_{0}+A_{1}\left(\alpha+\tan ^{-1} \frac{v}{\alpha}\right)\right]-v A_{1}\right\}\right)$
$F_{T_{v}}=2 v\left[B_{0}+B_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]+\alpha B_{1}$
$F_{T \alpha}=2 \alpha\left[B_{0}+B_{1}\left(\pi+\tan ^{-1} \frac{v}{\alpha}\right)\right]-v B_{1}+C_{\text {TORQ }}$

In Equations (4-50) and (4-51), TA is the value of TAU at current $\Delta \mathrm{t}$. TAU corresponds to valve opening. Equation (4-32) may be solved for $\Delta \mathrm{v}, \Delta \alpha$.

$$
\begin{equation*}
\Delta \alpha=\frac{\frac{F_{T}}{F_{T v}}-\frac{F_{H}}{F_{H v}}}{\frac{F_{H \alpha}}{F_{H v}}-\frac{F_{T \alpha}}{F_{T v}}} \tag{4-54}
\end{equation*}
$$

$\Delta v=-\frac{F_{H}}{F_{H v}}-\Delta \alpha \frac{F_{H_{\alpha}}}{F_{H v}}$

Then the new $\alpha$ and v values are evaluated.

### 4.5 Use of the Newton Raphson numerical method for determination of steady state parameters



Figure 4.10: Determination of operation point.


Figure 4-11: Reservoir-pump-pipeline system

Referring to Figure 4.10 and 4.11 , one may write the energy equation between suction reservoir A and downstream reservoir B.

$$
\begin{equation*}
z_{A}+\frac{P_{A}}{\gamma}+\frac{V_{A}{ }^{2}}{2 g}+H_{P}=z_{B}+\frac{P_{B}}{\gamma}+\frac{V_{B}{ }^{2}}{2 g}+h_{F}+h_{L} \tag{4-55}
\end{equation*}
$$

In which, $\mathrm{z}_{\mathrm{A}}$ and $\mathrm{z}_{\mathrm{B}}$ are the elevations at point A (upstream reservoir water surface) and point $B$ (downstream reservoir water surface), respectively. $\mathrm{P}_{\mathrm{A}} / \gamma$ and $\mathrm{P}_{\mathrm{B}} / \gamma$ terms are the pressure heads at point $A$ and point $B$, and equal to zero. $V_{A}{ }^{2} / 2 g$ and $V_{B}{ }^{2} / 2 g$ are the velocity heads and may be assumed to be equal to zero. $H_{P}$ is the head in meters added by pump. $h_{F}$ and $h_{L}$ represent the major head loss due to friction (Darcy-Weisbach formula) and minor head loss due to local elements, respectively. These losses may be written as follows,

$$
\begin{equation*}
h_{f}+h_{l}=\left(\sum \frac{f \cdot L}{D}+\sum K\right) \frac{V^{2}}{2 g}=\left(\sum \frac{f \cdot L}{2 g D A^{2}}+\sum \frac{K}{2 g A^{2}}\right) Q^{2}=C \cdot Q^{2} \tag{4-56}
\end{equation*}
$$

In which $f$ is Darcy Weisbach friction factor, $L$ is length of pipe $(m)$ and $K$ is the local head loss coefficient.

If $\mathrm{z}_{\mathrm{B}}-\mathrm{z}_{\mathrm{A}}=\mathrm{EL}$

The pipe characteristic may be expressed as
$H_{P}=E L+C \cdot Q^{2}$

One can write also

$$
\begin{equation*}
H_{P}=h \cdot H_{R}=H_{R}\left(\alpha^{2}+v^{2}\right)\left(A_{0}+A_{1} x\right) \tag{4-58}
\end{equation*}
$$

The operation point is found by intersecting the pump characteristic curve and the pipe characteristic curve as shown in Figure 4.10. For steady state $\alpha=1$, after combination of the relevant equations, one obtains

$$
\begin{equation*}
F_{H}=-E L-C Q^{2}+H_{R}\left(1+v^{2}\right)\left(A_{0}+A_{1} \cdot x\right) \tag{4-59}
\end{equation*}
$$

which is the equation for solving v corresponding to steady state conditions by means of Newton Raphson method.

### 4.6 Presence of disc type check valve in second system

The second system includes disc type check valve at node 1 . If a check valve is used at the pump discharge, the crude assumption may be made that the head loss is constant for all forward flow. For a more accurate treatment, the valve head loss may be expressed as a function of discharge found from valve tests.

To find the criteria for backward flow through the pump and check valve, we set $\mathrm{v}=0$ in Equation (4-49). There is no backward flow due to disc type check valve, after v has been decelerated from steady state value to zero.

## CHAPTER FIVE

## EXPERIMENTAL SET-UP

Experiments are carried out on two different systems.

### 5.1 The first experimental set up

In the first system, the pipeline length is 28 meters. Pipe diameter is constant along the pipeline and it is 0.125 meter. Roughness height, $\mathrm{k}_{\mathrm{s}}$ is taken as 0.00015 meter obtained from the table for steel pipe. The experimental set up is sketched in Figure 5.1.


Figure 5.1: Sketch of the first system.

There exist 9 elbows, 1 reduced bore ball valve and 1 tee in the first system.

There is main water tank (reservoir A) immediately at upstream of the pump. Main water tank is placed at upstream of the pump both in first and second systems, and constitutes a source to both systems and constructed in reinforced concrete.

V notched triangular weir is placed at the downstream end point of the systems. The discharge is measured by this V notched triangular weir of 90 degrees.

Triangular weir is calibrated before starting to experiments. It also serves as downstream reservoir B.

The sketch of the V notched triangular weir is given in Figure 5.2


Figure 5.2: The profile of V notched triangular weir of 90 degrees.

In the first pipeline system, pressure measuring point is placed immediately at 0.50 meter after the pump. The end of the measuring device (transducer) is assembled by screwing to this measuring point. The suction pipe is 0.60 meter long.

In Figure 5.3, the location of measuring point in the first system is shown.
(Reservoir A)


Figure 5.3: The location of measuring point in first system.

After spilling to triangular weir placed at downstream end, water is carried again to main upstream water tank by means of a pipe from downstream tank. Water circulates in the close circuit.

The distances of the pipeline segments from the pump and their elevations are given in Table 5.1. The plan view of the first system is given in Figure 5.4.

The plan view of main upstream water tank (reservoir A) and pump room have also been sketched in Figure 5.4. The dimensions of the main water tank and those of the pump room are $4.50 \mathrm{~m} * 4.50 \mathrm{~m}$ and $2.5 \mathrm{~m} * 2.5 \mathrm{~m}$ respectively. Water level inside the tank may be seen by the opening of the lid. The reinforced concrete walls of the pump room and the main water tank are 0.30 m thick. In Figure 5.4, 'e' letter represents the elbow. The direction of flowing water is shown on the Figure.

Table 5.1: Distances and elevations from the pump in the first system.

| Points | Distance from the pump <br> as origin (meters) | Elevation according to <br> datum (pump axis), m |
| :--- | :--- | :--- |
| A | 1.25 | 1.25 |
| B | 2.50 | 1.25 |
| C | 7.00 | 1.44 |
| D | 8.90 | 1.46 |
| E | 18.90 | 1.77 |
| F | 19.53 | 2.43 |
| G | 20.23 | 2.45 |
| H | 20.73 | 2.99 |
| I | 26.94 | 2.99 |
| J | 28.04 | 4.20 |

The characteristics of the pump used in each system are identical. Measuring point is placed immediately after the pump.


Figure 5.4: Plan view of the first system and the pump room

### 5.2 Second experimental set-up

In the second system, the length of the pipeline is 108 meters. The elevation difference between pump and downstream end point of the pipeline is 5.3 meters (Figure 5.5). Pipe diameter is 0.125 meter and remains constant along the pipeline reach. There are disc type check valve and reduced bore ball valve immediately at upstream of pump. Roughness height, $\mathrm{k}_{\mathrm{s}}$ is taken as 0.00015 m for steel pipe same as in first system. The characteristics of the pump used in each system are identical. The second system contains $1490^{\circ}$ elbows, 1 disc type check valve, 2 reduced bore ball valves and 19 junction points.


Figure 5.5: Sketch of the second pipeline system.

The location of first measuring point in second system is shown in Figure 5.6. The distances of the pipeline segment from the pump and their elevations are given in Table 5.2. The plan view of the second system is given in Figure 5.7.


Figure 5.6: The location of first measuring point in second system.

Table 5.2: Distances and elevations from the pump for the second system.

| Points | Distance from the pump as origin (meters) | Elevation according to datum (pump axis), $m$ |
| :---: | :---: | :---: |
| $\mathrm{A}^{\text {I }}$ | 1.250 | 1.250 |
| A | 2.750 | 1.250 |
| B | 7.700 | 1.410 |
| C | 9.600 | 1.240 |
| D | 18.800 | 1.510 |
| E | 19.800 | 2.590 |
| F | 20.950 | 2.620 |
| G | 27.950 | 2.540 |
| H | 34.150 | 2.560 |
| I | 40.250 | 2.550 |
| J | 46.350 | 2.520 |
| K | 52.350 | 2.540 |
| L | 58.350 | 2.480 |
| M | 58.750 | 2.480 |
| N | 64.550 | 2.610 |
| O | 64.725 | 2.770 |
| $\mathrm{N}^{1}$ | 64.900 | 2.930 |
| $\mathrm{N}^{\text {II }}$ | 66.420 | 2.900 |
| P | 70.800 | 2.800 |
| R | 71.600 | 2.800 |
| S | 77.550 | 2.800 |
| $\mathrm{S}^{1}$ | 83.450 | 2.780 |
| T | 89.400 | 2.810 |
| U | 95.300 | 2.800 |
| $\mathrm{U}^{1}$ | 101.300 | 2.810 |
| V | 104.800 | 2.810 |
| Y | 105.200 | 2.810 |
| Z | 107.300 | 5.310 |



Figure 5.7: Plan view of the second system and the pump room.

Losses due to junction points are not considered in the transient flow calculations. Loss coefficients for elbows and reduced bore ball valve are included according to measurements carried out in the laboratory.

Disc type check valve is placed immediately at upstream of the pump. During the run down period of the pump, due to flow, check valve is closed. Due to closed check valve, backward flows through the pump cannot occur.

The tank at the downstream end of the second pipeline is manufactured from sheet iron. Two pipes are connected to the elevated tank at downstream end to establish the inflow and outflow as shown in Figure 5.8.


Figure 5.8: The profile of downstream water tank for second system (drawing is not scaled).

Water is collected in the tank and spills to outlet pipe. The tank is square with dimensions $1 \mathrm{~m} * 1 \mathrm{~m}$ width and length each. Sheet iron has a thickness of 4 mm .30 $\mathrm{mm} * 30 \mathrm{~mm}$ square box profiles surround the sheet iron all around. In the second system, pressure is measured at two points. The first measurement point is located at
0.90 meter downstream of the pump as in the first system and second measurement point is an interior point located at 66.4 meters apart from the pump.

### 5.3 Pump Characteristic Curves

The geometric characteristics of the centrifugal pump are given in Figure 5.9.


Figure 5.9: The geometries of the pump used in the systems $(\mathrm{ABe}=125 \mathrm{~mm}, \mathrm{ABb}=125$ $\mathrm{mm}, \mathrm{A}=125 \mathrm{~mm}, \mathrm{~F}=470 \mathrm{~mm}, \mathrm{LC}=1052 \mathrm{~mm}, \mathrm{~N}=385 \mathrm{~mm}, \mathrm{Ls}=930 \mathrm{~mm}, \mathrm{Na}=415 \mathrm{~mm}$, $\mathrm{La}=650 \mathrm{~mm}, \mathrm{~s}=20 \mathrm{~mm}, \mathrm{a}=100 \mathrm{~mm}, \mathrm{~b}=90 \mathrm{~mm}, \mathrm{~g}=100 \mathrm{~mm}, \mathrm{~h}=300 \mathrm{~mm}, \mathrm{H}=580 \mathrm{~mm}$, $\mathrm{h} 2=280 \mathrm{~mm}$, weight $=184 \mathrm{~kg}$ ).

In Table 5.3, the numerical values read from the pump characteristic curves are presented. At the best efficiency point ( $\eta=79.3 \%$ ) the rated head and discharge values are 11 m and $36 \mathrm{lt} / \mathrm{s}$ respectively. In Figure 5.10, pump characteristic curves are given. In Table 5.4, torque and power values generated from pump characteristic curve values are presented.

The characteristics of the pump :
Trademark of the pumps: Standard (Made in Turkey), Type of pumps: SNK 100-200,
Rotational speed of the pump: 1450 rpm ,
Suction orifice diameter: 125 mm ,
Discharge orifice diameter: 125 mm ,
Diameter of the impeller: 210 mm ,
Power of motor: 5.5 KW:7.5 HP
Weight of the pump: 80 kg ,
Discharge, Q (at best efficiency point): $130 \mathrm{~m}^{3} / \mathrm{h}$,
Head, H (at best efficiency point): 11 m ,
Efficiency of the pump: 79.30 \%,
Effective power: 4.91 KW ,
Maximum effective power: 5.43 KW,

Table 5.3: Pump characteristic values ( $1 \mathrm{KW}=1.338 \mathrm{HP}$ ).

| Head, <br> $(\mathrm{meters})$ | Discharge, <br> $\left(\mathrm{m}^{3} / \mathrm{h}\right)$ | Discharge, <br> $\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Efficiency, <br> $(\%)$ | Power, <br> $(\mathrm{KW})$ | Power, <br> $(\mathrm{HP})$ |
| :--- | :--- | :--- | :--- | :--- | :--- |
| 14.00 | 0.00 | 0.00 | 0.00 | 3.100 | 4.140 |
| 13.50 | 80.00 | 0.022 | 68.50 | 4.253 | 5.690 |
| 13.10 | 90.00 | 0.025 | 72.50 | 4.431 | 5.928 |
| 12.70 | 100.00 | 0.027 | 76.00 | 4.550 | 6.087 |
| 12.20 | 110.00 | 0.030 | 77.00 | 4.740 | 6.342 |
| 11.70 | 120.00 | 0.033 | 79.00 | 4.794 | 6.414 |
| 11.00 | 130.00 | 0.036 | 79.30 | 4.886 | 6.537 |
| 10.30 | 140.00 | 0.038 | 78.00 | 4.922 | 6.585 |
| 9.50 | 150.00 | 0.041 | 74.50 | 5.128 | 6.861 |
| 8.60 | 160.00 | 0.044 | 70.00 | 5.303 | 7.095 |



Figure 5.10: Pump characteristic curves.

Table 5.4: Torque, T values corresponding to H and Q values in pump characteristic curve.

| $\mathrm{P},($ watt $)$ | Head (meters) | Disc., $\mathrm{Q}\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Torque, (Nt.m) |
| :--- | :--- | :--- | :--- |
| 3100.000 | 14.000 | 0.000 | 20.415 |
| 4253.000 | 13.500 | 0.022 | 28.009 |
| 4431.000 | 13.100 | 0.025 | 29.181 |
| 4550.000 | 12.700 | 0.027 | 29.965 |
| 4740.000 | 12.200 | 0.030 | 31.216 |
| 4794.000 | 11.700 | 0.033 | 31.571 |
| 4886.000 | 11.000 | 0.036 | 32.177 |
| 4922.000 | 10.300 | 0.038 | 32.414 |
| 5128.000 | 9.500 | 0.041 | 33.771 |
| 5303.000 | 8.600 | 0.044 | 34.924 |

Some photographs corresponding to pump room are given from Figure 5.11 to Figure 5.15.


Figure 5.11: Pump room at the left and water tank at the right.


Figure 5.12: pump room at the forward, water tank at the backward and man hole of the tank.


Figure 5.13: Inside of the pump room. Used pump for system 1 at the right and used pump for system 2 at the left. These pump are same. The view of first measuring points, valve and check valve.


Figure 5.14: Discharge adjustment valve, measuring node 1 for system 2 , suction flange and water tank.


Figure 5.15: To observe the water surface in the tank, the piezometric tube connected to suction flange and datum.

Figure 5.16 is about the triangular weir.


Figure 5.16: 90 degrees triangular V notched triangular weir to measure the steady state discharge for the first system.

Some views taken for first experimental setup are given from Figure 5.17 to 5.19.


Figure 5.17: The downstream part of the pipe for system 1, tee, discharge adjustment valve and triangular weir.


Figure 5.18: Gauge manometer to measure the pressure heads after valve and upstream connection point to differential glass manometer to find loss coefficient of discharge valve.


Figure 5.19: The view of tee, discharge regulation valve, upstream and downstream points to find local head loss coefficient of valve and gauge manometer for system 1.

The photographs of differential manometer are given in Figures 5.20 and 5.21.


Figure 5.20: Manufactured glass differential manometer using mercury as fluid.


Figure 5.21: Two connection points of glass differential manometer to the upstream and downstream points of the discharge adjustment valve to find local had loss coefficient of the valve.

The photographs of second system are presented from Figure 5.22 to 5.26 .


Figure 5.22: The view from the middle part of second pipeline and measuring point.


Figure 5.23: The view of second pipeline system.


Figure 5.24: The discharge adjustment valve, the end point of the pipe, the downstream water tank and 90 degrees triangular V notched weir for system 2.


Figure 5.25: The view of junctions in second system (welded and screwed).


Figure 5.26: 90 degrees triangular V notched weir to measure the steady state discharge in system 2, same as used in the system 1.

### 5.4 Pressure transient data logger

There are certain numbers of technological equipments to measure transient pressures in the pipelines. Pressure transient data logger equipment has been provided from Radcom Ltd, USA, is used in this study (Figure 5.27).


Figure 5.27: Pressure transient data logger, Radcom Ltd.

Pressures have been measured in meters (pressure heads) with the aid of this equipment which is connected upon the pipeline directly. Its clock is adjusted to local time to measure transient pressures at known specific time. After the clock of the equipment has been reached to known specific time, equipment is started to operate automatically and recorded the transient pressures. On the other hand equipment can start to measure using one button click. Equipment records the transient pressures data into the logger inside the cover. After the measurements have been completed, equipment is connected to computer and data recorded inside the logger are transmitted to computer as tabulated data and graphics family. Hence, recorded data can be seen upon the computer screen. This equipment should be calibrated before starting the measuring process. Calibration process has been realised before to start the measurements. Logger presents the results in meters vs. time (seconds). Measurements can be performed up to maximum 0.20 meter
sensitivity. Logger equipment is operated with Lithium batteries of 5 years service life. Data values stored into the logger up to 2 million readings.

Experiment begins by starting up the pumps. Discharges are adjusted with the help of valves in order to obtain desired discharges. After the adjustment of system discharge and the establishment of steady state, pressure transient kit begins to record. Pump electricity is shut off manually. After measurements, the equipment is taken out from the pipeline, and then, it is connected to computer and recorded data are stored in the computer, thus new experimental values and new graph are saved.

### 5.5 Computation of valve and elbow loss coefficients

### 5.5.1 The use of the differential manometer

Valve and elbow loss coefficients are measured in the hydraulic laboratory. Reduced bore ball valve loss coefficients for various discharge values are measured using differential manometer and gauge manometer. Differential manometer was manufactured according to needed dimensions. The dimensions and simple sketch of the differential manometer is shown in Figure 5.28.

If differential manometer is connected to points $A$ and $B$ on the pipe

$$
\begin{equation*}
P_{1}=P_{2} \tag{5-1}
\end{equation*}
$$

$P_{A}+\gamma_{w} \cdot z_{1}=P_{B}+\gamma_{w} \cdot z_{2}+\gamma_{M} \cdot \Delta z$
$P_{A}-P_{B}=\left(\gamma_{M}-\gamma_{W}\right)\left(z_{1}-z_{2}\right)=\left(\gamma_{M}-\gamma_{w}\right) \cdot \Delta z$

In Equation (5-3), $\mathrm{P}_{\mathrm{A}}-\mathrm{P}_{\mathrm{B}}$ is the pressure head difference, $\gamma_{\mathrm{M}}$ is the unit weight of mercury inside the manometer, $\gamma_{\mathrm{w}}$ is the unit weight of water and $\mathrm{z}_{1}-\mathrm{z}_{2}$ is the difference between the mercury columns.


Figure 5.28: Sketch of the manufactured glass differential manometer.

Mercury is placed inside the manometer. Then manometer columns are connected to points on the pipeline, these points are placed at upstream and downstream of the valve. During the measurement, valve opening position does not change. Deviation of mercury columns is read. Then pressure difference is calculated. If the pipe is horizontal the local loss is expressed as :

$$
\begin{equation*}
h_{L}=\frac{\left(P_{A}-P_{B}\right)}{\gamma_{W}}=\frac{\left(\gamma_{M}-\gamma_{W}\right) \cdot \Delta z}{\gamma_{W}} \tag{5-4}
\end{equation*}
$$

### 5.5.2 The loss coefficients of reduced bore ball valve

During the pump shutdown, the discharge (and consequently the velocity) decreases continuously. As the head loss coefficients vary in terms of flow velocity, these coefficients should be determined as the function of the velocity. On the other hand, the steady state discharge is adjusted by changing the opening of the regulating
valve. The measurements are performed for different valves used in two different experimental systems.

Several measurements are performed with different valve openings in the first system. Head loss coefficients calculated from the measurements are given in Table 5.5 realised on the first system. The graphs and analytical expressions obtained from the curve fitting analysis are given in Figure 5.29

Table 5.5: Discharges, Q, velocities, V and valve loss coefficients, K (variable valve opening for the first system).

| Discharge, $\mathrm{Q}\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Velocity, $\mathrm{V}(\mathrm{m} / \mathrm{s})$ | Valve loss coefficient, K <br> (dimensionless) |
| :--- | :--- | :--- |
| 0.041 | 3.36 | 5.26 |
| 0.040 | 3.28 | 5.91 |
| 0.030 | 2.46 | 21.82 |
| 0.022 | 1.80 | 54.04 |
| 0.008 | 0.65 | 428.53 |



Figure 5.29: The valve loss coefficients, K versus flow velocity, V (variable valve opening).

The obtained analytical expression is : $\mathrm{K}=1125.6 . \mathrm{e}^{(-1.607 . \mathrm{V})}$.

### 5.5.3 The loss coefficients of elbows

The results of experiments to determine the $90^{\circ}$ elbow loss coefficients are given in Table 5.6 and the graphs and analytical expressions obtained from the curve fitting analysis are given in Figure 5.30.

Table 5.6: Discharges, Q, velocities, V and elbow loss coefficients, K

| Discharge, $\mathrm{Q}\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Velocity, $\mathrm{V}(\mathrm{m} / \mathrm{s})$ | Valve loss coefficient, K <br> (dimensionless) |
| :--- | :--- | :--- |
| 0.024 | 1.97 | 1.76 |
| 0.019 | 1.61 | 1.90 |
| 0.012 | 1.02 | 4.67 |
| 0.007 | 0.56 | 15.38 |



Figure 5.30: The elbow loss coefficients, K versus flow velocity, V .

The obtained analytical expression is : $\mathrm{Ke}=-9.3404 . \mathrm{V}^{3}+47.498 . \mathrm{V}^{2}-80.348 . \mathrm{V}+47.12$.

### 5.6 The loss coefficients of the disc type check valve and valves for the second system.

The check valve loss coefficient is determined with the use of the pump characteristic curve. As there is reading after the valve and since the valve loss coefficient is known, the check valve loss coefficient may be deduced by subtraction from the relevant manometric head read from the pump characteristic curve.

During these measurements the full open valve loss coefficient (AKV1) and partially open valve loss coefficient (AKV2) are determined as well.

The results are collected in the Table 5.7. The graphs and analytical expressions obtained from the curve fitting analysis are given in Figure 5.31, Figure 5.32, Figure 5.33.

Table 5.7: Experimental values of local head loss coefficients.

| Steady state <br> discharge, $\mathrm{Q}(\mathrm{lt} / \mathrm{s})$ | Velocity, V, loss <br> $(\mathrm{m} / \mathrm{s})$ | CK, loss <br> coefficient of <br> Check valve | AKV2, <br> coefficient of <br> valve, node 1 | (losf <br> coefficient of <br> valve, node 42 |
| :--- | :--- | :--- | :--- | :--- |
| 6.60 | 0.54 | 10.87 | 17.29 | 585.92 |
| 9.01 | 0.74 | 9.35 | 11.32 | 303.40 |
| 16.50 | 1.35 | 6.62 | 4.98 | 74.00 |
| 19.14 | 1.57 | 6.23 | 4.06 | 44.02 |
| 20.55 | 1.70 | 6.12 | 3.69 | 29.31 |
| 24.52 | 2.01 | 2.00 | 2.90 | 15.62 |
| 25.16 | 2.06 | 1.00 | 2.80 | 14.23 |



Figure 5.31: The check valve loss coefficients, K versus flow velocity, V.

The obtained analytical expression is : $\mathrm{CK}=-3.1404 . \mathrm{V}^{2}+2.3912 . \mathrm{V}+9.9341$.


Figure 5.32: The full open valve loss coefficients, AKV1 versus flow velocity, V.

The obtained analytical expression is : $\mathrm{AKV} 1=7.5039 . \mathrm{V}^{-1.3581}$.


Figure 5.33: The partially open valve loss coefficients, AKV2 versus flow velocity, V.

The obtained analytical expression is : AKV2=1970. $\mathrm{e}^{-2.4214 . \mathrm{V}}$.

### 5.7 Calibration of triangular weir

V notched triangular weirs of 90 degrees are manufactured in hydraulic laboratory. Various discharge conditions are obtained adjusting the reduced bore ball valve. Scaled water tank is used in order to measure the volume of water passing through the triangular weir in the specified time interval. Thus discharges through the weir are determined. During calibration, weir head on the V notched is measured, therefore, different discharge values corresponding to weir heads on the V notch are drawn and curve fitting is performed.

In Figure 5.34, the calibration curves of our V notched weir of 90 degrees are sketched. The expression of the calibration curve is $\mathrm{Q}=1.53$. $(\mathrm{H})^{5 / 2}$. In this equation, Q is the discharge in $\mathrm{m}^{3} / \mathrm{s}$, H is the measured head on V notched in meters.


Figure 5.34: Calibration curves of V notched triangular weir of 90 degrees used in our systems. x axis represents the head in meters and y axis represents discharges in $\mathrm{m}^{3} / \mathrm{s}$.

The geometric characteristics of reduced bore ball valve and those of the disc type check valve, together with head loss coefficients provided from the literature are given in Appendix A.

## CHAPTER SIX <br> DESCRIPTION OF COMPUTER PROGRAMS

Fortran computer programs proposed by Wylie and Streeter are used after some modifications and adaptation. Using these modified versions of Fortran computer programs, several computations are realised and results are interpreted.

### 6.1 The flow chart of the computer program



Figure 6.1 : Flow Diagram of Fortran computer program.


Figure 6.1 : Flow diagram of Fortran computer program (continued).

### 6.2 Description of parameters used in the computer programs

### 6.2.1 Fortran computer program used for the first system

The parameters used in the computer program adopted for the first system are described below :

EL: The elevation difference between the upstream water surface elevation and downstream water surface elevation ( $=\Delta \mathrm{z}$ in the equations).

A : Wave celerity in $\mathrm{m} / \mathrm{s}$ ( $=\mathrm{a}$ in Equation ).
XL: Pipeline length in $m .(=\mathrm{L})$
CK: The dimensionless valve head loss coefficient.
TM: The maximum time (seconds) in which the Fortran calculations perform.
RN : Rotational speed at rated condition in $\mathrm{rpm}\left(=\mathrm{N}_{\mathrm{R}}\right)$
TR : Rated torque in Nt.m ( $=\mathrm{T}_{\mathrm{R}}$ )
HR : Rated head in meter $\left(=\mathrm{H}_{\mathrm{R}}\right)$
QR : Rated discharge in $\mathrm{m}^{3} / \mathrm{s}\left(=\mathrm{Q}_{\mathrm{R}}\right)$
WRR : Rotational moment of inertia in Nt.m ${ }^{2}\left(=\mathrm{W} . \mathrm{R}_{\mathrm{g}}{ }^{2}\right)$
TOL : Tolerance into Newton Raphson method to provide equality
N : Number of the equal reaches in the pipeline which is divided into N equal reaches.

F: Darcy Weisbach friction coefficient ( = f)
D : Diameter of the pipe, m
G : Acceleration due to gravity ( $=\mathrm{g}$ )
V : The ratio of the steady state discharge to discharge at rated condition.
VI : The ratio of the steady state discharge to discharge at rated condition.
V, V0, V00 : Dimensionless the ratio of steady state discharge to discharge at rated condition. $\mathrm{Q} / \mathrm{Q}_{\mathrm{R}}$. V0 corresponds to $\Delta \mathrm{t}$ time step earlier. V00 corresponds to $2 \Delta \mathrm{t}$ time step earlier.

PI : Number of $\pi(=3.1416)$
JPR : Integer number of time step iterations between each print
KIT : Maximum iteration number in Newton Raphson numerical procedure
DX : Dimensionless pump data space (Radians)
AL : Dimensionless pump rotational speed ratio.
AL0 : Dimensionless pump rotational speed ratio for $\Delta t$ time step before
AL00 : Dimensionless pump rotational speed ratio for $2 \Delta \mathrm{t}$ time step before
DT : Time increment in the successive time step (s)
K : Counter designates maximum calculation time period number
WH (I) : Dimensionless pump (head) data
WB(I) : Dimensionless pump (torque) data

ZH(I) : Elevations of nodes throughout the pipe according to datum (upstream reservoir level) in $m$

AKS : Roughness height of steel pipe wall in $m$
QQA : Steady state discharges in $\mathrm{m}^{3} / \mathrm{s}$
$\mathrm{H}(\mathrm{I})$ : Piezometric heads in m
HPH(I) : Pressure heads in m
NSS : Number of node in which there is a discharge adjustment valve
VN : Kinematic viscosity of water in $\mathrm{m}^{2} / \mathrm{s}$
HMIN : Minimum pressure head in $m$
$\mathrm{Q}(\mathrm{I})$ : Discharge in transient condition, $\mathrm{m}^{3} / \mathrm{s}$
$\mathrm{H}(\mathrm{I})$ : Piezometric head in transient condition, m
HP11: Pressure head in meters at node of valve immediately upstream of valve, m
HP1 : Piezometric head in meters at node of valve immediately upstream of valve, m
CP : The constant in $\mathrm{C}^{+}$equation given in Equation (3-20)
CM : The constant in $\mathrm{C}^{-}$equation given in Equation (3-21)
C3 : Local head loss equation constant for valve given in Equation (3-40)
C4: Constant in discharge equation in transient condition given in Equation (3-38)
C5 : Constant in discharge equation in transient condition given in Equation (3-39)
BP : The constant in $\mathrm{C}^{+}$equation
BM : The constant in $\mathrm{C}^{-}$equation
$\mathrm{B}:$ Pipeline impedance, $\mathrm{s} / \mathrm{m}^{2}=\mathrm{a} /(\mathrm{g} . \mathrm{AR})$
R : Resistance coefficient for one reach $=\left(\mathrm{f} . \Delta \mathrm{x} /\left(2 \mathrm{~g} . \mathrm{D} \cdot \mathrm{AR}^{2}\right)\right)$
H3 : Local head loss at valve for steady state
X : Angle theta
A1 : The constant in linear similarity head equation given in Equation (4-18).
A0 : The constant in linear similarity head equation given in Equation (4-18)
B1 : The constant in linear similarity torque equation given in Equation (4-28)
B0 : The constant in linear similarity torque equation given in Equation (4-28)
DAL : Changing of $\alpha$ value given in Equation (4-38)
DV : Changing of v value given in Equation (4-39)
F1 : Head Balance equation given in Equation (4-22)

F1V : Partial derivative of head balance equation with respect to V given in Equation (4-34).
F1A : Partial derivative of head balance equation with respect to $\alpha$ given in Equation (4-35)

F2 : Speed change equation given in Equation (4-29).
F2V : Partial derivative of speed change equation with respect to V given in Equation (4-36)
F2A : Partial derivative of speed change equation with respect to $\alpha$ given in Equation (4-37)

### 6.2.2 Fortran computer program used for the second system

The terms described in section 6.2.1 are generally valid for the first system. In this section, we will mention only the additional or modified terms.

CK : Disc type check valve head loss coefficient
AKV1 : Head loss coefficient of valve immediately after pump
AKV2 : Head loss coefficient of valve at the end of the pipe
TAU : Check valve opening
DTAU : Time increment between Tau values (s)
NTAU : Number of Tau values.
NSV : Node number of valve at the end of the pipe
CCT : Check valve closing time (s)
M2 : Node number of second measuring point

## CHAPTER SEVEN

## EXPERIMENTAL RESULTS

### 7.1 Results of experiments performed in the first system

Several experiments are realised to study unsteady flows due to the pump failure in first system. But three of these experiments are presented in the following pages corresponding to lower steady state discharge, moderate steady state discharge and upper steady state discharge.

### 7.1.1 Results of experiment performed for lower steady state discharge

The results correspond to a steady state discharge of $6.2 \mathrm{lt} / \mathrm{s}(0.5 \mathrm{~m} / \mathrm{s})$ are given in Figure 7.1, Table 7.1 and Figure 7.2. The experimental curve obtained for a steady flow discharge of $8.16 \mathrm{lt} / \mathrm{s}(0.67 \mathrm{~m} / \mathrm{s})$ is presented in Figure 7.3.


Figure 7.1: Pressure heads correspond to $6.2 \mathrm{lt} / \mathrm{s}$ lower steady state discharge.

Table 7.1: Measured pressure heads corresponding to $6.2 \mathrm{lt} / \mathrm{s}$ lower steady state discharge.
Time (minute-second-milli second ) Measured pressure heads (m)

| $04: 23: 50$ | 13.80 |
| :--- | :--- |
| $04: 23: 55$ | 14.50 |
| $04: 24: 00$ | 15.20 |
| $04: 24: 05$ | 15.20 |
| $04: 24: 10$ | 15.20 |
| $04: 24: 15$ | 14.70 |
| $04: 24: 20$ | 15.20 |
| $04: 24: 25$ | 15.40 |
| $04: 24: 30$ | 15.20 |
| $04: 24: 35$ | 14.50 |
| $04: 24: 40$ | 14.90 |
| $04: 24: 45$ | 13.60 |
| $04: 24: 50$ | 12.00 |
| $04: 24: 55$ | 9.80 |
| $04: 25: 00$ | 8.20 |
| $04: 25: 05$ | 8.00 |
| $04: 25: 10$ | 6.60 |
| $04: 25: 15$ | 6.10 |
| $04: 25: 20$ | 5.50 |
| $04: 25: 25$ | 4.60 |
| $04: 25: 30$ | 4.60 |
| $04: 25: 35$ | 3.70 |
| $04: 25: 40$ | 3.70 |
| $04: 25: 45$ | 3.40 |
| $04: 25: 50$ | 2.80 |
| $04: 25: 55$ | 2.80 |
| $04: 26: 00$ | 2.30 |
| $04: 26: 05$ | 2.30 |
| $04: 26: 10$ | 2.30 |
| $04: 26: 15$ | 2.50 |
| $04: 26: 20$ | 2.10 |
| $04: 26: 25$ | 2.10 |
| $04: 26: 30$ | 1.70 |
| $04: 26: 35$ | 1.70 |
| $04: 26: 40$ | 1.70 |
| $04: 26: 45$ | 2.30 |
| $04: 26: 50$ | 1.70 |
| $04: 26: 55$ | 1.70 |
| $04: 27: 00$ | 1.60 |
| $04: 27: 05$ | 1.60 |
| $04: 27: 10$ | 1.40 |
| $04: 27: 15$ | 1.20 |
| $04: 27: 20$ | 1.20 |
|  |  |



Figure 7.2: Pressure heads correspond to $6.2 \mathrm{lt} / \mathrm{s}$ lower steady state discharge.


Figure 7.3: Pressure heads correspond to $8.16 \mathrm{lt} / \mathrm{s}$ lower steady state discharge.

### 7.1.2 Results of experiment performed for moderate steady state discharge

In these experiments, moderate steady state discharges were measured as 12.31 $\mathrm{lt} / \mathrm{s}(1 \mathrm{~m} / \mathrm{s}), 18.32 \mathrm{lt} / \mathrm{s}(1.5 \mathrm{~m} / \mathrm{s})$ and $22.635 \mathrm{lt} / \mathrm{s}(1.85 \mathrm{~m} / \mathrm{s})$. The graphs are shown in Figures 7.4, 7.5 and 7.6, respectively.


Figure 7.4: Pressure heads correspond to $12.31 \mathrm{lt} / \mathrm{s}$ moderate steady state discharge.

In Figure 7.5, in steady state flow conditions pressure heads are approximately 13.5 meters. Then the pump is shut down manually and transient flow starts. In the graph, x axis represents the time in minutes and y axis represents the pressure heads in meters.

In Figure 7.6, the steady state pressure head is about 12.5 meters which corresponds to steady state discharge of $22.63 \mathrm{lt} / \mathrm{s}$. The tendency of the curves are similar in Figures 7.5 and 7.6.


Figure 7.5: Pressure heads versus time corresponding to $18.32 \mathrm{lt} / \mathrm{s}$ moderate steady state discharge.


Figure 7.6: Run down period correspond to $22.63 \mathrm{lt} / \mathrm{s}$ moderate steady state discharge.

### 7.1.3 Results of experiment performed for upper steady state discharge

In this experiment, upper steady state discharge is measured as $39.00 \mathrm{lt} / \mathrm{s}$ and the related curve is presented in Figure 7.7.


Figure 7.7: Run down period corresponding to $39 \mathrm{lt} / \mathrm{s}$ upper steady state discharge.

### 7.2 Results of experiments performed in the second system

In the similar manner as in Section 7.1, experiments were performed using lower, moderate and upper steady state discharges in the second system.

Although several experiments are realised to study unsteady flows due to the pump failure in second system, only some of them are presented.

### 7.2.1 Results of experiment performed for lower steady state discharge

In these experiments, lower steady state discharges were measured as 6.60 and 9.01 $\mathrm{lt} / \mathrm{s}$. The output measurement results and the graphics are shown below. In Table 7.2, pressure heads measured at the first and second measuring points in second system corresponds to $6.60 \mathrm{lt} / \mathrm{s}$ lower steady state discharge.


Figure 7.8: Pressure heads corresponding to $6.6 \mathrm{lt} / \mathrm{s}$ lower steady state discharge at point 1.

Table 7.2: Measured pressure heads corresponding to $6.6 \mathrm{lt} / \mathrm{s}$ lower steady state discharge for first and second measuring points.

| Time (Minute- | Measured | Time (Minute- | Measured |
| :---: | :---: | :---: | :---: |
| Second- | Pressure Heads, | Second- | Pressure Heads, |
| Milli second) | (meters), point 1 | Milli second) | (meters), point 2 |
| 08:50:15 | 13.20 | 07:54:35 | 11.10 |
| 08:50:20 | 13.40 | 07:54:40 | 10.90 |
| 08:50:25 | 13.40 | 07:54:45 | 10.70 |
| 08:50:30 | 13.40 | 07:54:50 | 10.70 |
| 08:50:35 | 13.80 | 07:54:55 | 11.30 |


| Table 7.2 : (Continued) |  |  |  |
| :---: | :---: | :---: | :---: |
| 08:50:40 | 13.60 | 07:55:00 | 11.10 |
| 08:50:45 | 13.60 | 07:55:05 | 11.30 |
| 08:50:50 | 14.00 | 07:55:10 | 11.10 |
| 08:50:55 | 13.60 | 07:55:15 | 11.10 |
| 08:51:00 | 14.00 | 07:55:20 | 10.70 |
| 08:51:05 | 13.60 | 07:55:25 | 10.90 |
| 08:51:10 | 13.60 | 07:55:30 | 11.30 |
| 08:51:15 | 12.00 | 07:55:35 | 10.70 |
| 08:51:20 | 10.40 | 07:55:40 | 10.00 |
| 08:51:25 | 8.60 | 07:55:45 | 8.90 |
| 08:51:30 | 7.10 | 07:55:50 | 7.90 |
| 08:51:35 | 6.60 | 07:55:55 | 6.80 |
| 08:51:40 | 5.00 | 07:56:00 | 5.70 |
| 08:51:45 | 4.40 | 07:56:05 | 5.20 |
| 08:51:50 | 3.70 | 07:56:10 | 4.30 |
| 08:51:55 | 3.00 | 07:56:15 | 3.70 |
| 08:52:00 | 2.60 | 07:56:20 | 3.20 |
| 08:52:05 | 2.30 | 07:56:25 | 2.80 |
| 08:52:10 | 2.10 | 07:56:30 | 2.30 |
| 08:52:15 | 2.10 | 07:56:35 | 1.90 |
| 08:52:20 | 1.60 | 07:56:40 | 1.70 |
| 08:52:25 | 1.60 | 07:56:45 | 1.40 |
| 08:52:30 | 1.20 | 07:56:50 | 1.40 |
| 08:52:35 | 1.20 | 07:56:55 | 1.20 |
| 08:52:40 | 1.00 | 07:57:00 | 1.00 |
| 08:52:45 | 1.00 | 07:57:05 | 0.70 |
| 08:52:50 | 1.00 | 07:57:10 | 0.70 |
| 08:52:55 | 0.70 | 07:57:15 | 0.50 |
| 08:53:00 | 0.70 | 07:57:20 | 0.50 |
| 08:53:05 | 0.70 | 07:57:25 | 0.50 |
| 08:53:10 | 0.70 | 07:57:30 | 0.30 |
| 08:53:15 | 1.40 | 07:57:35 | 0.50 |
| 08:53:20 | 1.40 | 07:57:40 | 0.50 |
| 08:53:25 | 1.40 | 07:57:45 | 0.80 |
| 08:53:30 | 1.40 | 07:57:50 | 0.80 |
| 08:53:35 | 1.60 | 07:57:55 | 0.80 |
| 08:53:40 | 1.70 | 07:58:00 | 0.70 |
| 08:53:45 | 2.10 | 07:58:05 | 0.70 |
| 08:53:50 | 2.10 | 07:58:10 | 0.70 |
| 08:53:55 | 2.50 | 07:58:15 | 0.80 |
| 08:54:00 | 2.60 | 07:58:20 | 1.00 |
| 08:54:05 | 3.40 | 07:58:25 | 1.20 |
| 08:54:10 | 3.70 | 07:58:30 | 1.60 |
| 08:54:15 | 4.30 | 07:58:35 | 1.70 |
| 08:54:20 | 4.60 | 07:58:40 | 1.90 |
| 08:54:25 | 5.20 | 07:58:45 | 1.90 |
| Table 7.2 : (Continued) |  |  |  |
| 08:54:30 | 5.50 | 07:58:50 | 2.10 |


| $08: 54: 35$ | 6.10 | $07: 58: 55$ | 2.30 |
| :--- | :--- | :--- | :--- |
| $08: 54: 40$ | 6.20 | $07: 59: 00$ | 2.60 |
| $08: 54: 45$ | 6.40 | $07: 59: 05$ | 2.80 |
| $08: 54: 50$ | 6.40 | $07: 59: 10$ | 3.00 |
| $08: 54: 55$ | 6.40 | $07: 59: 15$ | 2.80 |
| $08: 55: 00$ | 6.40 | $07: 59: 25$ | 2.80 |
| $08: 55: 05$ | 5.90 | $07: 59: 30$ | 2.60 |
| $08: 55: 10$ | 5.30 | $07: 59: 35$ | 2.30 |
| $08: 55: 15$ | 5.20 | $07: 59: 40$ | 2.10 |
| $08: 55: 20$ | 4.60 | $07: 59: 45$ | 1.10 |
| $08: 55: 25$ | 4.30 | $07: 59: 50$ | 1.70 |
| $08: 55: 30$ | 4.10 | $07: 59: 55$ | 1.70 |
| $08: 55: 35$ | 3.90 |  |  |
| $08: 55: 40$ | 3.50 |  |  |
| $08: 55: 45$ | 3.40 |  |  |
| $08: 55: 50$ | 3.00 |  |  |
| $08: 55: 55$ | 3.00 |  |  |
| $08: 56: 00$ | 2.80 |  |  |
| $08: 56: 05$ | 2.80 |  |  |
| $08: 56: 10$ | 2.80 |  |  |
| $08: 56: 15$ | 2.80 |  |  |
| $08: 56: 20$ | 3.00 |  |  |
| $08: 56: 25$ | 3.00 |  |  |
| $08: 56: 30$ | 3.40 |  |  |
| $08: 56: 35$ | 3.40 |  |  |
| $08: 56: 40$ | 3.50 |  |  |
| $08: 56: 45$ | 3.70 |  |  |
| $08: 56: 50$ | 3.90 |  |  |
| $08: 56: 55$ | 4.10 |  |  |
| $08: 57: 00$ | 4.30 |  |  |
| $08: 57: 05$ | 4.40 |  |  |
| $08: 57: 10$ | 4.60 |  |  |
| $08: 57: 15$ | 4.80 |  |  |
| $08: 57: 20$ | 5.00 |  |  |
| $08: 57: 25$ | 5.00 |  |  |
| 55 |  |  |  |

In the Figures 7.9 and 7.10, the unsteady state flow pressure signals can be seen for first and second measuring points.

The results concerning the steady state discharge of $9.01 \mathrm{lt} / \mathrm{s}$ are presented in Figure 7.11 and Figure 7.12.


Figure 7.9: Run down period for $6.6 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point.


Figure 7.10: Run down period for $6.6 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.


Figure 7.11: Run down period for $9.01 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point.


Figure 7.12: Run down period for $9.01 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.

### 7.2.2 Results of experiment performed for moderate steady state discharge

In these experiments, moderate steady state discharges were measured as 16.5 , 19.14 and $20.55 \mathrm{lt} / \mathrm{s}$. The output measurement graphics are shown below. In Figures 7.13 and 7.14, pressure heads were measured at the first and second measuring points corresponding to $16.5 \mathrm{lt} / \mathrm{s}$ normal steady state discharge.


Figure 7.13: Run down period for $16.5 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point.

The results relative to a steady state discharge of $19.14 \mathrm{lt} / \mathrm{s}$ are given in Figure 7.15 and Figure 7.16.

Figure 7.17 and Figure 7.18 show the results corresponding to a steady state discharge of $20.55 \mathrm{lt} / \mathrm{s}$.


Figure 7.14: Run down period for $16.5 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.


Figure 7.15: Run down period for $19.14 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point


Figure 7.16: Run down period for $19.14 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.


Figure 7.17: Run down period for $20.55 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point.


Figure 7.18: Run down period for $20.55 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.

### 7.2.3 Results of experiment performed for upper steady state discharge

In these experiments, upper steady state discharges were measured as 24.51 and $25.16 \mathrm{lt} / \mathrm{s}$. The output measurement graphics are shown in Figures 7.19, 7.20 and 7.21, 7.22, respectively.

According to the experimental curves, one can say that the check valve is not closed immediately after the shut down of the pump.

The water hammer pressure waves are registered after the closure of the check valve. The water hammer phenomenon starts with a negative pressure wave followed by positive pressure wave and the subsequence of negative / positive pressure waves. This phenomenon is damped and finished after certain time.


Figure 7.19: Run down period for $24.51 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point.


Figure 7.20: Run down period for $24.51 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.


Figure 7.21: Run down period for $25.16 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point.


Figure 7.22: Run down period for $25.16 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point.

## CHAPTER EIGHT

## THEORETICAL RESULTS

Several computations are realised for the first system. These computations correspond to experimental conditions, namely those performed for lower steady state discharge, moderate steady state discharge and upper steady state discharge.

### 8.1 Results of computations performed for the first system

### 8.1.1 Results of computation performed for lower steady state discharge

The first page of output relative to a discharge of $6.2 \mathrm{lt} / \mathrm{s}$ is given in Figure 8.1. The relevant flow average velocity is $0.50 \mathrm{~m} / \mathrm{s}$. The graph of pressure heads and discharge values at measurement point drawn according to the results of numerical solution are given in Figures 8.2, 8.3, 8.4 and 8.5, respectively.

```
```

EL (M)= 3.7000

```
```

EL (M)= 3.7000
HMIN= .0000
HMIN= .0000
NSS=23

```
```

NSS=23

```
```

A $(M 3 / S E C)=1331.0000$




$\begin{array}{ll}\text { TOL }= & .0002 \\ \text { DX (RAD) }= & V=\quad .1714 \\ \text { CK }= & 860.0000\end{array}$
$\mathrm{AL}=1.0000 \quad \mathrm{AL} 0=1.0000$
$\operatorname{AKS}(M)=.0002 \quad$ DT $(S)=\quad .0008$
$\mathrm{T}=\quad .0000$
$\mathrm{K}=0 \quad$ QQA $(\mathrm{M} 3 / \mathrm{S})=$. 0062

| XL (M) = 28.0000 |  |
| :---: | :---: |
| G (M/S2) = | $)=9.8060$ |
| TR (NT.M) = | M) $=32.1800$ |
| WRR (NT.M2) = | $. \mathrm{M} 2)=$. 8300 |
| $\mathrm{N}=28$ |  |
| $\mathrm{PI}=\quad 3.1416$ |  |
| AL00= 1.0000 |  |
| VN (M2/S) = | S $=.000001$ |
| JPR= 56 |  |
| $\mathrm{KIT}=5$ |  |



$\begin{array}{ll}\text { TOL }= & .0002 \\ \text { DX (RAD) }= & \text { V }= \\ \text { CK }= & .1714 \\ 860.0000\end{array}$
$\operatorname{AKS}(M)=.0002 \quad$ DT $(S)=\quad .0008$
$\mathrm{KIT}=5$
******************
*INITIAL VALUES
******************)
*****************

* RESULT VALUES
*****************

```
HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE
HP2=PIEZOMETRIC HEAD IN METERS AFTER VALVE
HP=HEADLOSS IN METERS AT VALVE (NODE NSS)
HP1 HP2 HP 14.8736 3.7120 11.1616
```

NUMBER OF NODES AND PIEZOMETRIC HEADS AT NODES AS METERS :

Figure 8.1: Theoretical pressure head results obtained for $6.2 \mathrm{lt} / \mathrm{s}$.

| $1:$ | 14.93 | $2:$ | 14.93 | $3:$ | 14.92 | $4:$ | 14.92 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | 14.92 | $7:$ | 14.91 | $8:$ | 14.91 | $9:$ | 14.91 | $10:$ |
| $11:$ | 14.90 | $12:$ | 14.90 | $13:$ | 14.90 | $14:$ | 14.90 | $15:$ |
| $16:$ | 14.89 | $17:$ | 14.89 | $18:$ | 14.89 | $19:$ | 14.88 | $20:$ |
| $21:$ | 14.88 | $22:$ | 14.88 | $23:$ | 3.71 | $24:$ | 3.74 .89 |  |
| $26:$ | 3.70 | $27:$ | 3.70 | $28:$ | 3.70 | $29:$ | 3.70 |  |

ZH(I) VALUES AS METERS :

| $1:$ | -.36 | $2:$ | .14 | $3:$ | .39 | $4:$ | .44 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | .49 | $7:$ | .54 | $8:$ | .59 | $9:$ | .59 | $10:$ |
| $11:$ | .64 | $12:$ | .69 | $13:$ | .71 | $14:$ | .74 | $15:$ |
| $16:$ | .79 | $17:$ | .84 | $18:$ | .85 | $19:$ | .89 | $20:$ |
| $21:$ | 1.59 | $22:$ | 2.14 | $23:$ | 2.14 | $24:$ | 2.14 | $25:$ |
| $26:$ | 2.14 | $27:$ | 2.14 | $28:$ | 2.14 | $29:$ | 3.35 | .96 |
|  |  |  |  |  |  |  | .14 |  |


| TIME | ALPHA | BETA | V | Q(1) | Q(NS ) | HPH(1) | HPH (NS | ) HP11 | HPH(NSS) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| . 000 | 1.000 | . 656 | . 172 | . 006 | . 006 | 15.289 | . 350 | 12.734 | 1.572 |
| . 002 | . 998 | . 653 | . 172 | . 006 | . 006 | 15.214 | . 350 | 12.734 | 1.572 |
| . 084 | . 878 | . 507 | . 159 | . 006 | . 006 | 11.853 | . 350 | 10.770 | 1.158 |
| . 168 | . 783 | . 403 | . 139 | . 005 | . 005 | 9.502 | . 350 | 8.501 | 1.137 |
| . 252 | . 707 | . 327 | . 120 | . 004 | . 004 | 7.816 | . 350 | 6.680 | 1.177 |
| . 337 | . 644 | . 271 | . 103 | . 004 | . 004 | 6.565 | . 350 | 5.287 | 1.219 |
| . 421 | . 591 | . 228 | . 088 | . 003 | . 003 | 5.612 | . 350 | 4.219 | 1.252 |
| . 505 | . 547 | . 195 | . 075 | . 003 | . 003 | 4.865 | . 350 | 3.390 | 1.275 |
| . 589 | . 509 | . 168 | . 062 | . 002 | . 002 | 4.268 | . 350 | 2.740 | 1.290 |
| . 673 | . 476 | . 146 | . 050 | . 002 | . 002 | 3.788 | . 350 | 2.230 | 1.299 |
| . 757 | . 447 | . 128 | . 038 | . 001 | . 001 | 3.397 | . 350 | 1.834 | 1.301 |
| . 842 | . 422 | . 113 | . 025 | . 001 | . 001 | 3.073 | . 350 | 1.538 | 1.293 |
| . 926 | . 399 | . 101 | . 013 | . 000 | . 000 | 2.802 | . 350 | 1.335 | 1.275 |
| 1.010 | . 379 | . 090 | -. 001 | . 000 | . 000 | 2.579 | . 350 | 1.240 | 1.240 |
| 1.094 | . 361 | . 079 | -. 017 | -. 001 | -. 001 | 2.388 | . 350 | 1.117 | 1.222 |
| 1.178 | . 346 | . 071 | -. 032 | -. 001 | -. 001 | 2.231 | . 350 | . 863 | 1.248 |
| 1.262 | . 332 | . 064 | -. 045 | -. 002 | -. 002 | 2.099 | . 350 | . 532 | 1.302 |
| 1.347 | . 319 | . 059 | -. 056 | -. 002 | -. 002 | 1.983 | . 350 | . 195 | 1.362 |
| 1.431 | . 307 | . 054 | -. 063 | -. 002 | -. 002 | 1.879 | . 350 | -. 101 | 1.415 |
| 1.515 | . 296 | . 051 | -. 069 | -. 002 | -. 002 | 1.784 | . 350 | -. 340 | 1.454 |
| 1.599 | . 285 | . 047 | -. 073 | -. 003 | -. 003 | 1.697 | . 350 | -. 527 | 1.481 |
| 1.683 | . 276 | . 044 | -. 076 | -. 003 | -. 003 | 1.618 | . 350 | -. 674 | 1.500 |
| 1.767 | . 267 | . 042 | -. 078 | -. 003 | -. 003 | 1.545 | . 350 | -. 791 | 1.512 |
| 1.852 | . 258 | . 039 | -. 080 | -. 003 | -. 003 | 1.477 | . 350 | -. 887 | 1.520 |
| 1.936 | . 250 | . 037 | -. 081 | -. 003 | -. 003 | 1.415 | . 350 | -. 969 | 1.525 |
| 2.020 | . 243 | . 035 | -. 082 | -. 003 | -. 003 | 1.359 | . 350 | -1.040 | 1.529 |
| 2.104 | . 235 | . 033 | -. 084 | -. 003 | -. 003 | 1.306 | . 350 | -1.103 | 1.532 |
| 2.188 | . 229 | . 032 | -. 084 | -. 003 | -. 003 | 1.258 | . 350 | -1.160 | 1.534 |
| 2.272 | . 222 | . 030 | -. 085 | -. 003 | -. 003 | 1.214 | . 350 | -1.211 | 1.536 |
| 2.356 | . 216 | . 029 | -. 086 | -. 003 | -. 003 | 1.173 | . 350 | -1.258 | 1.538 |
| 2.441 | . 210 | . 028 | -. 087 | -. 003 | -. 003 | 1.134 | . 350 | -1.301 | 1.539 |
| 2.525 | . 204 | . 027 | -. 087 | -. 003 | -. 003 | 1.099 | . 350 | -1.340 | 1.540 |
| 2.609 | . 199 | . 025 | -. 088 | -. 003 | -. 003 | 1.065 | . 350 | -1.377 | 1.541 |
| 2.693 | . 194 | . 024 | -. 088 | -. 003 | -. 003 | 1.034 | . 350 | -1.411 | 1.542 |
| 2.777 | . 188 | . 024 | -. 089 | -. 003 | -. 003 | 1.005 | . 350 | -1.443 | 1.543 |
| 2.861 | . 184 | . 023 | -. 089 | -. 003 | -. 003 | . 977 | . 350 | -1.474 | 1.543 |
| 2.946 | . 179 | . 022 | -. 090 | -. 003 | -. 003 | . 950 | . 350 | -1.502 | 1.544 |
| 3.030 | . 174 | . 021 | -. 090 | -. 003 | -. 003 | . 926 | . 350 | -1.529 | 1.544 |
| 3.114 | . 170 | . 021 | -. 091 | -. 003 | -. 003 | . 903 | . 350 | -1.554 | 1.545 |
| 3.198 | . 165 | . 020 | -. 091 | -. 003 | -. 003 | . 881 | . 350 | -1.577 | 1.545 |
| 3.282 | . 161 | . 020 | -. 091 | -. 003 | -. 003 | . 860 | . 350 | -1.599 | 1.546 |
| 3.366 | . 157 | . 019 | -. 092 | -. 003 | -. 003 | . 841 | . 350 | -1.620 | 1.546 |
| 3.451 | . 153 | . 019 | -. 092 | -. 003 | -. 003 | . 822 | . 350 | -1.640 | 1.547 |
| 3.535 | . 149 | . 018 | -. 092 | -. 003 | -. 003 | . 805 | . 350 | -1.658 | 1.547 |
| 3.619 | . 146 | . 018 | -. 092 | -. 003 | -. 003 | . 788 | . 350 | -1.676 | 1.547 |

Figure 8.1: Theoretical pressure head results obtained for $6.2 \mathrm{lt} / \mathrm{s}$ (continued).

Theoretical results for $6.2 \mathrm{lt} / \mathrm{s}$ steady state discharge


Figure 8.2: Theoretical pressure head results obtained for $6.2 \mathrm{lt} / \mathrm{s}$.

Theoretical discharges in unsteady flow at node 1 for 6.2 lt/s steady state discharge


Figure 8.3: Theoretical transient discharges obtained for $6.2 \mathrm{lt} / \mathrm{s}$.

Theoretical results for 8.16 lt/s steady state discharge


Figure 8.4: Theoretical pressure head results obtained for $8.16 \mathrm{lt} / \mathrm{s}$.

Theoretical discharges in unsteady flow at node 1 for 8.16 lt/s steady state discharge


Figure 8.5: Theoretical transient discharges obtained for $8.16 \mathrm{lt} / \mathrm{s}$.

### 8.1.2 Results of computation performed for moderate steady state discharge

In these computations, moderate steady state discharges are taken as $12.31 \mathrm{lt} / \mathrm{s}(1$ $\mathrm{m} / \mathrm{s}), 18.32 \mathrm{lt} / \mathrm{s}(1.5 \mathrm{~m} / \mathrm{s}), 22.64 \mathrm{lt} / \mathrm{s}(1.85 \mathrm{~m} / \mathrm{s})$. The output and results are shown below.

The numerical solution result is given in Figure 8.6. The graph of pressure heads and discharge values at measurement points drawn according to the results of numerical solution are given in Figures 8.7, 8.8, 8.9, 8.10, 8.11 and 8.12 respectively.

```
EL (M)= 3.7200
F= .0226
TM (S)= 8.0000
HR (M)= 11.0000
TOL= . }000
DX (RAD)= . }071
AL= 1.0000
AKS(M)= .0002
K= 0 .0000
K= 0
NSS= 23
\begin{tabular}{|c|c|}
\hline EL ( \(M\) ) = & 3.7200 \\
\hline \multicolumn{2}{|l|}{\(\mathrm{F}=\quad .0226\)} \\
\hline TM (S) = & 8.0000 \\
\hline HR ( \(M\) ) = & 11.0000 \\
\hline TOL= & . 0002 \\
\hline DX (RAD) & . 0714 \\
\hline AL= 1 & AL = 1.0000 \({ }^{\text {d }}\) \\
\hline AKS \((M)=\) & . 0002 \\
\hline HMIN= & . 0000 \\
\hline \(\mathrm{K}=0\) & \\
\hline NSS= 23 & \\
\hline
\end{tabular}
```

    ******************
    * INITIAL VALUES
    ******************
    A $(M 3 / S E C)=1331.0000$
D (M)= . 1250
RN (RPM) $=1450.0000$
QR (M3/S)= . 0360
$\mathrm{V}=\quad .3419$
$C K=201.0000$
AL0= 1.0000
DT (S)= . 0008
T= 0000.0008
QQA (M3/S)= . 0123

```
XL (M)= 28.0000
G (M/S2)= 9.8060
TR (NT.M)= 32.1800
WRR (NT.M2)= . }830
N= 28
PI= 3.1416
AL00= 1.0000
VN (M2/S)= .000001
JPR= 56
KIT= 5
```

HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE
HP2=PIEZOMETRIC HEAD IN METERS AFTER VALVE
HP=HEADLOSS IN METERS AT VALVE (NODE NSS)
$\begin{array}{llll}\text { HP1 HP2 HP } & 13.9314 \quad 3.7718 & 10.1596\end{array}$
NUMBER OF NODES AND PIEZOMETRIC HEADS AT NODES AS METERS :

| $1:$ | 14.13 | $2:$ | 14.12 | $3:$ | 14.11 | $4:$ | 14.10 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | 14.09 | $7:$ | 14.08 | $8:$ | 14.07 | $9:$ | 14.06 | $10:$ |
| $11:$ | 14.04 | $12:$ | 14.03 | $13:$ | 14.02 | $14:$ | 14.01 | $15:$ |
| $16:$ | 14.00 | $17:$ | 13.99 | $18:$ | 13.98 | $19:$ | 14.97 | $20:$ |
| $21:$ | 13.95 | $22:$ | 13.94 | $23:$ | 3.77 | $24:$ | 3.76 | $25:$ |
| $26:$ | 3.74 | $27:$ | 3.74 | $28:$ | 3.73 | $29:$ | 3.72 | 3.75 |
|  |  |  |  |  |  |  |  |  |

ZH(I) VALUES AS METERS :

| $1:$ | -.34 | $2:$ | .16 | $3:$ | .41 | $4:$ | .46 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | .51 | $7:$ | .56 | $8:$ | .61 | $9:$ | .61 | $10:$ |
| $11:$ | .66 | $12:$ | .71 | $13:$ | .73 | $14:$ | .76 | $15:$ |

Figure 8.6: Theoretical pressure head results obtained for $12.31 \mathrm{lt} / \mathrm{s}$.

| 16: | .81 | $17:$ | .86 | $18:$ | .87 | $19:$ | .91 | $20:$ |
| :--- | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $21:$ | 1.61 | $22:$ | 2.16 | $23:$ | 2.16 | $24:$ | 2.16 | $25:$ |
| $26:$ | 2.16 | $27:$ | 2.16 | $28:$ | 2.16 | $29:$ | 3.36 |  |


| TIME | ALPHA | BETA | V | Q(1) | Q(NS ) | HPH(1) | HPH(NS | ) HP11 | HPH(NSS) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| . 000 | 1.000 | . 710 | . 339 | . 012 | . 012 | 14.472 | . 357 | 11.771 | 1.612 |
| . 002 | . 997 | . 707 | . 339 | . 012 | . 012 | 14.394 | . 357 | 11.771 | 1.612 |
| . 084 | . 868 | . 548 | . 323 | . 012 | . 012 | 10.881 | . 357 | 10.269 | 1.038 |
| . 168 | . 766 | . 430 | . 293 | . 011 | . 011 | 8.532 | . 357 | 8.454 | . 891 |
| . 252 | . 685 | . 343 | . 259 | . 009 | . 009 | 6.897 | . 357 | 6.833 | . 887 |
| . 337 | . 620 | . 279 | . 228 | . 008 | . 008 | 5.723 | . 357 | 5.510 | . 929 |
| . 421 | . 567 | . 230 | . 199 | . 007 | . 007 | 4.862 | . 357 | 4.464 | . 980 |
| . 505 | . 522 | . 193 | . 172 | . 006 | . 006 | 4.211 | . 357 | 3.642 | 1.027 |
| . 589 | . 485 | . 163 | . 148 | . 005 | . 005 | 3.708 | . 357 | 2.994 | 1.067 |
| . 673 | . 453 | . 140 | . 125 | . 005 | . 005 | 3.309 | . 357 | 2.483 | 1.099 |
| . 757 | . 426 | . 122 | . 104 | . 004 | . 004 | 2.989 | . 357 | 2.078 | 1.122 |
| . 842 | . 402 | . 107 | . 084 | . 003 | . 003 | 2.727 | . 357 | 1.760 | 1.138 |
| . 926 | . 380 | . 095 | . 064 | . 002 | . 002 | 2.501 | . 357 | 1.510 | 1.144 |
| 1.010 | . 361 | . 085 | . 045 | . 002 | . 002 | 2.310 | . 357 | 1.323 | 1.143 |
| 1.094 | . 345 | . 076 | . 026 | . 001 | . 001 | 2.148 | . 357 | 1.192 | 1.135 |
| 1.178 | . 329 | . 068 | . 005 | . 000 | . 000 | 2.008 | . 357 | 1.119 | 1.116 |
| 1.262 | . 316 | . 060 | -. 016 | -. 001 | -. 001 | 1.890 | . 357 | 1.075 | 1.097 |
| 1.347 | . 304 | . 054 | -. 037 | -. 001 | -. 001 | 1.796 | . 357 | . 976 | 1.099 |
| 1.431 | . 293 | . 049 | -. 058 | -. 002 | -. 002 | 1.722 | . 357 | . 821 | 1.121 |
| 1.515 | . 283 | . 047 | -. 078 | -. 003 | -. 003 | 1.663 | . 357 | . 624 | 1.159 |
| 1.599 | . 273 | . 045 | -. 095 | -. 003 | -. 003 | 1.611 | . 357 | . 404 | 1.205 |
| 1.683 | . 264 | . 044 | -. 110 | -. 004 | -. 004 | 1.567 | . 357 | . 178 | 1.254 |
| 1.767 | . 255 | . 044 | -. 123 | -. 004 | -. 004 | 1.526 | . 357 | -. 037 | 1.301 |
| 1.852 | . 246 | . 044 | -. 134 | -. 005 | -. 005 | 1.482 | . 357 | -. 235 | 1.344 |
| 1.936 | . 236 | . 044 | -. 142 | -. 005 | -. 005 | 1.441 | . 357 | -. 411 | 1.380 |
| 2.020 | . 227 | . 045 | -. 150 | -. 005 | -. 005 | 1.399 | . 357 | -. 563 | 1.410 |
| 2.104 | . 217 | . 045 | -. 155 | -. 006 | -. 006 | 1.358 | . 357 | -. 693 | 1.435 |
| 2.188 | . 208 | . 045 | -. 160 | -. 006 | -. 006 | 1.315 | . 357 | -. 806 | 1.454 |
| 2.272 | . 198 | . 045 | -. 164 | -. 006 | -. 006 | 1.272 | . 357 | -. 903 | 1.468 |
| 2.356 | . 189 | . 046 | -. 167 | -. 006 | -. 006 | 1.231 | . 357 | -. 986 | 1.480 |
| 2.441 | . 179 | . 046 | -. 170 | -. 006 | -. 006 | 1.193 | . 357 | -1.059 | 1.489 |
| 2.525 | . 169 | . 047 | -. 172 | -. 006 | -. 006 | 1.156 | . 357 | -1.122 | 1.497 |
| 2.609 | . 159 | . 047 | -. 174 | -. 006 | -. 006 | 1.118 | . 357 | -1.179 | 1.502 |
| 2.693 | . 149 | . 047 | -. 176 | -. 006 | -. 006 | 1.081 | . 357 | -1.231 | 1.506 |
| 2.777 | . 140 | . 047 | -. 178 | -. 006 | -. 006 | 1.047 | . 357 | -1.278 | 1.509 |
| 2.861 | . 130 | . 047 | -. 179 | -. 006 | -. 006 | 1.015 | . 357 | -1.321 | 1.512 |
| 2.946 | . 120 | . 047 | -. 181 | -. 006 | -. 006 | . 984 | . 357 | -1.360 | 1.515 |
| 3.030 | . 110 | . 047 | -. 182 | -. 007 | -. 007 | . 956 | . 357 | -1.397 | 1.517 |
| 3.114 | . 100 | . 047 | -. 183 | -. 007 | -. 007 | . 931 | . 357 | -1.429 | 1.519 |
| 3.198 | . 090 | . 046 | -. 184 | -. 007 | -. 007 | . 906 | . 357 | -1.460 | 1.520 |
| 3.282 | . 080 | . 046 | -. 185 | -. 007 | -. 007 | . 882 | . 357 | -1.489 | 1.521 |
| 3.366 | . 071 | . 045 | -. 186 | -. 007 | -. 007 | . 857 | . 357 | -1.517 | 1.522 |
| 3.451 | . 061 | . 045 | -. 186 | -. 007 | -. 007 | . 835 | . 357 | -1.543 | 1.523 |
| 3.535 | . 052 | . 044 | -. 187 | -. 007 | -. 007 | . 815 | . 357 | -1.567 | 1.524 |
| 3.619 | . 043 | . 044 | -. 188 | -. 007 | -. 007 | . 795 | . 357 | -1.589 | 1.525 |
| 3.703 | . 033 | . 043 | -. 189 | -. 007 | -. 007 | . 775 | . 357 | -1.612 | 1.525 |
| 3.787 | . 025 | . 042 | -. 189 | -. 007 | -. 007 | . 756 | . 357 | -1.632 | 1.526 |
| 3.871 | . 016 | . 041 | -. 190 | -. 007 | -. 007 | . 740 | . 357 | -1.652 | 1.527 |
| 3.956 | . 007 | . 040 | -. 190 | -. 007 | -. 007 | . 727 | . 357 | -1.669 | 1.528 |
| 4.040 | -. 001 | . 039 | -. 191 | -. 007 | -. 007 | . 716 | . 357 | -1.684 | 1.529 |
| 4.124 | -. 009 | . 038 | -. 191 | -. 007 | -. 007 | . 703 | . 357 | -1.699 | 1.530 |
| 4.208 | -. 017 | . 037 | -. 192 | -. 007 | -. 007 | . 690 | . 357 | -1.713 | 1.530 |
| 4.292 | -. 024 | . 036 | -. 192 | -. 007 | -. 007 | . 677 | . 357 | -1.728 | 1.530 |
| 4.376 | -. 032 | . 035 | -. 193 | -. 007 | -. 007 | . 665 | . 357 | -1.741 | 1.531 |
| 4.460 | -. 039 | . 034 | -. 193 | -. 007 | -. 007 | . 655 | . 357 | -1.752 | 1.531 |
| 4.545 | -. 046 | . 034 | -. 193 | -. 007 | -. 007 | . 647 | . 357 | -1.763 | 1.532 |

Figure 8.6: Theoretical pressure head results obtained for $12.31 \mathrm{lt} / \mathrm{s}$ (Continued).

Theoretical results for $\mathbf{1 2 . 3 1} \mathbf{~ l t / s}$ steady state discharge


Figure 8.7: Theoretical pressure head results obtained for $12.31 \mathrm{lt} / \mathrm{s}$.

Theoretical discharges in unsteady flow at node 1 for $12.31 \mathrm{lt} / \mathrm{s}$ steady state discharge


Figure 8.8: Theoretical transient discharges obtained for $12.31 \mathrm{lt} / \mathrm{s}$.

Theoretical results for $18.32 \mathrm{lt} / \mathrm{s}$ steady state discharge


Figure 8.9: Theoretical pressure head results obtained for $18.32 \mathrm{lt} / \mathrm{s}$.

Theoretical discharges in unsteady flow at node 1 for 18.32 lt/s steady state discharge


Figure 8.10: Theoretical transient discharges obtained for $18.32 \mathrm{lt} / \mathrm{s}$.

Theoretical results for $22.64 \mathrm{lt} / \mathrm{s}$ steady state discharge


Figure 8.11: Theoretical pressure head results obtained for $22.64 \mathrm{lt} / \mathrm{s}$.

Theoretical discharges in unsteady flow at node 1 for 22.63 lt/s steady state discharge


Figure 8.12: Theoretical transient discharges obtained for $22.64 \mathrm{lt} / \mathrm{s}$.

### 8.1.3 Results of computation performed for upper steady state discharge

In this computation, upper steady state discharge was taken as $39.00 \mathrm{lt} / \mathrm{s}$ (3.20 $\mathrm{m} / \mathrm{s}$ ).

The numerical solution result is given in Figure 8.13. The graph of pressure head and discharge values at measurement points drawn according to the results of numerical solution are given in Figures 8.14 and 8.15 respectively.

```
```

EL (M)= 3.7500

```
```

EL (M)= 3.7500
F= . }021
F= . }021
TM (S)= 8.0000
TM (S)= 8.0000
HR (M)= 11.0000
HR (M)= 11.0000
TOL= . }000
TOL= . }000
DX (RAD)= . }071
DX (RAD)= . }071
AKS(M)= .0002
AKS(M)= .0002
HMIN= .0000
HMIN= .0000
K= 0
K= 0
NSS= 23

```
```

NSS= 23

```
```

    ******************
    * INITIAL VALUES
    ******************
    A (M3/SEC)= 1331.0000
D (M)= . 1250
RN (RPM) $=1450.0000$
QR (M3/S)= . 0360
$V=\quad 1.0833$
CK= 7.9000
AL0= 1.0000
DT (S)= . 0008
$\mathrm{T}=\quad .0000$
QQA $(M 3 / S)=.0390$

```
XL (M)= 28.0000
G (M/S2)= 9.8060
TR (NT.M)= 32.1800
WRR (NT.M2)= . }830
N= 28
PI= 3.1416
AL00= 1.0000
VN (M2/S)= .000001
JPR= 56
KIT= 5
```

HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE
HP2=PIEZOMETRIC HEAD IN METERS AFTER VALVE
HP=HEADLOSS IN METERS AT VALVE (NODE NSS)
HP1 HP2 HP
8.3156
4.2723
4.0433

HP1 HP2 HP
8.3156
4.2723
4.0433

NUMBER OF NODES AND PIEZOMETRIC HEADS AT NODES AS METERS :

| $1:$ | 10.23 | $2:$ | 10.15 | $3:$ | 10.06 | $4:$ | 9.97 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | 9.80 | $7:$ | 9.71 | $8:$ | 9.62 | $9:$ | 9.54 | $10:$ |
| $11:$ | 9.36 | $12:$ | 9.28 | $13:$ | 9.19 | $14:$ | 9.10 | $15:$ |
| $16:$ | 8.93 | $17:$ | 8.84 | $18:$ | 8.75 | $19:$ | 8.66 | $20:$ |
| $21:$ | 8.49 | $22:$ | 8.40 | $23:$ | 4.27 | $24:$ | 4.19 | $25:$ |
| $26:$ | 4.01 | $27:$ | 3.92 | $28:$ | 3.84 | $29:$ | 3.75 | 4.10 |
|  |  |  |  |  |  |  |  |  |

ZH(I) VALUES AS METERS :

| $1:$ | -.31 | $2:$ | .19 | $3:$ | .44 | $4:$ | .49 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | .54 | $7:$ | .59 | $8:$ | .64 | $9:$ | .64 | $10:$ |
| $11:$ | .69 | $12:$ | .74 | $13:$ | .76 | $14:$ | .79 | $15:$ |

Figure 8.13: Theoretical pressure head results obtained for $39 \mathrm{lt} / \mathrm{s}$.

| 16: | .84 | $17:$ | .89 | $18:$ | .90 | $19:$ | .94 | $20:$ | .98 |
| :--- | ---: | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 21: | 1.64 | $22:$ | 2.19 | $23:$ | 2.19 | $24:$ | 2.19 | $25:$ | 2.19 |
| 26: | 2.19 | $27:$ | 2.19 | $28:$ | 2.19 | $29:$ | 3.39 |  |  |


| TIME | ALPHA | BETA | V | Q(1) | Q(NS) | HPH(1) | HPH(NS) | HP11 | HPH(NSS) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| . 000 | 1.000 | . 999 | 1.080 | . 039 | . 039 | 10.545 | . 359 | 6.126 | 2.082 |
| . 002 | . 996 | . 991 | 1.080 | . 039 | . 039 | 10.431 | . 359 | 6.125 | 2.083 |
| . 084 | . 827 | . 668 | 1.055 | . 038 | . 038 | 5.743 | . 359 | 4.955 | 1.108 |
| . 168 | . 709 | . 475 | 1.000 | . 036 | . 036 | 3.369 | . 359 | 4.135 | . 675 |
| . 252 | . 622 | . 356 | . 934 | . 034 | . 034 | 2.155 | . 359 | 3.527 | . 509 |
| . 337 | . 556 | . 279 | . 866 | . 031 | . 031 | 1.506 | . 359 | 3.052 | . 460 |
| . 421 | . 503 | . 225 | . 800 | . 029 | . 029 | 1.158 | . 359 | 2.680 | . 466 |
| . 505 | . 459 | . 187 | . 737 | . 027 | . 026 | . 974 | . 359 | 2.378 | . 499 |
| . 589 | . 423 | . 159 | . 677 | . 024 | . 024 | . 881 | . 359 | 2.130 | . 540 |
| . 673 | . 392 | . 137 | . 621 | . 022 | . 022 | . 838 | . 359 | 1.919 | . 585 |
| . 757 | . 365 | . 120 | . 569 | . 020 | . 020 | . 823 | . 359 | 1.749 | . 628 |
| . 842 | . 341 | . 106 | . 520 | . 019 | . 019 | . 823 | . 359 | 1.607 | . 673 |
| . 926 | . 320 | . 095 | . 474 | . 017 | . 017 | . 830 | . 359 | 1.487 | . 708 |
| 1.010 | . 301 | . 085 | . 430 | . 015 | . 015 | . 839 | . 359 | 1.377 | . 734 |
| 1.094 | . 284 | . 077 | . 388 | . 014 | . 014 | . 848 | . 359 | 1.290 | . 766 |
| 1.178 | . 268 | . 070 | . 349 | . 013 | . 013 | . 859 | . 359 | 1.217 | . 793 |
| 1.262 | . 254 | . 064 | . 310 | . 011 | . 011 | . 870 | . 359 | 1.148 | . 810 |
| 1.347 | . 241 | . 058 | . 274 | . 010 | . 010 | . 875 | . 359 | 1.087 | . 833 |
| 1.431 | . 230 | . 053 | . 238 | . 009 | . 009 | . 873 | . 359 | 1.039 | . 843 |
| 1.515 | . 219 | . 047 | . 203 | . 007 | . 007 | . 865 | . 359 | . 996 | . 849 |
| 1.599 | . 210 | . 041 | . 169 | . 006 | . 006 | . 852 | . 359 | . 958 | . 854 |
| 1.683 | . 202 | . 036 | . 136 | . 005 | . 005 | . 838 | . 359 | . 924 | . 863 |
| 1.767 | . 195 | . 031 | . 102 | . 004 | . 004 | . 822 | . 359 | . 897 | . 867 |
| 1.852 | . 189 | . 026 | . 069 | . 002 | . 002 | . 809 | . 359 | . 884 | . 870 |
| 1.936 | . 184 | . 022 | . 037 | . 001 | . 001 | . 811 | . 359 | . 880 | . 876 |
| 2.020 | . 179 | . 020 | . 004 | . 000 | . 000 | . 804 | . 359 | . 870 | . 873 |
| 2.104 | . 175 | . 018 | -. 029 | -. 001 | -. 001 | . 800 | . 359 | . 871 | . 873 |
| 2.188 | . 172 | . 018 | -. 061 | -. 002 | -. 002 | . 813 | . 359 | . 861 | . 874 |
| 2.272 | . 168 | . 021 | -. 094 | -. 003 | -. 003 | . 847 | . 359 | . 859 | . 890 |
| 2.356 | . 163 | . 028 | -. 125 | -. 004 | -. 004 | . 906 | . 359 | . 851 | . 908 |
| 2.441 | . 156 | . 038 | -. 155 | -. 006 | -. 006 | . 986 | . 359 | . 842 | . 933 |
| 2.525 | . 146 | . 050 | -. 184 | -. 007 | -. 007 | 1.075 | . 359 | . 842 | . 958 |
| 2.609 | . 134 | . 064 | -. 211 | -. 008 | -. 008 | 1.167 | . 359 | . 833 | . 985 |
| 2.693 | . 119 | . 078 | -. 237 | -. 009 | -. 009 | 1.263 | . 359 | . 819 | 1.019 |
| 2.777 | . 102 | . 090 | -. 261 | -. 009 | -. 009 | 1.342 | . 359 | . 809 | 1.048 |
| 2.861 | . 081 | . 102 | -. 284 | -. 010 | -. 010 | 1.413 | . 359 | . 787 | 1.074 |
| 2.946 | . 059 | . 113 | -. 305 | -. 011 | -. 011 | 1.468 | . 359 | . 763 | 1.087 |
| 3.030 | . 034 | . 122 | -. 325 | -. 012 | -. 012 | 1.511 | . 359 | . 736 | 1.105 |
| 3.114 | . 008 | . 129 | -. 345 | -. 012 | -. 012 | 1.562 | . 359 | . 711 | 1.126 |
| 3.198 | -. 020 | . 134 | -. 362 | -. 013 | -. 013 | 1.600 | . 359 | . 682 | 1.138 |
| 3.282 | -. 049 | . 138 | -. 379 | -. 014 | -. 014 | 1.619 | . 359 | . 648 | 1.151 |
| 3.366 | -. 079 | . 143 | -. 395 | -. 014 | -. 014 | 1.638 | . 359 | . 616 | 1.162 |
| 3.451 | -. 109 | . 150 | -. 410 | -. 015 | -. 015 | 1.669 | . 359 | . 586 | 1.174 |
| 3.535 | -. 141 | . 153 | -. 424 | -. 015 | -. 015 | 1.709 | . 359 | . 559 | 1.187 |
| 3.619 | -. 174 | . 152 | -. 437 | -. 016 | -. 016 | 1.758 | . 359 | . 533 | 1.204 |
| 3.703 | -. 206 | . 151 | -. 449 | -. 016 | -. 016 | 1.817 | . 359 | . 514 | 1.212 |
| 3.787 | -. 237 | . 150 | -. 460 | -. 017 | -. 017 | 1.896 | . 359 | . 512 | 1.242 |
| 3.871 | -. 269 | . 148 | -. 469 | -. 017 | -. 017 | 1.983 | . 359 | . 502 | 1.270 |
| 3.956 | -. 300 | . 145 | -. 477 | -. 017 | -. 017 | 2.079 | . 359 | . 506 | 1.301 |
| 4.040 | -. 330 | . 141 | -. 483 | -. 017 | -. 017 | 2.180 | . 359 | . 508 | 1.322 |
| 4.124 | -. 359 | . 136 | -. 489 | -. 018 | -. 018 | 2.281 | . 359 | . 517 | 1.341 |
| 4.208 | -. 387 | . 130 | -. 492 | -. 018 | -. 018 | 2.373 | . 359 | . 524 | 1.363 |
| 4.292 | -. 413 | . 122 | -. 495 | -. 018 | -. 018 | 2.459 | . 359 | . 536 | 1.385 |
| 4.376 | -. 438 | . 114 | -. 497 | -. 018 | -. 018 | 2.544 | . 359 | . 555 | 1.416 |
| 4.460 | -. 462 | . 107 | -. 498 | -. 018 | -. 018 | 2.634 | . 359 | . 565 | 1.429 |
| 4.545 | -. 483 | . 099 | -. 498 | -. 018 | -. 018 | 2.718 | . 359 | . 590 | 1.447 |

Figure 8.13: Theoretical pressure head results obtained for $39 \mathrm{lt} / \mathrm{s}$ (continued).

## Theoretical results for 39 lt/s steady state discharge



Figure 8.14: Theoretical pressure head results obtained for $39 \mathrm{lt} / \mathrm{s}$.

Theoretical discharges in unsteady flow at node 1 for 39 lt/s steady state discharge


Figure 8.15: Theoretical transient discharges obtained for $39 \mathrm{lt} / \mathrm{s}$.

### 8.2 Results of computations performed for the second system

Similarly numerical solutions are performed using lower, moderate and upper steady state discharges in the second system. In these computations, experimental steady state discharge values are used in the computer programs. The results are given for two measurement points, namely the upstream point and the interior point.

### 8.2.1 Results of computation performed for lower steady state discharges

In these computations, lower steady state discharges are taken as $6.6 \mathrm{lt} / \mathrm{s}(0.54$ $\mathrm{m} / \mathrm{s}), 9.01 \mathrm{lt} / \mathrm{s}(0.74 \mathrm{~m} / \mathrm{s})$. The output and results are shown below.

The numerical solution result is given in Figure 8.16. The graph of pressure heads and discharge values at measurement points drawn according to the results of numerical solution are given from Figure 8.17 to Figure 8.20, respectively.

|  | * INITIAL VALUES |  |
| :---: | :---: | :---: |
| $E L \quad(M)=4.6000$ | A (M3/S) $=1331.0000$ | XL (M) = 108.0000 |
| $\mathrm{F}=\quad .0239$ | D $(\mathrm{M})=$. 1250 | $G(M / S 2)=9.8100$ |
| TM (S) = 12.0000 | RN (RPM) = 1450.0000 | TR (NT.M) = 32.1800 |
| $H R(M)=11.0000$ | QR (M3/S) = . 0360 | WRR (NT.M2) = . 8300 |
| TOL= . 0002 | $\mathrm{VI}=$. 1830 | $\mathrm{N}=42$ |
| DX (RAD) = . 0714 | CK $=10.8700$ | $\mathrm{PI}=3.1416$ |
| AKV1 = 17.2900 | AKV2= 585.9800 | DTAU (S)= . 1000 |
| $\operatorname{TAU}(1)=1.0000$ | $\mathrm{TAU}(2)=$. 9474 | $\mathrm{TAU}(3)=.8947$ |
| $\operatorname{TAU}(4)=.8421$ | $\mathrm{TAU}(5)=.7895$ | $\mathrm{TAU}(6)=.7368$ |
| $\operatorname{TAU}(7)=.6842$ | $\mathrm{TAU}(8)=.6316$ | $\mathrm{TAU}(9)=.5789$ |
| $\operatorname{TAU}(10)=.5263$ | $\mathrm{TAU}(11)=.4737$ | $\mathrm{TAU}(12)=.4211$ |
| $\operatorname{TAU}(13)=\quad .3684$ | $\mathrm{TAU}(14)=.3158$ | $\operatorname{TAU}(15)=.2632$ |
| $\operatorname{TAU}(16)=.2105$ | $\operatorname{TAU}(17)=\quad .1579$ | $\operatorname{TAU}(18)=.1053$ |
| $\operatorname{TAU}(19)=.0526$ | $\operatorname{TAU}(20)=\quad .0000$ | M2 $=27$ |
| AL= 1.0000 | AL0= 1.0000 | AL00= 1.0000 |
| HMIN= . 0000 | $\mathrm{T}=\quad .0000$ | JPR= 21 |
| $\mathrm{K}=0$ | NSV = 42 | KIT $=5$ |
| CCT (S) = 1.9000 | NTAU $=20$ | QQA (M3/S)= .0066 |
| DT $(S)=.0019$ | AKS (M) = . 0002 | $\mathrm{VN}(\mathrm{M} 2 / \mathrm{S})=.000001$ |

Figure 8.16: Theoretical pressure head results obtained for $6.6 \mathrm{lt} / \mathrm{s}$.

HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE
HP2=PIEZOMETRIC HEAD IN METERS AFTER VALVE
HP=HEADLOSS IN METERS AT VALVE (NODE NSV)
$\begin{array}{lllll}\text { HP1 HP2 HP } & \text { 13.3869 } & 4.6074 & 8.7795\end{array}$
NUMBER OF NODES AND PIEZOMETRIC HEADS AT NODES AS METERS :

| $1:$ | 13.69 | $2:$ | 13.68 | $3:$ | 13.67 | $4:$ | 13.67 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | 13.65 | $7:$ | 13.64 | $8:$ | 13.64 | $9:$ | 13.63 | $10:$ |
| $11:$ | 13.62 | $12:$ | 13.61 | $13:$ | 13.60 | $14:$ | 13.59 | $15:$ |
| $16:$ | 13.58 | $17:$ | 13.57 | $18:$ | 13.56 | $19:$ | 13.56 | $20:$ |
| $21:$ | 13.54 | $22:$ | 13.53 | $23:$ | 13.53 | $24:$ | 13.52 | $25:$ |
| $26:$ | 13.50 | $27:$ | 13.50 | $28:$ | 13.49 | $29:$ | 13.48 | $30:$ |
| $31:$ | 13.47 | $32:$ | 13.46 | $33:$ | 13.45 | $34:$ | 13.45 | $35:$ |
| $36:$ | 13.43 | $37:$ | 13.42 | $38:$ | 13.42 | $39:$ | 13.41 | $40:$ |
| $41:$ | 13.39 | $42:$ | 4.61 | $43:$ | 4.60 |  |  | 13.44 |
|  |  |  |  |  |  |  |  |  |

ZH(I) VALUES AS METERS :

|  |  |  |  |  |  |  |  |  |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $1:$ | -.30 | $2:$ | .05 | $3:$ | .12 | $4:$ | .21 | $5:$ |
| $6:$ | .10 | $7:$ | .15 | $8:$ | .30 | $9:$ | 1.39 | $10:$ |
| $11:$ | 1.39 | $12:$ | 1.34 | $13:$ | 1.35 | $14:$ | 1.36 | $15:$ |
| $16:$ | 1.35 | $17:$ | 1.35 | $18:$ | 1.33 | $19:$ | 1.32 | $20:$ |
| $21:$ | 1.34 | $22:$ | 1.34 | $23:$ | 1.30 | $24:$ | 1.28 | $25:$ |
| $26:$ | 1.41 | $27:$ | 1.70 | $28:$ | 1.60 | $29:$ | 1.60 | $30:$ |
| $31:$ | 1.60 | $32:$ | 1.60 | $33:$ | 1.60 | $34:$ | 1.60 | $35:$ |
| $36:$ | 1.61 | $37:$ | 1.61 | $38:$ | 1.60 | $39:$ | 1.60 | $40:$ |
| $41:$ | 1.61 | $42:$ | 1.61 | $43:$ | 4.10 |  |  | 1.50 |
|  |  |  |  |  |  |  | 1.60 |  |


| TIME | TA | ALPHA | BETA | V | Q(1) Q | (M2) Q | Q(NS) H |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| . 0000 | 1.000 | 1.000 | . 556 | . 185 | . 007 | . 007 | . 007 | 713.989 | 11.797 | 711.777 | 72.99 | 500 |
| . 0812 | . 957 | . 898 | . 455 | . 178 | . 006 | . 007 | . 007 | 11.243 | 10.654 | 11.777 | 2.997 | . 500 |
| . 1623 | . 915 | . 813 | . 380 | . 173 | . 006 | . 006 | . 006 | 9.189 | 9.316 | 10.675 | 2.905 | . 500 |
| . 2435 | . 872 | . 742 | . 318 | . 161 | . 006 | . 006 | . 006 | 7.661 | 8.238 | 9.978 | 2.940 | . 500 |
| . 3246 | . 829 | . 683 | . 271 | . 151 | . 005 | . 005 | . 005 | 6.466 | 7.153 | 8.838 | 2.883 | . 500 |
| . 4058 | . 786 | . 632 | . 231 | . 138 | . 005 | . 005 | . 005 | 5.554 | 6.300 | 8.092 | 2.926 | . 500 |
| . 4869 | . 744 | . 588 | . 200 | . 128 | . 005 | . 005 | . 005 | 4.817 | 5.491 | 7.162 | 2.885 | . 500 |
| . 5681 | . 701 | . 550 | . 174 | . 116 | . 004 | . 004 | . 004 | 4.236 | 4.855 | 6.543 | 2.926 | . 500 |
| . 6492 | . 658 | . 517 | . 153 | . 106 | . 004 | . 004 | . 004 | 3.753 | 4.264 | 5.832 | 2.891 | . 500 |
| . 7304 | . 616 | . 488 | . 134 | . 095 | . 003 | . 003 | . 003 | 3.367 | 3.799 | 5.359 | 2.929 | . 500 |
| . 8115 | . 573 | . 463 | . 119 | . 086 | . 003 | . 003 | . 003 | 3.039 | 3.371 | 4.829 | 2.899 | . 500 |
| . 8927 | . 530 | . 440 | . 106 | . 076 | . 003 | . 003 | . 003 | 2.774 | 3.034 | 4.480 | 2.933 | . 500 |
| . 9739 | . 487 | . 419 | . 095 | . 068 | . 002 | . 002 | . 002 | 2.545 | 2.725 | 4.090 | 2.905 | . 500 |
| 1.0550 | . 445 | . 401 | . 085 | . 058 | . 002 | . 002 | . 002 | 2.361 | 2.486 | 3.843 | 2.936 | . 500 |
| 1.1362 | . 402 | . 384 | . 077 | . 050 | . 002 | . 002 | . 002 | 2.197 | 2.266 | 3.562 | 2.910 | . 500 |
| 1.2173 | . 359 | . 369 | . 070 | . 041 | . 001 | . 002 | . 002 | 2.066 | 2.099 | 3.398 | 2.938 | . 500 |
| 1.2985 | . 317 | . 356 | . 064 | . 034 | . 001 | . 001 | . 001 | 1.953 | 1.952 | 3.205 | 2.912 | . 500 |
| 1.3796 | . 274 | . 344 | . 058 | . 025 | . 001 | . 001 | . 001 | 1.868 | 1.848 | 3.110 | 2.938 | . 500 |
| 1.4608 | . 231 | . 332 | . 053 | . 017 | . 001 | . 001 | . 001 | 1.793 | 1.760 | 2.993 | 2.914 | . 500 |
| 1.5419 | . 188 | . 322 | . 048 | . 009 | . 000 | . 000 | . 000 | 1.739 | 1.710 | 2.962 | 2.938 | . 500 |
| 1.6231 | . 146 | . 313 | . 044 | . 001 | . 000 | . 000 | . 000 | 1.684 | 1.673 | 2.914 | 2.913 | . 500 |
| 1.7042 | . 103 | . 304 | . 040 | -. 007 | . 000 | . 000 | . 000 | 1.665 | 1.647 | 2.925 | 2.936 | . 500 |
| 1.7854 | . 060 | . 296 | . 037 | -. 013 | . 000 | -. 001 | -. 001 | 12.155 | 1.727 | 2.864 | 2.919 | . 500 |
| 1.8665 | . 018 | . 289 | . 036 | -. 011 | . 000 | -. 001 | -. 001 | 16.354 | 2.660 | 2.882 | 2.986 | . 500 |
| 1.9477 | . 000 | . 286 | . 037 | . 000 | . 000 | . 000 | . 000 | 12.778 | 9.394 | 3.369 | 3.351 | . 500 |
| 2.0289 | . 000 | . 286 | . 037 | . 000 | . 000 | . 001 | . 001 | 7.874 | 3.129 | 3.092 | 2.994 | . 500 |
| 2.11 | 00 | 286 | 037 | . 000 | 000 | . 000 | . 000 | -2.77 | 3.458 | 2.648 | 2.630 | 500 |

Figure 8.16: Theoretical pressure head results obtained for $6.6 \mathrm{lt} / \mathrm{s}$ (continued).

|  | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | －． |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | ． 000 | ． 000 | 12.772 | 9 | 8 | 3.351 | ． 500 |
| 2.3535 | ． 0 | ． 2 | ． 037 | ． 000 | ． 00 | ． 00 | ． 00 | 7. | 3. | 3.0 | 2.9 | ． 500 |
|  |  |  | ． 037 | ． 000 | ． 000 | ． 000 | ． 000 |  | －3．453 | 2.648 |  | 500 |
| 2.5158 | ． 0 | ． 2 | ． 0 | ． 000 | ． 00 | －． 001 | －． 0 | 1. | 2.8 | 3.0 | 2. | 500 |
|  |  |  | ． 037 | ． 000 | ． 000 | ． 000 | ． 000 | 12． |  | 3.368 |  | 500 |
|  | ． 000 | ． 2 | 0 | ． 000 | ． 000 | 00 | ． 00 | 7.8 | 3.12 | 3.0 | 2.9 | 00 |
|  |  |  |  |  | ． 000 |  | ． 000 |  | －3．448 |  |  | 500 |
| 2.8404 | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | －． 001 | －． 0 | 1 | 2. | 3.084 | 2.98 | 500 |
|  |  |  | ． 037 |  | ． 000 |  | ． 000 |  |  | 8 |  | 00 |
|  | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | ． 001 | ． 001 | 7 | 3. | 3 | 2 | ． 500 |
|  |  |  |  |  | 000 | ． 000 | ． |  |  |  |  | 0 |
| 3.1650 | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | －． 0 | －． | 1.96 | 2.852 | 3.084 | 2.986 |  |
|  |  |  |  |  |  |  |  |  |  |  |  | 00 |
|  |  | ． 286 | ． 037 | ． 000 | ． 000 | 001 | ． 00 | 7.866 | 3.128 | 3.091 | 2.994 | ． 500 |
|  |  |  | ． |  | ． 000 |  | ． |  |  |  |  | 00 |
|  |  | ． 286 | ． 037 | ． 000 | ． 000 | －． 001 |  | 1.963 | 2.852 | 3.084 | 2.986 | 500 |
|  |  |  |  |  |  |  |  |  |  |  |  | 00 |
|  | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | ． 001 | ． 001 | 7. | 3 | 3 | 2.994 | ． 500 |
|  |  |  |  |  | ． 000 |  | ． |  |  |  |  | 0 |
|  |  | ． 286 | ． 037 | ． 000 | ． 000 | －． 001 | ， | 1 | 2.852 | 3.084 | 2.986 |  |
|  |  |  |  |  | ． 000 |  |  |  | 9.364 |  |  | 00 |
|  |  |  |  |  | ． 000 | 01 |  | 7.862 | 3.128 |  |  | 500 |
|  |  |  | ． 037 |  | ． 000 |  | ． |  | －3．429 | 2.649 |  | 00 |
|  |  |  |  |  |  |  |  |  |  | 3.08 |  |  |
|  | 迷 | ． 286 | ． 037 |  | ． 000 | 00 | ． 000 | 12. | 9．35 | 3.367 | 3. | 500 |
|  |  |  |  |  |  |  |  |  |  |  |  | 500 |
|  |  |  | ． 037 |  | ， | ． 000 | ． | 2． 738 |  | 2. | 2.631 | 00 |
|  |  |  |  |  | 00 | －． 001 |  |  |  | 3. | 2.986 | － |
|  | ． 000 | ． 286 | ． 037 |  |  | 00 | ． 000 | 12.730 | ． | 3.367 | 3. | 500 |
|  |  |  |  |  |  |  |  |  |  |  |  | 500 |
|  | 迷 | ． 286 |  |  |  | 迷 | ． | －2．73 |  | 2.649 |  | ． 500 |
|  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | ． 000 | ． 286 | ． 037 |  | ． 000 | ． 000 | ． 000 | 12. | 9.349 | 3.366 | 3.34 | 500 |
|  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | 迷 | ． 286 | ． 037 |  |  |  | ． | －2．727 |  | 2.650 | 2.632 | 500 |
|  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | ． 000 | ． 286 | ． 037 |  | ． 000 | 000 |  | 12.718 | 9.344 | 3.366 | 3.3 | 500 |
|  |  |  |  |  |  |  |  |  | 3．127 | 3.09 |  | 500 |
|  | ． 000 | ． 2 | ． 037 |  | ． 000 | ． 000 | ． 000 | －2．72 |  |  |  | 00 |
|  |  |  |  |  |  | －． 001 | －． 00 | 1.97 |  | ． 08 | ． | 00 |
|  | ． | ． 286 | ． 0 |  | ． 000 | ． 000 |  | 12.712 | 9.339 | 3.366 | 3.348 | 00 |
|  |  |  |  |  |  |  |  |  | 3.127 | 3.09 |  | ． 500 |
|  |  | ． 286 | ， |  |  |  | ． |  |  |  | ， 63 | ． 500 |
|  |  |  |  |  |  |  | － | 1.97 |  | ． 08 | 析 |  |
|  |  | ． 286 | ， |  |  |  |  | 12.707 | 9.334 | 3.366 | 3.348 | 500 |
|  |  |  |  |  |  |  | ， | ． 85 | 3.126 | 3.090 | 2.994 | 00 |
|  | ． 0 | ． 2 | ． 037 |  | ． 000 | ． 000 | ． 0 |  |  |  | 2.63 | 0 |
|  | 000 |  | 7 |  |  | －． 001 | －． 001 |  | 2.85 | 3.082 | 2.986 | 0 |
|  | ． 0 | ． 2 | ． |  | ． 000 | ． 000 |  | 2.701 | 9.329 | 3.365 | 3.348 | 0 |
|  | － | 6 | 7 |  | 000 | 1 | ． 001 | 7.848 | 3.126 | 3.090 | 2.994 | ． 500 |
|  | ． 0 | ． 286 | ． 037 | ． 00 | ． 000 | ． 000 | ． 000 | －2．705 | －3．396 | 2.650 | 2.633 | 500 |
|  | ． 000 | ． 286 | ． 037 | ． 000 | ． 000 | －． 001 | －． 001 | 1.981 | 2.851 | 3.082 | 2.986 | 500 |
| ． 4923 | 000 | 286 | 03 | 00 | ． 00 | ． 00 | 000 | 12. | 9. | 3.365 | 3.34 |  |

Figure 8．16：Theoretical pressure head results obtained for $6.6 \mathrm{lt} / \mathrm{s}$（continued）．


Figure 8.17: Theoretical pressure head results obtained for $6.6 \mathrm{lt} / \mathrm{s}$.


Figure 8.18: Theoretical transient discharges obtained for $6.6 \mathrm{lt} / \mathrm{s}$.


Figure 8.19: Theoretical pressure head results obtained for $9.01 \mathrm{lt} / \mathrm{s}$.


Figure 8.20: Theoretical transient discharges obtained for $9.01 \mathrm{lt} / \mathrm{s}$.

### 8.2.2 Results of computation performed for moderate steady state discharges

In these computations, moderate steady state discharges are taken as $16.49 \mathrm{lt} / \mathrm{s}$ $(1.35 \mathrm{~m} / \mathrm{s}), 19.14 \mathrm{lt} / \mathrm{s}(1.57 \mathrm{~m} / \mathrm{s})$ and $20.55 \mathrm{lt} / \mathrm{s}(1.68 \mathrm{~m} / \mathrm{s})$. The output and results are shown below.

The numerical solution result is given in Figure 8.21. The graph of pressure heads and discharge values at measurement points drawn according to the results of numerical solution are given from Figure 8.22 to Figure 8.27, respectively.

| EL ( $M$ ) = | 4.6000 |
| :---: | :---: |
| $\mathrm{F}=\quad .0221$ |  |
| TM (S)= | 10.0000 |
| HR ( $M$ ) = | 11.0000 |
| TOL= | . 0002 |
| DX (RAD) $=$ | . 0714 |
| AKV1 = | 4.9800 |
| TAU(1) $=$ | 1.0000 |
| $\operatorname{TAU}(4)=$ | . 8421 |
| $\operatorname{TAU}(7)=$ | . 6842 |
| $\operatorname{TAU}(10)=$ | . 5263 |
| $\operatorname{TAU}(13)=$ | . 3684 |
| $\operatorname{TAU}(16)=$ | . 2105 |
| $\operatorname{TAU}(19)=$ | . 0526 |
| 1.0000 |  |
| HMIN $=$ | . 0000 |
| $\mathrm{K}=0$ |  |
| CCT (S) = | 2.5000 |
| DT (S)= | . 0019 |



| XL (M)= 108.0000 |  |
| :---: | :---: |
| G (M/S2) = | $)=9.8100$ |
| TR (NT.M) = | M $=32.1800$ |
| WRR ( $\mathrm{NT} . \mathrm{M} 2$ ) $=$ | .M2) = . 8300 |
| $\mathrm{N}=42$ |  |
| PI= 3.1416 |  |
| DTAU (S)= | )= . 1316 |
| TAU(3)= | . 8947 |
| TAU(6)= | . 7368 |
| TAU (9)= | . 5789 |
| $\operatorname{TAU}(12)=$ | $=.4211$ |
| $\operatorname{TAU}(15)=$ | $=\quad .2632$ |
| TAU(18) $=$ | $=\quad .1053$ |
| M2=27 |  |
| AL00 $=21.0000$JPR= |  |
|  |  |
| KIT $=5$ |  |
| QQA ( $\mathrm{M} 3 / \mathrm{S}$ ) = | /S $=$. 0166 |
| VN (M2/S)= | $S)=.000001$ |

```
******************
* RESULT VALUES *
*****************
```

```
HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE
HP2=PIEZOMETRIC HEAD IN METERS AFTER VALVE
HP=HEADLOSS IN METERS AT VALVE (NODE 23)
HP1 HP2 HP 11.0854 4.6397 6.4457
```

PIEZOMETRIC HEADS AT NODES AS METERS :

| $1:$ | 12.71 | $2:$ | 12.67 | $3:$ | 12.63 | $4:$ | 12.59 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | 12.51 | $7:$ | 12.47 | $8:$ | 12.43 | $9:$ | 12.39 | $10:$ |
| $11:$ | 12.32 | $12:$ | 12.28 | $13:$ | 12.24 | $14:$ | 12.20 | $15:$ |
| $16:$ | 12.12 | $17:$ | 12.08 | $18:$ | 12.04 | $19:$ | 12.00 | $20:$ |
| $21:$ | 11.92 | $22:$ | 11.88 | $23:$ | 11.84 | $24:$ | 11.80 | $25:$ |
| $26:$ | 11.72 | $27:$ | 11.68 | $28:$ | 11.64 | $29:$ | 11.96 |  |
| $31:$ | 11.52 | $32:$ | 11.48 | $33:$ | 11.44 | $34:$ | 11.40 | $30:$ |


| $36:$ | 11.32 | $37:$ | 11.28 | $38:$ | 11.24 | $39:$ | 11.20 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $41:$ | 11.13 | $42:$ | 4.64 | $43:$ | 4.60 |  |  |

ZH(I) VALUES AS METERS :

| $1:$ | -.30 | $2:$ | .05 | $3:$ | .12 | $4:$ | .21 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | .10 | $7:$ | .15 | $8:$ | .30 | $9:$ | 1.39 | $10:$ |
| $11:$ | 1.39 | $12:$ | 1.34 | $13:$ | 1.35 | $14:$ | 1.36 | $15:$ |
| $16:$ | 1.35 | $17:$ | 1.35 | $18:$ | 1.33 | $19:$ | 1.32 | $20:$ |
| $21:$ | 1.34 | $22:$ | 1.34 | $23:$ | 1.30 | $24:$ | 1.28 | $25:$ |
| $26:$ | 1.41 | $27:$ | 1.70 | $28:$ | 1.60 | $29:$ | 1.60 | $30:$ |
| $31:$ | 1.60 | $32:$ | 1.60 | $33:$ | 1.60 | $34:$ | 1.60 | $35:$ |
| $36:$ | 1.61 | $37:$ | 1.61 | $38:$ | 1.60 | $39:$ | 1.60 | $40:$ |
| $41:$ | 1.61 | $42:$ | 1.61 | $43:$ | 4.10 |  |  | 1.50 |
| 1.60 |  |  |  |  |  |  |  |  |


| TIME | TAU | ALPHA | BETA | V | Q(1) | M2) | Q(NS) | $\mathrm{H}(1) \mathrm{H}(2$ | 27) H(NS | V1) H(N | NSV2) H(N |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0000 | 1.000 | 1.000 | . 736 | . 446 | . 016 | . 016 | . 016 | 3.012 | 29.980 | 9.475 | 3.030 | . 500 |
| . 0812 | . 968 | . 867 | . 581 | . 437 | . 016 | . 016 | . 016 | 9.457 | 8.482 | 9.475 | 3.030 | . 500 |
| . 1623 | . 935 | . 761 | . 470 | . 431 | . 016 | . 015 | . 015 | 6.958 | 7.416 | 8.885 | 2.897 | . 500 |
| . 2435 | . 903 | . 675 | . 381 | . 411 | . 015 | . 015 | . 015 | 5.256 | 6.429 | 8.609 | 2.938 | . 500 |
| . 3246 | . 870 | . 604 | . 315 | . 398 | . 014 | . 014 | . 014 | 3.988 | 5.742 | 7.947 | 2.848 | . 500 |
| . 4058 | . 838 | . 546 | . 261 | . 376 | . 014 | . 014 | . 014 | 3.113 | 5.039 | 7.609 | 2.902 | . 500 |
| . 4869 | . 805 | . 497 | . 221 | . 360 | . 013 | . 013 | . 013 | 2.423 | 4.564 | 6.997 | 2.833 | . 500 |
| . 5681 | . 773 | . 456 | . 188 | . 337 | . 012 | . 012 | . 012 | 1.951 | 4.047 | 6.678 | 2.890 | . 500 |
| . 6492 | . 740 | . 420 | . 161 | . 321 | . 012 | . 011 | . 011 | 1.549 | 3.706 | 6.151 | 2.832 | . 500 |
| . 7304 | . 708 | . 390 | . 139 | . 300 | . 011 | . 011 | . 011 | 1.280 | 3.311 | 5.877 | 2.888 | . 500 |
| . 8115 | . 675 | . 363 | . 122 | . 284 | . 010 | . 010 | . 010 | 1.035 | 3.061 | 5.435 | 2.837 | . 500 |
| . 8927 | . 643 | . 340 | . 106 | . 264 | . 010 | . 010 | . 010 | . 878 | 2.754 | 5.209 | 2.890 | . 500 |
| . 9739 | . 610 | . 320 | . 094 | . 249 | . 009 | . 009 | . 009 | . 721 | 2.568 | 4.843 | 2.843 | . 500 |
| 1.0550 | . 578 | . 302 | . 083 | . 231 | . 008 | . 008 | . 008 | . 627 | 2.326 | 4.662 | 2.893 | . 500 |
| 1.1362 | . 546 | . 286 | . 075 | . 217 | . 008 | . 008 | . 008 | . 521 | 2.187 | 4.360 | 2.850 | . 500 |
| 1.2173 | . 513 | . 272 | . 066 | . 199 | . 007 | . 007 | . 007 | . 464 | 1.993 | 4.217 | 2.896 | . 500 |
| 1.2985 | . 481 | . 259 | . 060 | . 186 | . 007 | . 007 | . 007 | . 390 | 1.889 | 3.968 | 2.856 | . 500 |
| 1.3796 | . 448 | . 247 | . 054 | . 170 | . 006 | . 006 | . 006 | . 356 | 1.733 | 3.858 | 2.899 | . 500 |
| 1.4608 | . 416 | . 237 | . 049 | . 157 | . 006 | . 006 | . 006 | . 302 | 1.656 | 3.654 | 2.861 | . 500 |
| 1.5419 | . 383 | . 228 | . 044 | . 142 | . 005 | . 005 | . 005 | . 284 | 1.532 | 3.571 | 2.901 | . 500 |
| 1.6231 | . 351 | . 219 | . 040 | . 130 | . 005 | . 005 | . 005 | . 242 | 1.475 | 3.405 | 2.866 | . 500 |
| 1.7042 | . 318 | . 211 | . 036 | . 115 | . 004 | . 004 | . 004 | . 236 | 1.376 | 3.345 | 2.903 | . 500 |
| 1.7854 | . 286 | . 205 | . 032 | . 103 | . 004 | . 004 | . 004 | . 202 | 1.336 | 3.212 | 2.870 | . 500 |
| 1.8665 | . 253 | . 198 | . 029 | . 089 | . 003 | . 003 | . 003 | . 205 | 1.259 | 3.172 | 2.903 | . 500 |
| 1.9477 | . 221 | . 193 | . 026 | . 078 | . 003 | . 003 | . 003 | . 178 | 1.235 | 3.067 | 2.873 | . 500 |
| 2.0289 | . 188 | . 188 | . 023 | . 065 | . 002 | . 002 | . 002 | . 191 | 1.177 | 3.044 | 2.903 | . 500 |
| 2.1100 | . 156 | . 183 | . 021 | . 053 | . 002 | . 002 | . 002 | . 174 | 1.169 | 2.966 | 2.876 | . 500 |
| 2.1912 | . 124 | . 179 | . 019 | . 040 | . 001 | . 001 | . 001 | . 207 | 1.131 | 2.958 | 2.902 | . 500 |
| 2.2723 | . 091 | . 175 | . 017 | . 029 | . 001 | . 001 | . 001 | . 206 | 1.143 | 2.905 | 2.878 | . 500 |
| 2.3535 | . 059 | . 172 | . 015 | . 017 | . 001 | . 001 | . 001 | . 296 | 1.135 | 2.911 | 2.901 | . 500 |
| 2.4346 | . 026 | . 169 | . 013 | . 006 | . 000 | . 000 | . 000 | . 417 | 1.198 | 2.884 | 2.883 | . 500 |
| 2.5158 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.780 | 1.380 | 2.907 | 2.906 | . 500 |
| 2.5969 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 6.905 | 4.874 | 3.098 | 3.097 | . 500 |
| 2.6781 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 7.022 | 4.423 | 3.075 | 3.074 | . 500 |
| 2.7592 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.897 | . 930 | 2.884 | 2.883 | . 500 |
| 2.8404 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.780 | 1.380 | 2.907 | 2.906 | . 500 |

Figure 8.21: Theoretical pressure head results obtained for $16.5 \mathrm{lt} / \mathrm{s}$ (continued).

|  | . 000 |  |  |  | . 000 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 3.0027 | . 000 | . 167 | . 013 | 0 | 0 | 0 | . 000 | 1 | 3 | 75 | 3.074 | . 500 |
| 39 | 00 | . 167 | . 013 | 00 | . 000 | 000 | . 000 | 2.898 | 931 | 2.88 | 2.883 | . 500 |
| 1650 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.782 | 1.380 | 2.907 | 2.906 | 500 |
| 3.2462 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 4 | 4.872 | 3.098 | 3.098 | 500 |
| 73 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 7.019 | 4.423 | 3.075 | 3.074 | 0 |
| . 4085 | . 000 | . 167 | . 013 | . 00 | . 000 | 000 | 000 | 2.899 | 932 | 2.883 | 2.882 | 500 |
| . 4896 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | 000 | 2.783 | 1.380 | 仡 | 2.906 | 500 |
| 08 | 00 | . 167 | . 013 | . 000 | 00 | . 000 | . 000 | 6.903 | 4.8 | 3.099 | 3.098 | 500 |
| 519 | 00 | 67 | . 13 | 00 | 000 | 000 |  | 7.019 | 4.42 | 3.075 | 74 | 500 |
| 37331 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | . 000 | 2.9 | . 933 | 2.88 | 2.882 | . 500 |
| 142 | . 000 | . 167 | . 013 | . 00 | . 000 | . 000 | . 000 | 2.783 | 1.380 | 2.907 | . 90 | . 500 |
|  | . 000 | . 167 | . 013 | . 000 | 000 |  |  |  |  |  |  |  |
| 3.9766 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | . 000 | 7.0 | 4.4 | 3.075 | 3.074 | . 500 |
| 0577 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | 000 | 2.901 | 933 | 2.883 | 2.883 | 500 |
| 89 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | . 00 | 2.783 | 1.38 | 2.907 | 2.90 | 500 |
| 00 | . 000 | . 167 | . 013 | . 000 | . 00 | . 000 | . 000 | 6.9 | 4.870 | . 098 | 997 | 500 |
| 3012 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | 000 | 7.018 | 4.421 | 3.075 | 3.073 | 0 |
| 3823 |  | . 16 | . 013 | . 000 | 00 | , |  | 2.902 | 93 | 2.88 | 2.88 | 00 |
| 4635 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.78 | 1.38 | 2.907 | 2.907 | . 500 |
| 46 | . 000 | . 16 | . 013 | 000 | . 000 | 000 | 000 | 6.899 | 4.8 | 3.098 | . 097 | 500 |
| 6258 | . | . 16 | . 013 |  | . 00 |  |  | 7.0 | 4.4 | 3.07 | 3.073 |  |
| 7069 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.903 | . 93 | 2.88 | 2.883 | 500 |
| 4.7881 | . 000 | . 16 | . 01 | . 000 | . 00 | . 000 | . 000 | 2.783 | 1.38 | 2.907 | 2.90 |  |
| 93 | . 000 | . 1 | . 013 |  | . 000 | . 000 |  | 6.8 | 4. | 8 | 97 |  |
| 504 | . 000 | . 167 | . 013 | . 000 | . 000 | . 00 | 000 | 7.019 | 4.421 | 3.074 | 3.073 | 0 |
| 316 | . 000 | . 16 | . 01 | 000 | . 000 | . 00 | . 000 | 2.903 | . 935 | 2.885 | 2.88 | . 500 |
| . 1127 | 000 | . 16 | . 013 | . 000 | . 00 | . 000 |  |  | 1.3 | 2.907 | 86 |  |
| 939 | . 000 | . 167 | . 013 | . 000 | . 000 | . 00 | 000 | 6.899 | 4.868 | 3.098 | . 097 |  |
| 2750 | . 000 | . 16 | . 013 |  | . 000 | .000 | . 000 | 7.01 | 4.420 | 3.074 | 3.07 | . 500 |
| 562 | . 000 | . 167 | . 013 | . 000 | . 000 | 00 |  | 2.9 | 36 | 2.885 | 2.88 | 500 |
| 5.4373 | . 000 | . 16 | . 01 | . 000 | . 00 | . 000 | 000 | 2.784 | 1.38 | 2.90 | 2.90 | 500 |
| 85 | . 000 | . 167 | . 013 |  | . 000 | , | , | 6.899 | 4.86 | 3.098 | . 09 | 500 |
| 5996 | . 000 | . 167 | . 013 | 000 | 000 | . 000 | . 000 | 7.018 | 4.42 | 3.07 | 3.074 | 00 |
| 808 | . 000 | . 167 | . 013 | 000 | . 00 | 000 | . 000 | 2.903 | . 936 | 2.88 | 2.88 | . 500 |
| . 7619 | . 000 | . 167 | . 013 |  | . 000 |  | , | 2.78 | 1.38 | 2.907 | 2.906 | . 500 |
| 1 | . 00 | 67 | . 013 | 000 | 00 | 00 | . 000 | 6.899 | 4.86 | 3.098 | 3.096 |  |
| . 9243 | . 000 | . 16 | . 013 | . 000 | . 00 | . 000 | . 000 | 7.017 | 4.419 | 3.07 | 3.074 | 500 |
| 054 | . 000 | . 167 | . 013 |  | . 000 | , | , | 2.903 | 93 | 2.88 | 2.88 | . 50 |
| 68 | 000 | . 167 | 13 | 000 | 000 | . 00 |  | 2.785 | , 38 | . 907 | . 906 | - |
| 7 | . 000 | . 16 | . 013 |  | . 000 | 000 | . 000 | 6.898 | 4.86 | 3.0 | 3.097 |  |
| 2489 | . 000 | . 167 | . 013 | . 000 | 000 | . 000 | . 000 | 7.017 | 4.419 | 3.07 | 3.074 | . 00 |
| 00 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | 000 | 2.904 | 938 | 2.88 | 2.883 | 500 |
| 12 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | , | 2.786 | 1.38 | 2.907 | 2.906 | . 500 |
| . 4923 | . 000 | 167 | . 013 | . 000 | 000 | . 000 | . 000 | 6.898 | 4.866 | . 09 | 3.097 | 00 |
| , | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 0 | 7.016 | 4.419 | 3.075 | 3.074 | 500 |
| 6546 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | , | 2.904 | . 939 | 2.88 | 2.883 | 500 |
| .7358 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | . 000 | 2.786 | 1.384 | 2.907 | 2.906 | . 500 |
| 6.8170 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 6.897 | 4.865 | 3.09 | 3.097 | , |
| 8981 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 7.015 | 4.418 | 3.075 | 3.074 | . 500 |
| . 9793 | . 000 | . 167 | . 013 | . 000 | . 000 | 000 | . 000 | 2.904 | . 939 | 2.884 | 2.883 | 500 |
| 7.0604 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 2.787 | 1.385 | 2.906 | 2.906 | 500 |
| 7.1416 | . 000 | . 167 | . 013 | . 000 | . 000 | . 000 | . 000 | 6.897 | 4.864 | 3.098 | 3.097 | 500 |
| . 2227 | 000 | 16 | . 01 | . 000 | 000 | . 000 | . 000 | 7.015 | 4.418 | 3.0 | 3.07 | , |

Figure 8.21: Theoretical pressure head results obtained for $16.5 \mathrm{lt} / \mathrm{s}$ (continued).


Figure 8.22: Theoretical pressure head results obtained for $16.5 \mathrm{lt} / \mathrm{s}$.


Figure 8.23: Theoretical transient discharges obtained for $16.5 \mathrm{lt} / \mathrm{s}$.


Figure 8.24: Theoretical pressure head results obtained for $19.14 \mathrm{lt} / \mathrm{s}$.


Figure 8.25: Theoretical transient discharges obtained for $19.14 \mathrm{lt} / \mathrm{s}$.


Figure 8.26: Theoretical pressure head results obtained for $20.55 \mathrm{lt} / \mathrm{s}$.


Figure 8.27: Theoretical transient discharges obtained for $20.55 \mathrm{lt} / \mathrm{s}$.

### 8.2.3 Results of computation performed for upper steady state discharges

In these computations, moderate steady state discharges are taken as $24.51 \mathrm{lt} / \mathrm{s}$ $(2.00 \mathrm{~m} / \mathrm{s}), 25.16 \mathrm{lt} / \mathrm{s}(2.06 \mathrm{~m} / \mathrm{s})$. The output and results are shown below.

The numerical solution result is given in Figure 8.28. The graph of pressure heads and discharge values at measurement points drawn according to the results of numerical solution are given from Figure 8.29 to 8.32 , respectively.

```
******************
* InItIAL VALUES *
******************
```

| EL (M) = | 4.6000 |
| :---: | :---: |
| $\mathrm{F}=\quad .0217$ |  |
| TM (S)= | 12.0000 |
| HR (M) = | 11.0000 |
| TOL= | . 0002 |
| DX (RAD) = | . 0714 |
| AKV1 = | 2.9000 |
| TAU(1)= | 1.0000 |
| TAU(4)= | . 8421 |
| $\operatorname{TAU}(7)=$ | . 6842 |
| TAU(10)= | . 5263 |
| TAU(13)= | . 3684 |
| $\operatorname{TAU}(16)=$ | . 2105 |
| TAU(19)= | . 0526 |
| 1.0000 |  |
| HMIN= | . 0000 |
| $\mathrm{K}=0$ |  |
| CCT (S)= | 6.0000 |
| DT (S)= | . 0019 |


| A (M3/S) $=1331.0000$ |  |
| :---: | :---: |
| D ( M ) $=$ | . 1250 |
| RN (RPM) = | $)=1450.0000$ |
| QR (M3/S) = | S $=$. 0360 |
| VI= . 7 | . 7000 |
| CK = 2 . | 2.0000 |
| AKV2= 15 | 15.6200 |
| $\operatorname{TAU}(2)=$ | . 9474 |
| TAU(5)= | . 7895 |
| TAU(8)= | . 6316 |
| TAU(11)= | $=.4737$ |
| TAU(14)= | $=.3158$ |
| TAU(17)= | $=.1579$ |
| TAU(20)= | $=\quad .0000$ |
| AL0= 1 . | 1.0000 |
| $\mathrm{T}=\quad .00$ | . 0000 |
| NSV = 42 |  |
| NTAU $=20$ |  |
| AKS (M)= | $=$. 0002 |


| XL (M) = 108.0000 |  |
| :---: | :---: |
| G (M/S2) = | $)=9.8100$ |
| TR (NT.M) = | M $=32.1800$ |
| WRR (NT.M2) = | . M 2$)=$. 8300 |
| $\mathrm{N}=42$ |  |
| $\mathrm{PI}=3.1416$ |  |
| DTAU (S)= | ) = . 3158 |
| TAU(3)= | . 8947 |
| $\operatorname{TAU}(6)=$ | . 7368 |
| $\operatorname{TAU}(9)=$ | . 5789 |
| $\operatorname{TAU}(12)=$ | $=.4211$ |
| $\operatorname{TAU}(15)=$ | $=.2632$ |
| TAU(18) = | $=$. 1053 |
| M2=27 |  |
| AL00= 1.0000 |  |
| JPR= 21 |  |
| KIT= 5 |  |
| QQA (M3/S)= | S $)=$. 0252 |
| VN (M2/S)= | $S)=.000001$ |

$* * * * * * * * * * * * * * * * *$
$*$ RESULT VALUES
$* * * * * * * * * * * * * * * * * * * *)$

```
HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE
HP2=PIEZOMETRIC HEAD IN METERS AFTER VALVE
HP=HEADLOSS IN METERS AT VALVE (NODE NSV)
HP1 HP2 HP 7.9775 4.6937 3.2838
```

NUMBER OF NODES AND PIEZOMETRIC HEADS AT NODES AS METERS :

| $1:$ | 11.82 | $2:$ | 11.72 | $3:$ | 11.63 | $4:$ | 11.54 | $5:$ | 11.44 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | 11.35 | $7:$ | 11.26 | $8:$ | 11.16 | $9:$ | 11.07 | $10:$ | 10.97 |
| $11:$ | 10.88 | $12:$ | 10.79 | $13:$ | 10.69 | $14:$ | 10.60 | $15:$ | 10.51 |
| $16:$ | 10.41 | $17:$ | 10.32 | $18:$ | 10.23 | $19:$ | 10.13 | $20:$ | 10.04 |

Figure 8.28: Theoretical pressure head results obtained for $24.51 \mathrm{lt} / \mathrm{s}$.

| 21: | 9.94 | $22:$ | 9.85 | $23:$ | 9.76 | $24:$ | 9.66 | $25:$ |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 26: | 9.48 | $27:$ | 9.38 | $28:$ | 9.29 | $29:$ | 9.20 | $30:$ |
| $31:$ | 9.01 | $32:$ | 8.91 | $33:$ | 8.82 | $34:$ | 8.73 | $35:$ |
| $36:$ | 8.54 | $37:$ | 8.45 | $38:$ | 8.35 | $39:$ | 8.26 | $40:$ |
| $41:$ | 8.07 | $42:$ | 4.69 | $43:$ | 4.60 |  | 8.63 |  |
|  |  |  |  |  |  |  |  |  |

ZH(I) VALUES AS METERS :

| $1:$ | -.30 | $2:$ | .05 | $3:$ | .12 | $4:$ | .21 | $5:$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $6:$ | .10 | $7:$ | .15 | $8:$ | .30 | $9:$ | 1.39 | $10:$ |
| $11:$ | 1.39 | $12:$ | 1.34 | $13:$ | 1.35 | $14:$ | 1.36 | $15:$ |
| $16:$ | 1.35 | $17:$ | 1.35 | $18:$ | 1.33 | $19:$ | 1.32 | $20:$ |
| $21:$ | 1.34 | $22:$ | 1.34 | $23:$ | 1.30 | $24:$ | 1.28 | $25:$ |
| $26:$ | 1.41 | $27:$ | 1.70 | $28:$ | 1.60 | $29:$ | 1.60 | $30:$ |
| $31:$ | 1.60 | $32:$ | 1.60 | $33:$ | 1.60 | $34:$ | 1.60 | $35:$ |
| $36:$ | 1.61 | $37:$ | 1.61 | $38:$ | 1.60 | $39:$ | 1.60 | $40:$ |
| $41:$ | 1.61 | $42:$ | 1.61 | $43:$ | 4.10 |  |  | 1.60 |
| 1.68 |  |  |  |  |  |  |  |  |


| TIME | TAU | ALPHA | BETA | V | Q(1) | Q(M2) Q | Q(NS) H | HPH(1) H | HPH(M2) | HP11 | HPH(NSV) | S) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| . 0000 | 1.000 | 1.000 | . 879 | . 692 | . 025 | . 025 | . 025 | 12.118 | 87.682 | 26.367 | 73.084 | . 500 |
| . 0812 | . 986 | . 844 | . 664 | . 682 | . 025 | . 025 | . 025 | 7.997 | 5.919 | 6.367 | 3.084 | . 500 |
| . 1623 | . 973 | . 725 | . 514 | . 675 | . 024 | . 024 | . 024 | 5.300 | 5.024 | 6.031 | 2.928 | . 500 |
| . 2435 | . 959 | . 633 | . 401 | . 652 | . 023 | . 024 | . 024 | 3.596 | 4.125 | 5.965 | 2.980 | . 500 |
| . 3246 | . 946 | . 561 | . 316 | . 637 | . 023 | . 023 | . 023 | 2.348 | 3.679 | 5.621 | 2.866 | . 500 |
| . 4058 | . 932 | . 503 | . 254 | . 610 | . 022 | . 022 | . 022 | 1.603 | 3.158 | 5.541 | 2.938 | . 500 |
| . 4869 | . 919 | . 456 | . 209 | . 593 | . 021 | . 021 | . 021 | 1.028 | 2.944 | 5.223 | 2.844 | . 500 |
| . 5681 | . 905 | . 417 | . 174 | . 565 | . 020 | . 020 | . 021 | . 705 | 2.598 | 5.146 | 2.919 | . 500 |
| . 6492 | . 892 | . 385 | . 148 | . 547 | . 020 | . 020 | . 020 | . 435 | 2.504 | 4.867 | 2.840 | . 500 |
| . 7304 | . 878 | . 357 | . 127 | . 521 | . 019 | . 019 | . 019 | . 300 | 2.246 | 4.804 | 2.909 | . 500 |
| . 8115 | . 865 | . 333 | . 110 | . 503 | . 018 | . 018 | . 018 | . 158 | 2.200 | 4.552 | 2.839 | . 500 |
| . 8927 | . 851 | . 312 | . 096 | . 478 | . 017 | . 017 | . 017 | . 109 | 1.994 | 4.501 | 2.906 | . 500 |
| . 9739 | . 838 | . 294 | . 085 | . 461 | . 017 | . 017 | . 016 | . 036 | 1.980 | 4.281 | 2.839 | . 500 |
| 1.0550 | . 824 | . 277 | . 076 | . 438 | . 016 | . 016 | . 016 | . 030 | 1.812 | 4.245 | 2.904 | . 500 |
| 1.1362 | . 811 | . 263 | . 068 | . 422 | . 015 | . 015 | . 015 | -. 005 | 1.815 | 4.054 | 2.845 | . 500 |
| 1.2173 | . 797 | . 250 | . 061 | . 400 | . 014 | . 014 | . 015 | . 012 | 1.676 | 4.023 | 2.906 | . 500 |
| 1.2985 | . 784 | . 238 | . 055 | . 385 | . 014 | . 014 | . 014 | -. 002 | 1.686 | 3.853 | 2.850 | . 500 |
| 1.3796 | . 770 | . 227 | . 050 | . 365 | . 013 | . 013 | . 013 | . 026 | 1.568 | 3.835 | 2.906 | . 500 |
| 1.4608 | . 757 | . 217 | . 046 | . 350 | . 013 | . 013 | . 013 | . 025 | 1.584 | 3.685 | 2.856 | . 500 |
| 1.5419 | . 743 | . 208 | . 043 | . 331 | . 012 | . 012 | . 012 | . 058 | 1.483 | 3.674 | 2.911 | . 500 |
| 1.6231 | . 729 | . 200 | . 039 | . 318 | . 011 | . 011 | . 011 | . 064 | 1.507 | 3.540 | 2.862 | . 500 |
| 1.7042 | . 716 | . 192 | . 036 | . 299 | . 011 | . 011 | . 011 | . 098 | 1.418 | 3.538 | 2.913 | . 500 |
| 1.7854 | . 702 | . 185 | . 034 | . 287 | . 010 | . 010 | . 010 | . 108 | 1.441 | 3.418 | 2.864 | . 500 |
| 1.8665 | . 689 | . 179 | . 032 | . 269 | . 010 | . 010 | . 010 | . 142 | 1.367 | 3.421 | 2.914 | . 500 |
| 1.9477 | . 675 | . 172 | . 029 | . 257 | . 009 | . 009 | . 009 | . 154 | 1.389 | 3.313 | 2.869 | . 500 |
| 2.0289 | . 662 | . 167 | . 028 | . 240 | . 009 | . 009 | . 009 | . 186 | 1.323 | 3.321 | 2.917 | . 500 |
| 2.1100 | . 648 | . 161 | . 026 | . 229 | . 008 | . 008 | . 008 | . 199 | 1.348 | 3.226 | 2.871 | . 500 |
| 2.1912 | . 635 | . 156 | . 024 | . 213 | . 008 | . 008 | . 008 | . 229 | 1.289 | 3.235 | 2.917 | . 500 |
| 2.2723 | . 621 | . 151 | . 023 | . 202 | . 007 | . 007 | . 007 | . 242 | 1.312 | 3.149 | 2.877 | . 500 |
| 2.3535 | . 608 | . 147 | . 022 | . 187 | . 007 | . 007 | . 007 | . 272 | 1.261 | 3.161 | 2.917 | . 500 |
| 2.4346 | . 594 | . 142 | . 020 | . 176 | . 006 | . 006 | . 006 | . 286 | 1.285 | 3.084 | 2.878 | . 500 |
| 2.5158 | . 581 | . 138 | . 019 | . 161 | . 006 | . 006 | . 006 | . 313 | 1.237 | 3.100 | 2.916 | . 500 |
| 2.5969 | . 567 | . 135 | . 018 | . 151 | . 005 | . 005 | . 005 | . 326 | 1.266 | 3.033 | 2.879 | . 500 |
| 2.6781 | . 554 | . 131 | . 017 | . 137 | . 005 | . 005 | . 005 | . 351 | 1.223 | 3.048 | 2.914 | . 500 |
| 2.7592 | . 540 | . 128 | . 016 | . 126 | . 005 | . 004 | . 004 | . 363 | 1.252 | 2.989 | 2.880 | . 500 |
| 2.8404 | . 527 | . 124 | . 015 | . 113 | . 004 | . 004 | . 004 | . 382 | 1.211 | 3.008 | 2.916 | . 500 |
| 2.9216 | . 513 | . 121 | . 014 | . 103 | . 004 | . 004 | . 004 | . 391 | 1.237 | 2.952 | 2.882 | . 500 |

Figure 8.28: Theoretical pressure head results obtained for $24.51 \mathrm{lt} / \mathrm{s}$ (continued).

| 3.0027 | . 50 | . 119 | . 013 | . 089 | . 003 | . 003 | . 003 | 409 | 1.198 | 2.975 | 2.917 | . 500 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 3.0839 | . 486 | . 116 | . 012 | . 079 | . 003 | . 003 | . 003 | . 417 | 1.228 | 2.925 | 2.884 | . 500 |
| 3.1650 | . 472 | . 114 | . 011 | . 066 | . 002 | . 002 | . 002 | . 431 | 1.195 | 2.949 | 2.917 | . 500 |
| 3.2462 | . 459 | . 112 | . 010 | . 057 | . 002 | . 002 | . 002 | . 437 | 1.220 | 2.904 | 2.886 | 500 |
| 3.3273 | . 445 | . 110 | . 008 | . 044 | . 002 | . 002 | . 002 | . 447 | 1.190 | 2.933 | 2.919 | . 500 |
| 3.4085 | . 432 | . 108 | . 008 | . 034 | . 001 | . 001 | . 001 | . 451 | 1.217 | 2.893 | 2.885 | . 500 |
| 3.4896 | . 418 | . 107 | . 006 | . 021 | . 001 | . 001 | . 001 | 456 | 1.189 | 2.921 | 2.918 | . 500 |
| 3.5708 | . 405 | . 106 | . 006 | . 012 | . 000 | . 000 | . 000 | 456 | 1.215 | 2.887 | 2.883 | . 500 |
| 3.6519 | . 391 | . 105 | . 005 | -. 001 | . 000 | . 000 | . 000 | . 455 | 1.190 | 2.918 | 2.918 | . 500 |
| 3.7331 | . 378 | . 104 | . 004 | -. 011 | . 000 | . 000 | . 000 | . 455 | 1.213 | 2.883 | 2.885 | 500 |
| 3.8142 | . 364 | . 103 | . 004 | -. 024 | -. 001 | -. 001 | -. 001 | . 463 | 1.189 | 2.915 | 2.918 | . 500 |
| 3.8954 | . 351 | . 102 | . 004 | -. 033 | -. 001 | -. 001 | -. 001 | . 474 | 1.215 | 2.877 | 2.882 | . 500 |
| 3.9766 | . 337 | . 101 | . 004 | -. 046 | -. 002 | -. 002 | -. 002 | . 498 | 1.196 | 2.906 | 2.918 | 500 |
| 4.0577 | . 324 | . 101 | . 005 | -. 055 | -. 002 | -. 002 | -. 002 | . 524 | 1.229 | 2.864 | 2.887 | . 500 |
| 4.1389 | . 310 | . 100 | . 006 | -. 067 | -. 002 | -. 002 | -. 002 | . 573 | 1.217 | 2.891 | 2.917 | 500 |
| 4.2200 | . 297 | . 098 | . 006 | -. 076 | -. 003 | -. 003 | -. 003 | . 621 | 1.256 | 2.850 | 2.892 | . 500 |
| 4.3012 | . 283 | . 097 | . 008 | -. 088 | -. 003 | -. 003 | -. 003 | . 701 | 1.252 | 2.868 | 2.922 | . 500 |
| 4.3823 | . 270 | . 095 | . 010 | -. 097 | -. 003 | -. 004 | -. 004 | . 778 | 1.301 | 2.831 | 2.896 | . 500 |
| 4.4635 | . 256 | . 093 | . 012 | -. 108 | -. 004 | -. 004 | -. 004 | . 898 | 1.311 | 2.849 | 2.924 | . 500 |
| 4.5446 | . 243 | . 090 | . 014 | -. 116 | -. 004 | -. 004 | -. 004 | 1.015 | 1.373 | 2.810 | 2.904 | . 500 |
| 4.6258 | . 229 | . 087 | . 016 | -. 127 | -. 005 | -. 005 | -. 005 | 1.194 | 1.406 | 2.823 | 2.928 | . 500 |
| 4.7069 | . 216 | . 084 | . 018 | -. 134 | -. 005 | -. 005 | -. 005 | 1.372 | 1.492 | 2.790 | 2.913 | . 500 |
| 4.7881 | . 202 | . 080 | . 020 | -. 143 | -. 005 | -. 005 | -. 005 | 1.632 | 1.554 | 2.801 | 2.935 | . 500 |
| 4.8693 | . 188 | . 076 | . 022 | -. 148 | -. 005 | -. 005 | -. 005 | 1.899 | 1.676 | 2.771 | 2.926 | . 500 |
| 4.9504 | . 175 | . 071 | . 024 | -. 156 | -. 006 | -. 006 | -. 006 | 2.281 | 1.785 | 2.788 | 2.950 | . 500 |
| 5.0316 | . 161 | . 066 | . 025 | -. 160 | -. 006 | -. 006 | -. 006 | 2.683 | 1.959 | 2.768 | 2.946 | . 500 |
| 5.1127 | . 148 | . 061 | . 026 | -. 165 | -. 006 | -. 006 | -. 006 | 3.243 | 2.137 | 2.789 | 2.971 | . 500 |
| 5.1939 | . 134 | . 056 | . 026 | -. 166 | -. 006 | -. 006 | -. 006 | 3.854 | 2.395 | 2.779 | 2.972 | 500 |
| 5.2750 | . 121 | . 050 | . 026 | -. 167 | -. 006 | -. 006 | -. 006 | 4.685 | 2.676 | 2.810 | 3.002 | . 500 |
| 5.3562 | . 107 | . 045 | . 025 | -. 165 | -. 006 | -. 006 | -. 006 | 5.615 | 3.067 | 2.823 | 3.012 | . 500 |
| 5.4373 | . 094 | . 040 | . 024 | -. 161 | -. 006 | -. 006 | -. 006 | 6.851 | 3.506 | 2.868 | 3.050 | . 500 |
| 5.5185 | . 080 | . 036 | . 021 | -. 152 | -. 005 | -. 006 | -. 006 | 8.263 | 4.082 | 2.909 | 3.076 | . 500 |
| 5.5996 | . 067 | . 031 | . 018 | -. 141 | -. 005 | -. 005 | -. 005 | 10.076 | 64.756 | 62.978 | 83.120 | . 500 |
| 5.6808 | . 053 | . 028 | . 014 | -. 124 | -. 004 | -. 005 | -. 005 | 12.152 | 25.589 | 9 3.052 | 23.164 | 00 |
| 5.7619 | . 040 | . 026 | . 010 | -. 102 | -. 004 | -. 004 | -. 004 | 14.668 | 86.550 | 03.146 | 63.224 | . 500 |
| 5.8431 | . 026 | . 024 | . 005 | -. 074 | -. 003 | -. 003 | -. 003 | 17.417 | 77.643 | 33.244 | 43.287 | . 500 |
| 5.9243 | . 013 | . 024 | . 001 | -. 039 | -. 001 | -. 001 | -. 002 | 20.312 | 28.780 | 03.346 | 63.355 | . 500 |
| 6.0054 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | 21.649 | 9.788 | 3.419 | 3.418 | . 500 |
| 6.0866 | . 000 | . 023 | . 000 | . 000 | . 000 | . 001 | . 002 | 4.808 | 2.390 | 2.634 | 2.623 | . 500 |
| 6.1677 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | -11.818 | -3.981 | 12.561 | 12.564 | . 500 |
| 6.2489 | . 000 | . 023 | . 000 | . 000 | . 000 | -. 001 | -. 002 | 4.992 | 3.420 | 3.372 | 3.355 | . 500 |
| 6.3300 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | 21.634 | 9.780 | 3.418 | 3.418 | . 500 |
| 6.4112 | . 000 | . 023 | . 000 | . 000 | . 000 | . 001 | . 002 | 4.810 | 2.392 | 2.636 | 2.623 | . 500 |
| 6.4923 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | -11.798 | -3.973 | 32.564 | 2.564 | . 500 |
| 6.5735 | . 000 | . 023 | . 000 | . 000 | . 000 | -. 001 | -. 002 | 4.991 | 3.418 | 3.367 | 3.358 | . 500 |
| 6.6546 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | 21.609 | 9.776 | 3.415 | 3.415 | . 500 |
| 6.7358 | . 000 | . 023 | . 000 | . 000 | . 000 | . 001 | . 002 | 4.808 | 2.395 | 2.637 | 2.623 | . 500 |
| 6.8170 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | -11.774 | -3.970 | 02.565 | 2.566 | . 500 |
| 6.8981 | . 000 | . 023 | . 000 | . 000 | . 000 | -. 001 | -. 002 | 4.992 | 3.415 | 3.368 | 3.355 | . 500 |
| 6.9793 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | 21.587 | 9.770 | 3.416 | 3.415 | . 500 |
| 7.0604 | . 000 | . 023 | . 000 | . 000 | . 000 | . 001 | . 002 | 4.809 | 2.394 | 2.636 | 2.623 | . 500 |
| 7.1416 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | -11.756 | -3.966 | 2.566 | 2.566 | . 500 |
| 7.2227 | . 000 | . 023 | . 000 | . 000 | . 000 | -. 001 | -. 002 | 4.988 | 3.417 | 3.367 | 3.353 | . 500 |
| . 3039 | . 000 | . 023 | . 000 | . 000 | . 000 | . 000 | . 000 | 21.570 | 9.769 | 3.415 | 3.412 | . 500 |

Figure 8.28: Theoretical pressure head results obtained for $24.51 \mathrm{lt} / \mathrm{s}$ (continued).


Figure 8.29: Theoretical pressure head results obtained for $24.51 \mathrm{lt} / \mathrm{s}$.


Figure 8.30: Theoretical transient discharges obtained for $24.51 \mathrm{lt} / \mathrm{s}$.


Figure 8.31: Theoretical pressure head results obtained for $25.16 \mathrm{lt} / \mathrm{s}$.


Figure 8.32: Theoretical transient discharges obtained for $25.16 \mathrm{lt} / \mathrm{s}$.

## CHAPTER NINE <br> COMPARISON OF EXPERIMENTAL AND THEORETICAL RESULTS

Experimental results and theoretical ones had been presented in chapter seven and chapter eight respectively. In this chapter, experimental results and theoretical results will be presented in the same figure for comparing.

### 9.1 Comparison of experimental results with those obtained from computations performed for the first system.

### 9.1.1 Comparison of experimental results with theoretical results performed for lower steady state discharges

In the case of discharges of $6.2 \mathrm{lt} / \mathrm{s}(0.5 \mathrm{~m} / \mathrm{s})$ and $8.16 \mathrm{lt} / \mathrm{s}(0.67 \mathrm{~m} / \mathrm{s})$, the graphs corresponding to experimental and numerical results are presented in Figures 9.1 and 9.2 respectively.
comparison of measurement and computation for $6.2 \mathrm{lt} / \mathrm{s}$ steady state discharge


Figure 9.1: Pressure heads correspond to $6.2 \mathrm{lt} / \mathrm{s}$ steady state discharge.

## Comparison of measurement and computation for $8.16 \mathrm{lt} / \mathrm{s}$ steady state discharge



Figure 9.2: Pressure heads correspond to $8.16 \mathrm{lt} / \mathrm{s}$ steady state discharge.

### 9.1.2 Comparison of experimental results with theoretical results performed for moderate steady state discharges

In these comparisons, moderate steady state discharges are taken as $12.31 \mathrm{lt} / \mathrm{s}(1$ $\mathrm{m} / \mathrm{s}), 18.32 \mathrm{lt} / \mathrm{s}(1.5 \mathrm{~m} / \mathrm{s})$ and $22.63 \mathrm{lt} / \mathrm{s}(1.85 \mathrm{~m} / \mathrm{s})$. The graphs corresponding to experimental and numerical results are presented from Figures 9.3 to 9.5 respectively.

## Comparison of measurement and computation for $\mathbf{1 2 . 3 1} \mathbf{l t / s}$ steady state discharge



Figure 9.3: Pressure heads correspond to $12.31 \mathrm{lt} / \mathrm{s}$ steady state discharge.

## comparison of measurement and computation for $18.32 \mathrm{lt} / \mathrm{s}$ steady state discharge



Figure 9.4: Pressure heads correspond to $18.32 \mathrm{lt} / \mathrm{s}$ steady state discharge.

## Comparison of measurement and computation for $22.63 \mathrm{lt} / \mathrm{s}$ steady state discharge



Time (s)

Figure 9.5: Pressure heads correspond to $22.63 \mathrm{lt} / \mathrm{s}$ steady state discharge.

### 9.1.3 Comparison of experimental results with theoretical results performed for

 upper steady state dischargeIn this comparison, upper steady state discharge was taken as $39.00 \mathrm{lt} / \mathrm{s}(3.2 \mathrm{~m} / \mathrm{s})$. The graph corresponding to experimental and numerical results is presented in Figure 9.6.

Similarly, red line corresponds to computation results and blue line corresponds to measurement results.

## comparison of measurement and computation for $39 \mathrm{lt} / \mathrm{s}$ steady state discharge



Figure 9.6: Pressure heads correspond to $39 \mathrm{lt} / \mathrm{s}$ steady state discharge.

### 9.2 Comparison of experimental results with those obtained from computations performed for the second system

In this section, experimental results and theoretical results will be compared to each other and exhibited on the same graph for system two. We have computations for lower, normal and upper steady state discharges for system two. However we have two measuring point in system two, therefore we drawn the graphics separately for first point and second point.

We realised comparisons for lower steady state discharges as 6.6 and $9.01 \mathrm{lt} / \mathrm{s}$, normal steady state discharges as $16.50,19.14$ and $20.55 \mathrm{lt} / \mathrm{s}$, upper steady state discharges as 24.51 and $25.16 \mathrm{lt} / \mathrm{s}$ according to system 2.

### 9.2.1 Comparison of experimental results with theoretical results performed for lower steady state discharge

The comparisons are made for two measurements points. As mentioned, the first point is placed just after the pump while the second one corresponds to the interior point which is 66.7 m far from the upstream end.

In this comparison, lower steady state discharges are taken as $6.6 \mathrm{lt} / \mathrm{s}(0.54 \mathrm{~m} / \mathrm{s})$ and $9.02 \mathrm{lt} / \mathrm{s}(0.74 \mathrm{~m} / \mathrm{s})$. The graphs corresponding to experimental and numerical results are presented in Figures 9.7, 9.8, 9.9 and 9.10 respectively.

## Comparison of measurement and computation for $6.6 \mathrm{lt} / \mathrm{s}$ steady state discharge for node 1



Figure 9.7: Pressure heads correspond to $6.6 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 1 (check valve closing time, $\mathrm{CCT}=1.9 \mathrm{~s}$ ).

Comparison of measurement and computation for $6.6 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point


Time (s)

Figure 9.8: Pressure heads correspond to $6.6 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 2 (check valve closing time, $\mathrm{CCT}=1.9 \mathrm{~s}$ ).
comparison of measurement and computation for $9.01 \mathrm{lt} / \mathrm{s}$ staedy state discharge for point 1


Figure 9.9: Pressure heads correspond to $9.01 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 1 (check valve closing time, $\mathrm{CCT}=2 \mathrm{~s}$ ).


Figure 9.10: Pressure heads correspond to $9.01 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 2.1 (check valve closing time, $\mathrm{CCT}=2 \mathrm{~s}$ ).

### 9.2.2 Comparison of experimental results with theoretical results performed for moderate steady state discharges

The comparisons are made for two measurements points. As mentioned in Section 9.2.1., the first point is placed just after the pump while the second one corresponds to the interior point which is 66.7 m far from the upstream end.

In these comparisons, moderate steady state discharges are taken as $16.5 \mathrm{lt} / \mathrm{s}$ $(1.35 \mathrm{~m} / \mathrm{s}), 19.14 \mathrm{lt} / \mathrm{s}(1.57 \mathrm{~m} / \mathrm{s})$ and $20.55 \mathrm{lt} / \mathrm{s}(1.68 \mathrm{~m} / \mathrm{s})$.

The graphs corresponding to experimental and numerical results are presented in Figure 9.11, 9.12, 9.13, 9.14, 9.15 and 9.16 respectively.
comparison of measurement and computation for $16.50 \mathrm{lt} / \mathrm{s}$ steady state discharge for point 1


Figure 9.11: Pressure heads correspond to $16.5 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 1 (check valve closing time, $\mathrm{CCT}=2.5 \mathrm{~s}$ ).
comparison of measurement and computation for $16.50 \mathrm{lt} / \mathrm{s}$ steady state discharge for point 2

time (s)

Figure 9.12: Pressure heads correspond to $16.5 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 2 (check valve closing time, $\mathrm{CCT}=2.5 \mathrm{~s}$ ).

## Comparison of measurement and computation for $19.14 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point



Figure 9.13: Pressure heads correspond to $19.14 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 1 (check valve closing time, $\mathrm{CCT}=2.75 \mathrm{~s}$ ).
comparison of measurement and computation for $19.14 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point


Figure 9.14: Pressure heads correspond to $19.14 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 2 (check valve closing time, $\mathrm{CCT}=2.75 \mathrm{~s}$ ).

## Comparison of measurement and computation for $20.55 \mathrm{lt} / \mathrm{s}$ steady state

 discharge for first measuring point

Figure 9.15: Pressure heads correspond to $20.55 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 1 (check valve closing time, $\mathrm{CCT}=3 \mathrm{~s}$ ).
comparison of measurement and computation for $20.55 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point


Figure 9.16: Pressure heads correspond to $20.55 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point 2 (check valve closing time, $\mathrm{CCT}=3 \mathrm{~s}$ ).

### 9.2.3 Comparison of experimental results with theoretical results performed for

 upper steady state dischargesIn these comparisons, upper steady state discharges are taken as $24.51 \mathrm{lt} / \mathrm{s}(2 \mathrm{~m} / \mathrm{s})$ and $25.16 \mathrm{lt} / \mathrm{s}(2.06 \mathrm{~m} / \mathrm{s})$. The graphs corresponding to experimental and numerical results are presented in Figure 9.17, 9.18, 9.19 and 9.20 respectively.

## Comparison of measurement and computation for $24.51 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point



Figure 9.17: Pressure heads correspond to $24.51 \mathrm{lt} / \mathrm{s}$ discharge for point $1(\mathrm{CCT}=6 \mathrm{~s})$.

Comparison of measurement and computation for $24.51 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point


Figure 9.18: Pressure heads correspond to $24.51 \mathrm{lt} / \mathrm{s}$ discharge for point $2(\mathrm{CCT}=6 \mathrm{~s})$.

Comparison of measurement and computation for $25.16 \mathrm{lt} / \mathrm{s}$ steady state discharge for first measuring point


Figure 9.19: Pressure heads correspond to $25.16 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point $1, \mathrm{CCT}=8 \mathrm{~s}$

Comparison of measurement and computation for $25.16 \mathrm{lt} / \mathrm{s}$ steady state discharge for second measuring point


Figure 9.20: Pressure heads correspond to $25.16 \mathrm{lt} / \mathrm{s}$ steady state discharge for measuring point $2, \mathrm{CCT}=8 \mathrm{~s}$.

## CHAPTER TEN CONCLUSIONS

In practice, it is of vital importance to know the unsteady flow conditions in the hydraulic systems whose designs are generally based on steady flow conditions. That's why, in this study it is aimed to build experimental systems involving a pump in order to study the behaviour of such systems by taking into consideration the boundary conditions described by pump, valve, check valve,...etc.

Two different experimental systems are installed in the Hydraulics Laboratory of Civil Engineering Department, Dokuz Eylül University. These two systems are built to study the unsteady flow due to pump failure. The discharge pipe of the second experimental set up is longer and there exist a check valve after the pump.

The minor losses due to valve, check valve and elbows are determined experimentally by using $U$ tube and gauge manometers. The discharges are measured by means of triangular weir manufactured in the Laboratory. The pressure heads are measured by pressure transient data logger. The pump is shut down manually.

In Wylie \& Streeter (1993), there exist 89 pump dimensionless parameters, WH and WB , in terms of various specific speeds, $\mathrm{N}_{\mathrm{s}}$. The 89 pump dimensionless parameters of WH and WB which are used in our study, are obtained by interpolation based on our pump's specific speed, $\mathrm{N}_{\mathrm{s}}$ ( 45.5 SI ). These founded pump dimensionless parameters are used in Fortran computations as input values. The interpolation is realised between pump dimensionless parameters of $\mathrm{N}_{\mathrm{s}}=25$ SI and $\mathrm{N}_{\mathrm{s}}=147$ SI.

The computer program given in Wylie \& Streeter, 1993 is used after the modifications required in order to adapt to our experimental systems. This program is based on the common Method of Characteristics which solves the ordinary differential equations by using the finite differences technique. The explicit approach is used with the Courant's criterion. After the calculation of piezometric heads in the
computer programs, elevations according to datum are subtracted from piezometric heads, therefore the results are transformed to pressure heads $(\mathrm{P} / \gamma)$, in order to compare directly with experimental results.

Some water hammer pressure waves are observed in the measurements in the second system where there exists a check valve.

It is observed that the theoretical and experimental results are in an acceptable accord in the first system. This accordance is better for lower and moderate discharges.

It is observed that the check valve is closing late for upper steady state discharges and early for lower steady state discharges in the second system.

In the second system, some discrepancies exist between theoretical and experimental results. They are due to the difficulty for reflecting the behaviour of the check valve. That's why, additional investigations, mainly about the time and shape of its closure need to be performed.

The oscillations of large amplitude observed at first point are likely to occur because of water hammer due to the closure of the check valve.

Some provisions need to be taken in order to decrease the fluctuations in the measurements, mainly at the point close to the pump and check valve.

The accord between theoretical and experimental results may be improved by translating the behaviour of the check valve more realistically. The computer program handled for the second system should be revised in order to translate the transient behaviour after the closure of the check valve.

It will be interesting to investigate the effect of the variable head loss coefficients involved during unsteady state flow determination in the Fortran programs.

## REFERENCES

A GFP Company. (n.d.). Retrieved June 01, 2007, from web address http://www.mechanicalroomsuplies.com

Advanced Valves. (n.d.). Retrieved June 01, 2007, from web address http://www.advancevalves.com

Alves, E. G. (2004). Hydraulic analysis of sudden flow changes in a complex piping circuit. Conference paper. Delaware, USA.

Bergant, A., Simpson, A.R. \& Tijsseling, A.S. (2005). Water hammer with column separation: A historical review.

Bryan Karney, W., \& McInnis, D., (1990). Transient analysis of water distribution system. Journal of American water works association (AWWA), 82, 7, 62-70.

Burmann, W. (1975). Water hammer in coaxial pipe systems. Journal of hydraulic division of ASCE, 101, 6, 699-715.

Champion Valves. (n.d.). Retrieved June 01, 2007, from web address http://www.wafercheck.com/Championvalves

Charles Darwin University, School of Engineering, Fluid Mechanics laboratory experiment, Eng 243, . (n.d.). Retrieved June 01, 2007, from web address http://www.cs.ntu.edu.au/homepages/jmitroy/eng243/FlowComplexPipeSystem

Chaudhry, M.H. (1987). Applied hydraulic transients. Van Nostrand Reinhold Company.

Clark, G. A., Smajstrla, A.G. \& Haman, D.Z. (2003). Water hammer in irrigation systems. Institute of Food and Agricultural Sciences, University of Florida, Gainesville.

Control Valve Monovar, hydraulic characteristics. (n.d.). Retrieved June 01, 2007, from web address http://www.4engr.com/product/catalog/1157/index.html

Diesselhorst, T., Neumann, U. (2005). Optimization of loads in piping systems by the realistic calculation method: Applying fluid-structure interaction (FSI) and dynamic friction. Journal of pressure vessel technology of ASME, 127, 1, 1-6.

Donsky, B. (1961). Complete pump characteristic and the effects of specific speeds on hydraulic transients. Journal of basic engineering, ASME, 685-699.

Duyar Valve (2006), Valve cataloque. Duyar valve mechanical industry Gaziosmanpaşa, İstanbul

Ezzeddine, H.T. \& Lili, T. (1998). Transient flow of homogeneous gas-liquid mixtures in pipelines. International Journal of numerical methods for heat and fluid flow, 8, 3, 350-368.

Fleming, K.K., Dugandzic, J.P., LeChevallier, M.W. \& Gullick, R.W. (2006). Susceptibility of potable water distribution systems to negative pressure transients. Research project summary. Division of science, research and technology, Trenton, NJ, USA.

Flowserve Flow Control (UK) Ltd. (n.d.). Retrieved June 01, 2007, from web address http://www.worcestercontrols.co.uk

Hayward Industrial Products, Inc. (n.d.). Retrieved June 01, 2007, from web address http://www.haywardindustrial.com

Kaliatka, A., Uspuras, E. \& Vaisnoras, M. (2005). RELAP5 code analysis of water hammer wave behaviour. Conference paper. Laboratory of Nuclear Installation Safety, Lithuanian Energy Institute, Kaunas, Lithuania.

Kameswara, R. C. V. \& Eswaran, K. (1999). Pressure transients in incompressible fluid pipeline networks. Journal of nuclear engineering and design, 188, 1, 1-11.

Kirkland, C. (1998). Controlling and understanding the effects of air in pipelines. Conference paper, Amiad Australia Pty. Ltd.

Koç, A.C. (2001). Computation of unsteady flows in hydraulic systems with hydraulic machinery. A thesis submitted to the graduate school of natural and applied sciences of Dokuz Eylül University, İzmir.

Kono, Y., Watanabe, M. \& Ito, T. (1998). Phase change analysis in water hammer by upstream finite difference method. Conference paper. Japan.

Larock, B.E., Jeppson, R.W., \& Watters, G.Z. (2000). Hydraulics of pipeline systems. CRC press.

Leishear, R.A. (2007). Dynamic pipe stresses during water hammer: A finite element approach. Journal of pressure vessel technology, ASME, 129, 2, 226-233.

LMNO Engineering. (n.d.). Retrieved June 01, 2007, from web address http://www.lmnoengineering.com

Mays, L.W. (1999). Hydraulic design handbook. McGraw-Hill.

Popescu, M., Arsenie, D., Vlase, P. (2003). Applied hydraulic transients. A.A. Balkema Publishers.

Rahmeyer, W. (1996). Dynamic flow testing of check valves. Journal of nuclear industry check valve group, winter meeting, St. Petersburg, Florida, USA.

Reduced Bore Valves, Engineering Toolbox. (n.d.). Retrieved June 01, 2007, from web address http://www.engineeringtoolbox.com/ball-valves-flow-coefficientsd_223.html

Streeter, V.L., \& Wylie, E.B. (1975). Transient analysis of offshore loading systems. Journal of manufacturing science and engineering, 97, (1), 259-265.

Sunrise Systems Ltd. (1999). Pipeline transient module. Computer program.

Thorley, A. R. D. (1989). Check valve behaviour under transient flow conditions: A state-of-the-art review. Journal of fluids engineering, (111), 178-183.

Thorley, A. R. D. (1991). Fluid transients in pipeline systems. D\&L George ltd.

Tullis, J. P. (1989). Hydraulics of pipelines. John Wiley \& Sons Inc.

Ward, D. (2004). Automatic and remotely controlled shutoff for direct flow liquid manure application systems.

Watters, G. Z. (1979). Modern analysis and control of unsteady flow in pipelines. Michigan: Ann Arbor Science.

White, F. (2003). Fluid Mechanics. $5^{\text {th }}$ edition, McGraw-Hill.

Wylie, E. B., \& Streeter, V. L. (1978). Fluid transients. Prentice hall.

Wylie, E. B., \& Streeter, V. L. (1993). Fluid transients in systems. Prentice hall.

Yanmaz, M. (2001). Applied water resources engineering. METU Press.

## APPENDICES

APPENDIX A: The geometric characteristics of Reduced Bore Ball valve and those of the disc type check valve, Head loss coefficients for minor losses.

APPENDIX B : The list of computer program adopted for the first system, Typical data file

APPENDIX C : The list of computer program adopted for the second system, Typical data file

## APPENDIX A

## A.1. The geometric characteristics of Reduced Bore Ball valve and those of the disc type check valve

In the experimental systems, PN6 DN125-100 reduced bore ball valve is used. It has an interior diameter of 100 mm and an exterior diameter of 125 mm (Figure A.1.1). The used disc type check valve is named as NEK NW 100 ND 16, 100 PN 10-16. The general view and the section of this valve is given in Figure A.1.2.

```
PN 16
```

PN 16
TS 3148/19
TS 3148/19
DIN 3357
DIN 3357
DN 20/15 ... DN 250/200

```
DN 20/15 ... DN 250/200
```




Figure A.1.1: The view of Reduced Bore Ball Valve (trademark: Asvan) .


Figure A.1.2: The Disc Type Check Valve.

## A.2. Head loss coefficients for minor losses.

Various loss coefficients for the Disc type check valve were investigated from the literature. The loss coefficients against to various steady state discharges are presented as tables and graphics in this section.

Table A.2.1: Check valve loss coefficients against to discharges obtained from 'Duyar valve' company. The valve diameter is equal to 125 mm .

| Steady state discharges, $\mathrm{Q}\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Check valve loss coefficient, K |
| :--- | :--- |
| 0.0075 | 10.50 |
| 0.010 | 9.45 |
| 0.015 | 9.19 |
| 0.020 | 8.86 |
| 0.030 | 8.21 |
| 0.040 | 7.39 |



Figure A.2.1: Pressure drop against to discharges for check valve corresponding to Table A.2.1 obtained from 'Duyar Valve Company'. x axis is pressure drop by bar and y axis is discharge in lt/s.

Loss coefficients of dual plate check valve


Figure A.2.2: Loss coefficients, $K$ of check valve against to steady state discharges corresponding to Table A.2.1 obtained from 'Duyar Valve' company. Diameter $=125 \mathrm{~mm}$.

The equation obtained from curve fitting will be $\mathrm{K}=4,1992 . \mathrm{Q}^{-0,1847}$.

## Head Loss v/s Flow Rate



Figure A.2.3: Head Losses vs Flow rate is drawn for check valve obtained from 'Advanced Valves' company.

Loss Coefficients of Dual plate check valve ("Advanced Valves")


Figure A.2.4: Against steady state discharges, Q to loss coefficients, K for check valve obtained from 'Advanced Valves' company, corresponding to Figure A.2.3. D $=125 \mathrm{~mm}$.

The equation found after curve fitting will be $K=\left(2.10^{6} . Q^{4}\right)$ $\left(338676 . Q^{3}\right)+\left(27474 . Q^{2}\right)-(962,66 . Q)+14,553$.

Table A.2.2: Check valve loss coefficients against to discharges obtained from 'Advanced valves' company, corresponding to Figure A.2.3. D $=125 \mathrm{~mm}$.

| Steady state discharge, $\mathrm{Q}\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Loss coefficient; K (for check valve) |
| :--- | :--- |
| 0.01 | 7.39 |
| 0.02 | 3.69 |
| 0.03 | 2.63 |
| 0.04 | 2.22 |
| 0.05 | 2.19 |
| 0.06 | 2.17 |
| 0.07 | 2.11 |
| 0.08 | 2.08 |

Pressure Drop Chart


Figure A.2.5: Head losses vs steady state discharges for check valve obtained from 'Champion valves' company.

Loss coefficients of dual plate check valve ("Champions Valve")


Figure A.2.6: Steady state discharges, Q against to loss coefficients, K for check valve obtained from 'Champion Valves' company, corresponding to Figure A.2.5. D=125 mm.

The equation obtained from curve fitting is $K=\left(7.10^{6} . Q^{4}\right)-\left(10^{6} . Q^{3}\right)+\left(79262 . Q^{2}\right)-$ (2122,7.Q)+23,543.

Table A.2.3: Check valve loss coefficients against to discharges obtained from 'Champion valves' company, corresponding to Figure A.2.5. D=125 mm.

| Steady state discharges, $\mathrm{Q}\left(\mathrm{m}^{3} / \mathrm{s}\right)$ | Loss coefficient, K |
| :--- | :--- |
| 0.010 | 8.62 |
| 0.012 | 6.79 |
| 0.025 | 3.25 |
| 0.037 | 3.14 |
| 0.056 | 2.79 |
| 0.063 | 2.75 |

Table A.2.4: Loss coefficients of different trademarks about disc type check valve against to steady state discharges (lt/s). Column 1 refers to steady state discharges (lt/s). Column 2, 3 and 4 refer to loss coefficients correspond to steady state discharges presented in Figures A.2.1, A.2.3 and A.2.5 respectively. In the last column at the right side refers to mean values of the summations of column 2 , 3 and $4 . D=125 \mathrm{~mm}$.

| Steady state <br> discharge (lt/s) | From <br> Figure A.2.1, <br> Loss coeff., K | From <br> Figure A.2.3, <br> Loss coeff., K | From <br> Figure A.2.5, <br> Loss coeff., K | Mean loss coeff., <br> with column 2, 3 <br> and 4 |
| :--- | :--- | :--- | :--- | :--- |
| 6.60 | 10.61 | 9.30 | 12.71 | 10.87 |
| 9.01 | 10.02 | 7.87 | 10.16 | 9.35 |
| 16.49 | 8.96 | 4.78 | 6.12 | 6.62 |
| 19.14 | 8.72 | 4.08 | 5.88 | 6.23 |
| 20.55 | 8.60 | 3.79 | 5.96 | 6.12 |
| 24.51 | 8.33 | 3.20 |  |  |
| 25.16 | 8.29 | 3.13 |  |  |

Table A.2.5: Flow coefficients Cv according to Valve sizes (Note that $\mathrm{Cv}=$ The number of US gpm of water that will flow thru the valve with a 1 psi pressure drop across the valve) obtained from 'A GFP Company' for Dual Disc Non Slam Check valve.

| Valve Size, inches | Valve sizes, mm | Flow Coefficient, Cv |
| :--- | :--- | :--- |
| 2.0 | 51 | 80 |
| 2.5 | 63 | 90 |
| 3.0 | 76 | 150 |
| 4.0 | 102 | 300 |
| 5.0 | 127 | 500 |
| 6.0 | 152 | 900 |
| 8.0 | 203 | 1700 |
| 10.0 | 254 | 3000 |
| 12.0 | 305 | 4000 |
| 14.0 | 356 | 5350 |
| 16.0 | 406 | 7400 |

Table A.2.6: Flow coefficients Cv according to Valve sizes (Note that $\mathrm{Cv}=$ The number of US gpm of water that will flow thru the valve with a 1 psi pressure drop across the valve) obtained from 'Hayward Industrial Products, Inc' for Swing Type Check valve ( $\Delta \mathrm{P}=(\mathrm{Q} / \mathrm{Cv})^{2}$ in which $\Delta \mathrm{P}=$ pressure drop in psi, $\mathrm{Q}=$ Flow rate in gpm, $\mathrm{Cv}=$ flow coefficient).

| Valve Size, inches | Valve sizes, mm | Flow Coefficient, Cv |
| :--- | :--- | :--- |
| 3 | 76 | 328 |
| 4 | 101 | 514 |
| 6 | 152 | 1278 |
| 8 | 203 | 2549 |

## A.3. Head loss coefficients for reduced bore ball valves in the literature

Typical flow characteristics


Figure A.3.1: \% Flow vs \% valve opening (from
Monovar control valve)


Figure A.3.2: Headloss coefficient K vs \% valve opening (from Monovar control valve).

Table A.3.1: Loss coefficients, K in valves (obtained from Charles Darwin University)

| \% Opening of valve | K for Gate valve | K for Globe valve |
| :--- | :--- | :--- |
| 100 | 0.2 | 10.0 |
| 75 | 0.9 | 11.0 |
| 50 | 5.0 | 12.5 |
| 25 | 24.0 | 50.0 |

Table A.3.2: Loss coefficient ratio of globe valve, $\mathrm{K} / \mathrm{K}_{\text {open }}$ corresponding to $\%$ opening of valve (obtained from White, 2003)

| \% Opening of valve | $\mathrm{K} / \mathrm{K}_{\text {open }}$ |
| :--- | :--- |
| 100 | 1.00 |
| 80 | 1.03 |
| 70 | 1.13 |
| 60 | 1.38 |
| 50 | 1.50 |
| 40 | 1.75 |
| 30 | 2.63 |
| 25 | 3.75 |

Table A.3.3: Typical flow coefficients, Cv (for US Customary Units) for reduced bore valves (obtained from 'Ball valves and typical flow coefficients' web page).

| Diameter of pipe in mm | Diameter of pipe in inches | Flow coefficient, Cv |
| :--- | :--- | :--- |
| 75 | 3 | 420 |
| 100 | 4 | 770 |
| 150 | 6 | 1800 |
| 200 | 8 | 2500 |
| 250 | 10 | 4500 |
| 300 | 12 | 8000 |
| 350 | 14 | 12000 |

Table A.3.3 : (Continued)

| 400 | 16 | 14000 |
| :--- | :--- | :--- |
| 450 | 18 | 18000 |
| 500 | 20 | 22000 |

Table A.3.4: Ball valve head loss coefficient, K vs \% opening of valve (obtained from 'LMNO Engineering' home web page).

| \% Opening of valve | Head loss coefficient, K |
| :--- | :--- |
| Angle, 100 | 5.00 |
| Ball, 100 | 0.05 |
| Ball, 67 | 5.50 |
| Ball, 33 | 210.00 |

Table A.3.5: Flow coefficients, Cv (for US Customary Units, i.e. Cv is flow in US gpm, pressure is psi ) and Kv (for SI Units, i.e. flow in $\mathrm{m}^{3} / \mathrm{h}$ and pressure is bar) versus valve sizes (obtained from 'Flowserve Flow Control (UK) Ltd’ web page).

| Valve size in mm | Valve size <br> in inches | Flow coefficient, <br> Cv | Flow coefficient, <br> Kv |
| :--- | :--- | :--- | :--- |
| 15 | 0.50 | 6.00 | 7 |
| 20 | 0.75 | 8.70 | 10 |
| 25 | 1.00 | 26.00 | 30 |
| 40 | 1.50 | 77.00 | 89 |
| 50 | 2.00 | 112.50 | 130 |
| 65 | 2.50 | 230.00 | 267 |
| 80 | 3.00 | 303.00 | 350 |
| 100 | 4.00 | 623.00 | 720 |
| 150 | 6.00 | 882.00 | 1020 |
| 200 | 8.00 | 1557.00 | 1800 |
| 250 | 10.00 | 2560.00 | 2970 |

Table A.3.6 : Flow coefficients, Cv (for US Customary Units, i.e. Cv is flow in US gpm, pressure is psi) versus valve sizes (obtained from Larock et al, 2000) for Ball valves.

| Valve size in inches | Valve size in mm | Flow coefficient, Cv |
| :--- | :--- | :--- |
| 6 | 152 | 5250 |
| 8 | 203 | 9330 |
| 10 | 254 | 14600 |
| 12 | 305 | 21000 |
| 14 | 356 | 28600 |
| 16 | 406 | 37300 |
| 18 | 457 | 47300 |
| 20 | 508 | 58300 |
| 24 | 610 | 84000 |
| 30 | 762 | 131300 |
| 36 | 914 | 189000 |
| 42 | 1067 | 257300 |
| 48 | 1219 | 336000 |
| 54 | 1372 | 425300 |
| 60 | 1524 | 525100 |

Table A.3.7 : Flow coefficients, Cv (for US Customary Units, i.e. Cv is flow in US gpm, pressure is psi) corresponding to degrees open (obtained from Larock et al, 2000) for Ball valves.

| Degrees Opening | Percentage of fully open Cv |
| :--- | :--- |
| 5 | 0.16 |
| 10 | 0.88 |
| 15 | 1.4 |
| 20 | 1.8 |
| 25 | 2.4 |
| 30 | 3.1 |
| 35 | 3.7 |
| 40 | 4.7 |

Table A.3.7 : (Continued)

| 45 | 5.9 |
| :--- | :--- |
| 50 | 7.2 |
| 55 | 9.0 |
| 60 | 11.2 |
| 65 | 14.1 |
| 70 | 18.0 |
| 75 | 24.5 |
| 80 | 41.5 |
| 85 | 73 |
| 90 | 100 |



Figure A.3.3 : Discharge coefficients for valves. V is the mean velocity and $\Delta \mathrm{H}$ is the head loss. We concern with $12^{11}$ reduced ball valves. ( $K=1 /(C d)^{2}$ where $K$ is the local head loss coefficient).

Table A.3.8 : Discharge coefficient, Cd corresponding to \% opening (obtained from Wylie \& Streeter, 1993) for Figure A.3.3 ( $\mathrm{K}=1 /(\mathrm{Cd})^{2}$ where K is the local head loss coefficient).

| \% Opening | Discharge Coefficient, Cd | Head loss coefficient, K |
| :--- | :--- | :--- |
| 0 | 0.000 | 0 |
| 20 | 0.030 | 1111.11 |
| 30 | 0.050 | 400.00 |
| 40 | 0.095 | 110.80 |
| 50 | 0.125 | 64.00 |
| 60 | 0.215 | 21.63 |
| 70 | 0.300 | 11.11 |
| 80 | 0.450 | 4.94 |
| 90 | 0.610 | 2.69 |
| 100 | 0.990 | 1.02 |

## APPENDIX B

## B.1. The list of computer program adopted for the first system

```
C FOR THE FIRST PIPELINE SYSTEM
C SINGLE PUMP,VALVE IS PERMANANTLY OPEN
C NO CHECK VALVE IN A LINE
C DISCHARGE VALVE IS NEARLY CLOSE TO DOWNSTREAM END POINT
        DIMENSION Q(150),H(150),WH(89),WB(89)
        DIMENSION HPH(150),ZH(150),HD(150)
        OPEN (5,FILE='WH316E1.DAT',STATUS='OLD')
        OPEN (6,FILE='WH316E1.OUT',STATUS='NEW')
    C EL=ELEVATION DIFFERENCE BETWEEN DOWNSTREAM RES SURF & DATUM (M)
    C WATER SURFACE ELEVATION OF THE RESERVOIR A IS DATUM
    C A=WAVE CELERITY (M/S), XL=PIPE LENGTH (M)
    C F=DARCY-WEISBACH FRICTION COEFFICIENT
    C D=PIPE DIAMETER (M), G=ACCELARATION DUE TO GRAVITY (9.806 M/S2)
    C CK=REDUCED BORE BALL VALVE HEADLOSS COEFFICIENT
    C TM=MAXIMUM CALCULATION TIME PERIOD (S)
    C RN=RATED ROTATIONAL SPEED (RPM)
    C TR=RATED TORQUE (NT.M)
    C HR=RATED HEAD (M)
    C QR=RATED DISCHARGE (M3/S)
    C WRR=ROTATIONAL MOMENT OF INERTIA (NT.M2)
    C REAL WRR VALUE=0.83 NT.M2
    C TOL=TOLERANCE INTO THE NEWTON-RAPHSON METHOD
    C V=THE RATIO OF VELOCITY OF WATER IN STEADY STATE TO VELOCITY OF WATER
    C IN RATED CONDITION OR DISCHARGE RATES
    C N=NUMBER OF REACHES IN THE PIPELINE
    C JPR=INTEGER NUMBER OF TIME STEP ITERATIONS BETWEEN EACH PRINT
    C KIT=MAXIMUM ITERATION NUMBER IN NEWTON-RAPHSON NUMERICAL PROCEDURE
    C DX=DIMENSIONLESS PUMP DATA SPACE (RADIAN)=PI/44=0.0714
    C PI=PI NUMBER=3.1416
    C AL,AL0,AL00=DIMENSIONLESS PUMP VELOCITY
    C T=TIME INCREMENT IN THE SUCCESSIVE TIME STEP
    C K=COUNTER DESIGNATES MAXIMUM CALCULATION TIME PERIOD
    C WH(I),WB(I)=DIMENSIONLESS PUMP DATA
    C ZH=ELEVATIONS OF MEASURING NODES ACCORDING TO DATUM (UPSTR RES LEVEL)
    C AKS=ROUGHNESS HEIGHT IN STEEL PIPE (M)
    C QQA=STEADY STATE DISCHARGE (M3/S)
    C H=PIEZOMETRIC HEAD IN METERS
    C HPH=PRESSURE HEAD IN METERS
    C NSS=NUMBER OF NODE IN WHICH THERE IS A DISCHARGE ADJUSTMENT VALVE
    C VN=KINEMATIC VISCOSITY OF WATER (M2/S)
        READ (5,*) EL,A,XL,D,G,CK,TM,RN,TR,HR,QR,WRR
        READ (5,*) TOL,N,JPR,KIT,VN
        READ (5,*) DX,PI,AKS,QQA,NSS
        READ (5,*) AL,AL0,AL00,T,K,HMIN
        READ (5,*) (ZH(I),I=1,N+1)
        READ (5,*) (WH(I), I=1, 89)
        READ (5,*) (WB(I),I=1,89)
        UU=QQA/(PI*D*D/4)
        RE=UU*D/VN
        F=(1./(1.14-(2.*ALOG10((AKS/D)+(21.25/(RE**0.9))))))**2.
        V=QQA/QR
        DT=XL/(A*N)
        WRITE (6,1)
        WRITE (6,2)
        WRITE (6,3) EL,A,XL
        WRITE (6,4) F,D,G
        WRITE (6,500) TM,RN,TR
```

```
        WRITE (6,501) HR,QR,WRR
        WRITE (6,502) TOL,V,N
        WRITE (6,503) DX,CK,PI
        WRITE (6,504) AL,AL0,AL00
        WRITE (6,509) AKS,DT,VN
        WRITE (6,505) HMIN,T,JPR
        WRITE (6,506) K,QQA,KIT
        WRITE (6,510) NSS
        WRITE (6,507)
        WRITE (6,508)
        IF (MOD(N,2).NE.0) N=N+1
        NS=N+1
        NG=(NS+1)/2
        KMAX=INT((0.5*TM+0.001)/DT)+1
        AR=0.7854*D*D
        R=F*A*DT/(2.*G*D*AR*AR)
        B=A/(G*AR)
        CTORQ=WRR*PI*RN/(G*30.*TR*DT)
    C STEADY STATE FLOW DETERMINATION
        DH=CK*((QR/AR)**2.)/(2.*G)
        C=((N*R)*QR*QR)+(DH)
        X=PI+ATAN2(V,AL)
        I=X/DX+1
        KK=0
    C BEGINNING OF LOOP TO FIND STEADY STATE FLOW AND PUMP OPERATING POINT
        10 A1=(WH(I+1)-WH(I))/DX
        A0=WH(I+1)-A1*I*DX
        DO 15 KI=1,KIT
        F1=-EL-C*V*V+HR* (1.+V*V)*(A0+A1*X)
        F1V=-2.*C*V+HR*(2.*V*(A0+A1*X)+A1)
        V=V-F1/F1V
        X=PI+ATAN2(V,AL)
        15 CONTINUE
        II=X/DX+1
        IF (II.NE.I) THEN
        KK=KK+1
        IF (KK.GT.3) THEN
        WRITE (6,701)
    701 FORMAT (' TROUBLE WITH STEADY STATE')
        STOP
        ENDIF
        I=II
        GO TO 10
        ENDIF
    C FIND STEADY STATE TORQUE AND STORE INITIAL VARIABLES
        Q0=V*QR
        B1=(WB(I+1)-WB(I))/DX
        B0=WB(I+1)-B1*I*DX
        BETA=(AL*AL+V*V)* (B0+B1*(PI+ATAN2(V,AL)))
        BET0=BETA
    C DETERMINATION OF Q AND H
        WRITE (6,400)
        400 FORMAT ('HP1=PIEZOMETRIC HEAD IN METERS BEFORE VALVE')
        WRITE (6,401)
        4 0 1 ~ F O R M A T ~ ( ' H P 2 = P I E Z O M E T R I C ~ H E A D ~ I N ~ M E T E R S ~ A F T E R ~ V A L V E ' ) ~
        WRITE (6,402)
    4 0 2 ~ F O R M A T ~ ( ' H P = H E A D L O S S ~ I N ~ M E T E R S ~ A T ~ V A L V E ~ ( N O D E ~ N S S ) ' , / ) ~
        H3=CK*(Q0**2)/(2.*G*AR*AR)
            DO 20 I=1,NS
            IF (I.GT.NSS) THEN
            H(I)=EL+(R*(NS-I)*Q0*ABS(Q0))
            ELSE IF (I.LE.NSS) THEN
            H(I)=EL+(R*(NS-I)*Q0*ABS(Q0))+H3
            ENDIF
            Q(I)=Q0
```

```
        IF (I.EQ.NSS) GO TO 699
        GO TO 20
    6 9 9 ~ H P 1 = H ( I )
        HP2=H(I)-H3
        H(I)=HP2
        HP=HP1-HP2
        WRITE (6,698) HP1,HP2,HP
    698 FORMAT ('HP1 HP2 HP',3F10.4,/)
    20 CONTINUE
        WRITE (6,403)
    403 FORMAT ('NUMBER OF NODES AND PIEZOMETRIC HEADS AT NODES AS METERS
        $:',/)
        WRITE (6,702) (I,H(I),I=1,NS)
        FORMAT (5(I2,':',1X,F8.2,3X))
        WRITE (6,404)
    404 FORMAT (/,'ZH(I) VALUES AS METERS :',/)
        WRITE (6,702) (I,ZH(I),I=1,NS)
C
        HMAX=EL
        V0=V
        V00=V
        HP11=HP1-ZH(NSS)
        DO 21 I=1,NS
        21 HPH(I)=H(I)-ZH(I)
C
        WRITE (6,405)
    405 FORMAT (/)
        WRITE (6,703)
    703 FORMAT (' TIME ALPHA BETA V Q(1) Q(NS) H(1) H(NS) H
        1(NSS1) H(NSS2)')
        WRITE (6,704) T, AL, BETA,V,Q(1),Q(NS),HPH(1),HPH(NS),HP11,HPH(NSS)
    704 FORMAT (F5.3,9F7.3)
C BEGINNING OF TRANSIENT LOOP
        DO 80 K=1,KMAX-1
        T=2.*DT*K
C COMPUTATION OF INTERIOR POINTS
        DO 30 I1=2,3
        DO 25 I=I1,N,2
        BP=R*ABS(Q(I-1))+B
        IF (I.EQ.(NSS-1)) GO TO 667
        IF (I.EQ.NSS) GO TO 668
        IF (I.EQ.(NSS+1)) GO TO 666
        GO TO 26
    667 Q(I)=(H(I-1)-HP1+B*(Q(I-1)+Q(I+1)))/(BP+B+R*ABS(Q(I+1)))
        H(I)=H(I-1)+B*Q(I-1)-BP*Q(I)
        GO TO 25
    666 IF ((NSS+1).EQ.(N+1)) GO TO 25
        Q(I)=(HP2-H(I+1)+B*(Q(I-1)+Q(I+1)))/(BP+B+R*ABS(Q(I+1)))
        H(I)=HP2+B*Q(I-1)-BP*Q(I)
        GO TO 25
    668 CM=H(I+1)-(B*Q(I+1))+(R*Q(I+1)*ABS(Q(I+1)))
    CP=H(I-1)+(B*Q(I-1))-(R*Q(I-1)*ABS(Q(I-1)))
    IF (V.LT.0.0) GO TO 996
        C3=CK/(2.*G*AR*AR)
        C4=(2*B)/C3
        C5=(CM-CP)/C3
        Q(I)=0.5*(-C4+(SQRT((C4*C4)-(4.*C5))))
        HP1=CP-(B*Q(I))
        HP2=CM+(B*Q(I))
        H(I)=HP2
        GO TO 25
    996 C3=CK/(2.*G*AR*AR)
        C4=(2.*B)/C3
        C5=(CM-CP)/C3
        Q(I)=0.5*(C4-(SQRT(((-C4)**2.)+(4.*C5))))
```

```
    HP1=CP-(B*Q(I))
    HP2=CM+(B*Q(I))
    H(I)=HP2
    GO TO 25
    Q(I)=(H(I-1)-H(I+1)+B*(Q(I-1)+Q(I+1)))/(BP+B+R*ABS(Q(I+1)))
    H(I)=H(I-1)+B*Q(I-1)-BP*Q(I)
    25 CONTINUE
    IF (N.LE.2) GO TO 35
    CONTINUE
C
C
C
C
35 CM=H(2)-Q(2)*B
    BM=B+R*ABS(Q(2))
C BALANCE HEAD AND TORQUE EQUATIONS
    AL=2. *AL0-AL00
    V=2. *V0-V00
    X=PI+ATAN2(V,AL)
    I=X/DX+1
C BEGINNING OF LOOP TO FIND PUMP FLOW AND SPEED
40 A1=(WH(I+1)-WH(I))/DX
    A0=WH(I+1)-A1*I *DX
    B1=(WB(I+1)-WB(I))/DX
    B0=WB(I+1)-B1*I *DX
C NEWTON RAPHSON
    DO 50 KI=1,KIT
    F1=(HR* (AL**2+V*V)* (A0+A1*X) - (CM+BM*QR*V))
    F2=(AL**2+V*V)* (B0+B1*X)+BET0+CTORQ* (AL-AL0)
    F1A=HR*(2. *AL* (A0+A1*X) - V*A1)
    F1V=(HR* (2.*V* (A0+A1*X)+AL*A1) - BM*QR)
    F2A=2. *AL* (B0+B1*X) -V*B1+CTORQ
    F2V=2. *V*(B0+B1*X)+AL*B1
    DAL=(F2/F2V-F1/F1V)/(F1A/F1V-F2A/F2V)
    DV=-F1/F1V-DAL*F1A/F1V
    AL=AL+DAL
    V=V+DV
    X=PI+ATAN2(V,AL)
    IF (ABS(DAL)+ABS(DV).LT.TOL) GO TO 55
50 CONTINUE
C CHECK TO SEE IF SOLUTION HAS MOVED TO ANOTHER PART OF PUMP CURVE
55 II=X/DX+1
    IF (II.NE.I) THEN
    I=II
    GO TO 40
    ENDIF
C SOLUTION HAS BEEN GENERATED
V00=V0
V0=V
AL00=AL0
AL0=AL
BETA=(AL** 2+V*V)*(B0+B1*X)
Q(1)=V*QR
H(1)=CM+BM*Q(1)
C DOWNSTREAM BOUNDARY CONDITION
C
C
    Q(NS)=(H(N)-H(NS )+B*Q(N))/(R*ABS(Q(N))+B)
    BET0=BETA
    DO 70 I=1,NS,2
    IF (H(I).GT.HMAX) HMAX=HD(I)
    IF (H(I).LT.HMIN) HMIN=HD(I)
    IF (H(I).GT.HMAX) HMAX=H(I)
    IF (H(I).LT.HMIN) HMIN=H(I)
CONTINUE
```

```
            IF (K.EQ.1) GO TO 5
            IF (MOD(K,JPR).EQ.0) GO TO 5
            GO TO 80
        5 HPH(1)=H(1)-ZH(1)
            HPH(NS)=H(NS)-ZH(NS)
            HP11=HP1-ZH(NSS)
            HPH(NSS)=H(NSS)-ZH(NSS)
            WRITE (6,704) T,AL,BETA,V,Q(1),Q(NS),HPH(1),HPH(NS),HP11,HPH(NSS)
            80 CONTINUE
OF TRANSIENT LOOP
            HMAX1=HMAX
            HMIN1=HMIN
            WRITE (6,705) HMAX1,HMIN1
            FORMAT ('HMAX HMIN ',2F8.3)
            FORMAT (26X,18('*'),26X,/,26X,'* INITIAL VALUES *',26X)
            FORMAT (26X,18('*'),26X,/,/)
                            F0RMAT ('EL (M)=',F10.4,8X,'A (M3/SEC)=',F10.4,7X,'XL (M)=',F10.4
            $)
    4 FORMAT ('F=',F10.4,13X,'D (M)=',F10.4,12X,'G (M/S2)=',F10.4)
500 FORMAT ('TM (S)=',F10.4,8X,'RN (RPM)=',F10.4,9X,'TR (NT.M)=',F10
    $4)
501 FORMAT ('HR (M)=',F10.4,8X,'QR (M3/S)=',F10.4,8X,'WRR (NT.M2)=',F
    $10.4)
5 0 2 ~ F O R M A T ~ ( ' T O L = ' , F 1 0 . 4 , 1 1 X , ' V = ' , F 1 0 . 4 , 1 6 X , ' N = ' , I 3 )
5 0 3 ~ F O R M A T ~ ( ' D X ~ ( R A D ) = ' , F 1 0 . 4 , 6 X , ' C K = ' , F 1 0 . 4 , 1 5 X , ' P I = ' , F 1 0 . 4 )
504 F0RMAT ('AL=',F10.4,12X,'AL0=',F10.4,14X,'AL00=',F10.4)
509 FORMAT ('AKS(M)=',F10.4,8X,'DT (S)=',F10.4,11X,'VN (M2/S)=',F10.6
    $)
5 0 5 \text { FORMAT ('HMIN=',F10.4,10X,'T=',F10.4,16X,'JPR=',I3)}
506 FORMAT ('K=',I3, 20X,'QQA (M3/S)=',F10.4,7X,'KIT=',I3)
510 FORMAT ('NSS=',I3)
507 FORMAT (/,/, 26X,17('*'),26X,/,26X,'* RESULT VALUES *',26X)
508 FORMAT (26X,17('*'),26X,/,/)
STOP
    END
```


## B.2. Typical data file

```
3.73 1331.0 28 0.125 9.806 84 8 1450 32.18 11 0.036 0.83
0.0002 28 56 5 0.000001
0.0714 3.1416 0.00015 0.0183 23
1 1 1 0 0 0
-0.39 0.17 0.42 0.47 0.47 0.52 0.57 0.62 0.62 0.65 0.67 0.72 0.74
0.77 0.79 0.82 0.87 0.88 0.92 0.96 1.62 2.17 2.17 2.17 2.17
2.17 2.17 2.17 3.38
0.411 0.4338 0.451 0.4618 0.4719 0.4793 0.4816 0.4763 0.4644 0.4557
0.451 0.4519 0.4637 0.4852 0.5034 0.5272 0.5617 0.6077 0.6634 0.725
0.7983 0.8754 0.9426 0.9935 1.0503 1.1094 1.1541 1.1884 1.2223
1.2388 1.2447 1.2573 1.2642 1.2729 1.2746 1.2926 1.3069 1.3281
1.3541 1.3733 1.3936 1.4066 1.4121 1.412 1.4029 1.3785 1.34 1.2833
1.2032 1.1154 1.0436 0.9395 0.8427 0.7349 0.6209 0.5 0.369 0.2317
0.1041-0.0229 -0.1495 -0.2382 -0.3444 -0.4481 -0.5513 -0.6419
0.715 -0.8423 -0.9085 -0.9429 -0.953 -0.9446 -0.9203 -0.9027 -0.8577
-0.8193 -0.7642 -0.71 -0.6624 -0.5907 -0.4982 -0.3722 -0.2236 -
0.0519 0.0887 0.1976 0.2889 0.3606 0.411
-0.808 -0.6785 -0.5482 -0.4207 -0.302 -0.2013 -0.1406 -0.0972 -
0.0291 0.0416 0.1028 0.1888 0.2693 0.3604 0.4437 0.5308 0.6201
```

```
0.7153 0.8096 0.8652 0.9321 1.0058 1.0696 1.1217 1.1545 1.1723
1.1731 1.1494 1.1259 1.0831 1.0149 0.9463
0.8702 0.7878 0.7056 0.6467 0.5994 0.5621 0.5407 0.5318 0.5374
0.5502 0.5659 0.5933 0.6252 0.6376 0.6384 0.6355 0.6346 0.6385
0.6413 0.6367 0.6181 0.5891 0.55 0.5 0.4278 0.3462 0.2641 0.1818
0.0728-0.0331-0.1426-0.2649 -0.376 -0.4762 -0.5721 -0.831 -1.0331
-1.1865 -1.2981 -1.403 -1.511 -1.6041 -1.6955 -1.7668 -1.8082
-1.8163 -1.866 -1.7494 -1.6793 -1.6008 -1.5076 -1.4177 -1.3428 -
1.2263 -1.1097 -0.9564 -0.808
```


## APPENDIX C

## C.1. The list of computer program adopted for the second system

```
C WH25.FOR=UNSTEADY FLOW CALCULATIONS FOR SECOND SYSTEM
    DIMENSION Q(50),H(50),WH(89),WB(89),TAU(50)
    DIMENSION HPH(50),ZH(50)
    OPEN (5,FILE='WH25B.DAT',STATUS='OLD')
    OPEN (6,FILE='WH25B.OUT',STATUS='NEW')
C DATUM=UPSTREAM RESERVOIR WATER SURFACE ELEVATION
C EL=ELEVATION DIFFERENCE BETWEEN UPSTREAM & DOWNSTREAM RESERVOIRS (M)
C A=WAVE CELERITY (M/S), XL=PIPE LENGTH (M)
C F=DARCY-WEISBACH FRICTION COEFFICIENT
C D=PIPE DIAMETER (M), G=ACCELARATION DUE TO GRAVITY (9.806 M/S2)
C CK=DISC TYPE CHECK VALVE HEADLOSS COEFFICIENT
C AKV1=HEADLOSS COEFFICIENT OF REDUCED BORE BALL VALVE IMMEDIATELY
C AFTER PUMP
C AKV2=HEADLOSS COEFFICIENT OF REDUCED BORE BALL VALVE AT THE END
C OF THE PIPE
C TM=MAXIMUM CALCULATION TIME PERIOD (S)
C RN=RATED ROTATIONAL SPEED (RPM)
C TR=RATED TORQUE (NT.M)
C HR=RATED HEAD (M)
C QR=RATED DISCHARGE (M3/S)
C WRR=ROTATIONAL MOMENT OF INERTIA (NT.M2)
C DTAU=TIME INCREMENT BETWEEN TAU VALUES (S)
C TOL=TOLERANCE INTO THE NEWTON-RAPHSON METHOD
C VI=THE RATIO OF VELOCITY OF WATER IN STEADY STATE TO VELOCITY OF WATER
C IN RATED CONDITION
C N=NUMBER OF REACHES IN THE PIPELINE
C JPR=INTEGER NUMBER OF TIME STEP ITERATIONS BETWEEN EACH PRINT
C KIT=MAXIMUM ITERATION NUMBER IN NEWTON-RAPHSON NUMERICAL METHOD
C TAU=CHECK VALVE OPENING
C DX=DIMENSIONLESS PUMP DATA SPACE (RADIAN)=PI/44=0.0713
C PI=PI NUMBER=3.1416
C AL,AL0,AL00=DIMENSIONLESS ROTATIONAL SPEED RATIO
C T=TIME INCREMENT IN THE SUCCESSIVE TIME STEP
C K=COUNTER DESIGNATES MAXIMUM CALCULATION TIME PERIOD
C WH(I),WB(I)=DIMENSIONLESS PUMP DATA
C ZH=ELEVATIONS OF MEASURING NODES ACCORDING TO DATUM (UPSTR WAT SUR)
C NSV=NODE NUMBER OF VALVE AT THE END OF THE PIPE
C NTAU=NUMBER OF TAU VALUES
C CCT=CHECK VALVE CLOSING TIME (S)
C AT PUMP NODE (i.e.NODE 1)
C M2=NODE NUMBER OF SECOND MEASURING POINT
C AKS=ROUGHNESS HEIGHT OF PIPE WALL (M)
C QQA=STEADY STATE DISCHARGE (M3/S)
C VN=KINEMATIC VISCOSITY OF WATER (M2/S)
READ (5,*) EL,A,XL,D,G,CK,AKV1,AKV2,TM,RN,TR,HR,QR,WRR
READ (5,*) CCT,NTAU,TOL,VI,N,JPR,KIT,AKS,VN
READ (5,*) DX,PI,NSV,M2
READ (5,*) AL,AL0,AL00,T,K,HMIN
READ (5,*) (ZH(I),I=1,N+1)
READ (5,*) (WH(I),I=1,89)
READ (5,*) (WB(I),I=1,89)
QQA=VI*QR
UU=QQA/(PI*D*D/4)
RE=UU*D/VN
F=(1./(1.14-(2.*ALOG10((AKS/D)+(21.25/(RE**0.9))))))**2.
IF (MOD(N,2).NE.0) N=N+1
NS=N+1
```

```
    DT=XL/(A*N)
    TAU(1)=1.
    DO 6 I=2,NTAU
    TAU(I)=1.-((I-1.)*(1./(NTAU-1.)))
    DTAU=CCT/(NTAU-1)
    WRITE (6,1)
    WRITE (6,2)
    WRITE (6,3) EL,A,XL
    WRITE (6,4) F,D,G
    WRITE (6,500) TM,RN,TR
    WRITE (6,501) HR,QR,WRR
    WRITE (6,502) TOL,VI,N
    WRITE (6,503) DX,CK,PI
    WRITE (6,511) AKV1,AKV2,DTAU
    WRITE (6,512) TAU(1),TAU(2),TAU(3)
    WRITE (6,513) TAU(4),TAU(5),TAU(6)
    WRITE (6,514) TAU(7),TAU(8),TAU(9)
    WRITE (6,515) TAU(10),TAU(11),TAU(12)
    WRITE (6,516) TAU(13),TAU(14),TAU(15)
    WRITE (6,517) TAU(16),TAU(17),TAU(18)
    WRITE (6,518) TAU(19),TAU(20),M2
    WRITE (6,504) AL,AL0,AL00
    WRITE (6,505) HMIN,T,JPR
    WRITE (6,506) K,NSV,KIT
    WRITE (6,519) CCT,NTAU,QQA
    WRITE (6,520) DT,AKS,VN
    WRITE (6,507)
    WRITE (6,508)
    KMAX=INT((0.5*TM+0.001)/DT )+1
    AR=0.7854*D*D
    R=F*A*DT/(2.*G*D*AR*AR)
    TA=TAU(1)
B=A/(G*AR)
V=VI
CTORQ=WRR*PI*RN/(G*30.*TR*DT)
C STEADY STATE FLOW DETERMINATION
DH=CK*((QR/AR)**2.)/(2.*G)
DHSV1=(AKV1)*((QR/AR)**2.)/(2.*G)
DHSV2=(AKV2)*((QR/AR)**2.)/(2.*G)
C=((N*R)*QR*QR)+(DH/TAU(1)**2.)+DHSV1+DHSV2
X=PI+ATAN2(V,AL)
I=X/DX+1
KK=0
C BEGINNING OF LOOP TO FIND STEADY STATE FLOW AND PUMP OPERATING POINT
10 A1=(WH(I+1)-WH(I))/DX
            A0=WH(I+1)-A1*I*DX
            DO }15\textrm{KI}=1,\textrm{KIT
            F1=-EL-C*V*V+HR*(1.+V*V)*(A0+A1*X)
            F1V=-2.*C*V+HR*(2.*V*(A0+A1*X)+A1)
            V=V-F1/F1V
            X=PI+ATAN2(V,AL)
        15 CONTINUE
            II=X/DX+1
            IF (II.NE.I) THEN
            KK=KK+1
            IF (KK.GT.3) THEN
            WRITE (6,701)
        701 FORMAT (' TROUBLE WITH STEADY STATE')
            STOP
            ENDIF
            I=II
            GO TO 10
            ENDIF
C FIND STEADY STATE TORQUE AND STORE INITIAL VARIABLES
            Q0=V*QR
```

```
        B1=(WB(I+1)-WB(I))/DX
        B0=WB(I+1)-B1*I*DX
        BETA=(AL*AL+V*V)*(B0+B1* (PI+ATAN2(V,AL)))
        BET0=BETA
C DETERMINATION OF Q AND H
C
    WRITE (6,400)
    4 0 0
    4 0 1 ~ F O R M A T ~ ( ' H P 2 = P I E Z O M E T R I C ~ H E A D ~ I N ~ M E T E R S ~ A F T E R ~ V A L V E ' ) ~
        WRITE (6,402)
    4 0 2 ~ F O R M A T ~ ( ' H P = H E A D L O S S ~ I N ~ M E T E R S ~ A T ~ V A L V E ~ ( N O D E ~ N S V ) ' , / )
C
    DH1=(AKV2*Q0**2.)/(2.*9.81*AR*AR)
    DO 20 I=1,NS
    IF (I.GT.NSV) THEN
    H(I )=EL+(R*(NS-I )*Q0*ABS(Q0))
    ELSE IF (I.LE.NSV) THEN
    H(I)=EL+(R*(NS-I )*Q0*ABS(Q0)) +DH1
    ENDIF
    Q(I)=Q0
        IF (I.EQ.NSV) GO TO 699
        GO TO 20
    6 9 9 ~ H P 1 = H ( I )
        HP2=H(I)-DH1
        H(I)=HP2
        HP=HP1-HP2
        WRITE (6,698) HP1,HP2,HP
        698 FORMAT ('HP1 HP2 HP',3F10.4,/)
    20 CONTINUE
C
        WRITE (6,403)
4 0 3 ~ F O R M A T ~ ( ' N U M B E R ~ O F ~ N O D E S ~ A N D ~ P I E Z O M E T R I C ~ H E A D S ~ A T ~ N O D E S ~ A S ~ M E T E R S ~
    $:',/)
        WRITE (6,702) (I,H(I),I=1,NS)
702 FORMAT (5(I2,':',1X,F8.2,3X))
    WRITE (6,404)
404 FORMAT (/,'ZH(I) VALUES AS METERS :',/)
    WRITE (6,702) (I, ZH(I),I=1,NS)
    HMAX=EL
    V0=V
    V00=V
    TA=TAU(1)
        HP11=HP1-ZH(NSV)
        DO 21 I=1,NS
    21 HPH(I)=H(I)-ZH(I)
        HPH(NSV)=HP2-ZH(NSV)
        WRITE (6,405)
405 FORMAT (/)
        WRITE (6,703)
703 FORMAT (' TIME TAU ALPHA BETA V Q (1) Q(M2)
    $ Q(NS) H(1) H(27) H(NSV1) H(NSV2) H(NS)')
        WRITE (6,704) T,TA,AL,BETA,V,Q(1),Q(M2),Q(NS),HPH(1),HPH(M2),HP11,
        $HPH(NSV), HPH(NS)
    704 FORMAT (F8.4,12F8.3)
C BEGINNING OF TRANSIENT LOOP
    DO 80 K=1,KMAX-1
        T=2. *DT*K
C COMPUTATION OF INTERIOR POINTS
    DO 30 I1=2,3
    DO 25 I=I1,N,2
    BP=R*ABS(Q(I-1))+B
    IF (I.EQ.(NSV-1)) GO TO 667
    IF (I.EQ.NSV) GO TO 668
    IF (I.EQ.(NSV+1)) GO TO 666
```

$667 \mathrm{Q}(\mathrm{I})=\left(\mathrm{H}(\mathrm{I}-1)-\mathrm{HP} 1+\mathrm{B}^{*}(\mathrm{Q}(\mathrm{I}-1)+\mathrm{Q}(\mathrm{I}+1))\right) /\left(\mathrm{BP}+\mathrm{B}+\mathrm{R}^{*} \mathrm{ABS}(\mathrm{Q}(\mathrm{I}+1))\right)$ $H(I)=H(I-1)+B * Q(I-1)-B P * Q(I)$ GO TO 25
IF ((NSV+1).EQ.(N+1)) GO TO 25
$\mathrm{Q}(\mathrm{I})=\left(\mathrm{HP} 2-\mathrm{H}(\mathrm{I}+1)+\mathrm{B}^{*}(\mathrm{Q}(\mathrm{I}-1)+\mathrm{Q}(\mathrm{I}+1))\right) /\left(\mathrm{BP}+\mathrm{B}+\mathrm{R}^{*} \mathrm{ABS}(\mathrm{Q}(\mathrm{I}+1))\right)$
$H(I)=H P 2+B^{*} Q(I-1)-B P * Q(I)$
GO TO 25
$668 C P=H(I-1)+\left(B^{*} Q(I-1)\right)-\left(R^{*} Q(I-1) * A B S(Q(I-1))\right)$
$C M=H(I+1)-\left(B^{*} Q(I+1)\right)+\left(R^{*} Q(I+1) * A B S(Q(I+1))\right)$
IF (V.LT.0.) GO TO 333
C3=AKV2/(2.*G*AR*AR)
C4 $=(2 * B) / C 3$
C5 $=(\mathrm{CM}-\mathrm{CP}) / \mathrm{C} 3$
$\mathrm{Q}(\mathrm{I})=0.5^{*}(-\mathrm{C} 4+(\operatorname{SQRT}((\mathrm{C} 4 * \mathrm{C} 4)-(4 . * \mathrm{C} 5))))$
HP1=CP-(B*Q(I))
HP2=CM+(B*Q(I))
GO TO 25
C4 $=(2 * B) / \mathrm{C} 3$
C5 $=(\mathrm{CM}-\mathrm{CP}) / \mathrm{C} 3$
$\mathrm{Q}(\mathrm{I})=0.5^{*}\left(\mathrm{C} 4-\left(\operatorname{SQRT}\left(((-\mathrm{C} 4) * * 2)+.\left(4 .{ }^{*} \mathrm{C} 5\right)\right)\right)\right)$
HP1=CP-(B*Q(I))
$H P 2=C M+\left(B^{*} Q(I)\right)$
GO TO 25
$26 \mathrm{Q}(\mathrm{I})=\left(\mathrm{H}(\mathrm{I}-1)-\mathrm{H}(\mathrm{I}+1)+\mathrm{B}^{*}(\mathrm{Q}(\mathrm{I}-1)+\mathrm{Q}(\mathrm{I}+1))\right) /\left(\mathrm{BP}+\mathrm{B}+\mathrm{R}^{*} \mathrm{ABS}(\mathrm{Q}(\mathrm{I}+1))\right)$
$H(I)=H(I-1)+B * Q(I-1)-B P * Q(I)$
25 CONTINUE
IF (N.LE.2) GO TO 35
30
CONTINUE
C PUMP BOUNDARY CONDITION
$35 \quad \mathrm{CM}=\mathrm{H}(2)-\mathrm{Q}(2) * \mathrm{~B}$
$B M=B+R^{*} A B S(Q(2))$
IF (TA.LT.0.0001) GO TO 60
C VALVE WAS OPEN; FIND CURRENT VALUE OF TAU I=T/DTAU +1
IF (I.GE.NTAU) THEN TA=TAU(NTAU)
ELSE
$\operatorname{TA}=\operatorname{TAU}(\mathrm{I})+(\mathrm{TAU}(\mathrm{I}+1)-\mathrm{TAU}(\mathrm{I})) *(\mathrm{~T}-(\mathrm{I}-1) * \mathrm{DTAU}) / \mathrm{DTAU}$
ENDIF
C Valve is closed
IF (TA.LT.0.0) TA=0.0
IF (TA.LT.0.0001) THEN
$\mathrm{V}=0$.
$Q(1)=0.0$
GO TO 60
ENDIF
C BALANCE HEAD AND TORQUE EQUATIONS
AL=2. *AL0-AL00
$\mathrm{V}=2 . * \mathrm{~V} 0-\mathrm{V} 00$
$\mathrm{X}=\mathrm{PI}+\mathrm{ATAN} 2(\mathrm{~V}, \mathrm{AL})$ I=X/DX+1
C BEGINNING OF LOOP TO FIND PUMP FLOW AND SPEED
40 A1 $=(W H(I+1)-W H(I)) / D X$
A0=WH(I+1)-A1*I*DX $B 1=(W B(I+1)-W B(I)) / D X$ $B 0=W B(I+1)-B 1 * I * D X$
C NEWTON RAPHSON DO $50 \mathrm{KI}=1, \mathrm{KIT}$ F1 $=T A^{* *} 2^{*}\left(H R^{*}(A L * * 2+V * V) *(A 0+A 1 * X)-(C M+B M * Q R * V)\right)-\left(D H^{*} V^{*} A B S(V)\right)-(D H$ \$SV1*V*ABS(V))
F2 $=(A L * * 2+V * V) *(B 0+B 1 * X)+B E T 0+C T O R Q *(A L-A L 0)$
F1A $=T A^{* *} 2^{*} H^{*}\left(2 . * A L *(A 0+A 1 * X)-V^{*} A 1\right)$
$F 1 V=T A * * 2 *\left(H R^{*}\left(2 . V^{*}(A 0+A 1 * X)+A L * A 1\right)-B M^{*} Q R\right)-\left(2 . * D H^{*} A B S(V)\right)-(2 . * D H S$

```
        $V1*ABS(V))
            F2A=2.*AL* (B0+B1*X) -V*B1+CTORQ
            F2V=2.*V*(B0+B1*X)+AL*B1
            DAL=(F2/F2V-F1/F1V)/(F1A/F1V-F2A/F2V)
            DV=-F1/F1V-DAL*F1A/F1V
            AL=AL+DAL
            V=V+DV
            X=PI+ATAN2(V,AL)
            IF (ABS(DAL)+ABS(DV).LT.TOL) GO TO 55
    50 CONTINUE
C CHECK to SEE IF SOLUTION HAS MOVED TO ANOTHER PART OF PUMP CURVE
    55 II=X/DX+1
            IF (II.NE.I) THEN
            I=II
            GO TO 40
            ENDIF
C SOLUTION HAS BEEN GENERATED
            V00=V0
            V0=V
            AL00=AL0
            AL0=AL
            BETA=(AL**2+V*V)* (B0+B1*X)
            Q(1)=V*QR
    60 H(1)=CM+BM*Q(1)
C DOWNSTREAM BOUNDARY CONDITION
            Q(NS)=(HP2-H(NS)+B*Q(N))/(R*ABS(Q(N))+B)
            BET0=BETA
            DO 70 I=1,NS,2
            IF (H(I).GT.HMAX) HMAX=H(I)
            IF (H(I).LT.HMIN) HMIN=H(I)
        70 CONTINUE
            IF (MOD(K,JPR).EQ.0) GO TO 5
            GO TO 80
        5 HPH(1)=H(1)-ZH(1)
            HPH(M2)=H(M2)-ZH(M2)
            HP11=HP1-ZH(NSV)
            HPH(NSV)=HP2-ZH(NSV)
            HPH(NS)=H(NS)-ZH(NS)
            WRITE (6,704) T,TA,AL,BETA,V,Q(1),Q(M2),Q(NS),HPH(1),HPH(M2),HP11,
            $HPH(NSV),HPH(NS)
        80 CONTINUE
C END OF TRANSIENT LOOP
                            HMAX1=HMAX
            HMIN1=HMIN
            WRITE (6,705) HMAX1,HMIN1
    705 FORMAT ('HMAX HMIN ',2F8.3)
        FORMAT (26X,18('*'),26X,/,26X,'* INITIAL VALUES *',26X)
        FORMAT (26X,18('*'),26X,/,/)
        FORMAT ('EL (M)=',F10.4,8X,'A (M3/S)=',F10.4,9X,'XL (M)=',F10.4)
        FORMAT ('F=',F10.4,13X,'D (M)=',F10.4,12X,'G (M/S2)=',F10.4)
        500 FORMAT ('TM (S)=',F10.4,8X,'RN (RPM)=',F10.4,9X,'TR (NT.M)=',F10.
    $4)
501 FORMAT ('HR (M)=',F10.4,8X,'QR (M3/S)=',F10.4,8X,'WRR (NT.M2)=',F
    $10.4)
5 0 2 ~ F O R M A T ~ ( ' T O L = ' , F 1 0 . 4 , 1 1 X , ' V I = ' , F 1 0 . 4 , 1 5 X , ' N = ' , I 3 ) ~
5 0 3 ~ F O R M A T ~ ( ' D X ~ ( R A D ) = ' , F 1 0 . 4 , 6 X , ' C K ~ = ' , F 1 0 . 4 , 1 4 X , ' P I = ' , F 1 0 . 4 ) ~
5 1 1 ~ F O R M A T ~ ( ' A K V 1 ~ = ' , F 1 0 . 4 , 9 X , ' A K V 2 = ' , F 1 0 . 4 , 1 3 X , ' D T A U ~ ( S ) = ' , F 1 0 . 4 )
5 0 4 ~ F O R M A T ~ ( ' A L = ' , F 1 0 . 4 , 1 2 X , ' A L 0 = ' , F 1 0 . 4 , 1 4 X , ' A L 0 0 = ' , F 1 0 . 4 )
5 0 5 ~ F O R M A T ~ ( ' H M I N = ' , F 1 0 . 4 , 1 0 X , ' T = ' , F 1 0 . 4 , 1 6 X , ' J P R = ' , I 3 ) ~
506 FORMAT ('K=',I3,20X,'NSV =',I3, 20X,'KIT=',I3)
5 1 2 ~ F O R M A T ~ ( ' T A U ( 1 ) = ' , F 1 0 . 4 , 8 X , ' T A U ( 2 ) = ' , F 1 0 . 4 , 1 1 X , ' T A U ( 3 ) = ' , F 1 0 . 4 ) ~
```



```
514 FORMAT ('TAU(7)=',F10.4,8X,'TAU(8)=',F10.4,11X,'TAU(9)=',F10.4)
515 FORMAT ('TAU(10)=',F10.4,7X,'TAU(11)=',F10.4,10X,'TAU(12)=',F10.4
    $)
```

```
516 FORMAT ('TAU(13)=',F10.4,7X,'TAU(14)=',F10.4,10X,'TAU(15)=',F10.4
    $)
517 FORMAT ('TAU(16)=',F10.4,7X,'TAU(17)=',F10.4,10X,'TAU(18)=',F10.4
    $)
518 FORMAT ('TAU(19)=',F10.4,7X,'TAU(20)=',F10.4,10X,'M2=',I2)
519 FORMAT ('CCT (S)=',F10.4,7X,'NTAU =',I3,19X,'QQA (M3/S)=',F10.4)
520 FORMAT ('DT (S)=',F10.4,8X,'AKS (M)=',F10.4,10X,'VN (M2/S)=',F10.6
    $)
507 FORMAT (/,/,26X,17('*'),26X,/,26X,'* RESULT VALUES *',26X)
508 FORMAT (26X,17('*'),26X,/,/)
        STOP
        END
```


## C.2. Typical data file

```
4.6 1331.0 108 0.125 9.81 6.62 4.98 74 10 1450 32.18 11 0.036 0.83
2.5 20 0.0002 0.46 42 21 5 0.00015 0.000001
0.0714 3.1416 42 27
1 1 1 0 0 0
-0.3 0.05 0.12 0.21 0.04 0.1 0.15 0.3 1.39 1.42 1.39 1.34 1.35
1.36 1.36 1.35 1.35 1.33 1.32 1.32 1.34 1.34 1.3 1.28 1.3 1.41 1.7
1.6 1.6 1.6 1.6 1.6 1.6 1.6 1.58 1.61 1.61 1.6 1.6 1.6 1.61 1.61 4.1
0.634 0.643 0.646 0.640 0.629 0.613 0.595 0.575 0.552 0.533 0.516
0.505 0.504 0.510 0.512 0.522 0.539 0.559 0.580 0.601 0.630 0.662
0.692 0.722 0.753 0.782 0.808 0.832 0.857 0.879 0.904 0.930 0.959
0.996 1.027 1.06 1.090 1.124 1.165 1.204 1.238 1.258 1.271 1.282
1.288 1.281 1.260 1.225 1.172 1.107 1.031 0.942 0.842 0.733 0.617
0.5 0.368 0.24 0.125 0.011 -0.102 -0.168 -0.255 -0.342 -0.423 -0.494
-0.556 -0.620 -0.655 -0.670 -0.670 -0.66 -0.655 -0.640 -0.6 -0.57 -
0.52 -0.47 -0.43-0.36 -0.275 -0.16 -0.04 0.130 0.295 0.430 0.55
0.620 0.634
-0.684 -0.547 -0.414 -0.292 -0.167 -0.105 -0.053 -0.012 0.042 0.097
0.156 0.227 0.3 0.371 0.444 0.522 0.596 0.672 0.738 0.763 0.797
0.837 0.865 0.883 0.886 0.877 0.859 0.838 0.804 0.758 0.703 0.645
0.583 0.520 0.454 0.408 0.370 0.343 0.331 0.329
0.338 0.354 0.372 0.405 0.450 0.486 0.520 0.552 0.579
0.603 0.616 0.617 0.606 0.582 0.546 0.5 0.432 0.36 0.288 0.214 0.123
0.037-0.053 -0.161 -0.248-0.314 -0.372 -0.58 -0.74 -0.88 -1 -1.12
-1.25 -1.37 -1.49 -1.59 -1.66 -1.69 -1.77 -1.65 -1.59 -1.520 -1.420
-1.320 -1.23 -1.1 -0.980 -0.82 -0.684
```

